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Volume II: Acrylic Submersibles

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Foreword

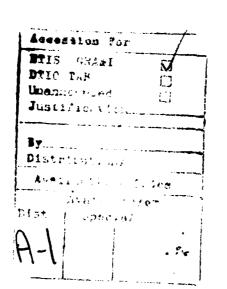
Exploration of hydrospace requires manned and unmanned underwater vehicles capable of carrying observers and/or electro-optical devices to the very bottom of the sea. In either case, the vehicles must be provided with viewports through which the occupants can observe, and the cameras can record, the environment around them. Windows in these viewports must not only be clear but also strong enough to withstand the external hydrostatic pressure exerted by a column of water extending from the vehicle to the water's surface.

Pressure-resistant, acrylic-plastic windows were introduced into submersibles in 1947 by Professor Piccard. Since then, these windows have seen extensive service on undersea vehicles where they provided the occupants a clear, but limited view of hydrospace. However, even vehicles equipped with multiple viewports do not afford the occupants the desired panoramic view of the environment outside the vehicle. On the contrary, they accentuate the occupants' feelings of being enclosed in an opaque box with multiple peepholes that allow only tantalizing glimpses of the colorful environment.

This burdensome obstacle to unimpeded visual exploration of hydrospace could be eliminated by providing the crew of the submersible with a pressure-resistant, transparent cockpit. This cockpit would be mounted on top or in front of the opaque housing that encloses the functional subsystems of the submersible. To convert this concept into reality, many technical problems had to be solved. A transparent material with desirable structural properties had to be selected; a pressure-resistant enclosure, compatible with the structural characteristics of the material, had to be designed; and one, or several, economical fabrication techniques had to be developed.

The Navy achieved the goal of a crew compartment that was transparent with panoramic visibility, when, in 1970, the Naval Facilities Engineering Command launched the world's first two-man transparent submersible, *Nemo*, that had an operational depth of 600 feet. The pioneering transparent cockpit design gave rise to a whole class of oceanographic submersibles with transparent compartments and with a depth rating that has gradually been extended to 3000 feet by improving the structural performance of the transparent enclosure.

To preserve and disseminate the new engineering knowledge gained during the development of the transparent, pressure-resistant crew compartments for oceanographic submersibles, all the technical reports published on this subject have been collected. They are presented to the ocean engineering community in Volumes 1 and 2 of this monograph. This information should prove very helpful to any engineer contemplating the design of transparent, pressure-resistant, spherical hulls for submersibles.



J. D. Stachiw Marine Materials Office Ocean Engineering Division



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IMPROVED FABRICATION PROCESS FOR SPHERICAL ACRYLIC PLASTIC SUBMERSIBLE HULLS

by

J. D. Stachiw
OCEAN TECHNOLOGY DEPARTMENT

December 1975



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NAVAL UNDERSEA CENTER, SAN DIEGO, CA. 92132

AN ACTIVITY OF THE NAVAL MATERIAL COMMAND

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Technical Director

ADMINISTRATIVE INFORMATION

The work described in this report was performed between June 1972 and June 1973 as part of an investigation into man-rated transparent submersibles for deep operation. It was funded through the Independent Research and Independent Exploratory Development Program at the Naval Undersea Center under subproject task area number ZF-61-412-001.

Released by H. R. TALKINGTON, Head Ocean Technology Department

ACKNOWLEDGMENTS

The fabrication and testing of the model 2000B spherical acrylic plastic hull represent the combined efforts of Adroit Engineering, San Diego, Calif., who designed the hatches and molds; Polymer Products, Oakland, Calif., who made the acrylic castings; and Southwest Research Institute, San Antonio, Texas, who tested the finished assembly. The successful completion of this work is due to the support of H. R. Talkington, Head of the Ocean Technology Department at the Naval Undersea Center, and Dr. W. B. McLean, retired technical director of the Naval Undersea Center. The report was reviewed for technical accuracy by K. O. Gray.

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consists of bonding together 12 thermoformed and machined spherical pentagonal shell sections. In addition, 90 percent fewer bonded joints are required, resulting in an order-of-magnitude improvement in optical qualities. A full-scale prototype with an outside diameter of 66.500 inches and an inside diameter of 58.000 inches has been constructed and shown to be acceptable for manned service to a depth of 2500 feet by hydrostatic testing under sustained loading at pressures of 900, 1350, 1800, and 4000 lb/in ² . Implosion occurred after 13 minutes of sustained loading at 4000 lb/in ² and 75°F (simulated depth of 9000 feet).				

SUMMARY

This report describes an improved process for fabricating spherical acrylic plastic pressure hulls within close dimensional tolerances. The process consists of casting acrylic plastic hemispheres in a precision mold assembly, machining their equatorial edge and cutting polar penetrations, bonding them together with a cast-in-place equatorial joint, polishing their inner and outer surfaces, and installing an aluminum hatch and penetration plate. The cost of the improved process is approximately 50 percent less than that of the standard process, which consists of bonding together 12 thermoformed and machined spherical pentagonal shell sections. In addition, 90 percent fewer bonded joints are required, resulting in an order-of-magnitude improvement in optical qualities. A full-scale prototype with an outside diameter of 66.500 inches and an inside diameter of 58.000 inches has been constructed and shown to be acceptable for manned service to a depth of 2500 feet by hydrostatic testing under sustained loading at pressures of 900, 1350, 1800, and 4000 lb/in². Implosion occurred after 13 minutes of sustained loading at 4000 lb/in² and 75°F (simulated depth of 9000 feet).

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INTRODUCTION

Since the spherical shape provides optimal resistance to external hydrostatic pressure (ref. 1-6), it was chosen for the first acrylic plastic pressure hull, used on the manned submersible NEMO (ref. 7-11). Though limited to an operational depth of 600 ft, NEMO proved the feasibility of this kind of hull and the value of panoramic visibility underwater (ref. 12). MAKAKAI, the second Navy submersible of this class, used the same hull design as NEMO and also had an operational depth limitation of 600 ft (ref. 13). When the Smithsonian Institution built JOHNSON SEA LINK I, a spherical acrylic plastic pressure hull similar to those used on NEMO and MAKAKAI was chosen; the design of hull and hatches was modified, however, to permit operation to a depth of 1000 ft (ref. 14-15). Finally, during design of JOHNSON SEA LINK II for the Harbor Branch Foundation, hull and hatches were further improved, and a depth capability of 3000 ft was obtained (ref. 16).

The only shortcoming of the model 600, 1000, and 2000 hulls used on NEMO. MAKAKAI, and JOHNSON SEA LINK was their fabrication from multiple spherical pentagonal units (figure 1a). This method of fabrication resulted in more than 50 feet of bonded joints, requiring a large amount of hand labor and producing frequent optical discontinuities in the hull (ref. 17). For this reason an experimental study was undertaken jointly by the Naval Undersea Center and the Harbor Branch Foundation to develop a method of fabricating spherical acrylic plastic hulls from two hemispherical sections with a single equatorial joint (figure lb).

The development of the improved fabrication process was planned to proceed in two steps. The objective of the first step was to show the feasibility of a precision casting process that would produce hemispheres requiring no machining of their spherical surfaces. This objective was accomplished by casting hemispherical windows with an inside diameter of 10 inches and outside diameter of 18 inches and testing them to show that they were acceptable for manned submersibles (ref. 4).

The second part of the study addressed itself to applying the casting process developed in the first part to full-sized spherical hulls. This report describes the work performed, which resulted in the model 2000B spherical acrylic plastic pressure hull with an operational depth rating of 2500 feet.

DESIGN AND FABRICATION

The focus of the study was the model 2000 hull developed for JOHNSON SEA LINK II. This hull, shown in figure 2, represented the most advanced model in the series and therefore had the greatest potential value to prospective users. In addition, the Harbor Branch Foundation had a requirement for two pressure hulls with at least a 2000-foot depth capability. Thus, if tooling could be developed for casting the model 2000 hull in hemispherical rather than pentagonal sections, it would be available without additional cost for fabrication of these hulls.

HULL DESIGN

The objective was to develop a hull design whose performance under hydrostatic loading would be identical to that of the model 2000 hull assembled from 12 thermoformed plexiglas-G spherical pentagons of four-inch thickness. This presented a problem, since tests conducted in phase I of the study showed that although the acrylic plastic resulting from the precision casting process developed by Polymer Products met Navy and ASME specifications (ref. 18) for man-rated acrylic windows and hulls, it had approximately 10 percent lower yield strength than Plexiglas G.

To approximate the structural performance of the model 2000 hull it became necessary to increase its minimum four-inch wall thickness by at least 10 percent (figure 3). Since decreasing the inner diameter of the hull would make the interior more cramped for the crew, most of the increase was accomplished by increasing the outer diameter of the spherical shell. No other changes were made at that time to the original design. As a result both the hatch and bottom penetration assemblies designed previously for the model 2000 capsule could be utilized in the model 2000B hull with only minor modifications. The modifications consisted of increasing the width of the polycarbonate insert from 4.875 inches to 5.000 inches and decreasing the thickness of the insert flange from 0.905 to 0.750 inch. These minor modifications accommodated the polycarbonate insert originally designed for a four-inch wall thickness to the 4.4-inch wall thickness of the model 2000B hull.

TOOLING

Tooling for the precision casting of acrylic plastic hemispheres for the model 2000B hull consisted of a mold assembly, an autoclave cart, and a strongback. All tooling components were designed by Adroit Engineering, San Diego, California.

Mold Assembly

The mold assembly consisted of a matched set of male and female molds (figure 4). Considerable thought went into the design of the mold assembly. It was to serve as the form for gelling and polymerization of the acrylic plastic and as a power assisted jig for separating the polymerized casting from the mold. In addition, the mold assembly had to fit into the autoclave, where polymerization took place under elevated temperature and pressure.

The mold assembly was patterned after the mold developed during phase 1 for the hemispherical window castings. The major components were the female mold (serving as the foundation for four flanged wheels), the male mold (serving as the foundation for six hydraulic lifting jacks and six elevation adjusting screws), and a manually operated hydraulic pump (for pressurization of the hydraulic jacks). The whole assembly was fabricated from welded low-carbon steel. The nominal wall thickness of the female mold was 0.5 inch and of the male mold 0.75 inch. Both molds were reinforced with meridional and circumferential stiffeners to maintain their sphericity during and after machining (figures 5 and 6). The male mold was made from thicker steel than the female mold because it was thought that it might buckle under external pressure, exerted by the shrinking plastic, during the polymerization process inside the autoclave.

The hydraulic jacks mounted on the extensions of the male mold served to separate it from the polymerized casting (figure 7). Since the shrinkage of the casting was known to be in the 5 to 10 percent range, a substantial grip would be exerted on the male mold. To overcome this grip each of the jacks was designed to exert up to 10 tons of thrust against the equatorial edge of the casting. A manually operated hydraulic pump provided pressurized oil through flexible hoses to the jacks.

The thrust of the hydraulic jacks was augmented by air pressure applied through a fitting in the bottom of the mold to the interface between mold and casting (figure 8a). This provision was found to be very helpful, since the jacks alone could not always insure separation between the mold and the casting.

The separation of the female mold from the casting was accomplished by pressurized water pumped into the annular space between mold and casting through a fitting located at the very bottom of the mold. After the separation was accomplished, further influx of water made the casting float up in the mold till a lifting jig could be attached to it (figure 8b).

The elevation adjusting screws, located on the extensions of the male mold, were used for adjusting the clearance between the bottom of the male mold and the female mold. They also helped to locate the center of the male mold in the center of the female mold. With the help of these screws it was possible to center the male mold within 0.030 inch of the desired location.

Flanged wheels, attached to the lower external circumferential stiffener on the female molu, were designed for moving the mold assembly on rails in and out of the autoclave located at Polymer Products. In this manner, the mold assembly could be easily filled with casting mix outside the autoclave and then moved on rails into the autoclave without disturbing the gelling mixture.

Autoclave Cart

The autoclave cart consisted of a box frame supported by four flanged wheels whose spacing matched that of the narrow track extending from the interior of the autoclave into the general work area. The cart was designed to support the assembled castings during bonding of the equatorial joint and subsequent polymerization of the adhesive inside the autoclave (figure 9).

Strongback

The strongback consisted of a circular frame with a lifting sling. The diameter of the circular frame was smaller than the outer diameter of the casting, permitting the casting to be lifted from the female mold after it was partially raised by water (figure 10). The strongback was attached to the casting by disassembling it into two halves, placing them around the casting, and clamping them together with bolts.

CASTING AND INSPECTION

Casting Process

The casting process developed previously for this purpose by Polymer Products of Oakland, California, consisted of five distinct steps: (1) mixing of resin with additives; (2) pouring of resin mix into the mold assembly; (3) gellation of resin mix in the mold at atmospheric pressure and temperature; (4) polymerization of gelled resin mix inside the autoclave under elevated temperature and pressure; and (5) removal of the polymerized casting from the mold.

Mixing of acrylic (figure 11) with the required additive took place under atmospheric pressure and temperature. The same ratio of acrylic to polymer powder, catalyst, and cross-linking initiator was used as in phase I of the study. The mixing was performed by hand with an electric rotary mixer in five gallon batches (figure 12), followed by degassing under vacuum. The degassed resin mix was then, after some further manual mixing (figure 13), poured into the mold assembly whose surfaces were scrupulously cleaned (figures 14, 15, 16) and protected from dust by plastic sheets. The mixing of batches was repeated until the mold was filled.

Gellation of the casting inside the mold assembly took place under atmospheric pressure and temperature. The length of time required for gellation varied with temperature, but as a rule several hours were sufficient.

Polymerization of the gelled resin mixture took place inside a horizontal autoclave. The resin-filled mold assembly was rolled into the autoclave on tracks extending from the general assembly area. The pressurization and heating schedule was the one developed (ref. 17) previously for this purpose by Polymer Products in phase I of the study. Record was kept of the pressures and temperatures during the polymerization process so that any malfunction of the autoclave system could be detected and its effect on the physical properties of the castings noted.

The crucial step in the polymerization process was the separation of the male mold from the already polymerized but still hot casting. The separation was achieved by simultaneously applying air pressure to the fitting in the bottom of the mold and hydraulic pressure to the six hydraulic jacks spaced around its circumference. After the mold had been raised about two inches, it was placed on wedges resting on the rim of the female mold. Upon completion of this step the door to the autoclave was closed again and the gradual lowering of ambient temperature initiated.

The lifting of the male mold generated a small clearance between it and the interior surface of the casting. Because of this clearance, cooling of the casting could take place without the generation of tensile hoop stresses in the rim of the casting. If the casting had been cooled to ambient atmospheric temperature without prior release of the male mold, tensile cracks would have appeared. Opening of the autoclave door and rolling out of the mold assembly completed the polymerization process (figure 17).

Removal of the casting from the mold assembly was accomplished in the general work area outside the autoclave. First, the male mold was removed with a forklift. Second, the casting was partially raised inside the female mold by injecting tap water through the

bottom of the mold into the interface between mold and casting (figure 18). Third, the split strongback frame was clamped around the casting protruding from the mold, and the casting was lifted with the forklift from the mold (figure 19).

Inspection

After removal from the mold, the casting was subjected to an inspection whose objectives were to determine its quality. The inspection included visual observation, dimensional measurement, and testing of material specimens to determine their physical properties.

Visual observation was conducted utilizing transmitted sunlight as the source of illumination. The observation focused on the smoothness of the casting surfaces, the clarity of the casting, and the size and number of voids.

The concave and convex surfaces possessed the same surface roughness (about 64 microinches rms) as the Teflon-coated metallic molds, and no further finishing was required except fine sanding and polishing (figure 20). Of the five hemispheres cast, only one exhibited surface irregularities caused by separation of the casting from the mold during the polymerization process (figure 21). The surface irregularity was repaired by casting a thin overlay followed by rough sanding that brought the finished surface to the required thickness and sphericity. One of the castings exhibited meridional cracks. They were caused by failure of temperature control in the autoclave, resulting in sudden cooling of the casting prior to removal of the male mold (figure 22).

The equatorial edge of the castings was in the form of a meniscus with two-inch depth caused by shrinking of the resin mix during the polymerization process (figure 23). This was as expected and allowed for in the design of the mold assembly.

The clarity of the casting was equivalent to that of Plexiglas G of similar thickness. After fine sanding and polishing of inner and outer surfaces, the casting was found to satisfy the proposed ASME requirement for clarity in acrylic plastic viewports.*

The number of voids varied from one casting to another. Some castings had none, while others had more than ten. The voids were in almost every case located in the midplane of the shell thickness and several inches below the equator. They varied in shape but as a rule were elongated, about one inch diameter and two to four inches long.

Since the presence of such voids (figure 24) was unacceptable from the structural and optical viewpoints, small holes were drilled from the equatorial edge of the casting and the voids filled with standard casting mix. The mix was subsequently polymerized by placing the casting back in the autoclave and subjecting it to the required pressure and temperature regimen. Visual inspection after polymerization showed a significant improvement (figure 25). Before the repair, the voids reflected most of the incident light; now they transmitted it in a largely coherent manner. However, even though the refilled voids represented an order-of-magnitude optical improvement, they still were not deemed acceptable. As a result, they were routed out completely and refilled again, utilizing the standard casting mix and polymerization process.

^{*}Clear print of size 7 lines per column inch and 16 letters to the linear inch shall be clearly visible when viewed from a distance of 20 inches through the thickness of the casting with opposite faces polished (ref. 19, 20).

The improvement achieved by the second recasting was still more significant (figure 26). The repair could now be detected only by moving a printed newspaper page along one surface while the observer watched for minute distortion of the print image at the boundary of the recast void. The credit for this result was given to routing, which created large cavities with smooth vertical walls and large openings. The vertical walls prevented entrapment of gas bubbles during pouring of the casting mix, while the large opening allowed for adequate degassing of the mix once it was poured.

Dimensional measurements were aimed at determining the actual wall thickness of the castings at all locations. There was little need to check sphericity, which always closely conforms to that of the mold. Since the mold surfaces were machined within ± 0.060 inch of the specified radius, the sphericity of the castings was more than adequate to meet U.S. Navy specifications for man-rated spherical pressure hulls of acrylic plastic.*

The reason for checking wall thickness was that the male mold might not have been aligned properly with the female mold. If the alignment was not proper, the wall thickness would vary from point to point on the hemisphere even though the sphericity of the surfaces was within specification.

The wall thickness of the hemispheres was found to vary from one location to another, with the largest deviation found at the pole. Thus, the minimum thickness at the pole was found to be 4.107 inches, while around the circumference at the equator it varied from 4.370 to 4.210 inches. Since the variation in thickness around the equator was less than the specified tolerance of 0.200 inch, it was considered to be within the range of permissible thickness tolerances imposed by machining tolerances of the molds, and thus acceptable. The variation in thickness between the equator and the pole of the hemisphere, however, was considered not to be acceptable, since it was approximately 0.093 inch below the minimum specified thickness of 4.200 inches.

To avoid this problem in future hemispherical castings produced in the existing 66-inch mold by Polymer Products, the elevation of the male mold inside the female mold will be raised by 0.125 inch. Thus, new castings will have a thickness that does not exceed the 4.200-to-4.400-inch range at any location.

The physical properties of the castings were determined by testing material specimens cut from the poles, the future location of metallic hatches. Two specimens were used per test for each hemisphere. The results of the tests (summarized in table 1) were satisfactory, and in every case the physical properties of the material met or surpassed Navy and ASME specifications for acrylic plastic in man-rated pressure resistant structures (appendix A).

^{*}For spherical hulls with 66-inch outside diameter, the maximum permitted deviation in sphericity is ±0.165 inch (ref. 19, 20).

Table 1. Properties of Acrylic Plastic Casting.

Property	Specified value	Actual value (average)
Ultimate tensile strength	9,000 lb/in ² min	9,670 lb/in ²
Tensile elongation at fracture	2 percent min	3.85 percent
Tensile modulus of elasticity Test method: ASTM D638	400.000 lb/in ² min	497,500 lb/in ²
Compressive yield strength	$15,000 lb/in^2 min$	16,150 lb/in ²
Compressive modulus Test method: ASTM D695	400,000 lb/in ² min	515,000 lb/in ²
Shear strength Test method: ASTM D732	8,000 lb/in ² min	9,775 lb/in ²
Ultimate flexural strength	$14,000 lb/in^2 min$	$15.150 lb/in^2$
Flexural modulus Test method: ASTM D790	420,000 lb/in ² min	490,000 lb/in ²
IZOD Impact strength Test method: ASTM D256	0.20 ft-lb/in min	0.29 ft-lb/in
Deformation under load at 4,000 lb/in ² and 122°F	1.0 percent max	0.385 percent
Test method: ASTM D621		
Rockwell M hardness Test method: Rockwell	90 min	105
Water absorption in 24-hour submersion Test method: ASTM D570	0.25 percent max	0.20 percent
Heat distortion temperature Test method: ASTM D648	205°F min	216°F
Refractive index Test method: ASTM D542	1.48-1.50	1.491
Specific gravity Test method: ASTM D792	1.18-1.20	1.182
Coefficient of linear thermal expansion Test method: ASTM D696	4.3 × 10 ⁻⁵ in/in at 80°F	4.6 × 10 ⁻⁵ in/in °F 77-105°F range

Table 1. Continued.

Property	Specified value	Actual value (average)	
Resistance to stress Test method: Table 1 of ASTM Methods	N.A.	2,000 lb/in ² ; no visual evidence of crazing or cracking	
Residual monomer (methyl methacrylate)	1.5 percent max	0.40 percent	
Test method: SPE Trans. 1962			

ASSEMBLY

Assembly of the model 2000B hull consisted of the machining and bonding together of two hemispherical castings, followed by polishing and inspection of the completed sphere. The finished hull was then fitted with aluminum inserts that served as hatch and penetration plate.

Machining of the hemispherical castings was preceded by rough grinding of the equatorial edge with a rotary file (figure 27). After the edge was ground to within an inch of its final dimension, the casting was mounted in a vertical mill and the polar opening machined (figure 28). It was then turned over in the mill and the equatorial edge machined to its final dimension.

Bonding of the hemispheres into a single structural entity was begun by placing one hemisphere on top of the other (figure 29). The width of the joint was controlled by placing small acrylic plastic spacers of 1/4-inch thickness between the hemispheres. The joint was subsequently covered with adhesive aluminum foil tape. To facilitate pouring of the bonding mix into the joint cavity, three pouring spouts were plumbed to openings in the tape covering provided for this purpose (figure 30).

The bonding mix was prepared by combining the same ingredients that made up the basic casting mix. The mix was poured concurrently into the three pouring spouts around the circumference of the sphere and into a separate test block joint. This block served later as a source of specimens for determination of joint strength. As soon as the mix gelled, the sphere assembly with the associated test block was placed in the autoclave and subjected to temperature and pressure until polymerization of the joint was completed (figure 31).

Upon removal of the assembly from the autoclave, extensive voids were found on the inner surface of the joint (figure 32). Careful examination of joint and polymerization procedure established shrinkage of the mix during polymerization to be the cause. The voids were not present on the outer surface of the joint because extra mix was provided by an outward bulge in the tape. This bulge was absent on the inner surface of the joint, resulting in the observed shrinkage voids. Such voids will be prevented in the future by forming the tape over the joint in such a manner that a bulge is present.

The voids in the joint decreased its bearing surface to such a degree (about 25 percent) that it became structurally unacceptable. This problem was corrected by removing the tape, rotating the sphere until the equatorial joint was in the vertical plane, and filling

the voids with room-temperature-polymerizing PS-30 adhesive. Since the adhesive could be placed properly only in the void at the lowest point of the vertically oriented joint, the sphere had to be rotated between fillings.

The resulting joint was still far from completely void free, but the cross section and number of remaining voids were so small that it could be considered structurally acceptable (figure 33). Because the room-temperature-polymerized PS-30 adhesive was somewhat softer than the high-temperature and high-pressure polymerized casting mix, differential compression of the joint was expected when the sphere was subjected to hydrostatic testing.

Polishing of the completed sphere consisted of rough sanding of the edges of the joint followed by fine sanding and polishing of both the internal and external surfaces (figure 34). Inspection consisted of detailed visual observation, dimensional measurements, and testing of bond samples. The objective of the visual observation was to ascertain the effect of the joint and repaired voids in the castings on the optical properties of the hull. The dimensional measurements were performed to determine the conformance of the completed sphere assembly to specified dimensional tolerances. The testing of bond samples served as quality control for the bonding technique used for joining the hemispheres.

The visual inspection showed that the optical properties of the sphere were generally more than adequate for underwater search, salvage, or work missions where panoramic visibility is of paramount importance. The only areas that showed optical distortion were the equatorial joint and the repaired voids in the castings (figure 35), both of which distorted images at their boundaries. The distortion was not severe enough to significantly lower the value of the sphere as a panoramic observation capsule. It was, however, sufficient to preclude photography through the sphere at those locations.

Dimensional measurements (figure 36) showed that the diameter and angle of the top and bottom polar openings, as well as the outside diameter of the capsule, were within specified tolerances. The thickness of the hull was found, however, to fall below the minimum specified thickness by 0.093 inch. As noted previously, the excessive variation in thickness was caused by improper centering of the male mold within the female mold during casting. The result was that the shell of the capsule was thinnest at the edges of penetrations, where the stresses are highest during external hydrostatic loading. Better alignment of the male mold within the female mold will forestall the recurrence of this problem.

Installation of polar inserts consisted of placing the top hatch and bottom penetration plate with associated polycarbonate gaskets into their respective polar openings and locking them in place by bolting on split retaining rings. The hatch and penetration plate assemblies used for testing the model 2000B hull were those used in the previous model 2000 hull test and evaluation program (figures 37 through 46). This meant that they had been previously pressurized to 1800 lb/in² and as a result might have experienced some local yielding.

Only minor modification was required to the polycarbonate gasket even though the cast hemispheres of the model 2000B hull were about 0.100 inch thicker at the polar opening than those of the model 2000 hull. This was feasible, however, only because the wall thickness at the polar opening was in the 4.1-to-4.2-inch range rather than the 4.2-to-4.4-inch range specified. If the wall thickness of the castings had been within the specified range, the model 2000 polycarbonate gasket would have had to be replaced with a new one (figure 46). The modified gasket actually used in the test is shown in figure 47.

HYDROSTATIC TESTING

The objective of hydrostatic testing was to determine the strains, displacements, and failure mode of the model 2000B hull so that its performance under pressure could be compared with that of the model 2000 hull. Since the polar inserts were the same for both hulls, differences in performance would be attributable only to the physical properties of the material and the slight difference in wall thickness.

TESTS AND INSTRUMENTATION

Hydrostatic testing was conducted in a 90-inch-diameter pressure vessel at the Southwest Research Institute, San Antonio, Texas (figure 48) in four discrete steps, each consisting of sustained loading and relaxation phases. By making the length of the loading and relaxation phases equal, viscoelastic strains were given an opportunity to return to zero before the material was subjected to higher strain levels. Prior to testing, the interior of the hull assembly was filled with water to mitigate the shock of implosion and to provide a means of determining the rate of volumetric contraction under loading. To maintain zero atmospheric pressure inside the sphere, a tube was connected to both the top hatch and the pressure vessel closure. As the acrylic sphere contracted, the water was forced out through this tube and its volume measured at the outlet with a 2000 ml graduate.

<u>Pressure Test 1</u>. The model 2000B assembly was pressurized to 900 lb/in^2 at $100 \text{ lb/in}^2/\text{min}$ at room temperature and maintained at this pressure for 24 hours; it was then depressurized to 0 lb/in^2 at $100 \text{ lb/in}^2/\text{min}$ and maintained at this pressure for 24 hours. Strains and displacements were recorded at 100 lb/in^2 intervals during pressurization and at 6 hour intervals during depressurization.

Pressure Test 2. Identical to test 1 except that the maximum pressure was 1350 lb/in².

Pressure Test 3. Identical to test 1 except that the maximum pressure was 1800 lb/in².

<u>Pressure Test 4.</u> The assembly was pressurized to 4000 lb/in² at 50 lb/in²/min and maintained at this pressure to implosion. Strains were recorded until the 4000-lb/in² pressure was reached, and displacements were recorded to the moment of implosion.

Instrumentation consisted of 90-degree biaxial strain rosettes bonded at critical locations to the acrylic hull and polar aluminum inserts (figures 49 and 50). The gage locations chosen for the model 2000B assembly were identical to those chosen for the model 2000 assembly tested previously. Because of this, direct comparison between strains on both assemblies could be made.

The acrylic hull was instrumented only on the equator and at the edges of the polar penetrations. Both locations were important; the magnitude of creep at the equator would give a fair indication of viscoelastic deformation over most of the hull, while that at the edge of penetrations would represent the maximum on the hull.

The aluminum inserts were instrumented only at locations that previous tests on the model 2000 hull had shown to be areas of high stress. Strains measured at these locations would indicate the onset of yielding as the model 2000B hull was pressurized to implosion.

STRAIN RESULTS

Strains measured on the model 2000B assembly, as expected, varied widely from one location to another, but in all cases they were very active (figures 51 through 62 and appendix B). In the hull itself the highest strains were recorded on the interior surface at the edges of polar openings. The maximum strain was in the longitudinal direction, and its magnitude was approximately 50 percent larger than that of strains measured on the interior surface at the equator.

During short term loading, the maximum strains at the polar openings were found to increase linearly with external hydrostatic pressure to about 1350 lb/in²; at higher pressures the strains increased faster than the external pressure. Their magnitudes were measured to be 4500, 9250, 14600, and 20300 microinches/inch at 450, 900, 1350, and 1800 lb/in² respectively. However, on the interior surface at the equator the maximum strains (hoop orientation) were only 2825, 5800, 9050, 12250, and 32500 microinches/inch at 150, 900, 1350, 1800, and 4000 lb/in² respectively.

During long term loadings of 24-hour duration there took place some viscoelastic creep whose magnitude varied with the location on the hull. On the interior surface at the equator, viscoelastic creep was approximately 15 percent of the short term strain at 900 and 1350 lb/in², while at 1800 lb/in² it increased to about 25 percent of the short term strain. At the polar penetrations, the magnitude was larger than at the equator, but in terms of short term strain the percentage was about the same. At the conclusion of the 24-hour sustained pressure loadings strains returned almost to zero. The difference between strain readings at the conclusion of relaxation and zero can be attributed to permanent deformation and errors in the strain recording system. Since the magnitude of permanent residual strains did not increase with the magnitude of sustained pressure, it can be postulated that the magnitude of residual strains measured on the sphere at 900, 1350, and 1800 lb/in² is not only a function of pressure but also of other unknown test variables. It is also interesting to note that the longitudinal strain measured on the equatorial bond was significantly higher than near the joint. The difference in readings indicates that the adhesive used in the bond had a lower modulus of elasticity and probably a lower compressive yield point than the hull material.

When the model 2000B assembly was tested to destruction, the magnitude of compressive strains measured on the interior at the equator was approximately 32500 microinches/inch. Since strain readings were not taken after the 4000-lb/in² pressure loading was reached, the magnitude of creep that took place during this test is not exactly known. However, by converting the change in displaced volume to strains on the acrylic hull, it is possible to calculate the average strains near the equator on the interior surface of the hull at the moment of failure. The magnitude of strain calculated in such a manner is 44500 microinches/inch.

The strains measured on the aluminum inserts (hatch and penetration plate) were similar to those measured when the inserts were tested as part of the model 2000 assembly, indicating that no yielding had taken place during the previous testing to 1800 lb/in².

Even when the model 2000B assembly was pressurized to 4000 lb/in², very little yielding was noted, though compressive strains in excess of 3000 microinches/inch were measured on the interior of the top hatch. When the maximum strain measured on the top hatch was converted to stress, it was found that it amounted to 39,423 lb/in².

DISPLACEMENT RESULTS

Displacement of water from the interior of the model 2000B assembly increased under load. The relationship between external pressure applied at 50 lb/in²/min and the volume of displaced water was linear to about 1350 lb/in² (figure 63). At higher pressures the relationship became markedly nonlinear, with the volume of displaced water increasing at a higher rate than the external pressure.

Under sustained loading, the displacement increased further than under short-term pressurization. The volume of displaced water under long-term loading was a function of both pressure and time. For the three nondestructive sustained pressure loadings of 24-hour duration at 900, 1350, and 1800 lb/in² the total volume of displaced water was 1.8, 3.0, and 4.2 percent, respectively, of the original volume. At the termination of each sustained loading test, the sphere returned to its original dimensions after the 24-hour relaxation period at 0 lb/in². The only sustained loading test that culminated in failure of the assembly took place at 4000 lb/in². The total volume of displaced water due to contraction of the sphere prior to failure was 13.5 percent of the original volume.

FAILURE MODE

The model 2000B assembly imploded by general plastic instability after being subjected to an external hydrostatic pressure of $4000 \, \mathrm{lb/in^2}$ for 13 minutes. Because of unforeseen mechanical problems with pumps, the average pressurization rate was $50 \, \mathrm{lb/in^2/min}$ instead of the $100 \, \mathrm{lb/in^2/min}$ specified.

The acrylic hull was fragmented into many irregularly shaped pieces whose size, as a rule, did not exceed three feet in length or width (figures 64 and 65). There was no indication that the bonded joint constituted a plane of weakness. As a matter of fact, most cleavage planes crossed the bonded joint at right angles rather than following it.

No crazing or radial cracks were found in the acrylic surfaces bearing against the polycarbonate gaskets (figure 66). This constitutes a significant improvement over the model 600 and 1000 hulls, which did not utilize polycarbonate gaskets and therefore exhibited many radial cracks after being tested to implosion.

The polycarbonate gasket for the penetration plate was found to be full of small cracks but intact, while the one for the hatch was fragmented into small pieces (figures 67 and 68). The high bearing stresses between the gasket and the aluminum inserts made the polycarbonate flow into the O-ring grooves on the outer circumference of the hatch ring and penetration plate (figure 69).

The aluminum inserts survived the implosion without any visible deformation (figures 70 and 71). This was significantly different from the results obtained for the model 600, assembly numbers 0 and 3, where the hatches buckled plastically at a pressure lower than the implosion pressure of the hull.

EVALUATION OF TEST RESULTS

The above test results show that the performance of the model 2000B assembly is comparable to that of the model 2000 assembly in distribution and magnitude of strains as well as in implosion depth. The basic difference is in the magnitude of permanent strains in the hull after sustained 24-hour pressure loadings. While in the model 2000 assembly the permanent strains were on the order of 50 to 100 microinches/inch, in the model 2000B they were about 200 to 1000 microinches/inch. Although some fraction of the permanent strains can be discounted as experimental error, it is impossible to discount them completely in this manner. In view of this, it appears that the model 2000B should be certified to a lesser depth than 3000 feet, the maximum operational depth of the model 2000. In order to arrive rationally at a safe maximum operational depth, however, it is necessary to review the pertinent design, fabrication, material, and test results involved in the evaluation of the model 2000B assembly.

The design of the model 2000B is identical to that of the model 2000 except for the location of bonded joints and hull thickness. While on the latter there is a multitude of bonded joints between the 12 spherical pentagons, on the former there is only a single bonded equatorial joint. The model 2000B hull is also thicker, 4.210 inches minimum thickness versus 4.050 inches for the model 2000.

Since both the single equatorial joint and additional hull thickness represent structural advantages, the total effect of design on the operational depth of the model 2000B is beneficial. If the acrylic castings of the model 2000B had the properties of Plexiglas G, from which the hull of the model 2000 was fabricated, the maximum operational depth could be increased by four percent to 3120 feet.

Although the fabrication techniques for the model 2000 and 2000B differ, the sphericity tolerances for the finished acrylic hull are the same. It can thus be postulated that the difference in fabrication technique would neither increase nor decrease the potential maximum operational depth capability of 3120 feet.

Although the materials used in constructing the model 2000 and 2000B possess physical properties that meet Navy specifications for acrylic plastics used in manned systems, there is a significant difference between their compressive yield strengths. Where, for example, the average compressive yield strength of the acrylic plastic pentagons used in the model 2000 was 18,416 lb/in², the hemispherical castings used in the model 2000B were found to have an average compressive strength of only 16,300 lb/in². The 11.5 percent lower yield strength of the hemispherical castings should decrease the potential maximum operational depth capability from 3120 to 2761 feet.

The compressive creep (time dependent strain under sustained constant loading) measured on the model 2000B was found to be approximately equal to that measured on the model 2000. For example, the values measured on the interior surface of the hull at the equator in the hoop direction were found to be 900, 1350, and 3000 microinches/inch for the model 2000B at 900, 1350 and 1800 lb/in² sustained loadings of 24-hour duration. This is approximately equal to the values of 500, 1850, and 3100 microinches/inch measured on the model 2000. Because the magnitude of the creep was approximately equal in both assemblies, it would appear that in this respect the maximum operational depth of the model 2000B assembly should be the same as that of the model 2000, that is 3000 feet.

Permanent deformation of the model 2000B was significantly higher than that of the model 2000 after identical sustained pressure loadings. For the former the deformations measured after 24 hours of sustained pressure loadings of 900, 1350, and 1800 lb/in² were -950, -700, +75 microinches/inch respectively. In contrast, the model 2000 showed a permanent deformation of only -100, -50, -150 microinches/inch after sustained pressure loadings of 900, 1350, and 1800 lb/in². Since there is no known well defined relationship between the magnitude of permanent deformation at the equator after several tests at different pressures and the fatigue life of the acrylic bearing surface at the polar penetration, the maximum operational depth of the model 2000B cannot be established on the basis of this data at some lesser depth than that of the model 2000 assembly. Instead, only an indirect approach can be used here, influenced to a large degree by other data not generated in the test program for the model 2000B hull.

The basic assumption underlying the indirect approach to the problem of permanent deformation is that the maximum operational depth of the model 2000B should be substantially less than that of the model 2000 until cyclic fatigue data is experimentally generated by tests of model 2000B scale models similar to those conducted during evaluation of the model 2000 pressure hull assembly. When such data is generated at some future time, it will be possible to establish with confidence the maximum operational depth at which fatigue cracks appear in the acrylic bearing surface at the polar penetrations after 1000 standard dives (four hours at maximum operational depth followed by four hours of relaxation). Until such cyclic fatigue data is generated, the maximum operational depth will of necessity be based on the cyclic pressure tests performed previously on cast acrylic hemispheres with a thickness-to-inside-radius (t/R_i) ratio of 0.8. Extrapolating the safe operational cyclic pressure of approximately 6100 lb/in^2 for hemispheres with $t/R_i = 0.8$ to $t/R_i = 0.147$ for the model 2000B, one arrives at a predicted safe cyclic pressure of approximately 1120 lb/ in² (equivalent to an operational depth of 2500 feet). Since extrapolating data from high to low t/R_i ratios is inherently conservative, the extrapolated cyclic fatigue depth for the model 2000B can be used without reservations until it can be increased on the basis of more complete cyclic fatigue tests.

The implosion pressure of the model 2000B system was found to be approximately the same as that of the model 2000 assembly. The comparison of implosion pressures could not be made directly as the model 2000B and 2000 assemblies tested to destruction were not of the same size, the former being a full-scale prototype and the latter a 1:4.4 scale model. Implosion of the model 2000 scale model took place after 23 minutes at 4000 lb/in², while that of the model 2000B full-scale assembly after only 13 minutes (figures 72). Since the hull thickness of the scale model was approximately 10 percent greater than specified, the projected time to implosion for the full-scale model 2000 is probably only about 15 minutes. In such a case there appears to be no significant difference between the model 2000B and 2000 in resisting external hydrostatic pressure under sustained long-term loading. In effect, then, the static fatigue of the model 2000B is comparable to that of the model 2000 and cannot be used as a factor for increasing or decreasing the maximum operational depth.

This discussion of the physical parameters and test results pertaining to the structural response of the model 2000B shows that, while in comparison to the model 2000 the hull is four percent thicker, has stronger bonded joints, and has approximately the same time-dependent strains, the residual strains observed at the conclusion of simulated dives are

significantly higher. In the absence of cyclic fatigue data for the model 2000B assembly at this time, the conservative approach is to consider the large residual strains as adequate reason for assigning the assembly temporarily an operational depth that is less than 3000 feet.

The temporary operational depth of 2500 feet (1120 lb/in²) assigned to the model 2000B is based on previously generated cyclic fatigue data for spherical sector windows cast by Polymer Products, Inc, utilizing the same resin mix and polymerization process as used in production of the model 2000B. If at some future time cyclic pressure testing of the model 2000B assembly establishes its ability to withstand 1000 simulated dives to 3000 feet without generation of cracks in the hull at the polar openings, the current operational depth limit of 2500 feet will be raised to 3000 feet.

FINDINGS AND CONCLUSIONS

The following are the specific findings of this study:

- 1. It is technically feasible to fabricate spherical pressure hulls of any size by bonding together acrylic plastic hemispheres precision cast in metallic mold assemblies composed of a male and a female component.
- 2. Precision cast hemispheres do not require any subsequent machining of spherical surfaces in order to satisfy Navy specified tolerances for sphericity and thickness.
- 3. The physical properties of the acrylic plastic castings produced by Polymer Products saxisfy Navy and ASME specifications for acrylic plastic used in man-rated external or internal pressure vessels.
- 4. Bonding with the same resin mix that was used in casting the hemispheres produces joints with a tensile strength in excess of 9000 lb/in^2 .
- 5. Voids in the acrylic plastic hemispheres can be successfully recast by filling them with standard casting resin mix and subjecting the hemisphere to the polymerization process for a second time.
- 6. The repaired hemisphere is structurally as strong under external hydrostatic loading as a hemisphere without recast voids.
- 7. The repaired voids are optically objectionable if located in the crew's main field of vision.
- 8. The model 2000B assembly, fabricated by bonding two precision cast acrylic plastic hemispheres with 66.5-inch outside diameter and 57.85-inch inside diameter, successfully withstood 24-hour simulated dives to 2000, 3000, and 4000 feet.
- 9. The model 2000B assembly imploded after 13 minutes of sustained loading at a simulated 9000-foot depth.
- 10. The aluminum polar inserts (hatch and penetration plate) withstood the simulated 9000-foot depth without losing their structural integrity.

On the basis of these findings it is concluded that the model 2000B assembly meets the applicable certification criteria for manned service. The maximum recommended safe operational depth of 2500 feet is based on a conservative interpretation of existing cyclic fatigue. Lata for the acrylic plastic material used in the tested prototype.

OPERATIONAL RECOMMENDATIONS

The following operational recommendations are based on the conservative interpretation of limited cyclic fatigue data. Thus the maximum operational depth and number of crack-free cycles at that depth, as well as the duration of individual cycles, are to be considered as temporary minimums, to be increased later when more definitive cyclic fatigue data is generated.

- 1. The model 2000B assembly during its operational life should never be subjected to depths greater than 2500 feet. The proof test should preferably utilize a test depth of less than or equal to 3000 feet. Under no conditions should the proof test depth exceed 3000 feet.
- 2. The cyclic crack-free fatigue life of the model 2000B assembly is considered to be in excess of 10,000,000 foot-hours (1000 cycles × 2500-foot depth × 4 hours duty). At the conclusion of each dive, the recorded foot-hours should be subtracted from the initial 10,000,000 foot-hour fatigue life. When the sum of foot-hour subtotals generated by dives equals 10,000,000, inserts and gaskets should be removed from the capsule and the entire hull subjected to a detailed visual examination. If no cracks are observed at the polar penetrations, the assembly should be strain-gaged, reassembled, proof-tested to the required depth, and recorded strains at the equator and penetrations compared to those generated during the first proof test conducted immediately after fabrication. Significant differences in strain behavior will be considered indicators of hull deterioration and should result in a significantly reduced depth rating. Cracks in the bonded joint originating at inclusions will be repaired if their length exceeds 0.5 inch. Severely cracked polycarbonate gaskets will be replaced with new gaskets.

If no significant difference in strain behavior is observed, the capsule assembly will be returned to service with a 2500-foot operational depth rating and an additional 10,000,000-foot-hour fatigue life. When the second 10,000,000-foot-hour life has been completed, the assembly will be subjected to the same inspection and proof-testing procedures conducted at the conclusion of the first 10,000,000-foot-hour period. If the results of the new inspection and proof-testing are satisfactory, the capsule will again return to service with a 2500-foot depth rating and additional 10,000,000-foot-hour life.

The recertification process will be repeated until cracks are observed in the bearing surfaces of the acrylic hull during one of the inspections or the strains change significantly. If cracks are observed, they will either be repaired by routing and recasting with resin prior to retesting of the hull, or they will be left in place and the hull's depth rating reduced to 600 feet.

Subsequently, the hull will be inspected without disassembly for signs of crack propagation every 100 dives. When the depth of any crack exceeds one inch, the capsule will be taken out of service immediately and the cracks repaired either by enlarging the polar opening or by recasting the cracked areas. If not repaired, such a hull can be recertified for service to 120 feet. If, during periodic inspections conducted every 100 dives, the depth of the crack at the penetration is found to exceed two inches, the acrylic hull will either be repaired or declared unfit for manned operation at any depth.

- 3. For applications where the presence of a bonded joint in the visual field of the crew is objectionable, the pair of polar penetrations should be moved from their present location to a new location, preferably close to the circumferential joint.
- 4. Attempts should be made to ensure that operators are seated inside the hull as close as possible to the center of the sphere in order to minimize optical distortion (ref. 21). Camera mountings should be located at the center of the hull if wide angle panning is to be performed.
- 5. Many functions of equipment mounted outside the pressure hull can be controlled by modulated light beams projected from the interior of the hull by the crew (ref. 22). This type of arrangement will eliminate the need for electrical connections through the penetration plate and make the control of externally stored scientific equipment an operationally easy matter.

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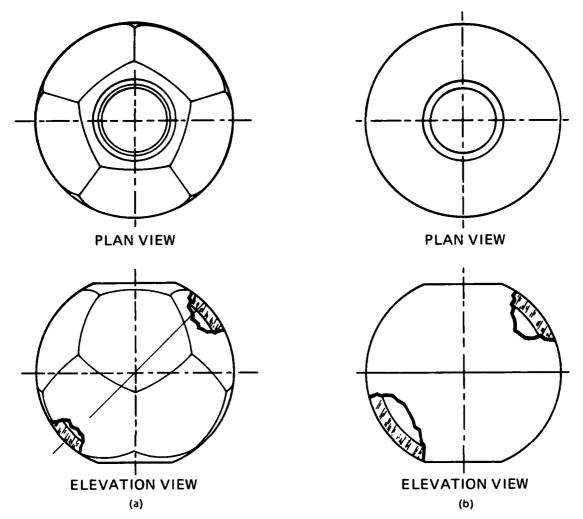


Figure 1. Two approaches to the fabrication of spherical acrylic plastic hulls.

(a) Assembly of twelve spherical pentagonal shell sections. (b) Assembly of two hemispherical shell sections.

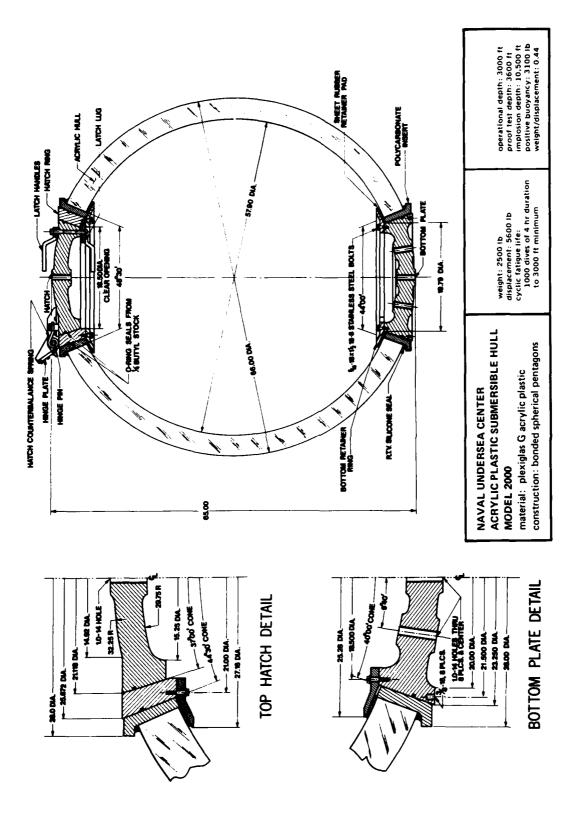


Figure 2. Model 2000 spherical acrylic plastic hull assembled from 12 spherical pentagonal shell sections.

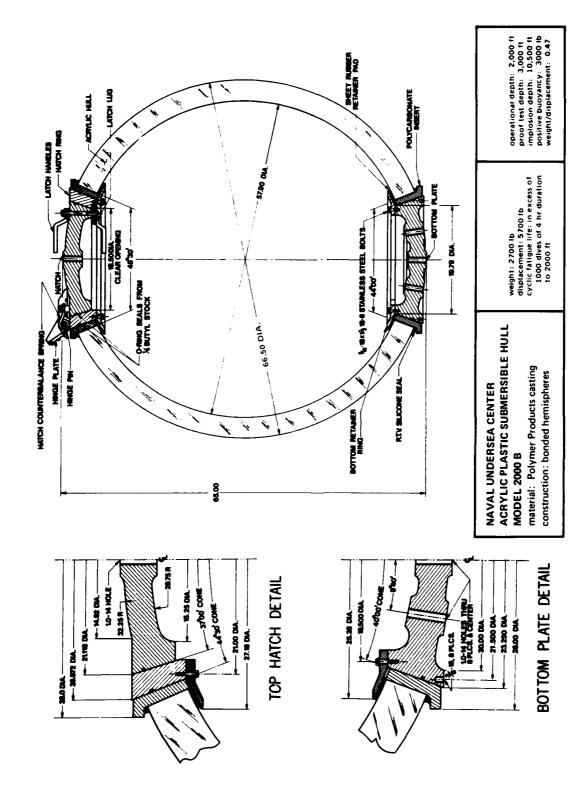


Figure 3. Model 2000B spherical acrylic plastic hull assembled from two hemispherical shell sections.

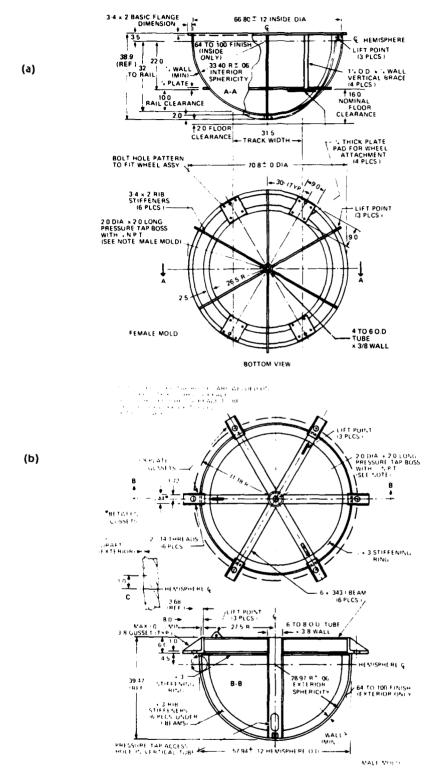


Figure 4. Steel molds for precision casting of hemispherical shells for model 2000B acrylic plastic hull.
(a) Female mold with pads for mounting wheels. (b) Male mold.

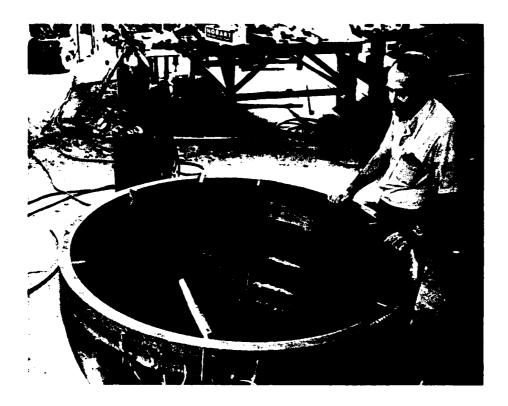
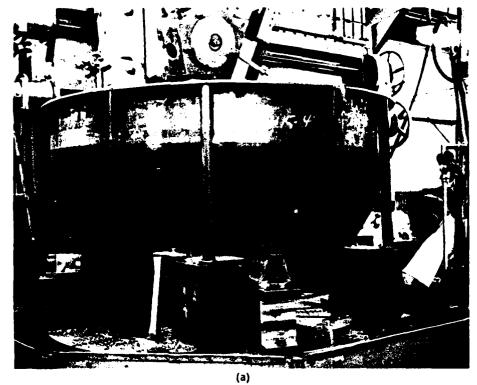


Figure 5. Male mold with inner surfener rings tacked in place for welding.



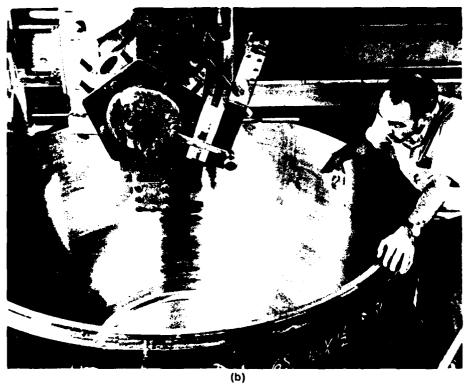


Figure 6. Female mold. (a) Ready for machining; note external longitudinal and circumferential stiffeners. (b) During machining.

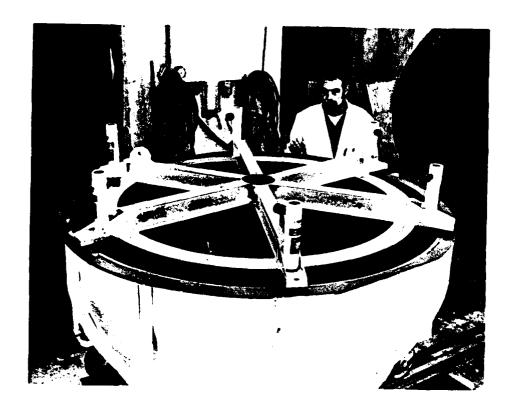
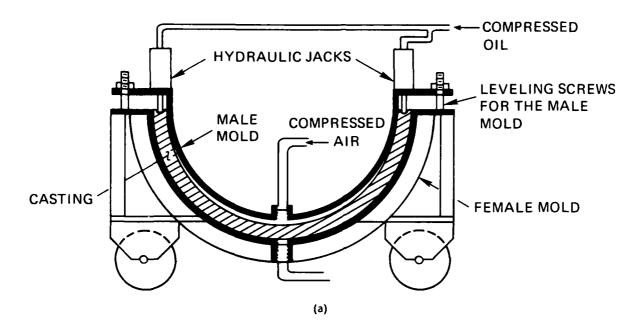


Figure 7. Completed mold assembly with hydraulic jacks and wheels in place.



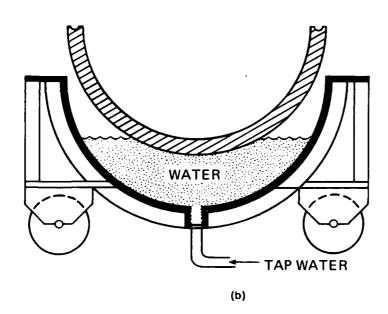


Figure 8. Method of separating molds from casting. (a) Separation of male mold with compressed air and hydraulic jacks. (b) Separation of female mold with compressed water.



Figure 9. Cart for transporting mold assembly into autoclave.

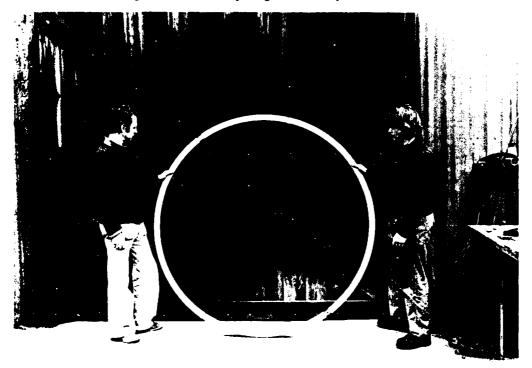


Figure 10. Circular strongback for lifting casting from mold.



Figure 11. Weighing acrylic polymer powder for a typical casting mix batch.



Figure 12. Mixing of ingredients used in a typical casting mix batch.



Figure 13. Stirring casting mix batch to the verge of gelling.



Figure 14. Cleaning of female mold.

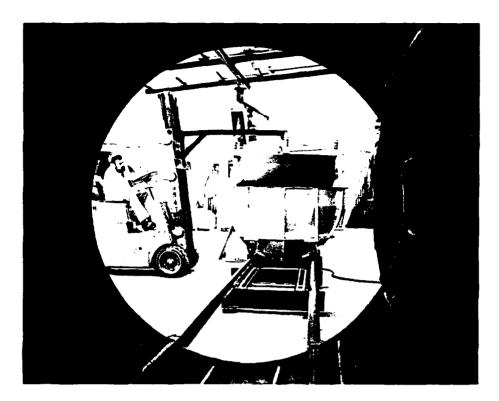


Figure 15. Assembly of cleaned molds to receive casting mix.

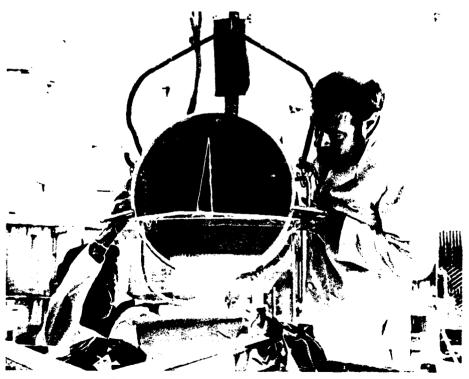


Figure 16. Pouring of casting mix into mold assembly.

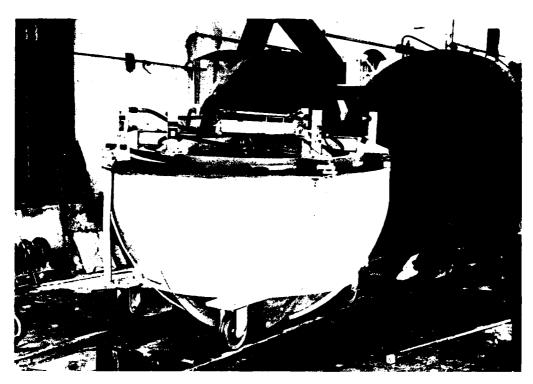


Figure 17. Mold assembly leaving autoclave after polymerization of casting, note hydraulic pump for applying pressure to hydraulic jacks and wooden blocks for keeping raised male mold from resting on casting.



Figure 18. Casting floating on water injected into female mold.

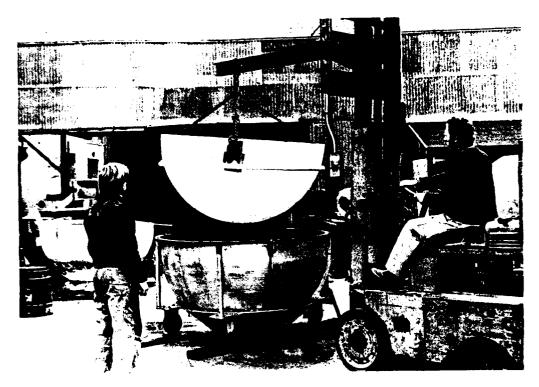


Figure 19. Lifting of casting with strongback.

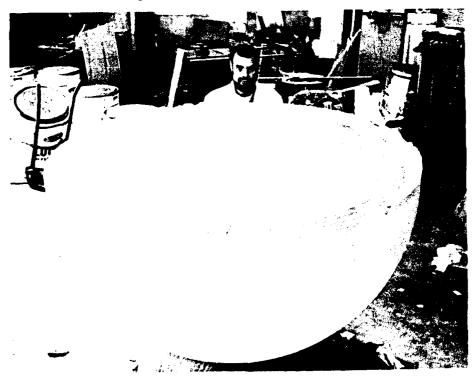


Figure 20. Typical casting after removal from mold.



Figure 21. Separation on surface of casting.



Figure 22. Meridional crack on casting subjected to interrupted polymerization cycle (temperature lowered to room temperature without removing male mold).



Figure 23. Typical two-inch meniscus on equatorial edge of casting.



Figure 24. Typical voids in castings.

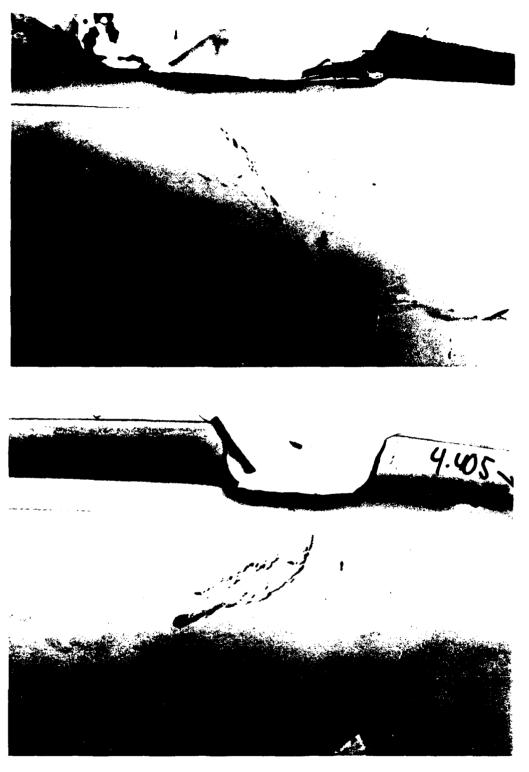


Figure 25. Typical voids after first repair, note numerous small voids.





Figure 26. Typical voids after second repair.

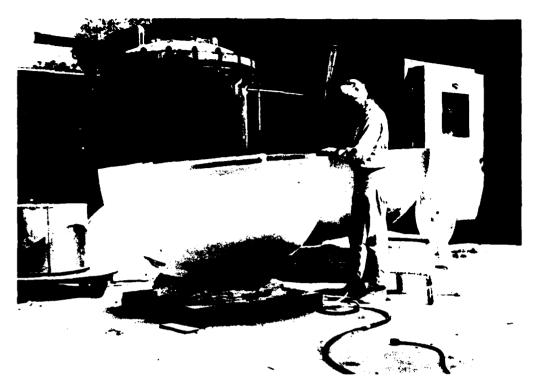


Figure 27. Grinding equatorial edge of casting before machining.

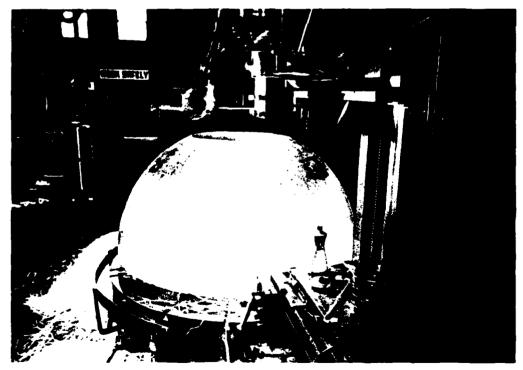


Figure 28. Casting after machining of polar opening.

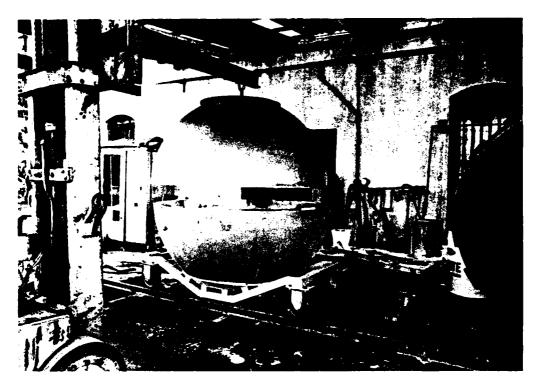


Figure 29. Assembly of castings for bonding together at equatorial joint.

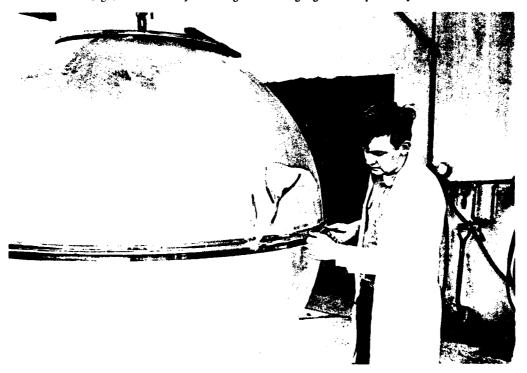


Figure 30. Taped equatorial joint ready for pouring of casting mix.

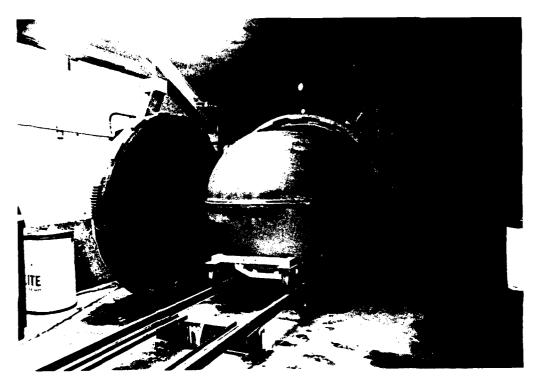


Figure 31. Gelled casting mix in equatorial joint ready for polymerization in autoclave.

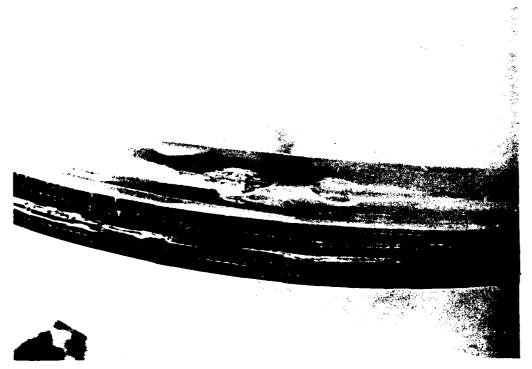


Figure 32. Polymerized equatorial joint with shrinkage voids on inner surface.

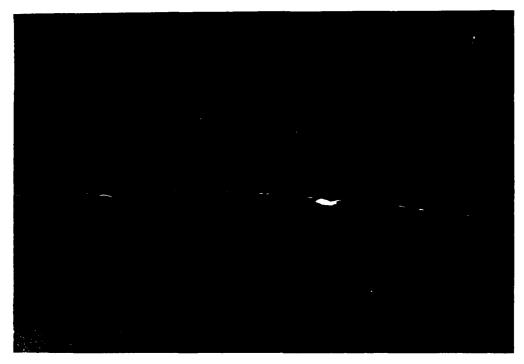
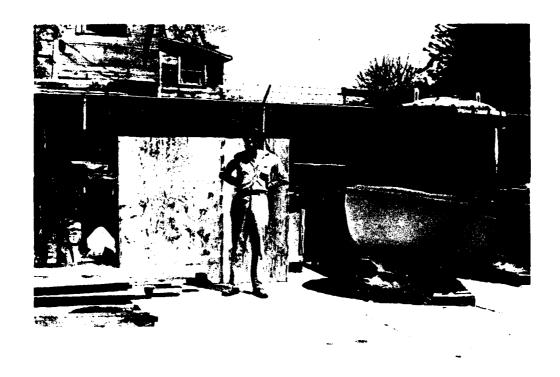
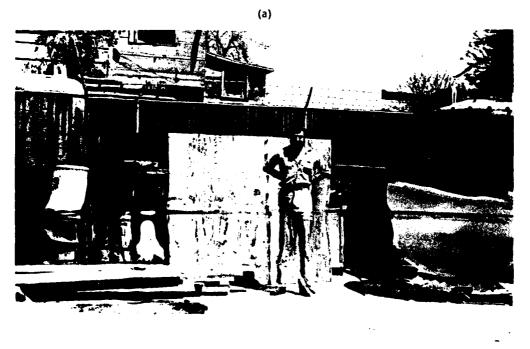


Figure 33. Equatorial joint after repair with room-temperature-polymerizing PS-30 adhesive.



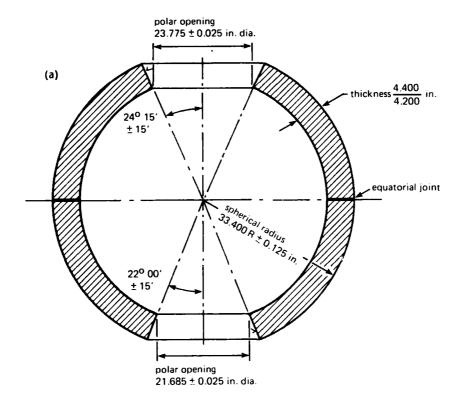
Figure 34. Completed acrylic plastic sphere ready for inspection.





(b)

Figure 35. View through assembled sphere. (a) Above equatorial joint. (b) At equatorial joint.



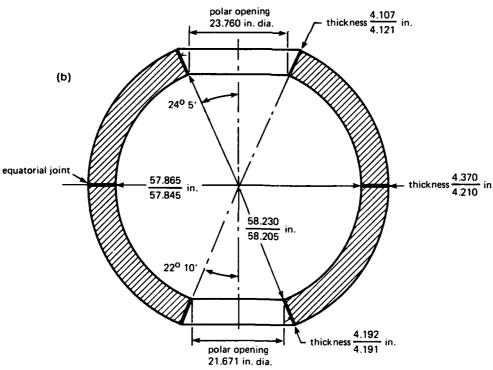


Figure 36. Dimensions of assembled sphere. (a) Specified dimensions. (b) Actual dimensions.

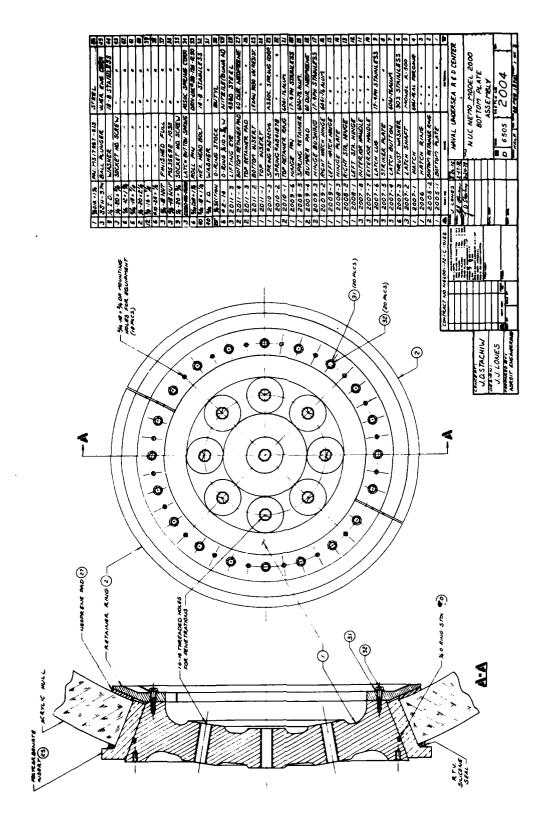


Figure 37. Penetration plate assembly.

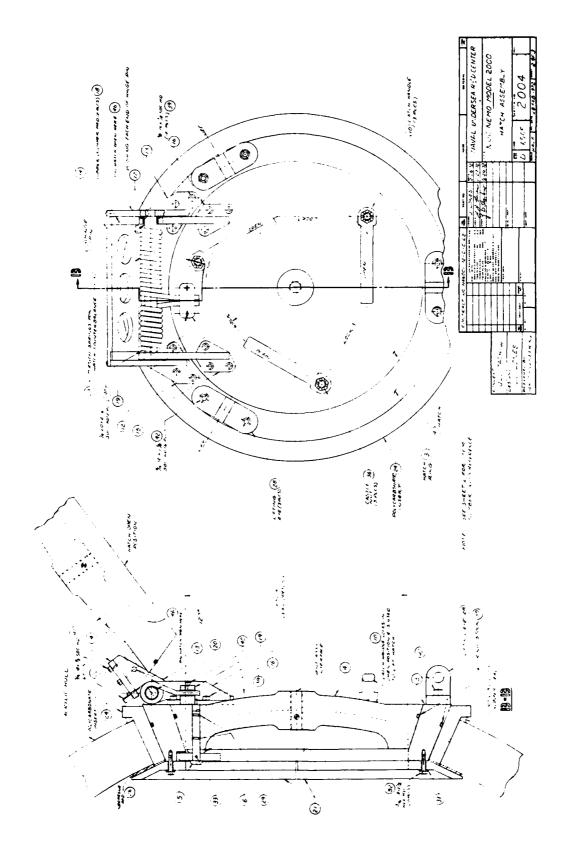


Figure 38. Hatch assembly.

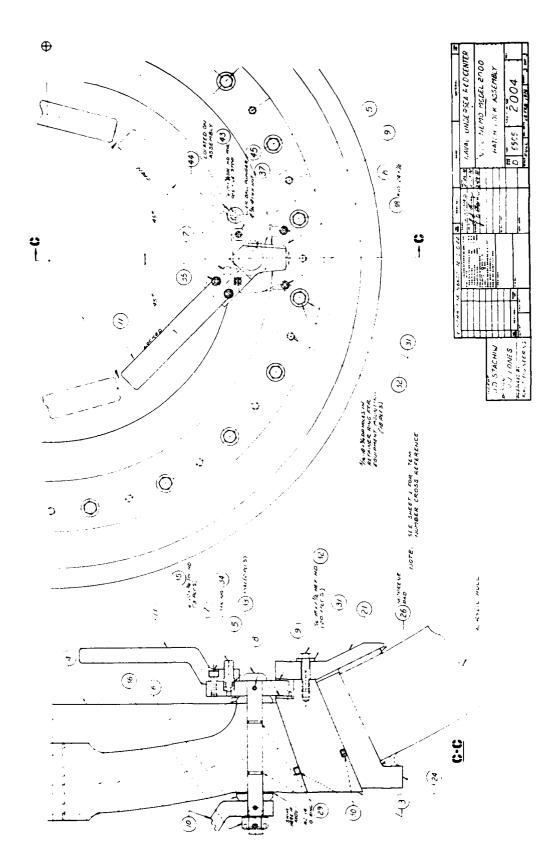


Figure 39. Hatch lock assembly.

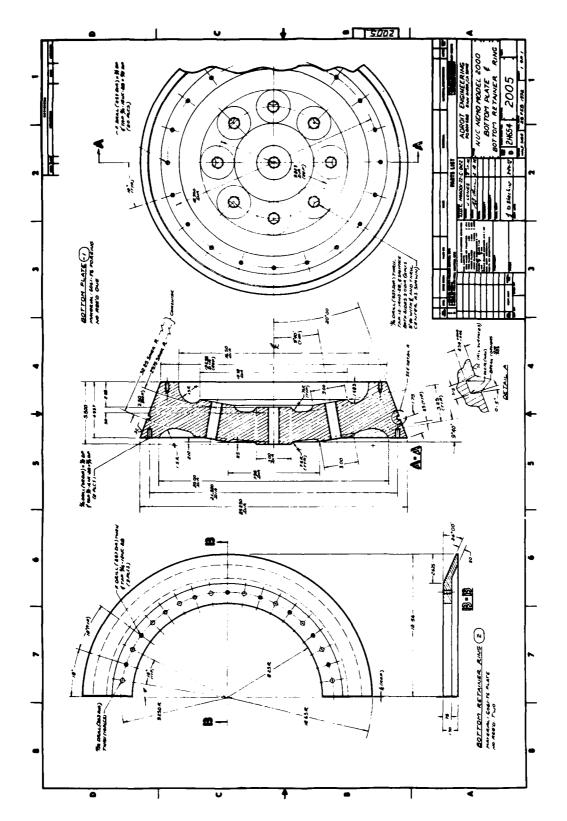


Figure 40. Penetration plate and retainer ring.

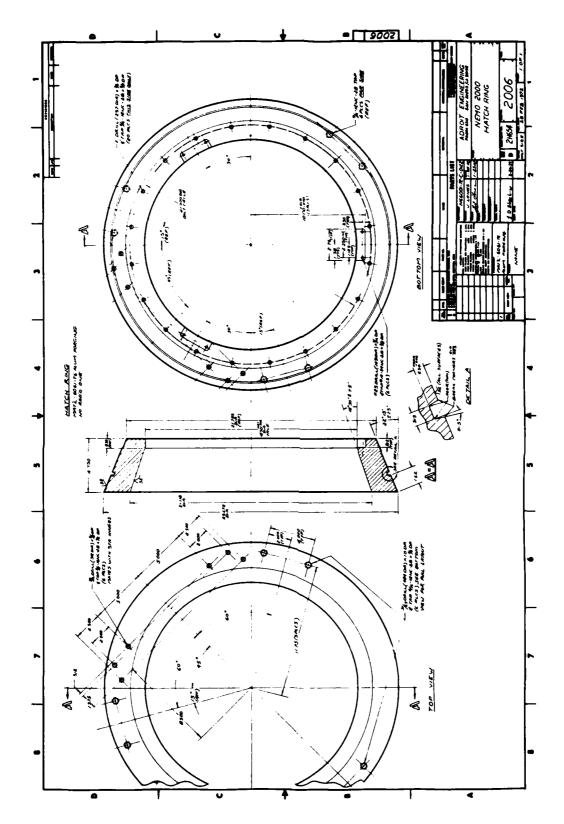


Figure 41. Hatch ring.

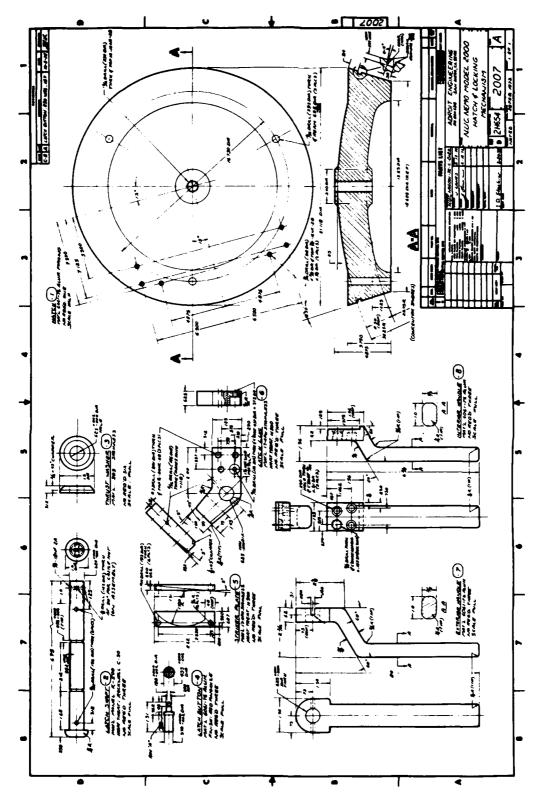


Figure 42. Hatch and locking mechanism.

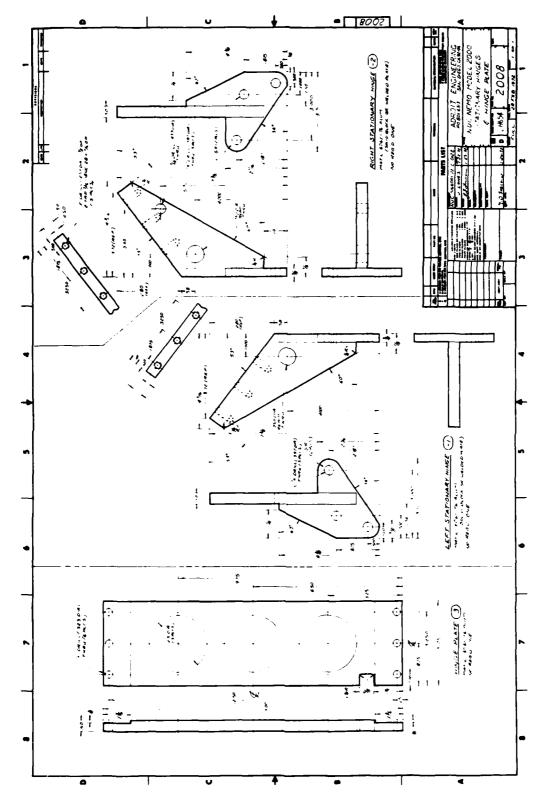


Figure 43. Stationary hinges and hinge plate.

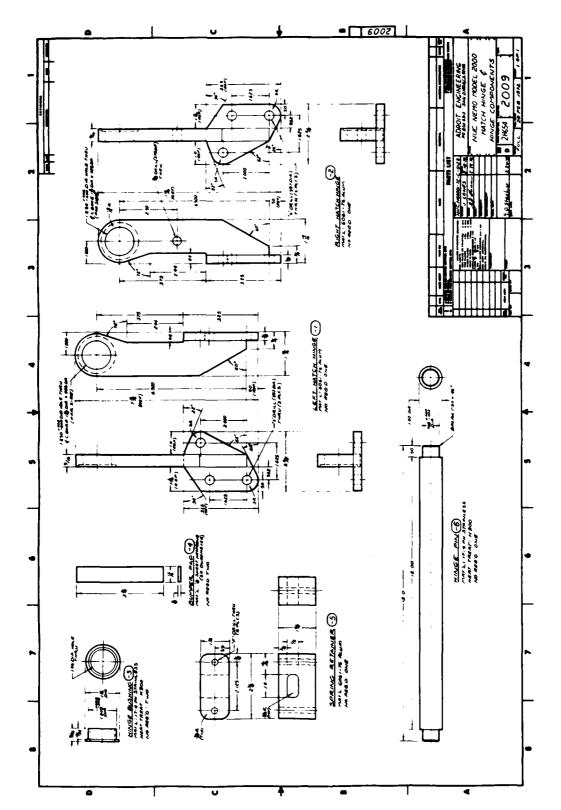


Figure 44. Hatch hinge and hinge components.

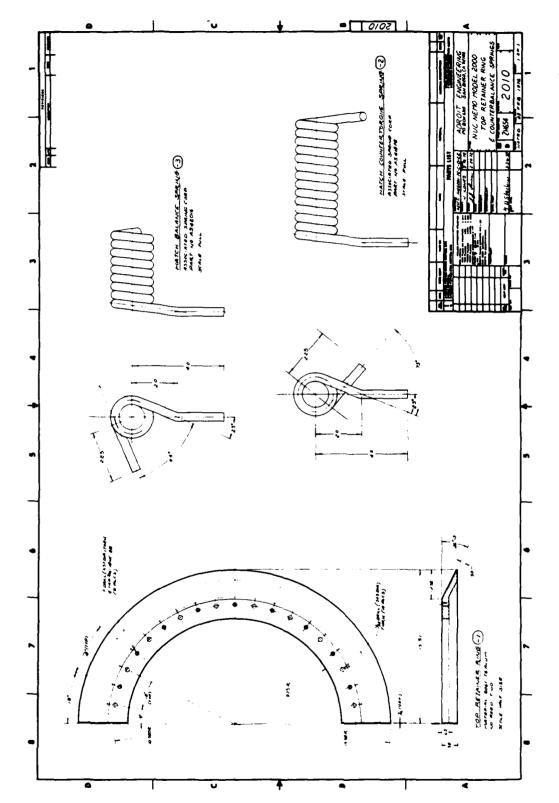


Figure 45. Hatch retainer ring and counterbalance springs.

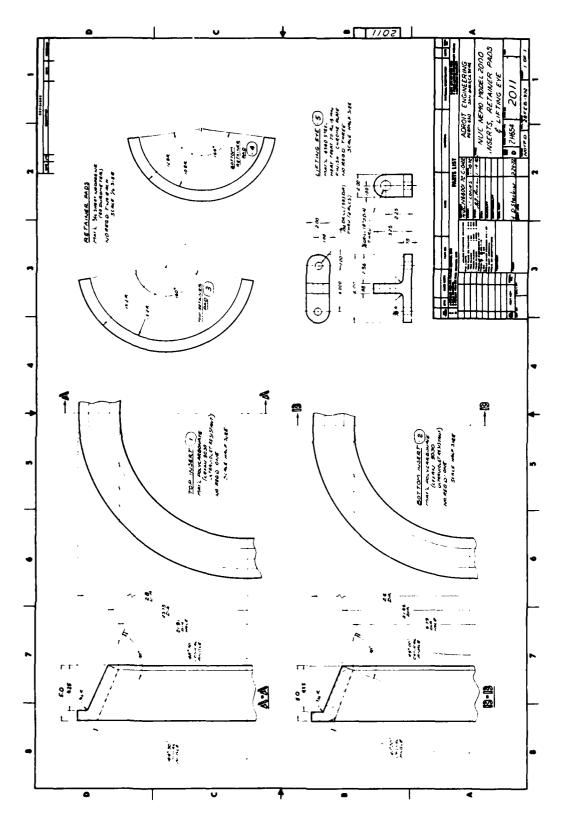
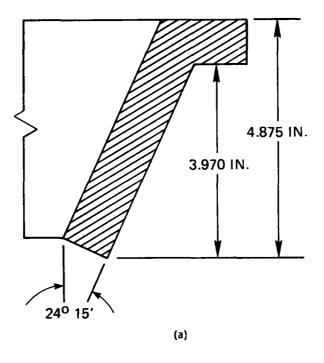


Figure 46. Inserts, retainer pads, and lifting eye.



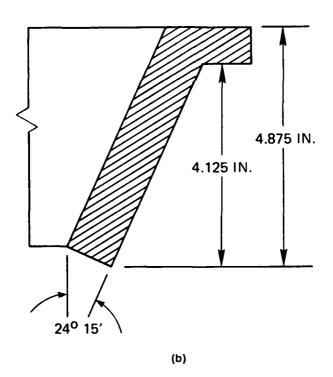


Figure 47. Polycarbonate gasket. (a) Before modification. (b) After modification.

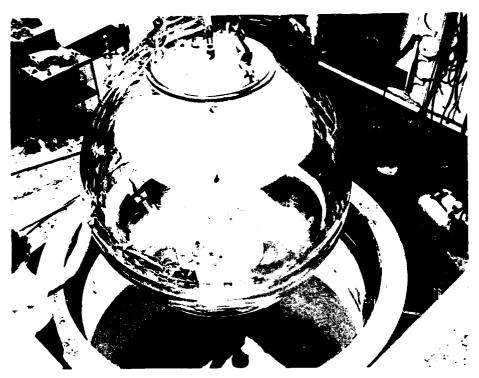
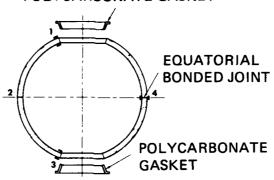


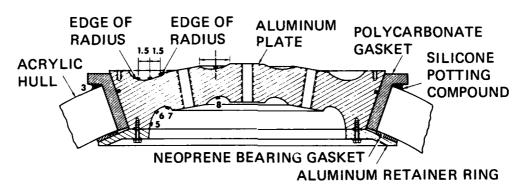
Figure 48. Placing the model 2000B assembly in the 90-inch-diameter pressure vessel at Southwest Research Institute.



Figure 49. Model 2000B assembly instrumented with electrical resistance strain gages and water displacement measurement tubing.

POLYCARBONATE GASKET





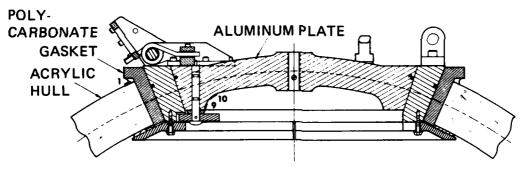


Figure 50. Location of electrical resistance strain gages on model 2000B assembly.

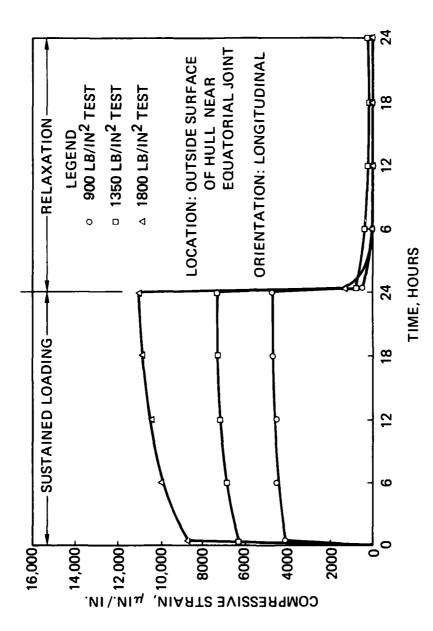


Figure 51. Longitudinal strain on outside surface of hull near equatorial joint (gage location 2).

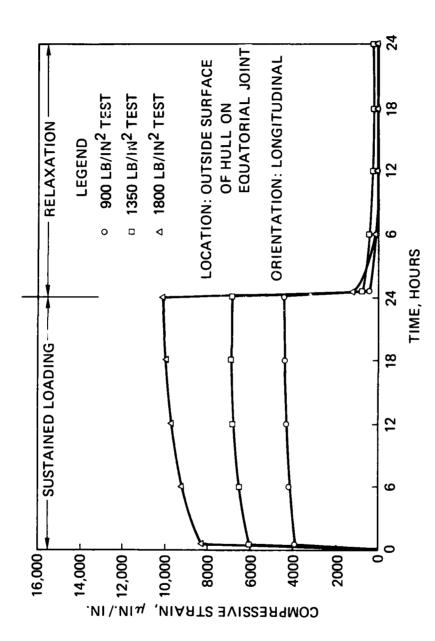


Figure 52. Longitudinal strain on outside surface of hull at equatorial joint (gage location 4).

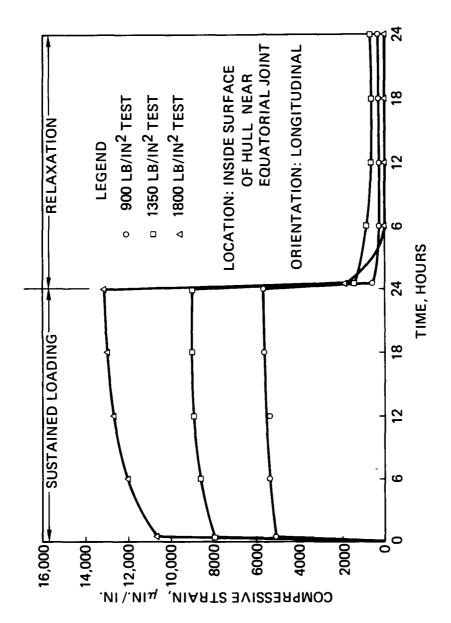


Figure 53. Longitudinal strain on inside surface of hull near equatorial joint (gage location 2).

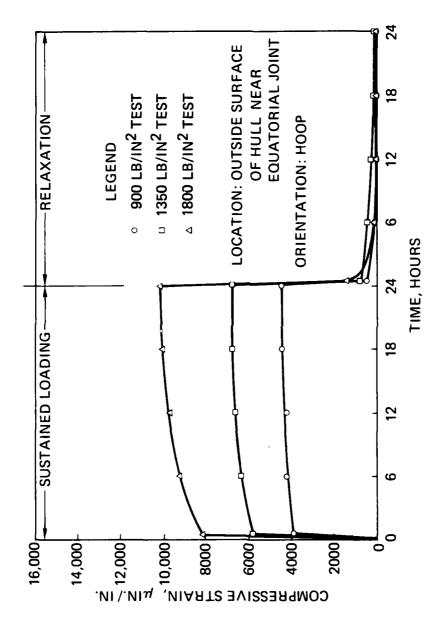


Figure 54. Hoop strain on outside surface of hull near equatorial joint (gage location 2).

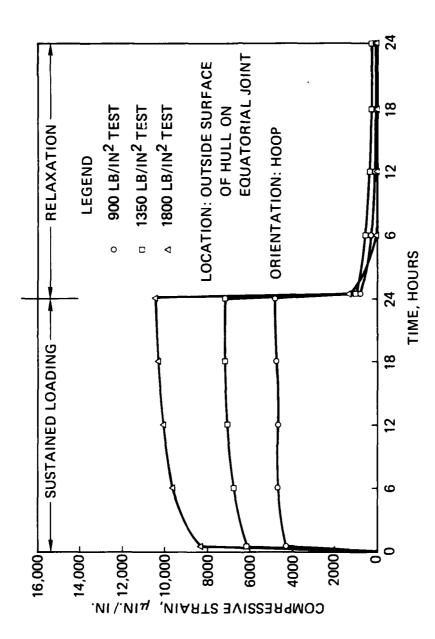


Figure 55. Hoop strain on outside surface of hull at equatorial joint (gage location 4).

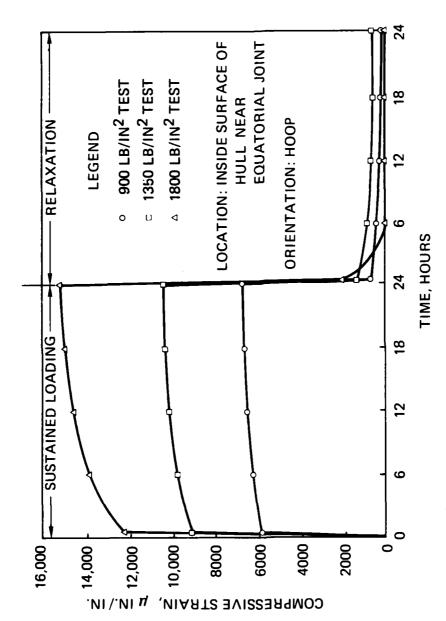


Figure 56. Hoop strain on inside surface of hull near equatorial joint (gage location 2).

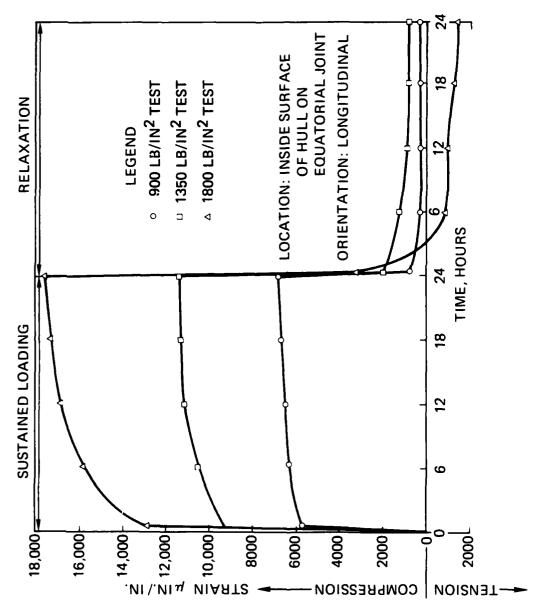


Figure 57. Longitudinal strain on inside surface of hull at equatorial joint (gage location 4).

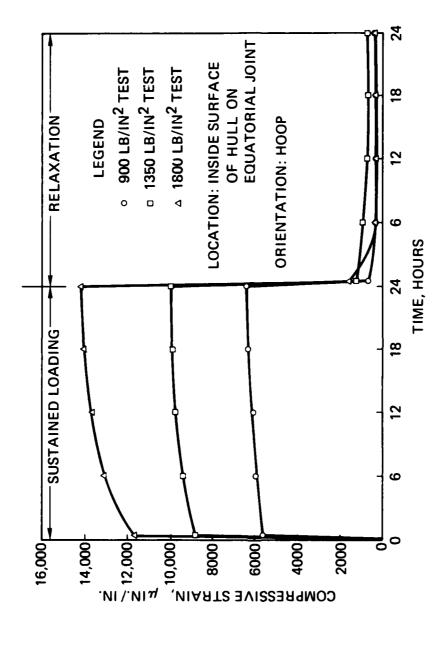


Figure 58. Hoop strain on inside surface of hull at equatorial joint (gage location 4).

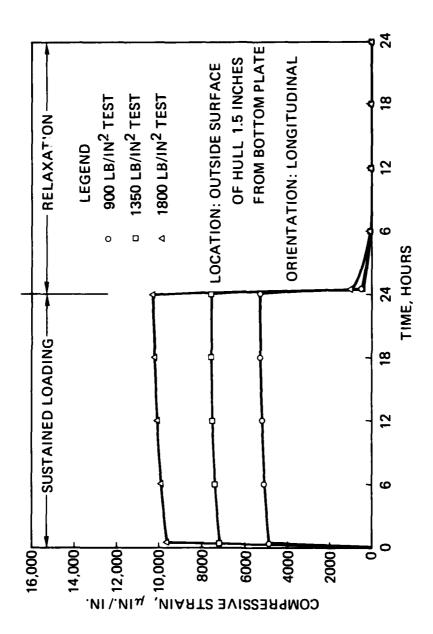


Figure 59. Longitudinal strain on outside surface of hull 1.5 inches from edge of penetration plate (gage location 3).

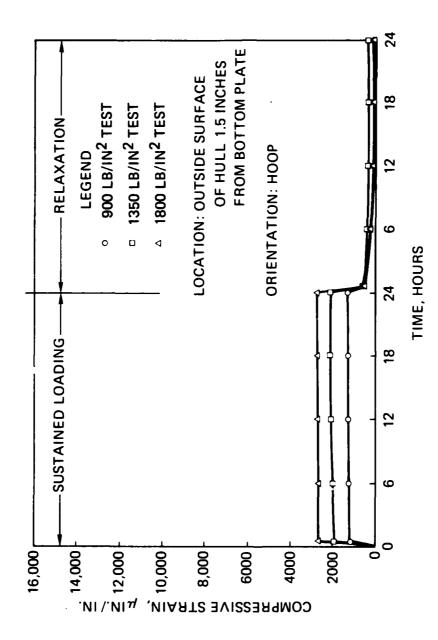


Figure 60. Hoop strain on outside surface of hull 1.5 inches from edge of penetration plate (gage location 3).

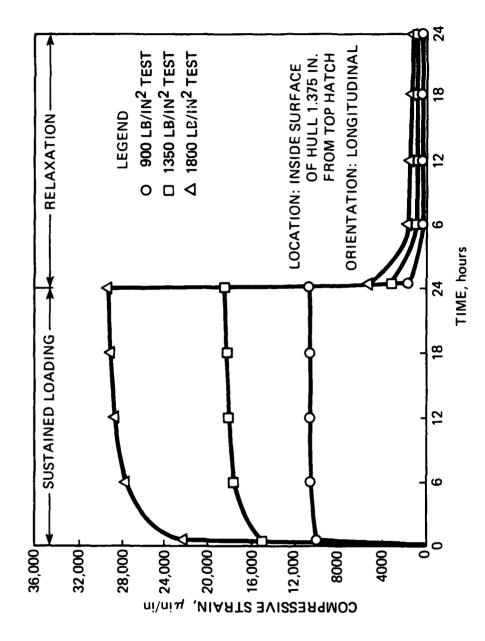


Figure 61. Longitudinal strain on inside surface of hull 1.375 inches from edge of hatch (gage location 1).

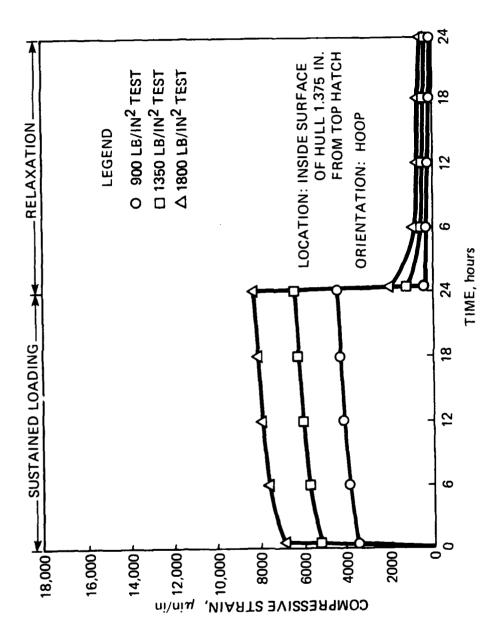


Figure 62. Hoop strain on inside surface of hull 1.375 inches from edge of hatch (gage location 1).

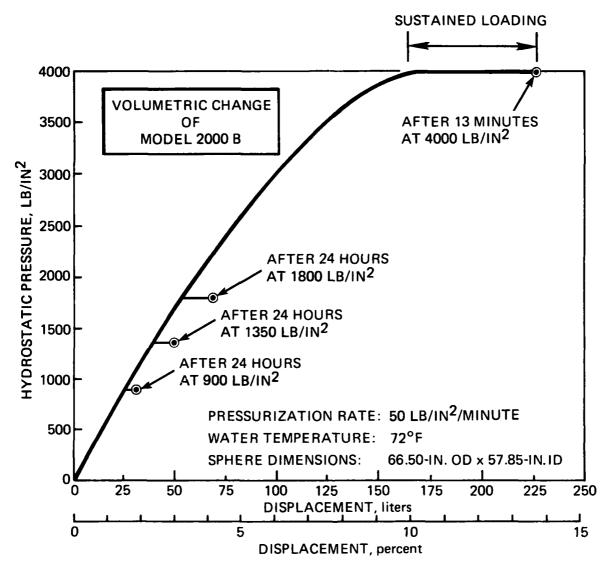


Figure 63. Displacement as a function of time and pressure.



Figure 64. Model 2000B hull assembly after implosion.



Figure 65. Fragments of imploded hull assembly.



Figure 66. Cross section of acrylic plastic bearing surface at bottom polar penetration.

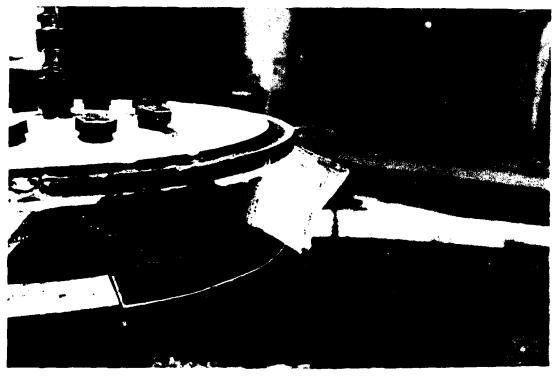


Figure 67. Penetration plate with piece of hull clinging to polycarbonate gasket.

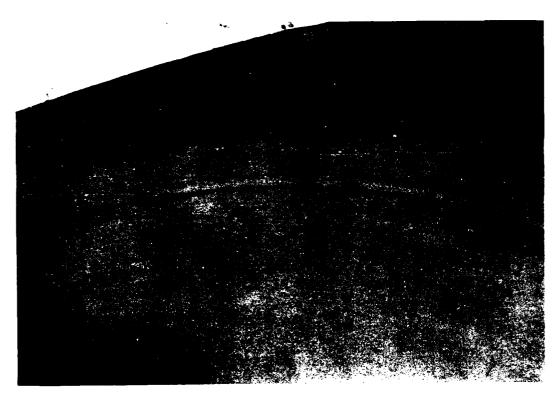
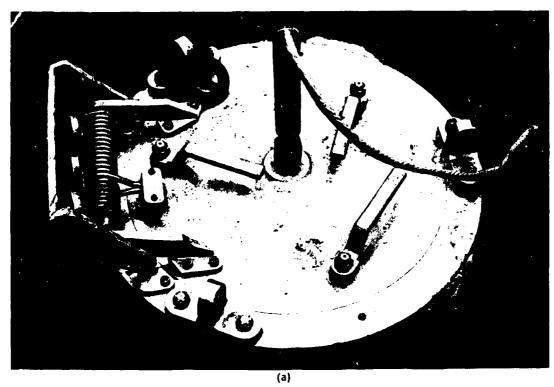


Figure 68. Typical shear cracks in bearing surface of polycarbonate gasket.



Figure 69. Polycarbonate gasket after intrusion into O-ring groove of hatch ring.



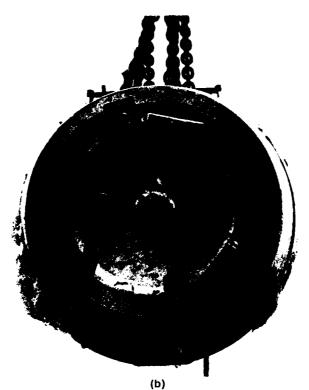


Figure 70. Undamaged aluminum hatch. (a) Top view. (b) Bottom view.

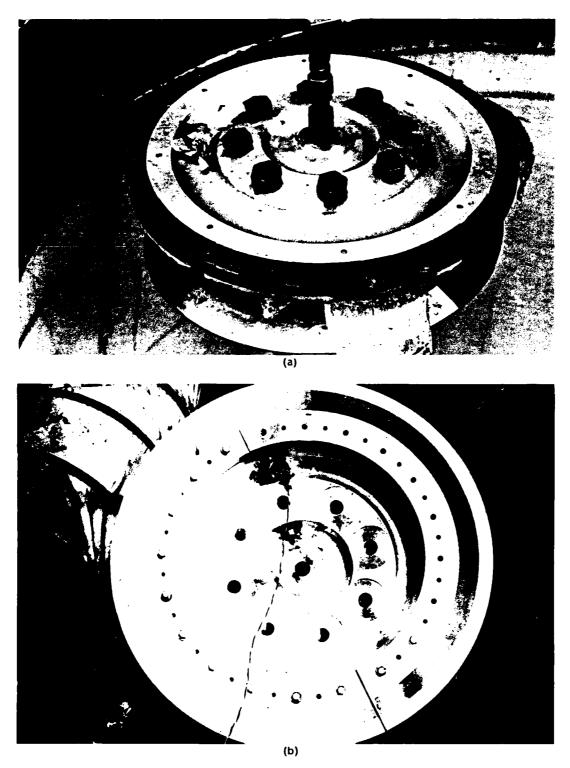


Figure 71. Undamaged aluminum penetration plate. (a) Top view. (b) Bottom view.

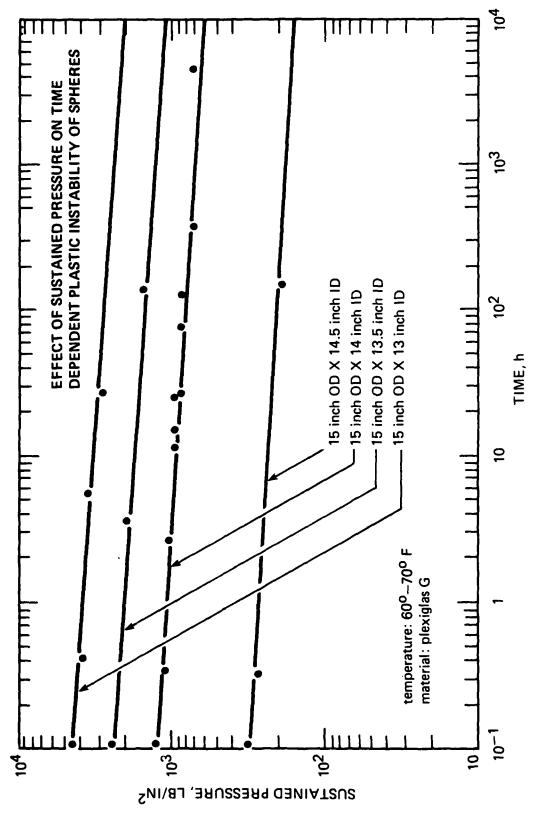


Figure 72. Implosion pressure of model 2000 hull assemblies as a function of time.

APPENDIX A PHYSICAL PROPERTIES OF TEST SPECIMENS

This appendix presents data from the material tests described in the Casting and Inspection section of the report.

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245-8517 245-4551



TEST REPORT

In account with	Date	
Haval Undersea Center	4/9/75	Page 1 of 7 Pages
South Rosecrans Street	W.O. No	P.O. No
San Diego, California 92132	T 11296	N66001-75-II-V390
	Identification	Shipper
	As noted	No number

IDENTIFICATION: Acylic Material

COMPRESSIVE YIELD STRENGTH AND MODULUS Tested as Received at Room Temperature Rate of Test: 0.05 Inch/Minute

TEST METHOD: ASTM D695

SI	PECIMEN	WIDTH INCHES	THICKNESS INCHES	YIELD LOAD POUNDS	COMPRESSIVE YIELD STRENGTH PSI	COMPRESSIVE MODULUS PSI x 100
A	1 2	0.492 0.493	0.490 0.491	3,960 3,770 AVERAGE:	16,400 15,600 16,000	5.3 5.1 5.2
В	1 2	0.504 0.505	0.501 0.504	4,160 4,110	16,500 16,100	5.1 5.0
				AVERAGE: REQUIREMENT:	16,300 15,000	5.1 4.0

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TENSILE STRENGH, MODULUS AND ELONGATION Tested at Room Temperature Rate of Test: 0.05 Inch/Minute

TEST METHOD: ASTM D638

SPECIMEN	THICKNESS INCHES	WIDTH INCHES	MAXIMUM LOAD POUNDS	TENSILE MODULUS PSI x 105	TENSILE STRENGTH PSI	TENSILE ELONGATION
A 1 2	0.236 0.242	0.483 0.483	1,142 1,125	5.1 4.9	10,020 9,620	4.1 3.5
_			AVERAGE:	5.0	9,820	3.8
B 1 2	0.241 0.236	0.482 0.475	1,113 1,060	4.9 5.0	9,580 9,460	4.1 3.6
			AVERAGE: REQUIREMENT:	4.6 4.0 Min	9,520 . 9,000 Min	3.9 . 2.0 Min.

SHEAR STRENGTH

Tested as Received at Room Temperature
Rate of Test: 0.05 Inch/Minute
Punch Diameter: 1.000 Inches

TEST METHOD: ASTM D732

SPECIMEN	THICKNESS INCHES	MAXIMUM LOAD POUNDS	SHEAR STRENGTH PSI
A 1 2 B	0.230 0.225	7,100 6,780 AVERAGE:	9,830 9,590 9,710
1 2	0.204 0.219	6,180 6,900 AVERAGE:	9,640 1,030 9,840
		REQUIREMENT:	8,000 min.

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FLEXURAL STRENGTH AND MODULUS Tested as Received at Room Temperature Rate of Test: 0.2 Inch/Minute Span: 7.180 Inch/Minute

TEST METHOD: ASTM D790

SPECIM	MEN THICKNESS INCHES	WIDTH INCHES	MAXIMUM LOAD POUNDS	TYPE OF FAILURE	FLEXURAL STRENGTH PSI	FLEXURAL MODULUS PSI x 107	
A 1 2	0.492 0.495	0.492 0.498	164.5 172.0	Fracture Fracture	14,900 15,200	4.9 4.9	
				AVERAGE:	15,100	4.9	
B 1 2	0.502 0.502	0.501 0.502	170.5 187.0	Frac ture Frac ture	14,500 15,900	4.9 4.9	
				AVERAGE:	15,200	4.9	
			RE	QUIREMENT:	14,000 Min	1. 4.2 Mi	in.

IZOD IMPACT STRENGTH Tested as Received at Room Temperature

TEST METHOD: ASTM D256, procedure B

SPECIMEN	WIDTH INCHES	BREAKING LOAD INCH-POUNDS	IZOD IMPACT STRENGTH FOOT-LBS/INCH OF NOTCH
A 1 2	0.490 0.494	1.5 1.7	0.26 0.29
		AVERAGE:	0.28
B 1	0.490	1.8	0.31
2	0.493	1.9 AVERAGE:	0.32 0.32
		REQUIREMENT:	0.20 Min.

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DEFORMATION UNDER LOAD
Tested at 24 Hours at 122°F + 20°F

TEST METHOD: ASTM D621

SPECIMEN	THICKNESS INCHES	WIDTH INCHES	LENGTH INCHES	APPLIED LOAD POUNDS	DEFORMATION %
A 1 2	0.500 0.505	0.502 0.495	0.499 0.496 AVERAGE:	1,000 1,000	0.38 0.40 0.39
B 1 2	0.501 0.500	0.500 0.500	0.492 0.494 AVERAGE:	1,000	0.39 0.38 0.38
	1.0 Max.				

SPECIFIC GRAVITY Tested as Received at Room Temperature

TEST METHOD: ASTM D792

APPARENT WEIGHT WEIGHT LOSS SPECIFIC GRAVITY

GRAMS GRAMS 6.7247 1.182

REQUIREMENT: 1.18 - 1.20

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COEFFICIENT OF LINEAR THERMAL EXPANSION

TEST METHOD: ASTM D696

ROCKWELL "M" HARDNESS
Tested as Received at Room Temperature

A

104 105 105 105 105

REQUIREMENT:

AVERAGE:

90

WATER ABSORBTION
Tested at Room Temperature in Distilled Water for 24 Hours

TEST METHOD: ASTM D570

SPECIMEN INITIAL WEIGHT WEIGHT GAIN WATER ABSORBTION GRAMS

A 1 10.3149 0.021 0.20
REQUIREMENT: 0.25 Max.

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HEAT DISTORTION TEMPERATURE Span: 4 Inches

TEST METHOD: ASTM D648

SPECIMEN	THICKNESS INCHES	WIDTH INCHES	APPLIED LOAD POUNDS	HEAT DISTORTION
A 1	0.507	0.505	2,409	216
			REQUIREMENT:	205 Min.

RESISTANCE TO STRESS Tested at 75 F

TEST METHOD: Table I of ASTM methods

SPECIMEN	width	THICKNESS	LOAD	STRESS
	Inches	INCHES	LBS.	PSI
A 1	1.010	0.232	4.53	2,000

Result: There was no visual evidence of crazing, cracking, or other chemical degradation.

Note: Applied Fiber Stress of 2,000 PSI.

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REFRACTIVE INDEX

TEST METHOD: ASTM D542

Refractive Index was taken on sample A, per the above test method and was found to be 1.491.

RESIDUAL MONOMER

Sample A, methyl methacrylate monomer $0.40 \pm 0.005 \ \%$.

Respectfully submitted.

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APPENDIX B STRAIN DATA

The strain data generated by the electrical resistance strain gages mounted on the model 2000B assembly during hydrostatic testing is reproduced in this appendix for the benefit of personnel who may be involved in certification of the model 2000B and engineers interested in the design of acrylic plastic submersibles. The data is presented in order of increasing pressure (900, 1350, 1800, and 4000 lb/in²). The gage numbers correspond to locations on the assembly indicated in figure 50. Some of the readings at interior locations 1 and 3 are irregular, and it is surmised that they were caused by sliding of the compressed neoprene gasket on the gage.

		STRAIN KFUUCTION	HEDUCTION OF A TWO GAGE ROSETTE		
5 → •		84088104	RATIOs , 40	GAGE NO.	1-0015106
LUAD	# G.	ବଧ ପ ଇଧ	SICEA	W Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z	M M M M M M M M M M M M M M M M M M M
c		c	.	c	5
100	005-	527.	-242	15.5	*50
900	•5n0	(155*	M = M =	-357	^
300	004-	-1050	.D. D.	-614	* 3
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500	• 600	0061-	***	P101•	186
900	40	⊕ 2 3 5 C	-257	£ 421 •	のナで
700	-650	00#2•	m + cc +	-1+12	302
800	0690	0026	# T # 1	G + 4 T =	354
000	-450	-3650	-1005	-1862	5 €
900	-625	05/6-	-1014	-1405	£ + +
006	009-	-3750	•1000	0061-	054
6 00	9610	DUBE	•105¢	-1471	471
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GAGE NO. 8 2-0UTSIDE	SIGMA TAU MIN MAX	c		-		<u>⊶ (N) (N)</u>	<u>⊶ (0.100 100</u>	<u>⊶ (N) (N) (N) (N)</u>	~ • • • • • • • • • • • • • • • • • • •	— (a) (b) (b) (c) (c)	~ • • • • • • • • • • • • • • • • • • •	~ N N N N N N N	~ (N	~ € € ™ ™ ™ ™ W W W W W W W	~ (A) (B) (B) (B) (B) (B) (B) (B) (B) (B) (B	~ (A) (B) (B) (B) (B) (B) (B) (B) (B) (B) (B	(A) (B) (B) (B) (B) (B) (B) (B) (B) (B) (B	~ • • • • • • • • • • • • • • • • • • •	~ € W € € W W W W W W W W W W	• A EWEEWWANNAMININ	9 A EWEEWWANNUMNNNNNNNNNNNN
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THAIN REDUCTION OF A THO GAGE HOSETTE

		STRAIN REDUCTION OF	A THU GAGE HOSETTE		
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POISSONS RATIOS . 40
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9-INSIDE -----15058 -4615 19961--17481 -2115 -462 496. 296. -11346 -21453 -22885 460P. -16058 MAX GAGE NO.B -5385 -7637 -23022 -17692 0 0585 10662--12588 -14560 -16978 -20055 -23077 SICHA STRAIN REDUCTION OF A TWO GAGE ROSETTE -10302 -27.7 -6758 -14615 -21691 -35275 -35275 **09* -2747 -2747 -2747 -66933 -69670 E 6 0 6 1 = -56014 -63407 -66923 -41868 SIGNA 90 POISSONS RATIOS 00000 LPP -5650 -250 0 -1300 -1400 0044--3750 -5000 -3150 EPI t = 10,00 0 200 5 500 6 600 1040

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GAGE NO. B 6-INSIDE 46. MAM •1758 •2148 •2555 •2912 04940 -5110 -3654 -1813 1976--5027 -6126 P + S = ----6291 SIGMA STRAIN REDUCTION OF A THU GAGE ROSETTE •2527 •3159 •3516 -3874 -6401 -8187 •645b -3846 -165 -5137 -7033 -6648 -7088 -7637 385 -165 BIGMA 90 POISSONS RATION • 100 • 125 • 150 • 175 00** -400 -150 50 50 50 50 50 50 •50 •50 EP2 -200 -275 •625 •\$00 •525 •575 • 275 • 50 E P 1 £= 10,00 1040

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STRAIN REDUCTION OF A TWO GAGE ROSETTE

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GAGE NO. # 3-0UTSIDE BOART CHATTER CONTROL AND CONTROL CONT TAN STRAIN REDUCTION OF A THO GAGE ROSETTE BIGHA • POISSONS RATIOS EPZ EP1 LUAD

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STAAIN REDUCTION	Polssons	# 8	0	059•	0041-	0082•	05464	0015-	•6 300	0564-	0058-	0096	-10800	00911-	-13000	00681-	-14600	-15000	-14600	-13200	0548-	-7500	000	0566-	004	-200	004•	055	-550	•\$50
		EP1	Đ	-550	000	-1100	0961-	-1650	0561-	-2150	-2400	-2700	-2400	-3550	03/6-	-+500		- 5400	-10500	00001-		0564+	•	-6300		057+-	3800	000F =	-2700	-2500
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GAGE NO. # 7-0UTSIDE TAN 2440 444 8848 BIGNA STRAIN REDUCTION OF A TWO GAGE ROSETTE -3840 BIGMA 6. POISSONS RATIOS £ P 2 EP1 LOAD Fs 10.00

F 10.00		POISSONS RATIOS	08.		GAGE NO. # 7-INGIDE
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OF A THO GAGE ROBETTE	RATIOM .30	OIGHA Max	D	5 + 3 E +	-4121	-3925 -	000.**	-545-	-7637	-7167	9906-	いまです!	0456	5/2010	•12253	-12967	94867	-13352	-12088	-12088	+4ETT-	+579=	4564	-2637	-3052	-2857	m	E 5 5 7 -	*	CE 920
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GAGE NO. # 10-INBIDE TAR TAR BICHA STRAIN REDUCTION OF A TWO GAGE ROSETTE BIGNA 30 POISSONS RATION EP1 LOAD E# 10,00

GAGE NO. # 2-1-SIDE BMCESTCOBETCOSEMPLANT TAU 67 til 8 1 Gr A STRAIN REDUCTION OF A TWO GAGE ROSETTE SIGWA POISSONS RATIUS 20000 EP1 LOAD •

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GAGE AG. 8-0UTSIVE A A C 15017 15400 15400 -4657 STRAIN REDUCTION OF A TAU GAGE RUSETTE SIGMA • POISSONS RATIOS EP2 EP1 ? 1040

	GASE NO. # 3-UUTSIDE	3× 44 5×	0		162	E + F	246	₽ 8 ±	204	F 7.5	543	657	5 N	721		7 6		1 M T T T T T T T T T T T T T T T T T T	# ±01	70.07	1057	1006	er nu	727		***	M + 1	T	*** !	2 mil 8	S. T.	~~	e i	
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GAGE NO. # S-INSIDE DE MENTE CONTRACTOR DE LA COMBRES DE CARDON DE LA CARDON DEL CARDON DE LA CARDON DEL CARDON DE LA CARDON DE L STRAIN REDUCTION OF A TWO GAGE ROSETTE COMPAGE COUNTRANCION MEDICA PRO SE A CALLA MA CALLA PRO CALLA PROPERTO CALLA SIGNA 30 POISSONS RATIOS EP1 1040 Em 10,00

		STRAIN REDUCTION OF	A THO GAGE ROSETTE		
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700	5	0	-16+8	56 **	-677
00	2	0	9512		24C
00	13	0	-16+8	56 72	-573
9	2	0	-2146	P 5 9 P	-169
2	25	0	C+C2-	+28 -	-462
2	5	0	C+C2=	*~ *	296+
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5	•	0	95M+1	-1314	-1538
8	\$	0	SFR+1	5161-	
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8	•	001-	5445	87 + 2 =	+517-
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SPHERICAL ACRYLIC PLASTIC HULLS UNDER EXTERNAL EXPLOSIVE LOADING

by J. D. Stachiw Ocean Technology Department March 1976



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NAVAL UNDERSEA CENTER, SAN DIEGO, CA. 92132

AN ACTIVITY OF THE NAVAL MATERIAL COMMAND

R. B. GILCHRIST, CAPT, USN

HOWARD L. BLOOD, PhD

Commander

Technical Director

ADMINISTRATIVE INFORMATION

This report describes research performed between June 1973 and June 1975 as part of the investigation into man-rated transparent submersibles. The program was funded under a Project Order from the Naval Material Command through the Independent Research and Independent Exploratory Development program at the Naval Undersea Center under Subproject Task Area Number ZF-61-412-001.

Released by

H. R. Talkington, Head, Ocean Technology Department

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The tests have shown that an acrylic sphere will fracture in the 0- to 50-ft depth range under dynamic peak overpressures that are smaller in magnitude than static pressures required for general implosion of the sphere. At the depth that is equal to 0.2 of static implosion pressure, the magnitude of dynamic peak overpressures must be in excess of the static implosion pressure before fracture of the acrylic sphere is initiated.

Fractures were generally initiated on the internal surface of the sphere at two locations; (a) at a point closest to the explosive and (b) at a point farthest from the explosive. The fractures were generally in the shape of a star.

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SUMMARY

PROBLEM

Manned submersibles with spherical acrylic plastic hulls have been known since the NEMO hull was designed in 1961 to provide greater panoramic vision at lower cost and weight-to-displacement ratio than steel hulls (of the same shape, size, and depth capability) equipped with many small viewports. Several submersibles with NEMO-type hulls have been built since that time by the U. S. Navy. After more than 5 years of service, the acrylic hulls have been found to be virtually maintenance free and have shown no sign of weathering. There is, however one area of uncertainty that currently restricts the choice of missions for submersibles with acrylic hulls; it is not known how resistant the spherical acrylic hull is to hydrodynamic impulse loadings generated by explosive-actuated tools like cable cutters, stud guns, explosive anchors, corers, and others. If the resistance of NEMO-type hulls to underwater explosions were known, acrylic submersibles could be utilized in missions for which explosive tools are mandatory for meeting the mission objective.

RESULTS

An exploratory test program has shown that spherical hulls of acrylic plastic can wighstand dynamic impulses of considerable magnitude before fracture of the hull is initiated. Increasing the depth of operations was found to increase significantly the resistance of the acrylic hull to fracture initiation by dynamic impulses. The NEMO Mod 2000 hull, with a 66-in. outside diameter and 4-in. shell thickness, has been found to withstand explosion-generated peak dynamic overpressure of 4,927 psi without initiation of fracture.

RECOMMENDATION

Manned submersibles with NEMO-type spherical hulls of acrylic plastic may be safely utilized in search, rescue, salvage, and work missions where explosive-actuated work tools are routinely utilized for achievement of mission objectives, provided that the peak dynamic overpressure impinging on the acrylic hull is less than 25 percent of static implosion pressure of the hull.

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INTRODUCTION

Underwater visibility is extremely important to crews of submersibles engaged in search, salvage, or work missions. Panoramic visibility can be provided in many ways, but large spherical acrylic plastic windows are considered to provide the most cost effective and reliable way of meeting this operational requirement (Ref. 1). An even better way is to use a transparent acrylic plastic hull of spherical shape (Ref. 2). Not only does it provide unlimited visibility in all directions, but it also generates a significant amount of buoyancy. Furthermore, such a hull is non-magnetic, provides unsurpassed thermal and sound insulation, and is virtually maintenance free. Because of its transparency, it can be inspected for incipient cracks visually by its crew at any time. This feature alone makes acrylic pressure hulls inherently safer than those fabricated from opaque materials that require expensive and time-consuming inspection procedures for detection of cracks.

The performance of spherical acrylic pressure hulls under short-term, long-term, and cyclic pressure loadings has been experimentally established over the years by U. S. Navy (Refs. 3-12) so that submersibles with spherical acrylic plastic hulls can be economically built and operated in the 0- to 3300-ft depth range. Several submersibles with acrylic plastic hulls have been already built and are operating in that depth range (Refs. 13, 14, 15). Their performance record is excellent, and acrylic plastic hulls that have been exposed to sun, seawater, and heat for 6 years still show no signs of deterioration.

In many undersea missions, the submersible may be exposed to explosions generated intentionally by, for example the firing of a stud gun or cable cutter during a typical underwater work sequence. During some missions, the submersible may be subjected to severe explosions unintentionally, e.g., during recovery or neutralization of underwater ordnance (Ref. 1). The effect of underwater explosions on submersible hulls made of steel is fairly well understood and the resistance to dynamic overpressures can be readily calculated. This is not the case with acrylic plastic pressure hulls or large spherical sector windows.

This report summarizes findings from the first exploratory study conducted by the U. S. Navy on the effect of underwater explosions on acrylic plastic spherical hulls of NEMO-type design and construction.

DESCRIPTION OF STUDY

The objective of the study was to provide operators of existing acrylic plastic submersibles (NEMO, Makakai, Johnson-Sea-Link I and Johnson-Sea-Link II) with operational guidelines for missions in which the submersible may be exposed to underwater explosions.

The approach selected for meeting the objective of the study was experimental in nature. It was felt that the experimental approach was, in this case, more direct, more reliable, and less expensive than an analytical approach, which would, subsequently, have to be experimentally validated before it could be used with confidence by operators of submersibles.

The scope of the study was limited to spherical hulls of NEMO design and construction with 1000 and 3000 ft maximum operational depths. Only two sizes of hulls were to be tested: the 66-in.-OD full-size and the 15-in.-OD scale-size spheres. The NEMO-type design uses a sphere with two penetrations located at opposite poles of the sphere; each penetration is closed with a metallic closure equipped with a conical seating surface. In NEMO-type construction, the sphere is assembled from 12 spherical-sector pentagons bonded together with self-polymerizing acrylic cement.

EXPERIMENTAL DESIGN

TEST SPECIMENS

Two NEMO capsules, one 66-in.-OD full-size and six 15-in.-OD scale-size, served as test specimens (Figs. 1 and 2). Both NEMO Mod 600 with 1000-ft operational depth and NEMO Mod 2000 with 3000-ft operational depth were utilized (Table 1 and Appendix A). Both the full-size and scale-size NEMO capsules have been exposed previously to cyclic fatigue testing and thus can be considered to be equivalent to submersibles with several years of field service (Ref. 11).

All specimens were fabricated from Plexiglas G, whose physical properties met the U. S. Navy and ASME requirements. The spherical hulls were assembled in every case from thermoformed spherical pentagons that were bonded together with either PS-18 or PS-30 self-polymerizing adhesive. The scale-size hulls had polar inserts machined either from stainless steel or titanium (Figs. 3-6), while the full-size hull utilized aluminum, both for top hatch and bottom penetration plate (Fig. 7).

TEST ARRANGEMENT

Scale-Size NEMO Capsules

The testing of scale-size models took place in a 30-in.-ID pressure vessel, 20 ft long, located at the Southwest Research Institute. The test specimen was placed in a test jig that held it approximately 120 in. below the end closure and 120 in. above the bottom closure (Fig. 8). To prevent point contact between the test specimen and the steel test jig, the specimen was wrapped in a wire net that, in turn, was fastened to the three longitudinal members of the test jig.

The explosive was suspended above the test specimen by means of two horizontal wires stretched between the longitudinal members of the test jig. It was centered directly above the center of the test specimen, with the standoff being defined as the distance between the center of explosive and the outer surface of the test specimen facing the charge (Fig. 9).

The instrumentation consisted solely of two tourmaline piezoelectric transducers for measurement of dynamic overpressures. The transducers were positioned adjacent to the model and were the same distance from the explosive charge as was the outer surface of the model. Transducer response was transmitted through differential amplifiers and displayed on a dual-beam oscilloscope, where it was photographed. It was considered advantageous to

use two transducer systems so that the validity of pulse characteristics could be ascertained by noting the similarity in response of the two independent monitoring systems.

The output of the piezoelectric transducers was displayed on an oscilloscope and recorded photographically by a Polaroid camera. The oscilloscope was triggered by a small breakwire wrapped around the charge. The breaking of the wire by the explosion generated a pulse which energized the oscilloscope for a single sweep. In initial tests, a time-delay pulse generator was not available, so the sweep speed of the oscilloscope had to be such that transducer response to shock overpressure was appropriately displayed during the single sweep. In later tests, by using a delayed trigger pulse, it was possible to eliminate the initial straight-line portion of the display and obtain greater detail of shock-pulse characteristics.

Full-Size NEMO Capsules

Testing of the full-size NEMO Mod 2000 capsule took place in a 12-ft-diameter, 100-ft-deep, water-filled well located on the premises of Southwest Research Institute (Fig. 10). The test specimen was securely wrapped with Nylon webbing and suspended within a steel cage by means of steel cables (Fig. 11). The cage itself was kept suspended at 50 ft depth by means of a cable attached to a large mobile crane.

For the first three shots, the explosive (Fig. 12) was held above the test specimen. For the subsequent two tests it was placed below the test specimen. Changing the location of the explosive was made necessary by the generation of large downward force upon the crane by pressure waves radiating from explosive held above the specimen. When the explosive was placed below the specimen, the pressure wave would tend to decrease the load on the crane, rather than increase it.

Instrumentation consisted of two electric resistance strain gages and two pressuresensitive transducers. The strain gages were mounted on the interior of the hull midway between the polar inserts and directly below the explosive.

The pressure transducers, PCB Model 113A23 acceleration-compensated ultra-rigid quartz element pressure probes with built-in amplifiers, were positioned the same distance from the explosive charge as the apex of the test specimen (Fig. 13). Pressure gage outputs were displayed on a Tektronix Model 454 split-beam oscilloscope, and strain gage outputs were displayed on a Tektronix Model 502 dual-beam oscilloscope. Both scopes were set to trigger in a single sweep mode, with the trace being recorded on Polaroid film. A small-diameter breakwire, wrapped around the charge, broke when the charge detonated, thereby creating a voltage change and triggering the oscilloscopes; the scope sweeps were delayed by a time slightly less than the time required for an acoustic pulse in water to travel the distance between the charge and the apex of the model. *

TEST PROCEDURE

Scale-Size NEMO Capsules

Each of the scale-size NEMO capsules were tested individually. Since the objective of the testing program for scale-size capsules was to determine the effect of the depth and capsule shell thickness on the resistance of capsules to damage caused by dynamic overpressure, some of the test parameters, like sizes of explosive charges and standoff distances, were

kept constant. The sizes of charges chosen were 1.1, 8.2 and 14.6 grams. Standoff distances were set at 48, 36, 24 and 12 in.

The procedure (Tables 2 and 3) followed during testing of any given test specimen was to start with the smallest charge (1.1 grams) placed at the longest standoff distance (48 in.). If no damage to the test specimen was observed, an identical charge would be placed at the next shorter standoff distance (36 in.). The standoff distances chosen for each shot were progressively shorter until the shortest standoff (12 in.) was reached.

If the smallest charge did not initiate failure of the test specimen at the shortest standoff distance, the next larger charge (8.2 grams) would be placed at the longest standoff. The larger charges would be set off following the test procedure already described for the smallest charge. If the larger charge did not initiate cracks at the shortest standoff, the series of tests would be repeated again, utilizing, however, the largest charge (14.6 grams).

The 15-in.-OD by 14-in.-ID scale-size NEMO test specimens were tested at simulated depths of 10, 100, or 1000 ft. The 10-ft depth represented the typical surface cruising depth of a submersible, while 1000 ft represented maximum operational depth of NEMO capsules with $t/R_0 = 0.067$ ratio.

The 15-in.-OD by 13-in.-ID scale-size NEMO test specimens were tested at depths of 10, 100, or 2000 ft. Here again, 10 ft represented the typical surface cruising depth, while 1000 and 2000 ft represented depths of typical deep submergence operational missions.

Full-Size NEMO Capsules

The test procedure for full-size capsules (Table 4) differed from the test procedure used for scale-size capsules. While for scale-size capsules both the size of the charge and the standoff distance were experimental variables, for the full-size capsule only the charge size was varied, while the standoff was held constant at 52.8 in. This standoff distance was determined by multiplying the shortest standoff distance of 12 in. by 66/15, the ratio representing the relationship between the size of the full-size NEMO and that of the scale-size NEMO.

The charge weights used against the full-size NEMO capsule were 1.1, 5.6, 14.5, 169.9, 387.8, and 688.6 grams. The first three charges were of the same weight as those used in the explosive testing of scale-size NEMO capsules. They were used primarily to calibrate pressure transducers and strain gage readout equipment. The last three charges were scaled-up versions of charges previously used against scale-model NEMOs. Thus, the 169.9-gram charge is the scaled-up version of the 1.1-gram charge, the 387.8-gram charge is the scaled-up version of 8.25-gram charge. The scaled-up charges were supposed to generate the same peak overpressures on the full-scale NEMO from a 0.8R_O standoff as were generated previously on the scale-size NEMO capsules from a 0.8r_O standoff by 1.1-, 4.6-, and 8.2-gram charges.*

^{*}R_o - external radius of the full-size NEMO

r - external radius of the scale-size NEMO

TEST OBSERVATIONS

Scale-Size NEMO Capsules

The testing of the scale-size NEMO capsules was very destructive to the 30-in. pressure vessel in which the testing was conducted. Seals in the vessel end closures as well as hydraulic piping were repeatedly damaged. Because of it, seals and hydraulic fittings had to be replaced every second or third shot.

Pressure transducers were also damaged repeatedly. After several days of testing, the project ran out of transducers, and further shots were conducted without any instrumentation. Thus, for some of the shots during which the capsules failed, both the peak overpressure and the impulse intensity had to be calculated. There is, however, a very high confidence in the calculated values, since it was found that during the shots in which instrumentation functioned, the correlation between calculated and experimental values was quite good (Figs. 14, 15, 16, and 17).

Failure of model-size NEMO capsules was manifested by formation of either tensileor flexure-type cracks. As a rule, the flexure cracks were present on the interior surface of the equator directly facing the charge, on the opposite side, or on both sides, while the tensile cracks extended radially from the edges of penetrations. If the underwater explosion was severe, there would be several long flexure-type cracks joined together in a form of a star, very similar in appearance to the pattern of cracks observed in spherical sector windows under point impact loading (Ref. 16). Severe explosions would also generate tensile meridional cracks at the penetrations.

<u>Light damage</u> was observed on test specimens J and 26 (Figs. 18 and 19). In both cases, there were only one or two small flexure cracks at the equator, no leakage of water took place and the capsule was considered to have withstood the explosion without endangering its potential cargo. These capsules could have completed their mission successfully.

Medium severe damage was noted on test specimens M and 24 (Figs. 20 and 21). In both cases, there were several short flexure cracks present at the equator and at least one long tensile crack at the penetration. Only a few drops of water leaked into the interiors of the capsules, but not in sufficient quantity to endanger the potential cargo. Still, the missions of the capsules would have to be terminated immediately to avoid endangering the crew.

Very severe damage was observed on test specimens K and 25 (Figs. 22 and 23). In both cases, a large star-shaped flexure crack at the equator and several tensile cracks radiating from penetrations were produced. Because of the many cracks, water leaked into the interiors of these two capsules. There would have been severe jeopardy for any cargo. It is very probable that capsules in such condition could not return from their missions since they would fill with water prior to reaching the mother ship.

Full-Size NEMO Capsule

The full-size NEMO capsule withstood all the explosions without initiation of cracks in the acrylic plastic. However, during the last three shots, the capsule was torn loose from its fastenings. On the last shot, the capsule broke free of its 0.25-in. steel cable netting and

rose rapidly to the surface of the well shaft, where it struck a protruding steel beam. The point impact broke off a large chip from the capsule surface, thus terminating any further tests on this capsule (Fig. 24).

Dynamic strains measured on the interior of the capsule facing the 387.8-gram charge indicated considerable tension immediately followed by compression of approximately the same magnitude (Fig. 25). Still, the strains were not of such magnitude as to suggest failure during the following 688.6-gm shot.

Dynamic pressure readings were obtained only with the initial two small charges (Figs. 26 and 27). No experimental pressure readings were obtained with the following four larger charges because the breaking loose of the capsule destroyed, in every case, the pressure pickups and associated wiring. However, such a good correlation was obtained between the experimental and calculated peak overpressure values during the initial shots that peak overpressures for the last three shots could be calculated with confidence.

DISCUSSION OF TEST RESULTS

Although the data generated during the testing program are far from complete, several definite relationships between the force of explosion and capsule's resistance to failure can be formulated.

EFFECT OF SHELL THICKNESS

It appears that the resistance to fracture of acrylic plastic spheres subjected to underwater explosions is directly related to shell thickness, provided that the method of construction and outside radius remain the same. This postulate is based on the observation that to initiate cracking in 1-in.-thick scale-size NEMO capsules required a unit impulse and peak dynamic overpressure twice as large as those required to produce similar results in 0.5-in.-thick capsules. Both tests were conducted at the same depth. For example, test specimen No. 26 with a 0.5-in.-thick wall failed at 1000 ft under 0.1 psi.-sec unit impulse and 2816 psi peak dynamic overpressure, while test specimen No. K with 1.0-in.-thick wall required 0.206 psi-sec unit impulse and 6176 psi peak dynamic overpressure to initiate cracking at the same depth. A similar relationship can be seen, although less clearly, between specimen No. 25 and No. M.

EFFECT OF DEPTH

The data show quite clearly that the resistance to fracture of acrylic plastic spheres subjected to underwater explosions increases significantly with depth. This conclusion is based on the observation that it required a 3 to 5 times larger peak overpressure and unit impulse to fracture an identical test specimen at 1000 ft depth than it did at 10 ft. For example, test specimen No. M failed at 10 ft depth under 0.045 psi-sec unit impulse, and 1434 psi peak dynamic overpressure, while test specimen No. K at 1000 ft depth required 0.206 psi-sec unit impulse and 6176 psi peak dynamic overpressure to generate a fracture. Similar relationship can be seen between specimen No. 25 and No. 26.

EFFECT OF SCALING

There are insufficient experimental data to establish the validity of using scale-size models for determining the resistance of full-size NEMO capsules to underwater explosions. The few data points generated during the study seem to indicate, however, that extrapolating data from scale-size models is on the conservative side and, thus, acceptable. This conclusion is based on the observation that the full-size NEMO capsule did not crack when subjected to peak dynamic overpressure of 4927 psi generated by a 688.56-gm charge with 0.8R₀ standoff, while the same peak dynamic overpressure generated by a scaled-down charge of 8.2 gm with 0.8 r₀ standoff would, without a doubt, have cracked the 15-in.-OD by 13-in.-ID scale-size NEMO capsule.

EFFECT OF MOUNTING

During the testing of model-size capsules, there was no problem with retaining the capsules inside the test jig to which they were mounted. The mounting, which consisted of chicken wire mesh wrapped around the capsule and fastened securely with wires to the jig frame, was substantial and capable of withstanding the thrust exerted upon the capsule by dynamic pressure. This was not the case with the full-size NEMO capsule. Although the nylon netting was substantial, and the net was fastened to the frame with 0.25-in. steel cables, the thrust exerted by dynamic pressure upon the 66-in.-diameter capsule was much higher than what the cables could withstand. As a result, the capsule was torn loose from its mounting during the firing of shots No. 4, 5, and 6. (Table 4)

The beneficial effect of depth on the resistance of pressure hulls to dynamic overpressure has been previously observed in other brittle materials besides acrylic plastic, materials whose tensile strength is significantly less than their compressive strength, e.g., glass, ceramics, and concrete. The beneficial effect of depth derives its action from the compressive membrane prestressing imposed on the hull by the static external pressure loading. The compressive prestress must be overcome by the tensile flexure stress generated by the underwater explosion before the brittle material can fail in tension on the interior surface of the hull.

Needless to say, imposing compressive prestress on the hull by static external pressure has its limits for all brittle materials. The limit for the beneficial depth effect is reached when the material in the pressure hull begins to fail during dynamic pressure loading in compression rather than in tension. This happens when the sum of the dynamic compressive stress (equal in magnitude to, and following immediately after, the tensile flexure stress phase) and static compressive stress exceeds either the yield or ultimate compressive strength (depending on which one is the smaller value) of the brittle material.

For acrylic plastic hulls designed to fail by general plastic instability, the maximum allowable depth for static precompression purposes is approximately 25 to 30 percent of their short-term critical pressure (based on compressive strains generated in the hull after 8 hours of sustained loading at maximum operational depth). Since the maximum operational depth of acrylic hulls is, as a rule, set at 25 to 30 percent of their short-term critical pressure, the beneficial depth effect is active through the whole depth range of operations for acrylic submersibles.

The breaking loose of the full-size NEMO from its mounting as a result of underwater explosion points up to a very serious practical problem for a submersible containing a NEMO capsule. It appears that unless the NEMO capsule is restrained in some very ingenious manner, the primary damage to the acrylic capsule will be caused either by impact against the framework of the submersible after the capsule has broken loose from a weak mounting, or by excessive dynamic stresses generated by very strong, but rigid mounting. Since the NEMO capsules are generally attached to the submersible framework by their metallic end closures, it is highly probable that when subjected to a severe underwater explosion, the capsule will crack around the penetrations because of unacceptably high bearing stresses. For this reason, it is desirable that the capsule also be supported at other locations by large elastomeric pads that would tend to distribute and absorb some of the capsule's thrust caused by impulse loading.

FINDINGS

- 1. Acrylic plastic spherical pressure hulls will fracture when exposed to underwater explosions whose peak dynamic overpressure may be less than the static critical pressure of the hull.
- 2. Underwater explosions generate cracks primarily on the interior surface of the sphere at locations directly facing and opposite the charge.
- 3. Cracks on the interior surface of the sphere indicate localized external dynamic pressure loading, very similar to a mechanical point-impact loading (Ref. 16).
- 4. Dynamic strains measured on the interior shell surface facing the charge alternate rapidly from tension to compression.
- 5. Increasing the thickness of the acrylic plastic sphere also increases its resistance to underwater explosions; doubling the thickness appears to double the unit impulse and peak dynamic overpressure required for crack initiation.
- 6. Increasing the depth of operation also increases the resistance of the acrylic plastic sphere to underwater explosions; increasing the depth by 1000 feet appears to at least triple the unit impulse and peak dynamic overpressure required for crack initiation.
- 7. Mountings for acrylic plastic spheres tend to fail sooner than the spheres themselves when subjected to underwater explosions.

CONCLUSIONS

Submersibles with NEMO-type acrylic plastic spherical hulls can successfully withstand underwater explosions of considerable magnitude. Increasing the depth of operation significantly increases the resistance of spherical acrylic plastic hull to underwater explosions.

RECOMMENDATIONS

Operational

Submersibles with spherical acrylic plastic pressure hulls should not be exposed to underwater explosions of such magnitude that cracks will be initiated in the hull, or the whole hull torn away from its mounting in the submersible structure. This means that (a) the explosive charges in cable cutters or stud drivers carried routinely by an acrylic plastic submersible should not exceed a certain size if the tools are to be activated in the immediate vicinity of the submersible, and (b) the submersible should not be involved in search missions for unexploded underwater ordnance whose warhead exceeds a critical size for given underwater visibility (i.e., good visibility allows discovery of an unexploded item of ordnance without getting close to it, while poor visibility requires the submersible to be almost in physical contact with the item of ordnance before it is recognized as such).

The maximum sizes of permissible explosive charges for work tools or devices carried routinely by a work submersible have been calculated (Ref. 19) on the basis of the largest charge used in the testing program against full-size NEMO Mod 2000 at 50 ft depth that did no damage to the hull or its mounting (Shot No. 3 of Table 4). Charges equal to, or less than those shown in Fig. 28 can be used repeatedly in performance of work missions in the 10- to 3000-ft depth range by a submersible equipped with NEMO Mod 2000 (Ref. 11) or 2000B hull (Ref. 12) ($t/R_0 \ge 0.121$).

The minimum safe standoff distance for missions involving search and/or disposal of underwater ordnance have been calculated (Ref. 19) on the basis of the largest charge used in the testing program against full-size NEMO Mod 2000 at 50 ft depth that did no damage to the hull but considerable damage to the capsule mounting (Shot No. 6 of Table 4). Standoff distances equal to or larger than those shown in Fig. 29 must be maintained between the submersible with NEMO Mod 2000 or 2000B hull and the unexploded underwater ordnance in the 50- to 3000-ft depth range if fracture of the acrylic hull due to explosion is to be avoided. The standoff distances shown in Fig. 29 are very conservative for depths in excess of 1000 ft.

It is understood, however, that unless a mounting is provided that is capable of restraining the NEMO Mod 2000 hull against a thrust of at least 10⁶ lb, the hull may be torn loose from its mounting when subjected to the explosions and standoff distances shown in Fig. 29.

Design

Typical mountings for work submersibles with NEMO Mod 2000 or 2000B acrylic hulls are generally configured (Fig. 30) to withstand forces generated only by vertical buoyancy or dead weight and horizontal hydrodynamic drag of the sphere. The magnitudes of these forces are low, approximately 3000 lb vertical static force and 1000 lb horizontal drag. Since, however, the submersible is also subjected to dynamic forces during docking and retrieval, a well-designed mounting will, as a minimum, restrain an acrylic hull against 100,000 lb of downward thrust, 100,000 lb horizontal thrust, and 10,000 lb vertical pull (for some types of mountings the vertical pull is also about 100,000 lb).

Unfortunately, even well-designed mountings for typical work missions do not provide adequate restraint against severe underwater explosions at the standoffs plotted in Fig. 29. To withstand the thrust of severe explosions, the mountings must be designed with this specific objective in mind. Unfortunately, proven mounting designs do not exist at the present time for submersibles with acrylic spheres routinely engaged in missions in which severe underwater explosions may be encountered. A conceptual design for such service has been prepared, however, and is shown in Fig. 31.

Although Figs. 28 and 29 have been developed specifically for NEMO Mod 2000 and 2000B acrylic plastic hulls, they are also applicable to other acrylic spheres with $t/R_0 \ge 0.12$. There is sufficient structural similarity between spheres and spherical sectors with included angle $\alpha \ge 120^\circ$ to make Figs. 28 and 29 applicable also to spherical acrylic plastic sector bow windows in submersibles. Some experimental data exist which confirm this belief.

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TABLE 1. ACRYLIC PLASTIC SPHERES SERVING AS TEST SPECIMENS

			Top .		Bottom	m	Short-term
	Outside Diameter, in.	Inside Diameter, in.	Penetration minor diameter (included angle)	Inserts	Penetration minor diameter (included angle)	Inserts	static critical pressure, *
Model 24	15	14	4.793 in. 40°	316 stainless steel hatch no gasket	4.793 in. 40°	316 stainless steel hatch no gasket	1,650
Model 25	15	14	5.150 in. 43°	316 stainless steel hatch polycarbonate gasket	5.150 in. 43°	acrylic plastic spherical sector no gasket	1,650
Model 26	15	14	5.150 in. 43°	acrylic plastic spherical sector no gasket	no penetration no hatch	no hatch no gasket	1,650
Model J	15	13	5.285 in. 48°	6A14V titanium hatch polycarbonate gasket	4.445 in. 40°	6A14V titanium hatch no gasket	4,750
Model K	15	13	. 5.285 in. 48°	6A14V titanium hatch polycarbonate gasket	4.445 in. 40°	6A14V titanium hatch no gasket	4,750

*Based on experimental data from previous studies (Refs. 3, 11).

TABLE 1. ACRYLIC PLASTIC SPHERES SERVING AS TEST SPECIMENS (Continued)

			Top		Bottom	ш	Short-term
	Outside Diameter, in.	Inside Diameter, in.	Outside Inside Penetration Diameter, Diameter, included angle)	Inserts	Penetration minor diameter (included angle)	Inserts	static critical pressure,* psi
Model M	15	13	5.285 in. 48°	6A14V titanium hatch	4.445 in. 40°	6A14V titanium hatch	4,750
				polycarbonate gasket		no gasket	
Model NEMO MOD 2000	99	57.9	23.822 in. 48° 30′	6061-T6 aluminum 21.727 in. hatch	21.727 in. 44°	6061-T6 aluminum penetration plate	4,750
				polycarbonate gasket		polycarbonate gasket	

*Based on experimental data from previous studies (Refs. 3, 11).

TABLE 2. RESISTANCE OF 15-in.-O.D. BY 14-in.-I.D. NEMO SCALE MODELS TO DYNAMIC PRESSURE IMPULSES

Size	Model 25 10 ft Depth	Model 24 100 ft Depth	Model 26 1000 ft Depth
Charge, grams		Standoff, in.	
1.1	48	48	48
	36	36	36
	24	24	24
	12	12	12
8.2	48.†	48	48
	1035. psi peak overpressure	36	36
	0.033 psi-sec unit impulse	24 **	24
į	Severe cracking of hull at equator facing and opposite charge; also severe radial cracks around the pene- trations (Fig. 23)	2250. psi peak overpressure 0.067 psi-sec unit impulse Minor meridional cracks near pene- trations facing and opposite charge. (Fig. 21)	
14.6			48 36 24 * 2816 psi peak overpressure 0.1 psi-sec unit impulse Small crack on equator opposite charge. (Fig. 19)

- Notes: The standoff is measured between the tip of the charge and the surface of the NEMO model.
 - Explosive used is cast explosive composed of 50% PETN and 50% TNT.
 - Failure is indicated by presence of cracks.
 - Shock wave parameters are calculated values.
 - *Denotes light damage.
 - **Denotes medium severe damage.
 - †Denotes very severe damage.

TABLE 3. RESISTANCE OF 15-in.-O.D. BY 13-in.-I.D. NEMO SCALE MODELS TO DYNAMIC PRESSURE IMPULSES

Size of Charge, grams	Model M 10 ft Depth	Model K 1000 ft Depth Standoff, in.	Model J 2000 ft Depth
1.1	48	48	48
1.1	36	36	36
	24	24	24
	12	12	12
8.2	48	48	48
0.2	36 **	36	36
	1434 psi peak overpressure	24	24
	0.045 psi-sec unit impulse		
	Cracks on equator facing charge; also radial crack at the penetration. (Fig. 20)		
14.6		48	48
		36	36
		24	24
		12 †	12 *
		6170. psi peak overpressure	6170 psi peak overpressure
ĺ		0.208 psi-sec unit impulse	0.208 psi-sec unit
		Star shaped cracks on equator facing charge; also radial crack at penetration. (Fig. 22)	Small incipient cracks on equator facing and opposite the charge. (Fig. 18)

- Notes: The standoff is measured between the tip of the charge and the surface of the NEMO model.
 - Explosive used is cast explosive composed of 50% PETN and 50% TNT.
 - Failure is indicated by presence of cracks.
 - Shockwave parameters are calculated values.

^{*}denotes light damage; **denotes medium severe damage, †denotes very severe damage

TABLE 4. RESISTANCE OF 66-in.-O.D. BY 57.90-in.-I.D. FULL-SIZE NEMO MOD 2000 TO DYNAMIC PRESSURE IMPULSES

Shot No.	Size of Charge, grams	Standoff, in.	Peak Overpressure, psi	Unit Impulse psi-sec	Comments
1	1.10*	52.9	435	0.0074	No damage
2	5.62*	52.9	805	0.0228	No damage
3	14.50*	52.9	1,150	0.0436	No damage
4	169.87*	52.9	2,906	0.2347	No damage; capsule broke loose from test jig.
5	387.77**	52.9	3,967	0.412	No damage, capsule broke loose from test jig.
.6	688.56**	52.9	4,927	0.611	No damage, capsule broke loose from test jig.

- Notes: The standoff is measured between the tip of the charge and the surface of the NEMO capsule.
 - Explosive used is cast explosive composed of 50% PETN and 50% TNT.
 - Damage is indicated by presence of cracks.
 - Shock wave parameters are calculated values.
 - All tests were conducted at 50 ft depth.
 - *Explosive located above the capsule.
 - **Explosive located below the capsule.



Figure 1. Scale-size NEMO-type hulls tested to destruction under dynamic impulse loading.

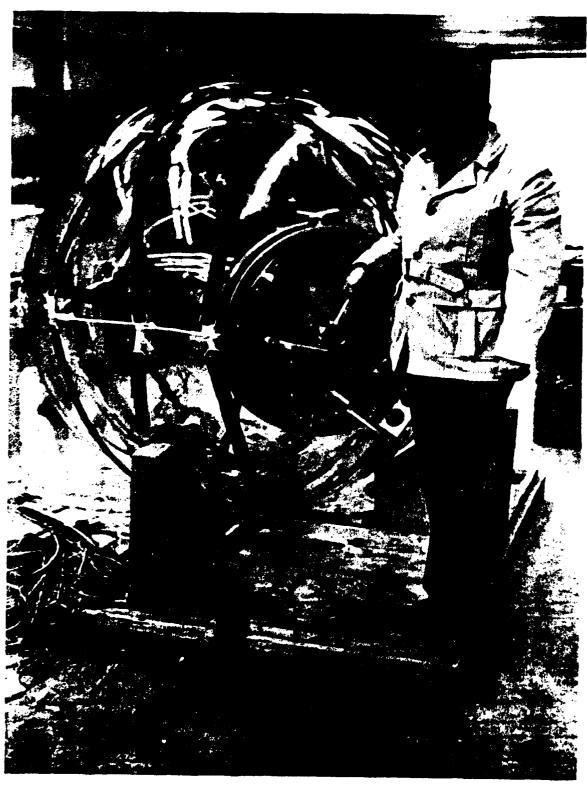


Figure 2. Full-size NEMO Mod 2000 hull tested to destruction under dynamic impulse loading.

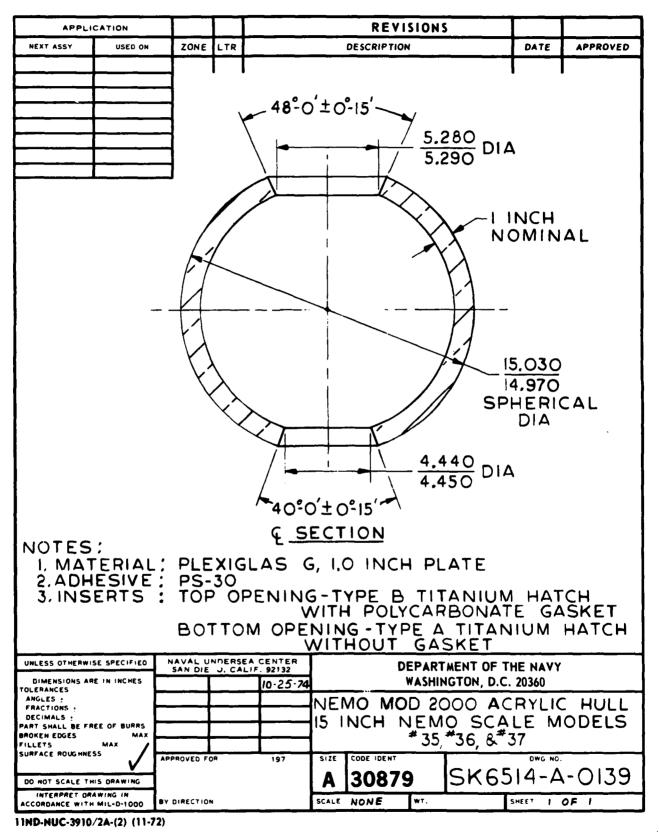
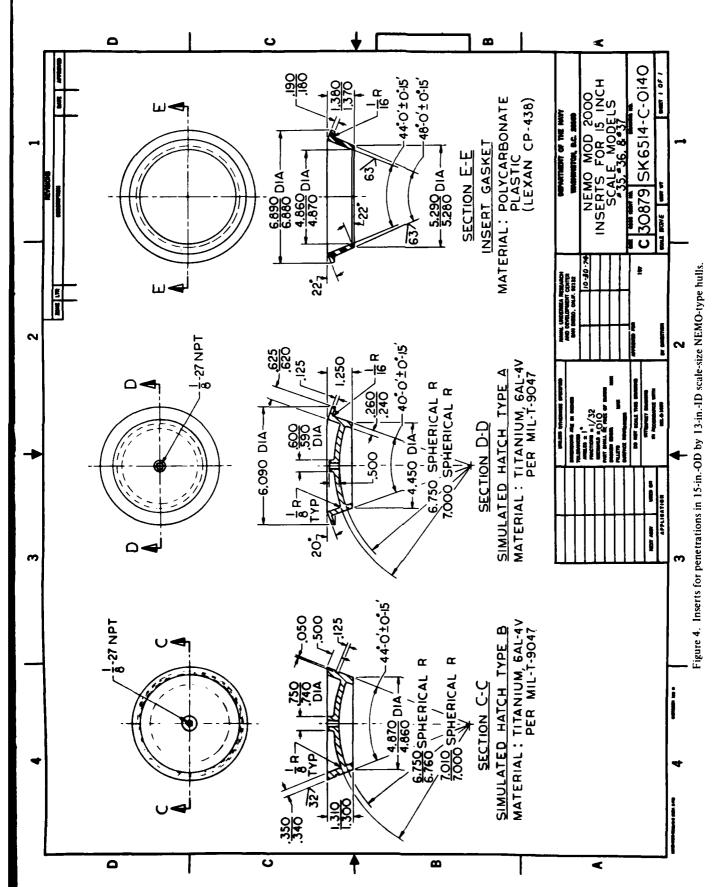
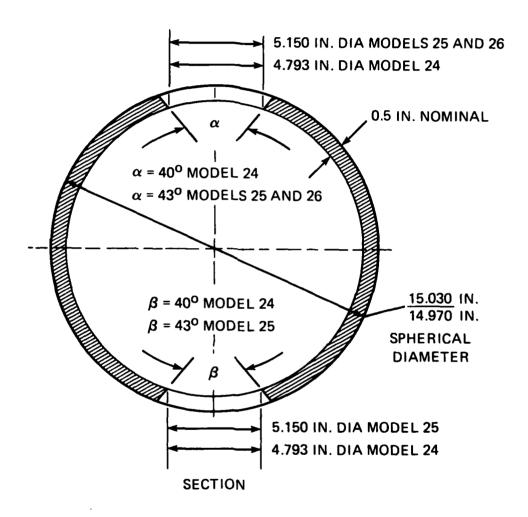


Figure 3. Typical dimensions of 15-in. OD by 13-in. ID scale-size NEMO-type hulls.





NOTES:

1. MATERIAL: PLEXIGLAS G, 0.5 IN. PLATE

2. ADHESIVE: PS-18

Figure 5. Typical dimensions of 15-in.-OD by 14-in.-ID scale-size NEMO-type hulls.

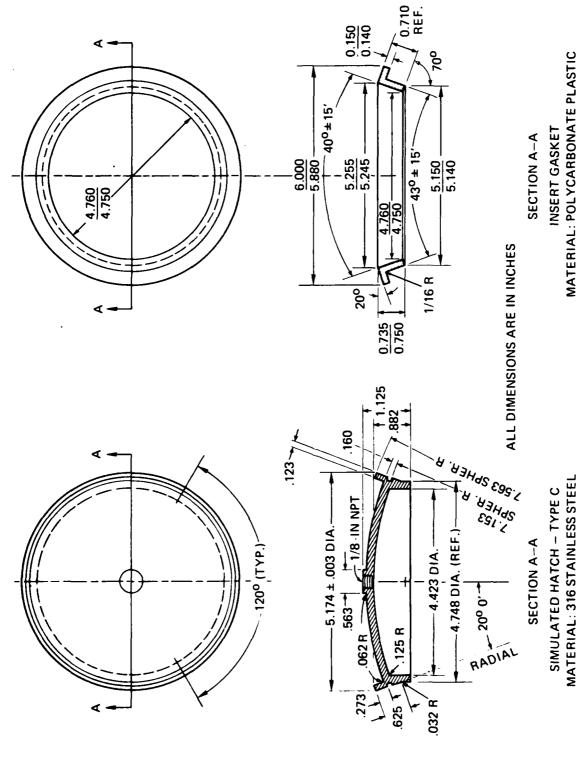
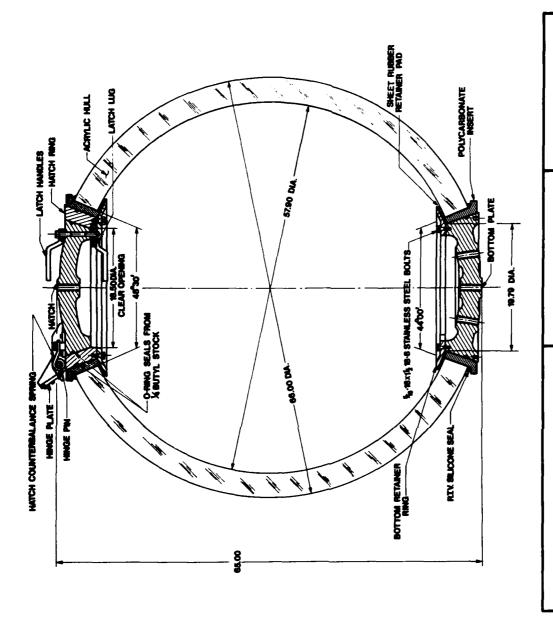


Figure 6. Inserts for penetrations in 15-in.-OD by 14-in.-ID scale-size NEMO-type hulls.





construction: bonded spherical pentagons

Figure 7. Dimensions of 66-in.-OD by 57.9-in.-ID full-size NEMO Mod 2000 capsule.

cyclic fatigue life: in excess of 1000 dives of 4 hr duration to 3000 ft weight: 2500 lbs displacement: 5600 lbs

operational depth: 3000 ft proof test depth: 3600 ft implosion depth: 10,500 ft positive buoyancy: 3100 lbs weight/displacement: 0.44

BOTTOM PLATE DETAIL

10-14 HOLES THEN

B, BPLCS

-21.500 DIA. -23.250 DIA -20.00 DIA.

28.00 DIA.

26

4000,core

TESOO DIA

26.28 DIA.

TOP HATCH DETAIL

- sybol cone - 15.25 DIA.

APS COME

2100 DIA 27.16 DIA.

10-14 HOLE

12.25 A

- KR DA

-- 21.TIS DIA.

- Men or 100

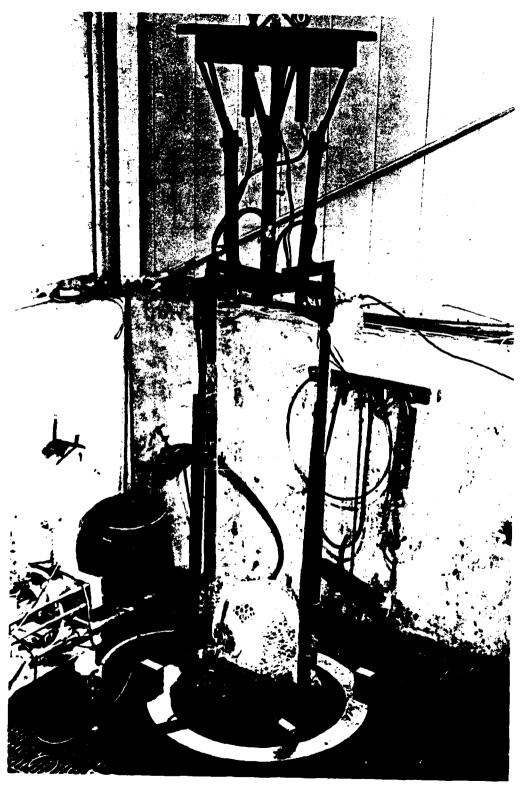


Figure 8. Test jig for holding the scale-size acrylic capsules in the pressure vessel during detonation of explosive.

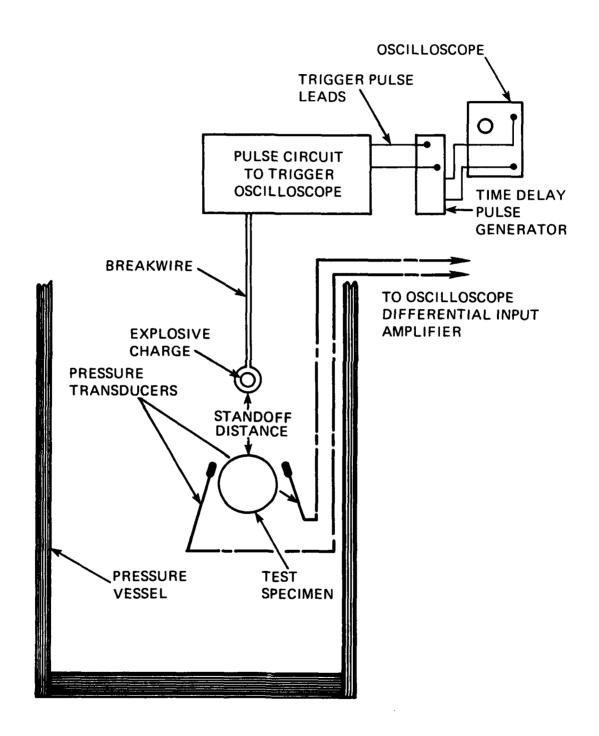


Figure 9. Schematic of instrumentation used for measurement of peak pressures impinging on the acrylic capsules.

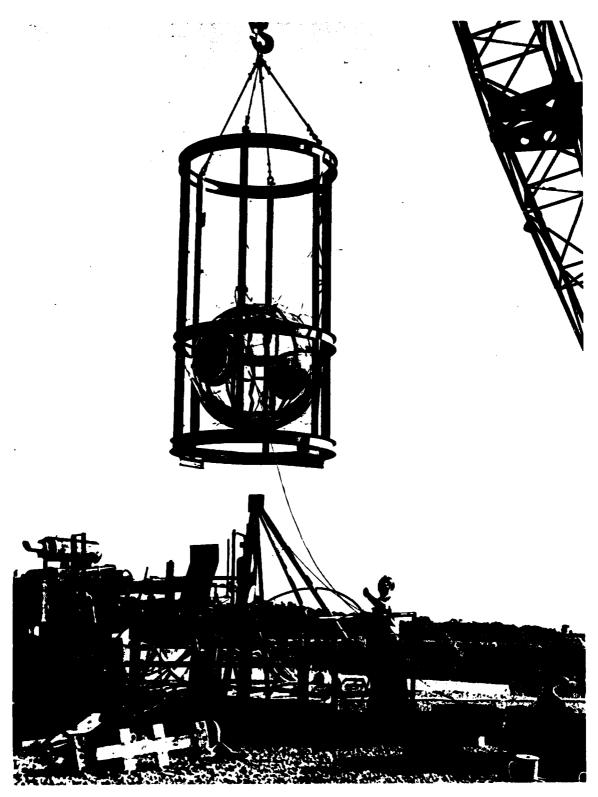


Figure 10. Test jig for holding the full-size NEMO Mod 2000 capsule in the well during detonation of explosive.



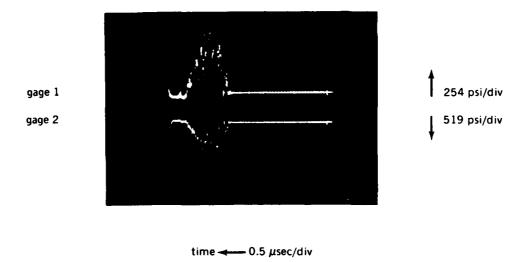
Figure 11. Mounting of the NEMO Mod 2000 capsule inside the test jig.



Figure 12. Typical size and shape of explosive charge used against full-size NEMO Mod 2000 capsule.



Figure 13. Closeup of pressure transducer used in tests with the full-size NEMO Mod 2000 capsule.



Model: 25 NEMO (15" X 14")

Charge: 8.2 grams

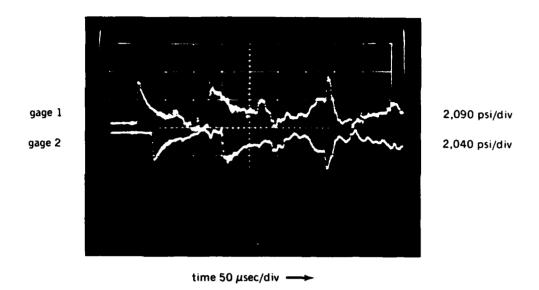
Standoff: 48 inches

Hydrostatic Pressure: 10 psi

	Gage 1	Gage 2	Calculated
Peak Shock Overpressure, psi	1,020	1,035	1,035
Unit Impulse, psi-sec	0.175	0.18	.0325
Duration, μsec	1,350	1,350	_

Note — Model failed.

Figure 14. Peak pressure measured at the scale-size capsule No. 25 during the explosion that fractured the capsule.



Model: NEMO 26 (15" X 14")

Charge: 14.6 grams

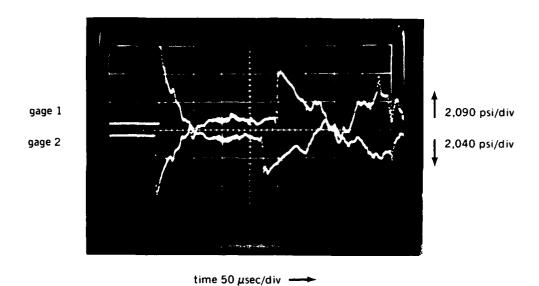
Standoff: 24" (gage 1), 26" (gage 2), 24" (model)

Hydrostatic Pressure: 450 psi

			Calculated	
	Gage 1	Gage 2	Gage 1	Gage 2
Peak Shock Overpressure, psi	2,820	2,250	2,810	2,580
Unit Impulse, psi-sec	.111	.0654	.101	.094
Duration, µsec	80	75	_	_

Note — Model failed. Gage 2 was farther from model and charge than gage 1 and gives a lower than anticipated value.

Figure 15. Peak pressure measured at the scale-size capsule No. 26 during the explosion that fractured the capsule



Model: NEMO J (15" \times 13")

Charge: 14.6 grams

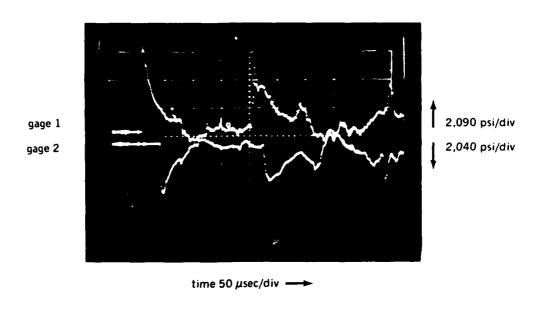
Standoff: 13" (gage 1), 12" (gage 2), 12" (model)

Hydrostatic Pressure: 1,000 psi

			Calculated	
	Gage 1	Gage 2	Gage 1	Gage 2
Peak Shock Overpressure, psi	5,660	4,480	5,660	6,200
Unit Impulse, psi-sec	.1276	.1108	.194	.215
Duration, μsec	50	47.5		_

Note - Model failed. Gage 2 gives lower than anticipated value.

Figure 16. Peak pressure measured at the scale-size capsule No. J during the explosion that fractured the capsule.



Model: NEMO K (15" X 13")

Charge: 14.6 grams

Standoff: 12" (gage 1), 13" (gage 2), 12" (model)

Hydrostatic Pressure: 450 psi

			Calculated	
	Gage 1	Gage 2	Gage 1	Gage 2
Peak Shock Overpressure, psi	6,280	4,080	6,200	5,660
Unit Impulse, psi-sec	.1856	.085	.215	.194
Duration, μsec	75	50	-	_

Note — Model failed. Gage 2 reads lower than anticipated.

Figure 17. Peak pressure measured at the scale-size capsule No. K during the explosion that fractured the capsule.

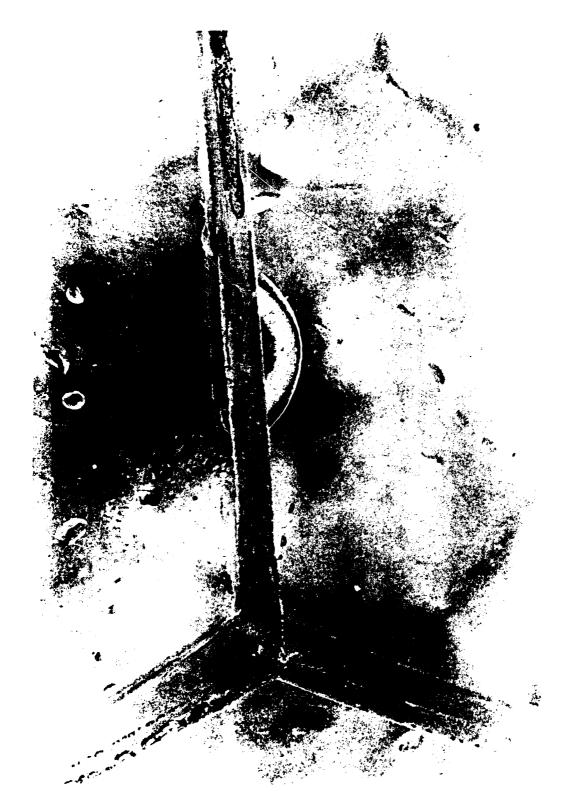


Figure 18. Fractures in capsule No. J after being subjected (at 2000-ft depth) to a 14.6-gram charge at 12 in. standoff: (a) crack is facing the explosive. (Sheet I of 2)



Figure 18. Fractures in capsule No. J after being subjected (at 2000-ft depth) to a 14.6-gram charge at 12 in. standoff: (b) crack is on the opposite side of capsule. (Sheet 2 of 2)



Figure 19. Fracture in capsule No. 26 after being subjected (at 1000-ft depth) to a 14.6-gram charge at 24 in. standoff. Crack is facing the explosive.

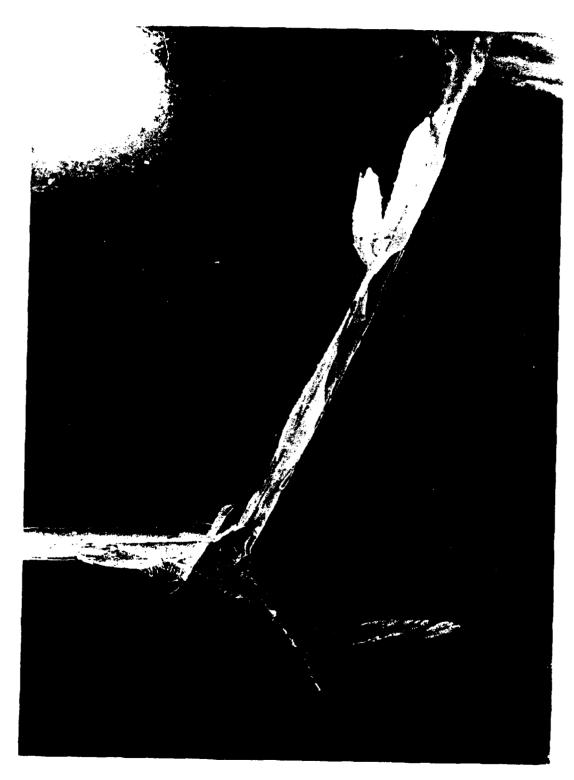


Figure 20. Fractures in capsule No. M after being subjected (at 10-ft depth) to a 14.6-gram charge at 36 in. standoff: (a) crack is facing the explosive. (Sheet 1 of 2)

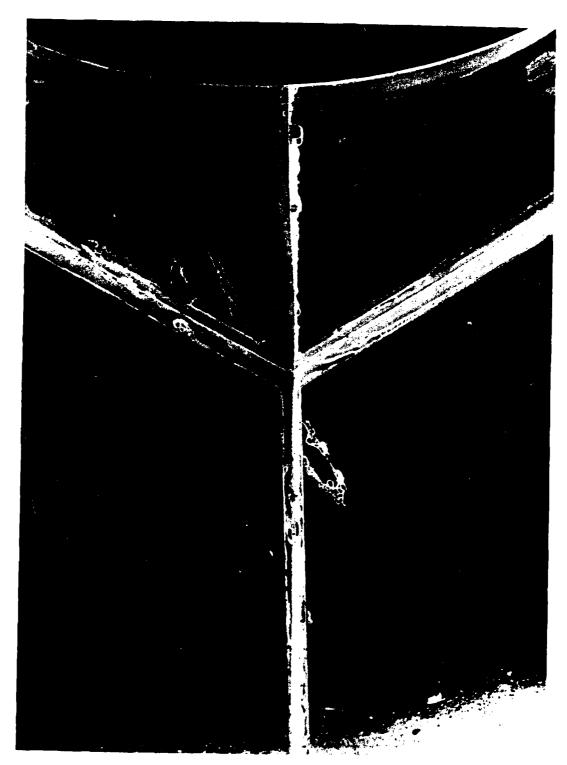


Figure 20. Fractures in capsule No. M after being subjected (at 10-ft depth) to a 14.6-gram charge at 36 in. standoff: (b) crack is at the penetration. (Sheet 2 of 2)

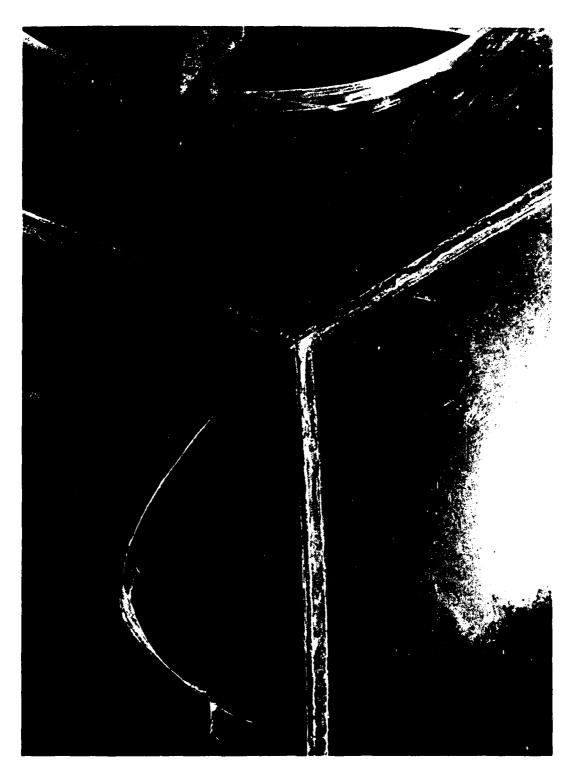


Figure 21. Fractures in capsule No. 24 after being subjected (at 10-ft depth) to a 8.2-gram charge at 24 in, standoff: (a) crack is facing the explosive. (Sheet 1 of 2)



Figure 21. Fractures in capsule No. 24 after being subjected (at 10-ft depth) to a 8.2-gram charge at 24 in. standoff: (b) crack is on the opposite side of capsule near penetration. (Sheet 2 of 2)

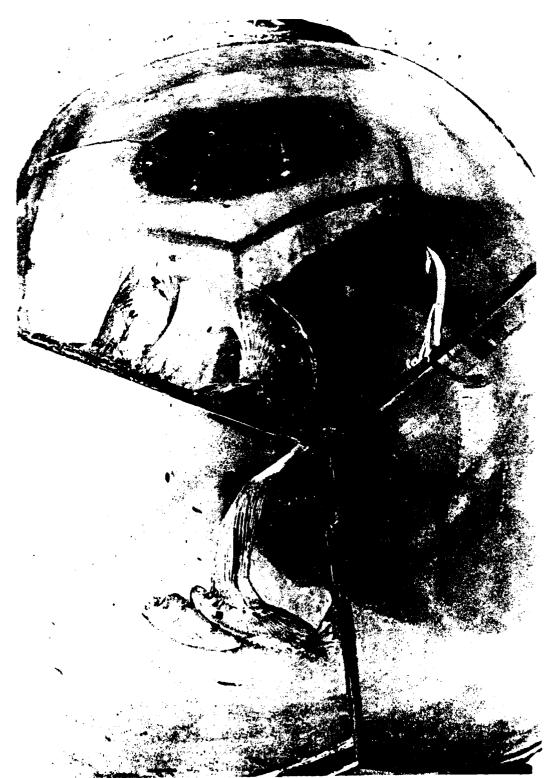


Figure 22. Fractures in capsule No. K after being subjected (at 1000-ft depth) to a 14.6-gram charge at 12 in. standoff: (a) crack is facing the explosive. (Sheet 1 of 2)



Figure 22. Fractures in capsule No. K after being subjected (at 1000-ft depth) to a 14.6-gram charge at 12 in. standoff: (b) crack is on the opposite side of capsule near penetration. (Sheet 2 of 2)



Figure 23. Fractures in capsule No. 25 after being subjected (at 10-ft depth) to a 8.2-gram charge at 48 in, standoff: (a) crack is facing the explosive. (Sheet 1 of 2)

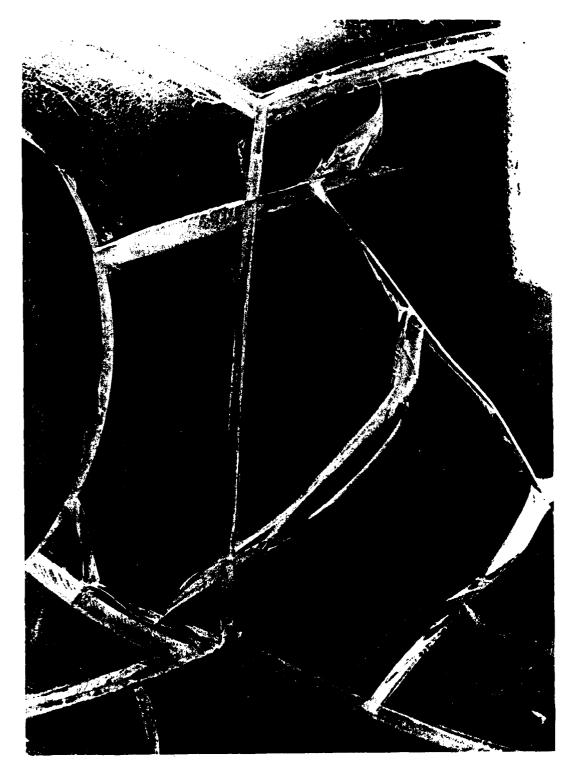
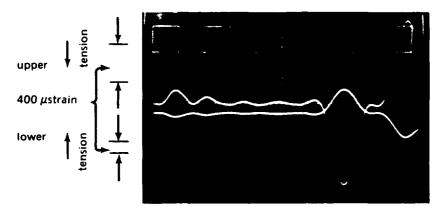


Figure 23. Fractures in capsule No. 25 after being subjected (at 10-ft depth) to a 8.2-gram charge at 48 in. standoff: (b) crack is on the opposite side of capsule near penetration. (Sheet 2 of 2)



Figure 24. Point-impact fracture in the NEMO Mod 2000 capsule after striking a steel beam in the test jig. The charge was 688.56 grams at 52.9 in. standoff in 50 ft of water.

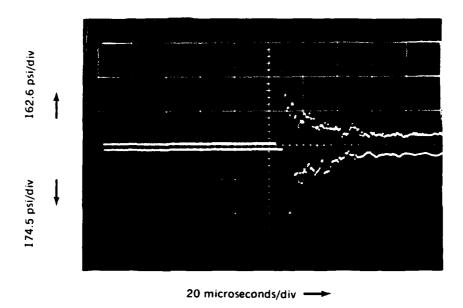


time 50 µsec/div ---

NOTES:

- a) 1/4 inch strain gage was positioned internally on apex of sphere closest to point of
- b) Gage output was recorded at two voltage deflection levels on scope since strain level was was not known
- c) 7/8 pound of pentolite was detonated 53 inches from outer surface of sphere
- Maximum strain was initially 930 microinches in tension followed by an equal strain in compression
- e) Scope trigger was delayed 500 microseconds after detonation

Figure 25. Dynamic strains measured on the interior surface of the NEMO Mod 2000 capsule directly below the charge of 387.8 grams at 52.9 in. standoff.



Charge: E81 Electric Blasting Cap

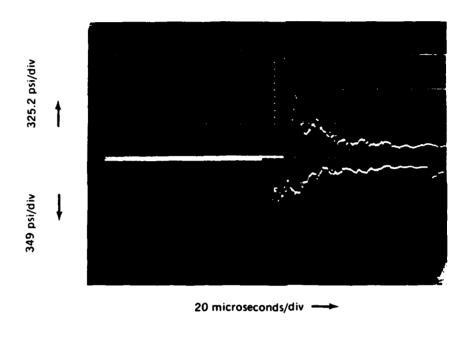
Standoff: 4.41 Feet
Depth: 50 Feet

Peak Measured Pressure: Upper Trace 504 Psi

Lower Trace 502 Psi Not Readable

Measured Unit Impulse: Not Readable Calculated Peak Pressure: 435 Psi Calculated Unit Impulse: .00745 Psi-Sec Comments: No Damage

Figure 26. Peak pressure measured at the full-size NEMO Mod 2000 capsule when the 1.1-gram charge was fired at 52.9 in. standoff.



Charge: .01242 lbs
Standoff: 4.41 Feet
Depth: 50 Feet

Peak Measured Pressure: Upper Trace 975 Psi

Lower Trace 1,012 Psi

Measured Unit Impulse: Not Readable Calculated Peak Pressure: 805 Psi Calculated Unit Impulse: 0.0228 Psi-Sec Comments: Nos Damage

Figure 27. Peak pressure measured at the full-size NEMO Mod 2000 capsule when the 5.62-gram charge was fired at 52.9 in. standoff.

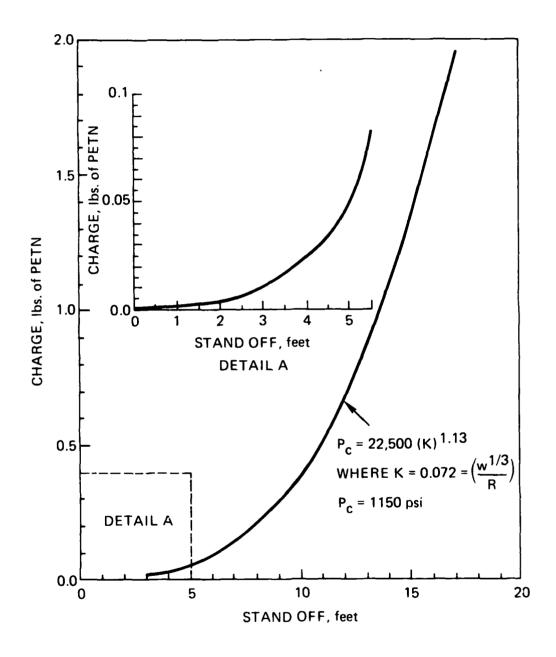


Figure 28. Weight of charges that can be repeatedly exploded underwater in the vicinity of NEMO Mod 2000 capsule without any damage to the acrylic hull or its mounting in the structure of the submersible.

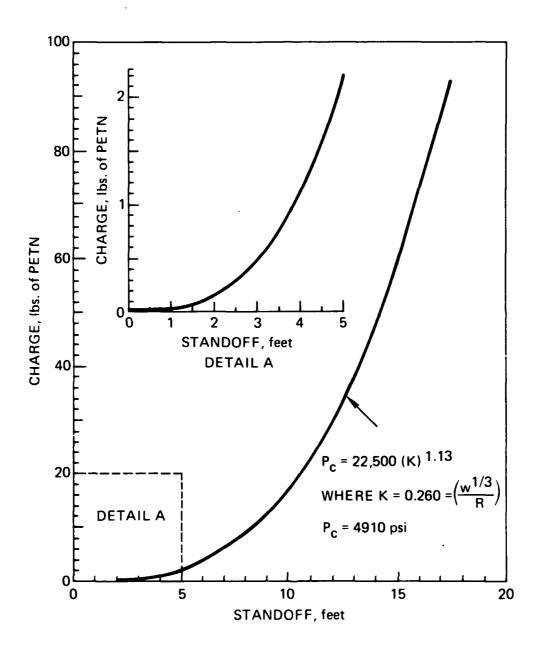
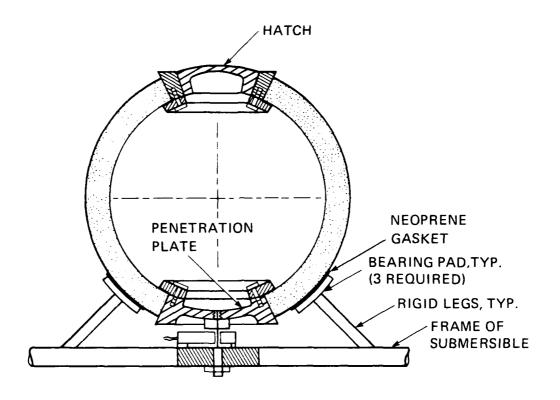
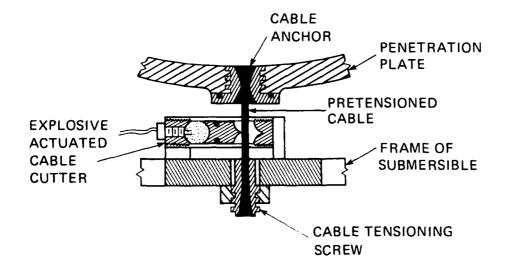


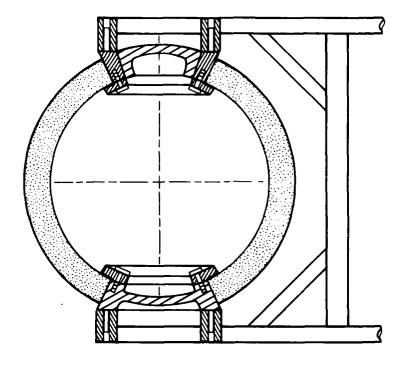
Figure 29. Maximum weight of charges that can be set off underwater in the vicinity of NEMO Mod 2000 without fracturing the hull. Such charges may, however, tear the capsule from its mounting in the structure of the submersible.

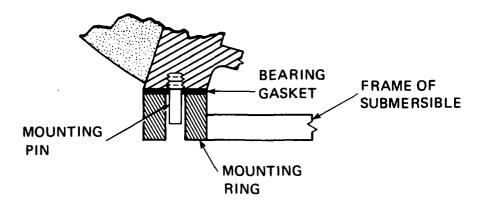




(a) SINGLE-POLE MOUNTING

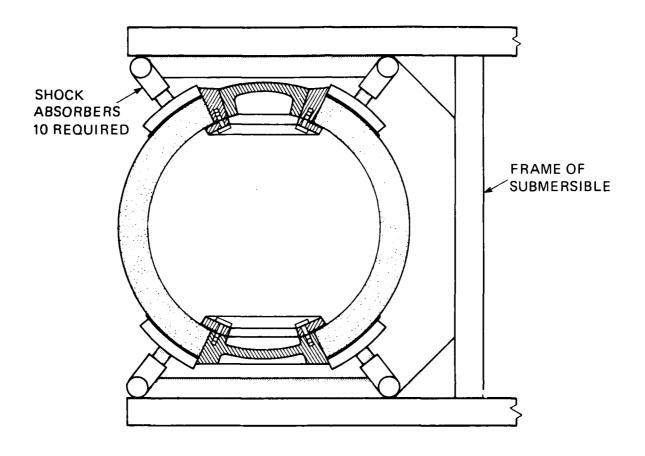
Figure 30. Typical mountings for NEMO type hulls. The single-pole mounting provides better upward visibility and is very suitable for emergency release of the capsule, while the twin-pole mounting provides a more secure attachment to the frame. (Sheet 1 of 2)





(b) TWIN-POLE MOUNTING

Figure 30. Typical mountings for NEMO type hulls. The single-pole mounting provides better upward visibility and is very suitable for emergency release of the capsule, while the twin-pole mounting provides a more secure attachment to the frame. (Sheet 2 of 2)



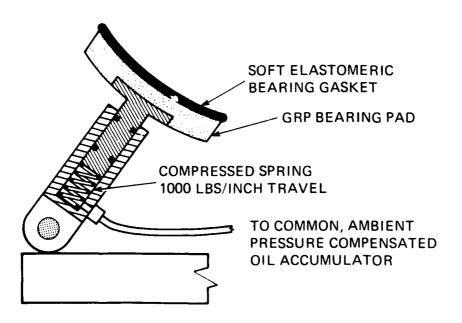


Figure 31. Concept of a mounting providing a secure, but shock-mitigating attachment to the frame.

APPENDIX DESCRIPTION OF TEST SPECIMENS

MODEL J

Outside diameter:

15 in.

Inside diameter:

13 in.

Shell thickness:

0.040 - 0.980 in.

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-30

adhesive (Fig. 3).

Penetrations:

top - 5.285 in. minor diameter with 48° included angle

bottom - 4.445 in. minor diameter with 40° included angle

Inserts:

top - Type B (Fig. 4), 6A14V titanium

bottom - Type A (Fig. 4), 6A14V titanium

Insert gasket:

top - polycarbonate gasket (Fig. 4)

bottom - none

Life history:

- Pressure cycled 1000 times to 1000 psi in tap water at 61-74°F ambient temperature. Typical pressure cycle consisted of pressurizing to 1000 psi, holding at 1000 psi for 4 hours, depressurizing to 0 psi, and relaxing for 4 hours at 0 psi.

MODEL K

Outside diameter:

15 in.

Inside diameter:

13 in.

Shell thickness:

0.935 - 0.975 in.

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-30

adhesive (Fig. 3)

Penetrations:

top - 5.285 in. minor diameter with 48° included angle

bottom -4.445 in. minor diameter with 40° included angle

Inserts:

top - Type B (Fig. 4) 6A14V titanium

bottom - Type A (Fig. 4) 6A14V titanium

Insert gasket:

top - polycarbonate gasket (Fig. 4)

bottom - none

Life history:

Pressure cycled 1000 times to 1500 psi in tap water at 61-74°F ambient temperature. Typical pressure cycle consisted of pressurizing to 1500 psi, holding at 1500 psi for 4 hours, depressurizing to

0 psi, and relaxing for 4 hours at 0 psi.

MODEL M

Outside diameter:

15 in.

Inside diameter:

13 in.

Shell thickness:

0.930 - 0.990

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-30

adhesive (Fig. 3)

Penetrations:

top - 5.285 in., minor diameter with 48° included angle

bottom - 4.445 in., minor diameter with 40° included angle

Inserts:

top - Type B (Fig. 4), 6A14V titanium

bottom - Type A (Fig. 4), 6A14V titanium

Insert gasket:

top – polycarbonate gasket (Fig. 4)

bottom - none

Life history:

Pressure cycled 1056 times to 500 psi in tap water at 61-74°F

ambient temperature. Typical pressure cycle consisted of

pressurizing to 500 psi, holding at 500 psi for 4 hours, depressuriz-

ing to 0 psi, and relaxing for 4 hours at 0 psi.

MODEL 24

Outside diameter:

15 in.

Inside diameter:

14 in.

Shell thickness:

0.460 - 0.490 in.

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-18

adhesive (Fig. 5).

Penetrations:

top - 4.793 in., minor diameter with 40° included angle

bottom -4.793 in., minor diameter with 40° included angle

Inserts:

top - Type C (Fig. 6), 316 stainless steel

bottom - Type C (Fig. 6), 316 stainless steel

Insert gasket:

top - none

bottom - none

Life history:

Pressure cycled 1056 times to 500 psi in tap water at 61-74°F ambient temperature. Typical pressure cycle consisted of pressurizing to 500 psi, holding at 500 psi for 4 hours, depressurizing to

0 psi, and relaxing for 4 hours at 0 psi.

MODEL 25

Outside diameter:

15 in.

Inside diameter:

14 in.

Shell thickness:

0.460 - 0.490 in.

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-18

adhesive (Fig. 5).

Penetrations:

top-5.150 in., minor diameter with 43° included angle

bottom - 5.150 in., minor diameter with 43° included angle

Inserts:

top - Type C (Fig. 6), 316 stainless steel

bottom - Spherical shell sector, 0.5 in. thick with 43° included angle, acrylic

plastic

Insert gasket:

top – polycarbonate gasket, (Fig. 6)

bottom - none

Life history:

- Pressure cycled 1056 times to 500 psi in tap water at 61-74°F

ambient temperature. Typical pressure cycle consisted of pressurizing to 500 psi, holding at 500 psi for 4 hours, depressuriz-

ing to 0 psi, and relaxing for 4 hours at 0 psi.

MODEL 26

Outside diameter:

15 in.

Inside diameter:

14 in.

Shell thickness:

0.460 - 0.500 in.

Material:

Plexiglas G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-18

adhesive (Fig. 5)

Penetrations:

top - 5.150 in., minor diameter with 43° included angle

bottom - none

Inserts:

top - Spherical shell sector, 0.5 in. thick with 43° included angle, acrylic

plastic

bottom - none

Insert gasket:

top - none

bottom - none

Life history:

- Was not subjected to any hydrostatic tests prior to explosive

testing.

MODEL NEMO 2000

Outside diameter:

66 in.

Inside diameter:

57.900 in.

Shell thickness:

4.050 in.

Material:

Plexiglass G

Construction:

Assembly of 12 thermoformed pentagons bonded with PS-30

adhesive (Fig. 7).

Penetrations:

top - 23.822 in., minor diameter with 48°30' included angle

bottom - 21.727 in., minor diameter with 44° included angle

Inserts:

top - Working hatch, 6061-T6 aluminum (Appendix A)

bottom – penetration plate, 6061-T6 aluminum (Appendix A)

Insert gasket:

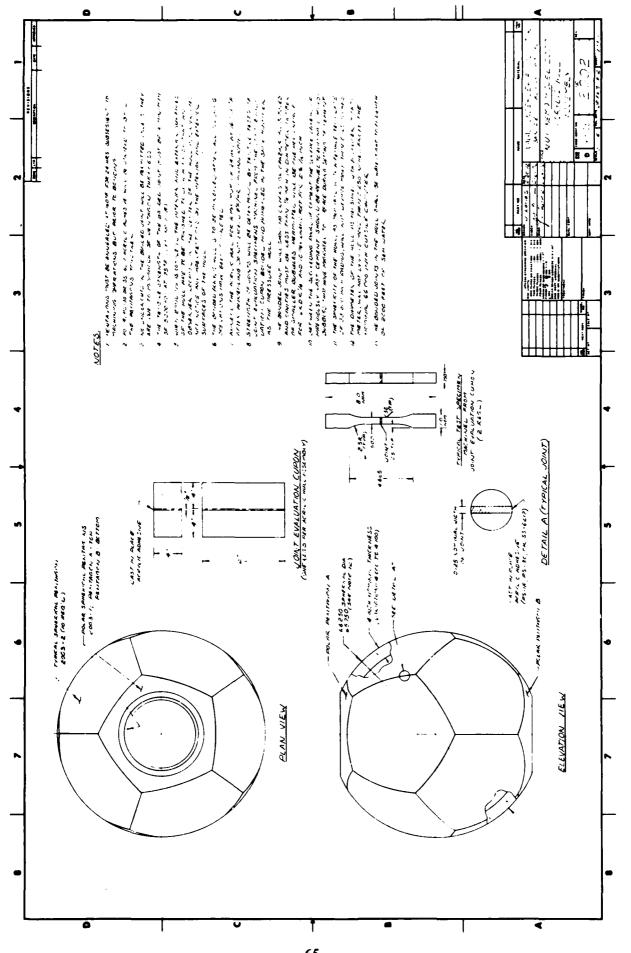
top – polycarbonate gasket (Appendix A)

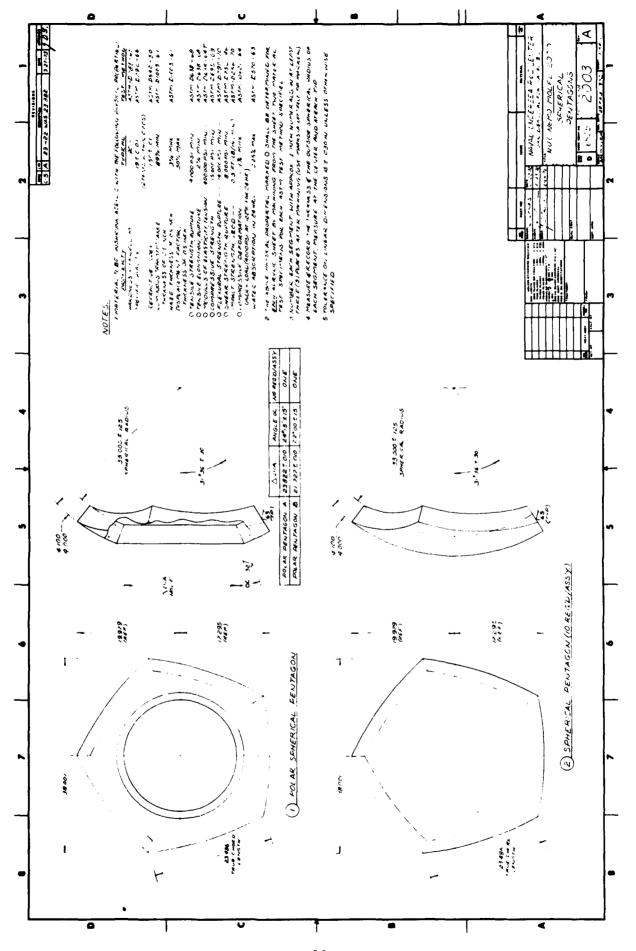
bottom – polycarbonate gasket (Appendix A)

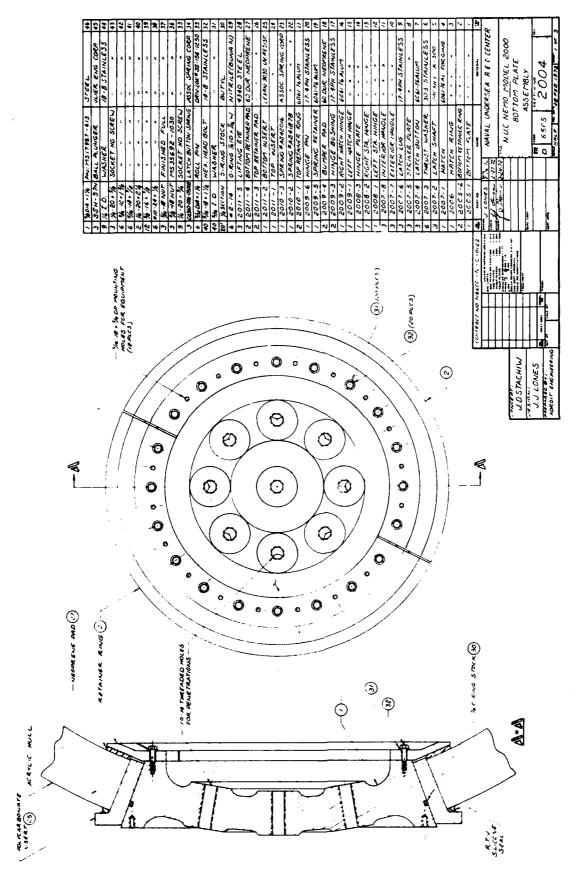
Life history:

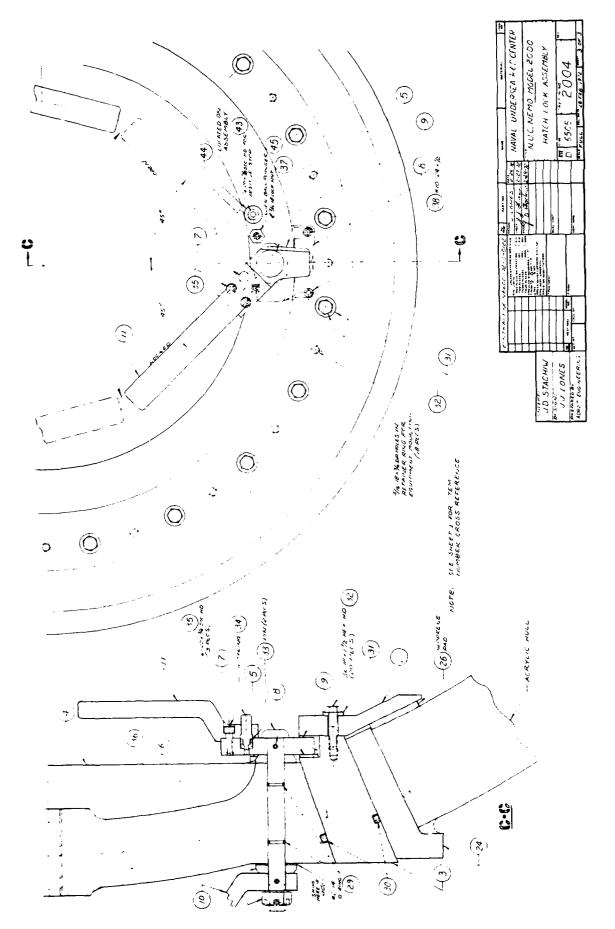
 Pressure cycled one time each to 450 psi, 900 psi, 1350 psi, and 1800 psi. Each pressure cycle consisted of pressurizing to maximum pressure, holding at that pressure for 24 hours, depressurizing

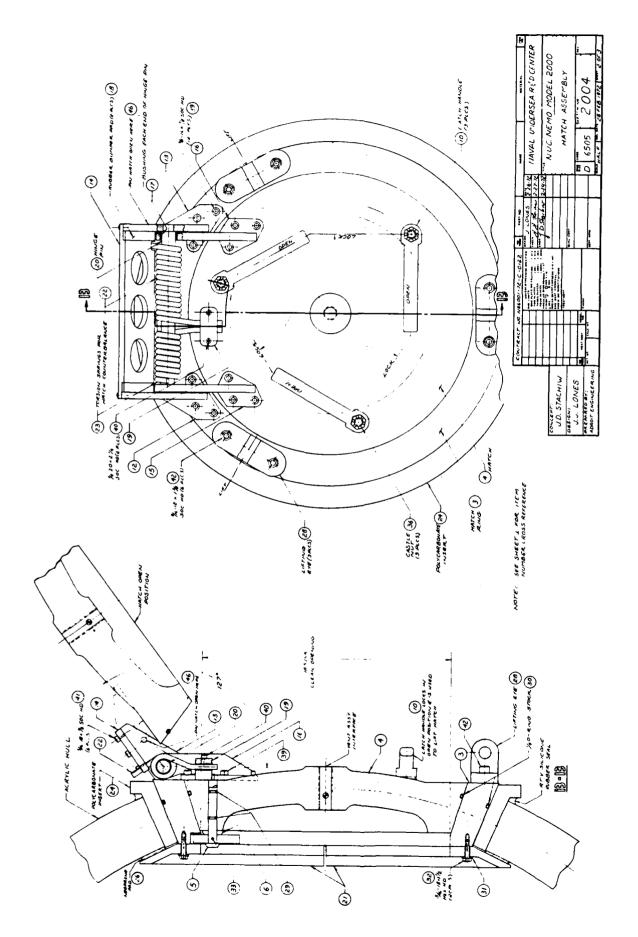
to 0 psi, and relaxing for 24 hours at 0 psi.

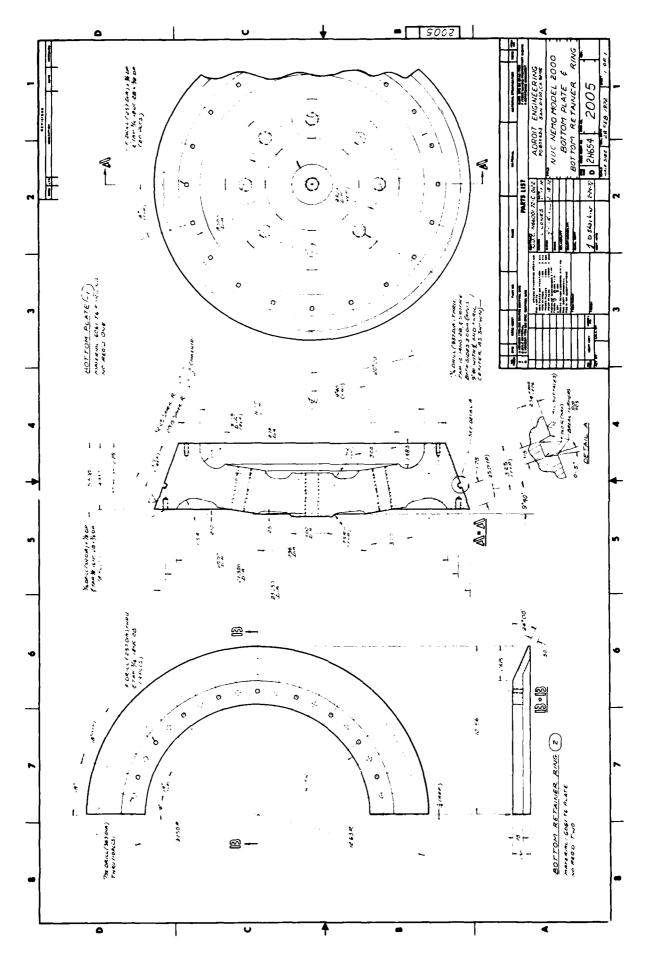


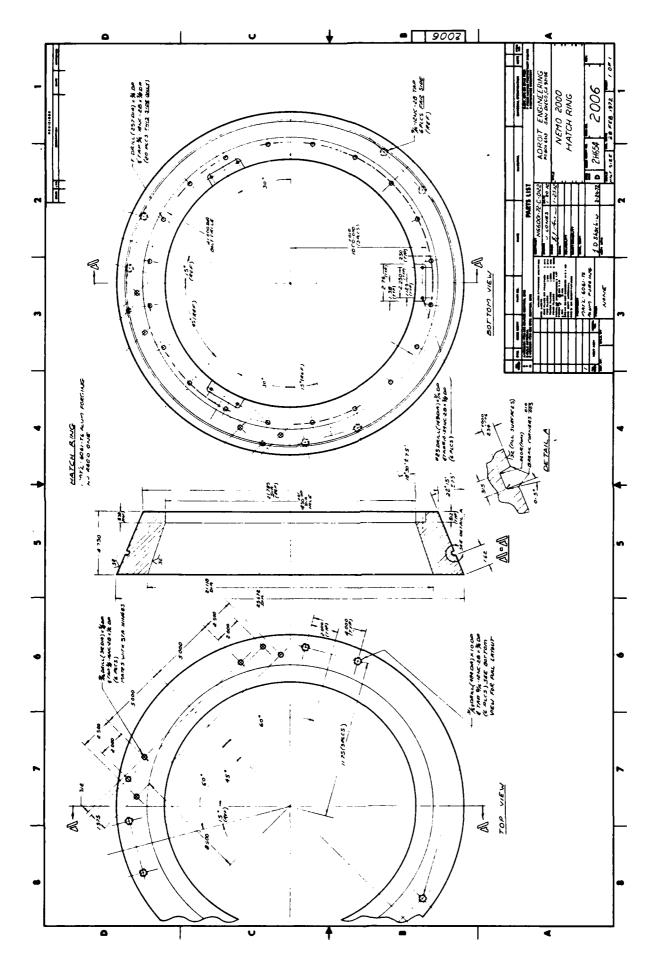


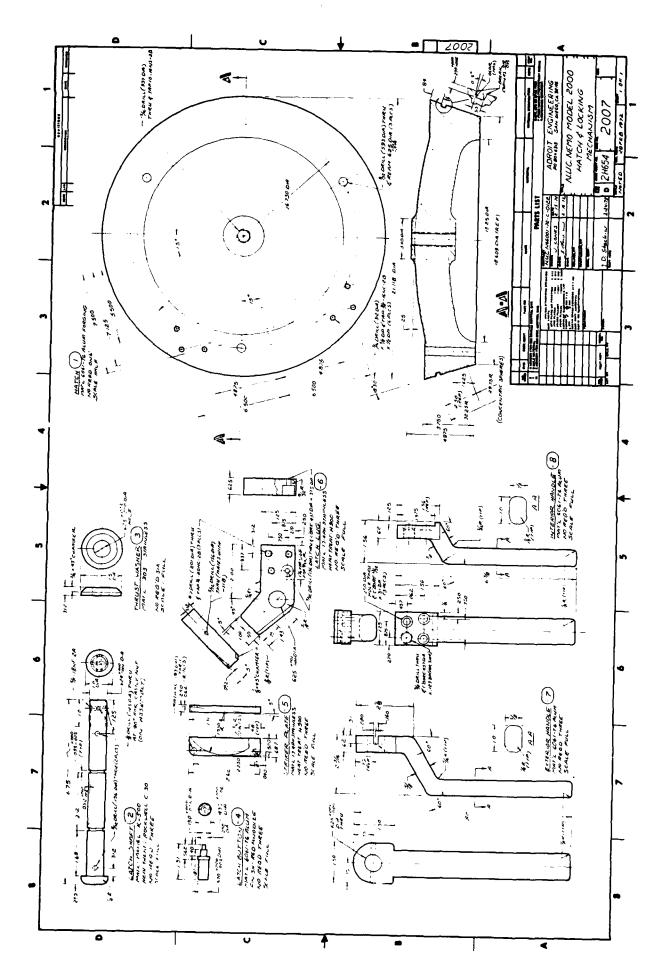


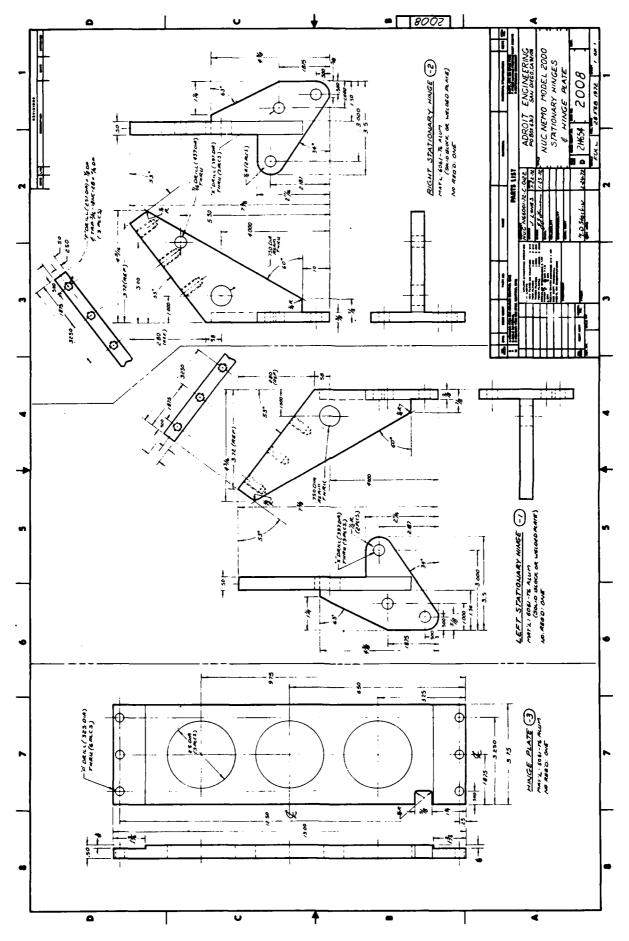


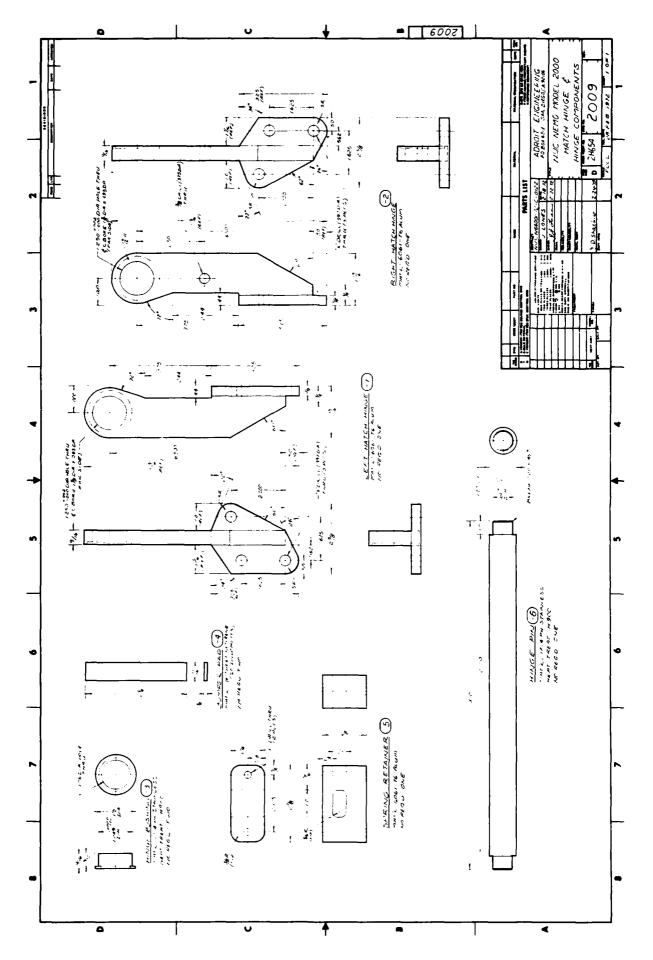


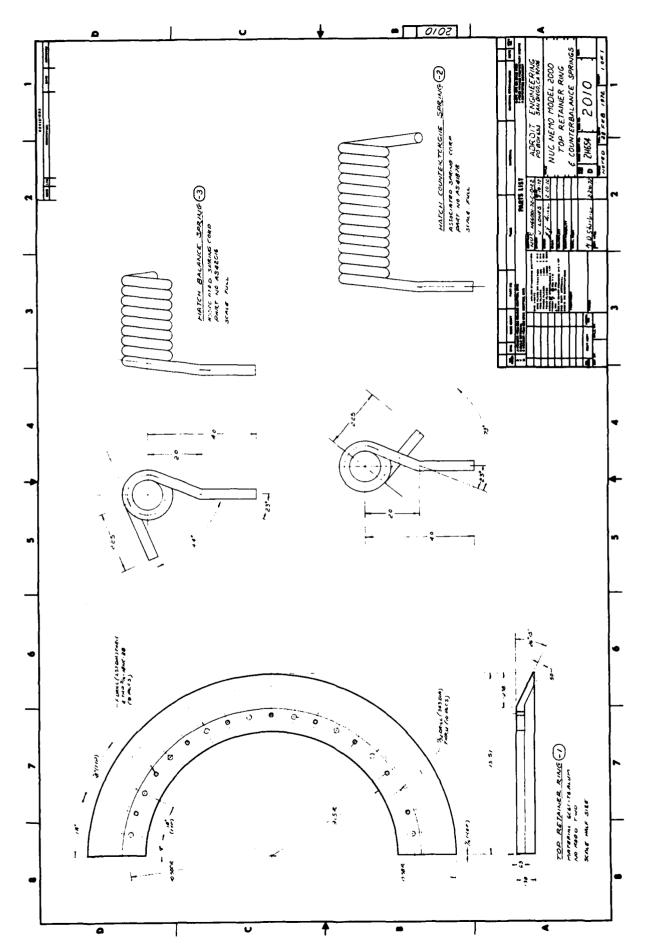


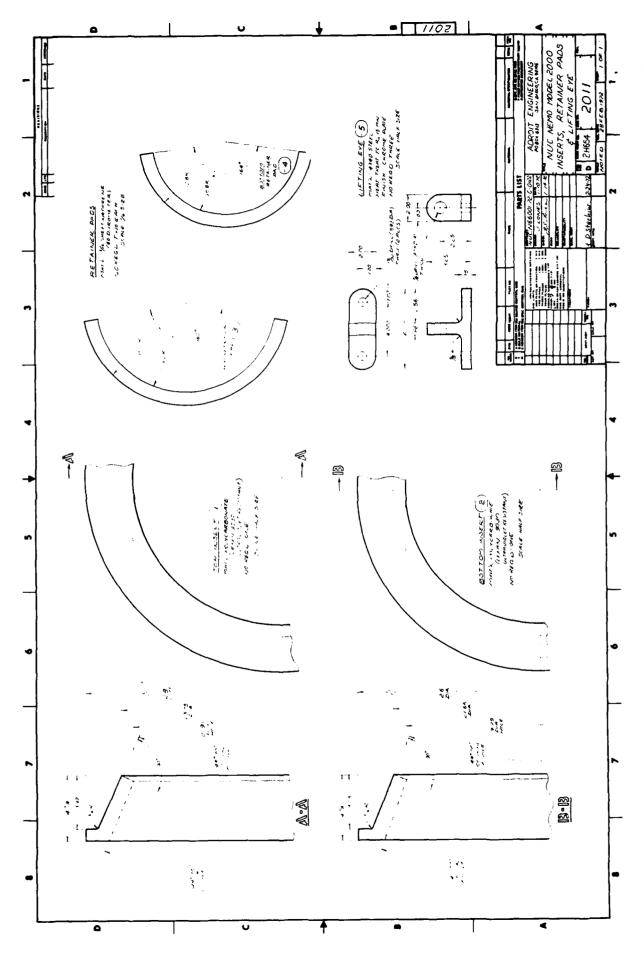














NEMO MODEL 2000 ACRYLIC PLASTIC SPHERICAL HULL FOR MANNED SUBMERSIBLE OPERATION AT DEPTHS TO 3000 FEET

by

Jerry D. Stachiw
OCEAN TECHNOLOGY DEPARTMENT
December 1974



Approved for public release; distribution unlimited.

ADMINISTRATIVE INFORMATION

This report describes research performed between June 1972 and December 1974 as part of the investigation into man-rated transparent submersibles for maximum depth capabilities. Program efforts were requested by the Director of Navy Laboratories and were funded under a Project Order from the Naval Material Command through the Independent Research and Independent Exploratory Development program at the Naval Undersea Center under Subproject Task Area Number ZF-61-412-001.

Released by H. R. TALKINGTON, Head Ocean Technology Department

ACKNOWLEDGMENTS

The successful completion of an acrylic plastic hull for submersible mission operation to 3000 feet represents the combined effort of many individuals and companies. The hatches were designed by Adroit Engineering of San Diego, CA; the hull was fabricated by Swedlow Incorporated of Los Angeles, CA; and the finished assembly was tested by Southwest Research Institute of San Antonio, TX. This unified study result owes its achievement to the administrative and moral support of H. R. Talkington, Head – NUC Ocean Technology Department, and Dr. Wm. B. McLean, retired NUC Technical Director.

UNCLASSIFIED

SECURITY CLASSIFICATION OF THIS PAGE When Para Entereds

Γ	REPORT DOCUME	NTATION PAGE	READ INSTRUCTIONS BEFORE COMPLETING FORM	
1	REPORT NUMBER TP 451	2 GOVT ACCESSION NO.		
•	Nemo Model 2000 Acrylic Plastic Spherical Hull for Manned Submersible Operation at Depths to 3000 Feet		TP — June 1972 to December 1974 F PERFORMING ORG REPORT NUMBER CNM — DLP	
7	Jerry D. Stachiw		B CONTRACT OR GRANT NUMBER(*)	
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19	Plastics Structural Engineering Submarine Engineering Acrylic Technology	Submersible Hulls Acrylic Hull Underwater Vehicles Structural Materials	Transparent Materials Pressure Hulls	
20	the latest addition to the Ner hull assemblies. The 66 inch hatches has successfully with 450, 900, 1350 and 1800 psi	ylic plastic pressure hull assen mo hull series represented by OD × 58 inch ID spherical a stood 24 hour long external l i. Pressure cycling and short t	crylic hull with aluminum hydrostatic pressurizations to	

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of 1000 pressure cycles at depths to 3000 feet and the short term implosion pressure is in the range of 10,000 to 11,000 feet. Stress wave emissions have been found to be a good indicator of incipient failure.				
Nemo Model 2000 spherical pressure hulls with panoramic visibility are considered to be acceptable for manned submersibles with an operational depth capability to 3000 feet. The cyclic fatigue life of such hulls is conservatively predicted to be at least 12×10^6 feet hours.				

SUMMARY

Problem

Manned submersibles with spherical acrylic plastic hulls have been known since the conception of the Nemo Hull in 1961 to provide greater panoramic vision at lower cost than steel hulls of the same shape and size equipped with many small portholes. To utilize this concept in Fleet design, the Nemo Hull was officially approved for minimum depth dives to 600 feet and demonstrated capabilities to 1000 feet in 1971. To further benefit Fleet operation, design and fabrication techniques were required to ameliorate fatigue factors inherent to structural joints between plastic and metal parts and thereby achieve the maximum operating depth allowed by the physical properties of acrylic plastic.

Results

Successive technological innovations have yielded three Nemo Hull designs that can be incorporated into existing or planned submersible systems for certified man-rated operation to ocean depths of 1000, 2000, and 3000 feet. New hatch design details have decreased bending moments at metallic hatch/acrylic hull interface and the use of polycarbonate gaskets in the acrylic plastic hatch seat has eliminated shear cracking. The latest of the Nemo Hull series, Model 2000, has a 66-inch outside diameter and a 58-inch inside diameter that yields a fatigue life of 12,000,000 feet hours over a projected 10-year life span and is capable of operation to the maximum depth allowed by the properties of acrylic plastic.

Recommendations

The Model 2000 Nemo Hull is recommended for manned operation at depths to 3000 feet. This latest design can now provide the Navy with a transparent hull for a wide variety of applications in undersea warfare, search, salvage, surveillance, and recovery missions.

DEFINITION OF TERMS

ASTM American Society for Testing of Materials

critical pressure external hydrostatic pressure at which catastrophic failure of the

pressure hull takes place

ft feet; equals 30.48 cm

hoop stress principal membrane stress oriented at right angles to the longitu-

dinal stress

ID inside diameter of acrylic sphere

in inches; equals 2.54 centimeters

longitudinal stress principal membrane stress whose direction passes through poles

of the sphere

NCEL Naval Civil Engineering Laboratory, Port Hueneme, CA

Nemo Hull Acrylic plastic hull with one atmosphere interior for manned sub-

mersibles with the following primary characteristics:

(1) spherical shape

(2) modular assembly of bonded pentagons

(3) polar openings closed by inserts, a hatch at the top and a

plate at the bottom

(4) bearing surfaces on metallic hatch form a spherical angle

whose apex is at the sphere

OD outside diameter of acrylic sphere

PMR Pacific Missile Range, Point Mugu, CA

psi pounds per square inch; equals 0.070 kg/cm²

R_i inside diameter of acrylic sphere

R_O outside diameter of acrylic sphere

short-term pressure pressurization at 100 psi/minute rate

t thickness of acrylic sphere

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INTRODUCTION

The idea of transparent pressure hulls for manned submersible operation became a reality in 1968 with the successful fabrication of the first full scale Nemo Hull shown in Figure 1. Since then, acrylic plastic pressure hulls of spherical shape have demonstrated competitive cost, unsurpassed panoramic visibility, and a large depth safety margin, which has given the Navy a cost-effective solution to submersible hull design.

Testing and evaluation of models and full scale Nemo Hulls continued until in 1970 the Model 600 Nemo Hull was officially approved by the U. S. Navy for manned operation to 600 feet, ¹ although it demonstrated capabilities to 1000 feet, as pictured in operation in Figure 2. At that time, Nemo's endurance range beyond the standard 1000-foot minimum was inhibited by stress limitations inherent to joints between the acrylic hull and metallic hatches. ²⁻⁵ Further design studies were initiated to relieve this fatigue factor and extend the depth limit of the Nemo Hull to the maximum allowed by physical properties of the material for future work in deep ocean environments.

Subsequent work efforts during 1971-1972 resulted in the Model 1000 Nemo Hull with an optimized aluminum hatch and aluminum seating ring assembly. This model has been designed and stress analyzed for safe operations to 2000 feet. Both Models 600 and 1000 Nemo Hulls are recommended as cost-effective transparent hulls for submersible operations to, respectively, 1000 and 2000 feet.

The apex of development for the Nemo Hull culminated in 1974 with the Model 2000, the presentation of which is the purpose of this report. The results of experimental and analytical studies are presented for this acrylic plastic pressure hull, which is capable of manned submersible operation at depths to 3000 feet for at least 1000 times without fatigue cracks. The Model 2000 Nemo Hull provides the maximum safe operational depth with minimum weight to displacement and cost/depth ratios for use with manned submersibles in undersea warfare, search, salvage, surveillance and recovery missions to 3000 feet.

New design techniques and fabrication procedures are presented along with documented test results. The main body of the report is concise but terse to afford the reader a summary of study highlights. A more detailed description of design studies, fabrication procedures, and hydrostatic tests are presented in, respectively, Appendixes A, B, C, and D. The report documents a major breakthrough in technology for submersible hulls with panoramic visibility.

BACKGROUND

The failure of an acrylic plastic spherical Nemo Hull can occur in three ways. The hull can implode instantaneously because of overpressurization accompanying the catastrophic loss of the depth control system in a submersible; this type of failure is classified as short term failure and the pressure at which the failure occurs as short term critical pressure. The hull can also implode after it has been subjected for a long period of time to a

depth which is greater than the operational pressure, but less than the short term critical pressure. This type of failure is classified as *long term creep failure*. Finally, the failure of the hull can be initiated by cracks resulting from repeated dives to operational depth; this type of failure is classified as *fatigue failure*.

Previous experimental studies have already generated the data for prediction of short term and long term creep failures in Nemo Hulls with t/D_0 ratios in the 0.015–0.017, 0.03, 0.047–0.05 and 0.064–0.67 ranges. In conjunction with short term and long term implosion studies some data was also generated on the initiation of fatigue cracks in the acrylic bearing surfaces for metallic hatches on hulls with t/D_0 ratios in the 0.03–0.035 range (references 2, 3, 4, 5). On the basis of that data it was concluded that for all practical applications in acrylic plastic Nemo Hulls the operational depth limitation is imposed primarily by the fatigue life of acrylic bearing surfaces supporting the metallic hatch. It is this fatigue life that imposes the 1000-foot operational depth limit on the 2.5-inch thick Model 600 Nemo Hull assembly and the 2000-foot limit on the nominally 4-inch thick Model 1000 Nemo Hull; the fatigue life being in both cases 1000 dives of 4 hour duration each to the maximum operational depth without initiation of fatigue cracks.

Since the short term collapse depth of the nominally 2.5-inch thick hull is 4150 feet and of the nominally 4-inch thick hull is 10,000 feet, there was no doubt that there existed a sufficient reservoir of potential structural strength to warrant research on improving the fatigue life of the hatch/hull interface. The effort to improve the fatigue life of Nemo Hulls was focused primarily on the Model 1000 Nemo Hull with a nominal 4-inch wall thickness, as the potential operational depth improvement appeared to be higher here than in the Model 600 Nemo Hull with a nominal 2.5-inch thickness.

DISCUSSION

The objective of the study was to maximize the operational depth of the 66-inch diameter Nemo Hull with a 4-inch nominal wall thickness and maintain the minimum 1000 cycle fatigue life requirement specified for all acrylic plastic hulls.^{5,6}

The approach selected was to (1) redesign the top and bottom aluminum inserts, (2) redesign the interface between the insert and the hull, and (3) improve the fabrication process for the hull. All of the modifications to the Nemo Hull design were to be evaluated experimentally and analytically.

The scope was to limit the study to two acrylic hull sizes; the 66-inch diameter, 4-inch thick, full scale operational hull and the 15-inch diameter, 1-inch thick model scale hull. The cyclic fatigue tests and the short term implosion tests were performed on the 15-inch diameter model scale hulls while the experimental determination of stress distribution and comparison with analytical stress calculations were conducted on the 66-inch diameter full scale operational hull.

DESIGN OF POLAR INSERTS

Although the existing design for polar inserts in the 66-inch diameter, nominally 4-inch thick Model 1000 Nemo Hull incorporated on the JOHNSON SEA-LINK submersible was satisfactorily tested to a prooftest of 2000 feet, without yielding and probably could be used to greater depths, it contained a large number of undesirably high stress concentrations and thus was considered questionable for 3000-foot service. For a minimal cost, these hatches were modified for safe operation to 3000 feet.

The redesign of existing aluminum polar inserts for the 66-inch diameter, 4-inch thick hull (currently in service with the JOHNSON SEA-LINK submersible; see Ref. 7) consisted of redistributing the metal in the hatch and penetration plate so that the resultant of compressive membrane stresses in the acrylic hull would pass as close as possible to the centroid of the insert and thus generate only moderate bending moments in the insert; see Figure 3. A more detailed description of system design including schematic drawings is presented in Appendix A.

The new hull assembly was stress analyzed utilizing the finite element stress analysis program for arbitrary axisymmetric structures, called ZP-13, 8 as illustrated in Figure 4.

For this program the top hatch was idealized into 244 nodes and 399 elements, as shown in Figure 5, while the bottom plate was idealized into 228 nodes and 367 elements, shown in Figure 6. The maximum stresses in the aluminum hatch and bottom plate were calculated to occur at the central penetrations and at the junctions between the spherical aluminum shells and the circular flanges (again see Figs. 5 and 6). The highest stress in the plastic hull was predicted to occur on the interior surface at the point of contact between the aluminum flanges and the polycarbonate gasket.

The magnitude of the peak principal stresses in the top and bottom aluminum inserts when operating at a depth of 3000 feet was predicted to be 24,250 and 25,400 psi, respectively (values derived from extrapolation of stresses shown in Figs. 5 and 6). In both cases the peak principal stresses were located on the inside of the central hatch penetrations. These peak compressive hoop stresses were not considered dangerous at that location since the steel bulkhead penetrators would serve as reinforcements and carry some of the compressive stresses. The stresses at the juncture between the spherical surfaces on the hatch and bottom plate and their flanges were also substantial (20,000-23,000 psi), but they were well below the yielding point of aluminum. From an extrapolation of the stress values shown in Figures 7 and 8, the maximum compressive stress in the plastic hull was predicted to be longitudinal and of 10,900 psi magnitude. These stress levels were considered acceptable for operation of the manned Model 2000 Nemo Hull submersible to 3000 feet since the associated calculated safety factors, based on yielding of the aluminum and acrylic plastic, were approximately 1.5 for both materials.

DESIGN OF INSERT/HULL INTERFACE

The insert/hull interface successfully utilized in existing acrylic plastic submersibles NEMO, MAKAKAI, and JOHNSON SEA-LINK #1, consisted of direct contact between the metallic insert flange and the acrylic plastic. 2,9-11 Since the orientation of the interface was radial, shear stresses were minimized. Still, because of differences in structural rigidity

between the metallic inserts and the plastic hull, some differential movement between the inserts and the acrylic hull, as well as bending of the hull, takes place which causes shear cracks to appear in those areas of the acrylic plastic that come in contact with the metallic inserts.

There were two options available for elimination of the shear cracks. One feasible approach was to replace the metallic inserts with discs made of acrylic plastic with the same rigidity and compressibility as the plastic hull. The other approach was to place a compliant gasket between the rigid metallic insert and the limber plastic hull. Of these two approaches, the use of gaskets was operationally more appealing as it allowed the retention of operationally proven metallic inserts with the desirable high heat transfer coefficients. This left only the selection of the right type and thickness of gasket.

The gasket finally selected for the Model 2000 Nemo Hull assembly was one-inch thick polycarbonate plastic (see Fig. 3, Appendix A). Polycarbonate plastic was chosen because of its toughness, resistance to crack propagation, and modulus of elasticity that matches that of acrylic plastic. Because of the 1-inch wall thickness, the gasket possessed sufficient inherent structural strength to prevent it from being extruded into the hull interior by outside hydrostatic pressure. Also, the circumferential flange on the exterior edge of the gasket serves as a seal retainer, which keeps the water from entering the joint between the polycarbonate gasket and the acrylic plastic hull. The seal consisted of room temperature vulcanizing silicone rubber dispensed from a tube into the space between the gasket flange and the acrylic hull.

IMPROVEMENT OF FABRICATION PROCESS

The fabrication process 12 developed for the first acrylic plastic Nemo submersible 1 was also used in the fabrication of the acrylic plastic hulls for MAKAKAI 9 and the JOHN-SON SEA-LINK 7 and focused primarily on the attainment of tight dimensional tolerances for the sphericity of the hull exterior. The variation in thickness of individual spherical pentagon modules remained a function of commercial casting tolerances for individual plates. Because of this dependency, the thickness of individual spherical pentagons varied as much as ± 0.250 of an inch.

The large variation in thickness was acceptable so long as it was economically acceptable to rate the operational depth capability of a Nemo Hull on the basis of minimum hull thickness, and the mismatch in thickness between individual spherical pentagons was not considered optically objectionable. However, when emphasis was placed on cost effectiveness of the acrylic hull as a structure and as an optical system, it became economically untenable to tolerate such a large variation in wall thickness. The thicker portions of the hull would have constituted additional ballast that detracted from the payload and added nothing to the depth capability. In addition, the mismatches in thickness between individual spherical pentagons would have created a noticeable optical distortion for the crew of the vehicle.

Obviously, the key to uniformity of wall thickness was to use thick acrylic plastic spherical pentagons of uniform thickness in the construction of the hull. Basically, there were three techniques available for attainment of uniformly thick spherical pentagons:

(1) custom casting of acrylic plates to very tight dimensional tolerances, (2) grinding

off-the-shelf acrylic plates to a uniform thickness, and (3) grinding spherical sectors to uniform curvature and thickness after thermoforming of plates. Of these three techniques the last one was found to produce the smallest thickness variation at approximately the same or less cost than required by the other techniques. The tolerances on hull thickness attained by grinding of formed spherical sectors were ±0.050 inches.

FABRICATION

Full Scale Assembly

Acrylic Hull

The 66-inch OD × 58-inch ID Model 2000 acrylic plastic Nemo Hull (see Figs. 9 and 10) was fabricated by Swedlow, Inc., basically in the same manner as the previously built Nemo Hulls. 12.13 The only improvements over the previous fabrication technique were the contour grinding of formed spherical sectors and placement of adhesive into the joints between spherical pentagons by means of hydrostatic pressure. Fabrication techniques, dimensional drawings, and relevant contract correspondence are presented in Appendix B.

Acrylite® plates manufactured by Monsanto served as basic construction material. The stringent material quality control procedures developed by NCEL and PMR for the prototype Model 600 Nemo Hull were applied here also. Testing of material specimens showed that the 4.125 nominally thick Acrylite met the physical properties criteria listed in Table 1, as established by the Navy for acrylic plastic windows and pressure hulls in manned service. 2-5

Because of improved fabrication techniques, not as many dimensional tolerances were required in building the Model 2000 Nemo Hull as were required for the Model 1000 Nemo Hull fabricated for the JOHNSON SEA-LINK submersible. Thus, whereas the Model 1000 Nemo Hull thickness varied from 3.844 to 4.030 inches, the Model 2000 Nemo Hull varied only from 4.000 to 4.100 inches. Similarly, whereas the sphericity of Model 1000 Nemo Hull varied from 66.250 to 65.800, the Model 2000 Nemo Hull varied only from 66.095 to 65.900 inches. Assembly dimensions for the Nemo Model 2000 are listed in Table 2.

The only area that did not realize a significant improvement in the fabrication process was the bonding of joints between individual pentagons. A comparison of the 5456 to 7804 psi bond strength achieved for the 4-inch thick Model 1000 Nemo Hull with that of the 5123 to 9116 psi bond strength attained for the Model 2000 Nemo Hull indicates that the strength of bonded joints in both hulls is about the same. This holds true also for the quality of the joint. Both the number and size of inclusions was about the same as shown in Figure 11 a and b. This indicates that although the technique of emplacing the adhesive into the joint and the polymerization regimen have been drastically changed since the fabrication of the first Model 600 Nemo Hull in 1968² the performance of the bonded joint has not. Since the entire Nemo Hull is under compression when submerged to operational depth, very little incentive exists to effect further improvements in the tensile qualities of the joints.

Table 1. Physical Properties of Acrylic Plastic Hull for 66 Inch OD × 58 Inch ID Nemo Model 2000

		Minimum	Average	Maximum
1.	Properties of Plastic*			
	ASTM D-638			
	Ultimate Tensile Strength, psi Elongation, percent Modulus of Elasticity, psi	9,545 3.0 428,000	10,972 5.291 465,583	12,331 7.0 505,000
	ASTM D-790			
	Ultimate Flexural, psi Modulus of Elasticity, psi	15,238 415,000	17,736 463,125	18,686 487,000
	ASTM D-732			
	Ultimate Shear Strength, psi	9,880	10,088	11,500
	ASTM D-695			
	Compressive yield, psi Compressive modulus, psi	17,700 500,000	18,416 520,416	19,600 570,000
	ASTM D-621			
	Deformation under load; 4000 psi, 122°F for 24 hours	0.42	0.55	0.72
2.	Properties of bonded joints**			
	ASTM D-638			
	Ultimate tensile strength, psi	5,123	7,815	9,116
		e	ŧ	

^{*}Total of 120 specimens taken from 12 acrylic plastic plates with $4.125 \times 48 \times 60$ inches nominal dimensions.

^{**}Total of 12 specimens taken from test blocks bonded for quality control purpose.

Table 2. Dimensions of the 66 Inch OD X 58 Inch ID Hull for Nemo Model 2000 Assembly

1. Individual Pentagons

	Thickness*		Contour Deviation**	
Pentagon	Maximum	Minimum	Maximum	Minimum
Α	4.070	4.035	0.075	0.005
В	4.075	4.040	0.070	0.010
C	4.070	4.020	0.070	0.010
D	4.100	4.050	0.100	0.030
E	4.070	4.050	0.070	0.010
F	4.110	4.005	0.040	0.020
G	4.060	4.030	0.150	0.050
Н	4.065	4.010	0.100	0.020
I	4.090	4.060	0.100	0.010
J	4.070	4.050	0.100	0.020
K	4.050	4.000	0.100	0.020
L	4.030	4.000	0.090	0.010

2. Sphere Assembly

Spherical Deviations	Maximum	Minimum	
Total of 5 Measurements	-0.100	+0.095	

^{*}Total of 6 measurements per pentagon

Metallic Polar Inserts

The top hatch, top hatch ring, and bottom plate for the 66-inch OD hull were fabricated by machining 6061-T6 aluminum forgings, as pictured in Figures 12a, b and 13a, b (also see Appendix B). Material quality control of the aluminum forgings indicated a compressive yield strength of 36,300 psi and ultimate compressive strength of 43,100 psi. Special attention was paid to the machining of beveled seal seating surfaces to assure positive sealing and good bearing contact.

^{**}Total of 5 measurements per pentagon

Polycarbonate Bearing Gasket

The bearing gaskets shown in Figure 14 between the metallic polar inserts and acrylic plastic hull were machined from polycarbonate plastic plates. Material quality control of the polycarbonate plates used as machining stock showed that the material was acceptable for this application. The physical properties are summarized in Table 3.

Table 3. Physical Properties of Polycarbonate Plastic Gaskets for 66 Inch OD X 58 Inch ID of Nemo Model 2000

	Minimum	A verage	Maximum
ASTM D-638			
Ultimate Tensile Strength, psi	6,640	7,170	7,690
Elongation, percent Modulus of Elasticity, psi	2.3 320,000	2.6 320,000	2.8 320,000
ASTM D-790			
Ultimate Flexural, psi	11,700	11,900	12,000
<u>ASTM D-732</u>			
Ultimate shear strength, psi	10,400	10,400	10,400
ASTM D-695			
Compressive yield, psi	12,500	12,800	13,000
Compressive modulus, psi	360,000	370,000	380,000
ASTM D-621			
Deformation under load, percent 4000 psi, 122°F for 24 hours	0.12	0.13	0.14
ASTM D-256			
Izod impact strength	0.67	0.76	0.85
ASTM D-570			
Water absorption, percent 24 hours	0.14	0.15	0.15

Since a polycarbonate plate of 6-inch thickness is currently not fabricated by industry, several half-inch thick plates were bonded together to form a 6-inch thick block. Although the bonding was performed by General Electric, the developer of polycarbonate plastic, the quality and strength of the joints were less than of the parent material. Additional efforts by Southwest Research Institute (SWRI) were made to improve the bonding technique, although these too failed to yield perfect joints. However, in view of the fact that the gasket is in compression when incorporated into the Model 2000 Nemo Hull assembly, less than perfect bonded joints were considered as acceptable for compressive loading service.

Scale Model Assembly

Acrylic Hull

The 15-inch OD × 13-inch ID acrylic plastic hulls were fabricated by the Technical Support Department of the Pacific Missile Range; Figure 15 pictures one of these hulls. The same thermoforming, machining, and bonding techniques were used to fabricate this scale model as were used in the fabrication of prototype models developed for the Nemo research program in 1965.² Quality control of acrylic plastic and of bonded joints showed that the scale model materials met the same specifications as did the full scale 66-inch diameter hull. Four scale models have been fabricated.

Metallic Polar Inserts

The polar inserts for the 15-inch OD model scale hulls were fabricated by machining aluminum and titanium forgings (see Appendix A). The 6061-T6 aluminum inserts were structural scale models of the aluminum polar inserts in the 66-inch diameter hull; see Figures 16a, b and 17a, b. The Ti-6Al-4V titanium inserts shown in Figure 18a, b, c, d represented simplified scale models of an alternate design for the 66-inch OD hull polar inserts, except that titanium was utilized instead of aluminum.

Polycarbonate Bearing Gaskets

The polycarbonate bearing gaskets (see Appendix A) were machined from 1-inch thick polycarbonate plates (Lexan CP-438). Common machine shop practices were used to achieve the desired finish and tolerances.

TEST PROGRAM

Model Scale Hulls

Static Tests

One 15-inch OD X 13-inch ID scale model hull, shown in Figure 19, was subjected to a series of hydrostatic tests which culminated in the implosion of the hull. The objectives of the static tests were to (1) establish the validity of aluminum hatch design for ocean depth operation to 3000 feet, (2) measure the stress wave emissions of the acrylic hull, (3) measure the creep of the hull at depths to 3000 feet, and (4) determine the short term implosion depth of the Model 2000 Nemo Hull assembly. To accomplish these objectives the scale model of the Model 2000 Nemo Hull was instrumented with 10 electric resistance strain gages and an acoustic transducer with a 160 kHz response capability, as shown in Figure 20.

Static tests were conducted at a room temperature range between 70-75°F in the pressure test facilities of the Southwest Research Institute, San Antonio, Texas.

- 1. Pressurize the 15-inch OD X 13-inch ID Nemo Model 34 to 1350 psi at 100 psi/minute rate, hold for 4 hours at that pressure, depressurize at 100 psi/minute rate to 0 psi, allow to relax for 4 hours prior to next test.
- 2. Pressurize the 15-inch OD X 13-inch ID Nemo Model 34 to 900 psi at 100 psi/minute rate, hold for 4 hours at that pressure, depressurize at 100 psi/minute rate to 0 psi, allow to relax for 4 hours prior to next test.
- 3. Pressurize the 15-inch OD × 13-inch ID Nemo Model 34 to implosion at 100 psi/minute rate.

Cyclic Tests

Three 15-inch OD X 13-inch ID Nemo Hull scale models were subjected to a series of pressure cycling tests. The objective of the pressure cycling tests was to establish (1) the fatigue life of bearing surfaces in acrylic plastic hulls that are in direct contact with metallic polar inserts and (2) the fatigue life of bearing surfaces in acrylic plastic hulls that are not in direct contact with the metallic polar inserts but interface through a polycarbonate bearing gasket; an assembly drawing of these contact points is shown in Figure 21. To achieve these objectives, the bearing surfaces of the acrylic hulls were to be inspected at the conclusion of the pressure cycle tests.

Pressure cycling of the scale model consisted of a series of tests conducted at a room temperature range between 70-75°F in the pressure test facilities of the Naval Civil Engineering Laboratory (NCEL), Port Hueneme, California, performed as follows.

1. Pressurize the 15-inch OD X 13-inch ID Nemo Model No. 35 to 500 psi at 100 psi/minute rate; hold at this pressure for 4 hours; depressurize at 100 psi/minute rate to zero and relax at that pressure for 4 hours before initiating the next pressure cycle. Repeat the pressure cycle 1000 times.

- Pressurize the 15-inch OD X 13-inch ID Nemo Model No. 36 to 1000 psi at 100 psi/minute rate; hold at this pressure for 4 hours, depressurize at 100 psi/minute rate to zero and relax at that pressure for 4 hours before initiating the next pressure cycle. Repeat the pressure cycle 1000 times.
- 3. Pressurize the 15-inch OD × 13-inch ID Nemo Model No. 37 to 1500 psi at 100 psi/minute rate; hold at this pressure for 4 hours; depressurize at 100 psi/minute rate to zero and relax at this pressure for 4 hours before initiating the next pressure cycle. Repeat the pressure cycle 1000 times.

Full Scale Hull

Static Tests

The 66-inch OD X 58-inch ID full scale Model 2000 Nemo Hull assembly was subjected to a series of hydrostatic tests at SWRI, which culminated with a 4000-foot depth proof test; Figure 22 pictures the Model 2000 Nemo Hull with the SWRI pressure vessel. The objectives of the static tests were to (1) establish experimentally the strains and stresses imposed on the Model 2000 Nemo Hull assembly at a 3000-foot operational depth for comparison with the analytically generated data and (2) prove that the full scale Model 2000 Nemo Hull assembly could withstand pressures to a depth of 3000 feet without permanent deformation. To accomplish these objectives, strains were to be recorded during all of the tests at 20 locations; the location of the strain gages on the Model 2000 Nemo Hull is shown in Figure 23.

The static test series consisted of the following tests conducted at a room temperature which ranged between 65-75°F in the pressure test facilities of the Southwest Research Institute, San Antonio, Texas:

- 1. Pressurize to 450 psi at 100 psi/minute rate, hold at that pressure for 24 hours, depressurize to 0 psi at 100 psi/minute rate, and relax at that pressure for 24 hours prior to beginning of next test.
- 2. Pressurize to 900 psi at 100 psi/minute rate, hold at that pressure for 24 hours, depressurize to 0 psi at 100 psi/minute rate, and relax at that pressure for 24 hours prior to beginning of next test.
- 3. Pressurize to 1350 psi at 100 psi/minute rate, hold at that pressure for 24 hours, depressurize to 0 psi at 100 psi/minute rate, and relax at that pressure for 24 hours prior to beginning of next test.
- 4. Pressurize to 1800 psi at 100 psi/minute rate, hold at that pressure for 24 hours, depressurize to 0 psi at 100 psi/minute rate, and relax at that pressure for 24 hours prior to beginning of next test.

No cyclic tests were performed on the full scale Model 2000 Nemo Hull assembly.

TEST OBSERVATIONS

Model Scale Tests

Stresses

The 15-inch OD X 13-inch ID Nemo Model 34 assembly performed satisfactorily at simulated depths to 3000 feet. The highest measured principal stress of -5086 psi in the acrylic hull was on the interior, located at the edge of the top polar opening (0.500 inches away from aluminum hatch) and orientated along the meridian of the sphere; recorded stress values are listed in Table 4 (also see Figure 19). It is worth noting, however, that its magnitude was approximately only 10 percent larger than the average stress of -4642 psi measured at the equator on the interior of the sphere. This can be explained by the fact that the stress riser effect of the metallic insert decays rapidly with distance from the hatch. Since the strain gage was located approximately 3 degrees away from the edge it did not measure the peak stress but rather the tail end of it.

The maximum stress of -15,714 psi in the polar aluminum inserts was measured on the inside surface, adjacent to the flange, in the bottom plate, and its orientation was in the longitudinal direction.

The highest measured stresses, both on the acrylic hull and the aluminum inserts during the simulated dive to 3000 feet, were well below the yield points of their respective materials. All strains were observed to return to zero upon completion of the relaxation period following the simulated dive to 3000 feet (see Table 4). For both the acrylic plastic and aluminum, the apparent safety factors, based on the short term yielding of material, were well in excess of 2. On the basis of these stress measurements it was concluded that (1) the proposed design of hatches was adequate for dives to 3000 feet and that (2) the whole full scale Model 2000 Nemo Hull assembly could be safety tested at least at depths to 3000 feet.

Implosion Resistance

The 15-inch OD X 13-inch ID Model 34 Nemo Hull imploded under short term pressure loading at a simulated depth of 10,600 feet. The assembly failed by general plastic instability: the fragmented model is shown in Figure 24. The highest measured stresses in the aluminum hatches, prior to implosion, were found to be at locations #4 and #7 (see Figure 19), and their magnitude was in the -35,000 to -39,000 psi range, as shown in Figure 25. The 10,600-foot short term implosion depth of Model 34 gives the scale model a 3.5 safety margin for catastrophic dives. Data reduction of the hydrostatic tests is documented in Appendix C.

Acoustic Emission

The 15-inch OD X 13-inch ID Model 34 was a good source of acoustic emission during the first pressurization to 1350 psi; Figure 26 presented a histogram of stress wave

Table 4. Strains in 15 Inch OD X 13 Inch ID Nemo Model #34 during Simulated Dive at Depths to 3000 Feet

Gages			Strain micro inches/inch		Stress (psi)	
No.	Location		Ноор	Longitudinal	Ноор	Longitudinal
la	Equator, outside	A B C	-5,900* -6,700 0	-5,700* -6,300 -100	-3,345*	-3,284*
lb	Equator, inside	A B C	-8,200* -9,550 +10	-8,000* -8,900 +25	-4,681*	-4,604*
2b	Edge of top polar opening, inside	A B C	-6,900* -7,700 -50	-9,500* -1,100 -100	-4,286*	-5,086*
3b	Lip of flange, bottom plate; inside	A B C	-900* -900 0	+150* +150 0	-9,386*	-1,319*
4 a	Root of flange, bottom plate; outside	A B C	-275* -275 0	-50* +100 0	-3,187*	-1,456*
4b	Root of flange, bottom plate; inside	A B C	-600* -700 0	-1,250* -1,350 0	-10,714*	-15,714*
5a	Root flange, top hatch, outside	A B C	-100* -100 0	malfunc- tioning s. gage 0	-1,099*	-330*
5b	Root of flange, top hatch, inside	A B C	-1,100* -1,100 0	-200* -200 0	-12,747*	-5,824*
6b	Edge of bottom, polar opening, inside	A B C	-8,300* -9,200 0	-6,900* -7,800 +250	-4,558*	-4,127*
7b	Lip of flange, top hatch, inside	A B C	-1,300* -1,300 0	+300* +300 0	-13,297*	-989*

Note A.* Immediately after pressurization to 3000 foot depth

B. After four (4) hours at 3000 foot depth

C. After 16 hours of relaxation at 0 depth

emissions. When, after relaxation at 0 psi, Model 34 was pressurized to 900 psi no further acoustic emission bursts were recorded which indicated that the acrylic hull exhibits a very marked Kaiser effect.

During the final pressure test to implosion, Model 34 emitted significant numbers of acoustic emissions, although only after the pressure passed the 9500-foot depth mark. Thus, between 0 and 9500 feet there were less than 50 emissions, as shown in Figure 27. Obviously then, the impending implosion of the acrylic hull could have been stopped during the simulated dive at about 500 feet above implosion depth on the basis of the acoustic emission recording (see Fig. 26).

Cyclic Fatigue Crazing

Observation of 15-inch OD × 13-inch ID Models 35, 36 and 37, after 1000 simulated 4-hour long dives, shown in Figure 28, revealed that only Model 37 which was pressure cycled to a depth of 3360 feet had slight indication of cyclic fatigue, whereas Models 35 and 36 which were pressure cycled to, respectively, 1120 and 2240 feet showed no signs of cyclic fatigue. The cyclic fatigue in Model 37 exhibited slight crazing of its conical bearing surface in the polar opening of the acrylic hull, which was exposed to direct contact with the metallic insert; see Figure 29. The other polar opening in Model 37, in which the acrylic bearing surface was not in direct contact with the metallic insert did not craze. From this, it can be concluded that at a cyclic history of less than 1000 dives the polycarbonate gasket has a significant effect only when the maximum pressure in a dive is 3360 feet or more. At lesser depths the polycarbonate gasket also increases the fatigue life of the acrylic hull, although more than 1000 dives are required to show experimentally the beneficial effect of the polycarbonate gasket.

It is important to point out here that even in Model 37 which was the only specimen with signs of cyclic fatigue on the acrylic bearing surface, the fatigue exhibited itself in the form of barely noticeable crazing. Based on past experience,⁵ it can be conservatively predicted that it would take at least another 1000 dives to 3360 feet before the crazing would deteriorate into cracks 1/2-inch deep and thus require remachining of the bearing surface.

Creep

The creep observed during 4 hour sustained loading to 1350 psi was significantly higher than during sustained loading to 900 psi, as shown in Figures 30 and 31. The magnitude of creep in both cases was about 15 percent of short term strain. As expected (magnitude of creep is a function not only of time but also of short term strain), creep was more substantial at the edges of polar openings than at the equator. Similarly, it was larger on the interior of the hull than on its exterior,

The creep returned to zero after several hours of relaxation at zero pressure, indicating that the creep observed did not represent permanent deformation of plastic.

Full Scale Tests

General Performance

The 66-inch OD × 58-inch ID Nemo Model 2000 withstood successfully the four successive 24-hour hydrostatic pressure loadings to 450, 900, 1350 and 1800 psi without any appearance of cracks in the acrylic and only minor surface cracking in the polycarbonate plastic bearing surfaces at the polar openings.

Strains

The magnitude of strains observed during the 24-hour pressurization tests is shown in Figure 32; recorded stress values are listed in Table 5. Stress range was predicted by (1) the ZIP-13 finite element computer program and (2) strains generated during the hydrostatic testing of the 15-inch OD X 13-inch ID Model 34. The fact that the acrylic hull of Model 34 was approximately 10 percent thicker than required by the 1:4.4 scaling factor had to be taken into consideration during comparison of strains measured on the 15-inch and 66-inch diameter hulls.

The highest strains in acrylic plastic were measured on the interior of the hull at the edge of the top polar opening. The strains at the edge of the bottom polar opening were about 10 percent less, reflecting the fact that the bottom aluminum plate is significantly less stiff than the top hatch. The ratios between longitudinal and hoop strains at both locations were in the 3:1-4:1 range.

The interior longitudinal strain at the top polar opening was 100 percent greater than the interior longitudinal strain at the equator, while the interior hoop strain at the top polar opening was 50 percent less than the interior hoop strain at the equator. The exterior longitudinal strain at the top polar opening was only 20 percent greater than the exterior longitudinal strain at the equator, while the exterior hoop strain at the top polar opening was 70 percent less than the exterior hoop strain at the equator. On the aluminum polar inserts the highest strain was measured on the interior surfaces of (1) the bottom plate at location #6 in longitudinal direction and (2) the top plate at location #13 in longitudinal direction (see Fig. 23).

Magnitude of *Creep* at the equatorial surfaces of the hull was approximately the same as that recorded for Model 34. It was for all practical purposes absent during the 24 hour pressurizations to 450 and 900 psi, but it became noticeable (20-25 percent increase over short term strain) during 1350 psi pressurization and was significant (25-30 percent increase) during 1800 psi pressurization, as can be seen in Figure 32a through 1. The numerical value of strains on the interior surface at the equator after 24 hours of sustained loading was in the 2500-3000, 5000-6000, 9000-11,000 and 13,000-15,000 micro inches/inch range for, respectively, 450, 900, 1350 and 1800 psi pressurizations. (See Figs. 32i and 32j.)

The numerical values of creep on the interior hull surface at the edges of top and bottom penetrations were higher than at the equator, but in terms of short term strain percentage they were not different from those at the equator. After 24 hours of sustained loading the longitudinal strains at penetrations were in the 4000-4500, 8000-10,000, 15,000-19,000 and 22,000-27,000 micro inches/inch range for, respectively, 450, 900, 1350 and 1800 psi

Table 5. Stresses in 66 Inch OD × 58 Inch ID Nemo Model 2000 Assembly During the 24 Hour Dive to a Depth of 3000 Feet

	Gages	Stress (psi)
Hull Location	Orientation	Upon reaching 3000 feet	After 24 hours
Inside #1 Outside Inside #2 Outside	Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal	-4,986 -7,914* -2,348 -3,819 -5,476 -5,290 -4,214 -4,186	Stresses cannot be calculated
Inside #3 Outside Inside #4	Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal	-4,486 -6,714 -1,900 -3,400 -5,595 -5,438	because of creep in acrylic
Outside	Hoop Longitudinal	-4,086 -4,014	
Inside #5 Outside	Hoop Longitudinal Hoop Longitudinal	-10,495 -11,648 -5,000 -5,000	-9,396 -11,319 -4,835 -4,451
Inside #6 Outside Inside #7 Outside	Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal Hoop Longitudinal	-13,626 -17,088** -6,264 -5,879 -13,297 -10,989 -6,429	-13,956 -18,187** -5,549 -5,165 -12,967 -9,890 -6,429 -6,429

^{*}Highest stress in acrylic hull (during conversion of strains to stresses E = 400,000 psi and $\mu = 0.4$ were applied).

^{**}Highest stress in polar aluminum inserts (during conversion of strains to stresses E = 10,000,000 psi and $\mu = 0.3$ were applied).

Table 5. (Continued).

G	ages	Stress (psi.)
Hull Location	Orientation	Upon reaching 3000 feet	After 24 hours
Inside	Ноор	-13,022	-13,187
#8	Longitudinal	-3,407	-3,956
Outside	Hoop	-3,516	-2,967
o atora c	Longitudinal	- 55	+110
Inside	Hoop	-10,549	-9,341
#9	Longitudinal	-15,165	-12,802
Outside	Hoop	-4,780	-4,780
Outside	Longitudinal	-5,934	-5,934
Inside	Ноор	-10,549	-9,835
#10	Longitudinal	-10,165	-9,451
Outside	Ноор	-4,670	-4 ,670
Outside	Longitudinal	-3,901	-3,901
Inside	Ноор	-10,165	-9,066
#11	Longitudinal	-10,549	-10,220
#11 Outside	Hoop	-3,736	-3,736
Outside	Longitudinal	-4 ,121	-4,121
Inside	Hoop	-11,429	-11,429
#12	Longitudinal	-11,429	-11,429
	Hoop	-3,956	-3,956
Outside	Longitudinal	-3,187	-3,187
1: 4	Ноор	-11,758	-13,352
Inside	Longitudinal	-12,527	-14,505
#13	Ноор	-3,956	-6,813
Outside	Longitudinal	-3.187	-6,044
Incida	Hoop	-12,692	-14,286
Inside	Longitudinal	-12,308	-14,286
#14	Ноор	-4, 670	-7,527
Outside	Longitudinal	-3,901	-6,758

pressurizations (see Figs. 32b and 32f). Strains in acrylic returned essentially to zero after a 24-hour period of relaxation indicating that the creep in acrylic was not of a permanent nature even after the 24 hour sustained loading to 1800 psi hydrostatic pressure.

Stresses

The maximum stress measured on the acrylic hull (see Table 5) at the beginning of 24 hour pressurizations was -2339, -5043, -7914 and -10,962 psi at, respectively, 450, 900, 1350 and 1800 psi pressure loadings. The maximum stress, analyzed as typical for Nemo hulls, ¹⁴ was located on the interior surface of the hull at the edge of top polar opening and was oriented in the longitudinal direction. The stress on the interior equatorial surface was measured simultaneously as -1804, -3610, -5595 and -7757 psi. The magnitude of stress on the acrylic hull at the conclusion of the 24 hour pressurization periods is not known since there was considerable creep in the plastic which would make any classical stress calculations inaccurate.

The maximum stress on the aluminum inserts was measured on the interior of the bottom plate at location No. 6 in the longitudinal direction. The magnitude of the stress at the beginning of 24 hour pressurizations was -4967, -9890, -17,088 and -21,044 psi at, respectively, 450, 900, 1350 and 1800 psi loadings. At the conclusion of the 24-hour pressurization periods the magnitude of the stress had changed to -3198, -9890, -18,187 and -18,846 psi, respectively. After the 24-hour relaxation periods following pressurizations to 450, 900 and 1350 psi, all stresses in aluminum returned essentially to zero, as listed in Table 6. A different case presented itself at the conclusion of the relaxation period following the 24-hour pressurization to 1800 psi. Here the stresses at interior location Nos. 6, 13 and 14 on aluminum inserts not only failed to return to zero (see Table 6) but showed residual positive stresses of significant magnitude, and the reasons for their presence are not known. A more detailed listing of stresses is presented in Appendix C.

The comparison of stresses calculated on the basis of experimental data and the ZIP-13 finite element computer program show good agreement for all locations on the acrylic hull. For locations on aluminum inserts the agreement is not as good. It appears that for the locations on the exterior of aluminum inserts the calculated stresses are generally lower than measured values, whereas for locations on the interior of the inserts the calculated values are generally higher. However, since the highest stresses measured on aluminum inserts were on the interior surface, the calculated values tend to be conservative in nature and, thus, useful in the design of pressurized Nemo Hulls. A complete listing of computer generated strains and stresses for the Model 2000 Nemo Hull assembly is presented in Appendix D.

Table 6. Residual Strains in Aluminum Plates and Hatches of the 66 Inch OD X 58 Inch ID Nemo Model 2000 after Repeated 24 Hour Long Pressurizations

		Gage Locations						
		No	o. 6 Inside	No. 13 Inside		No. 14 Inside		
Test		Ноор	Longitudinal	Ноор	Longitudinal	Ноор	Longitudinal	
450 psi	Α	-240	-380	-190	-200	-250	-140	
	В	-170	-240	-120	-140	-250	-140	
	C	+120	+150	+100	+110	+10	+80	
	D	+160	+180	+130	+210	+80	+60	
900 psi	Α	-500	-750	-450	-500	-500	-500	
	В	-500	- 750	- 450	-500	-500	-500	
	C	0	+100	+50	+0	+50	0	
	D	+50	+150	+100	+0	+100	50	
1350 psi	Α	-850	-1300	-800	-900	-900	-850	
	В	-850	-1400	-900	-1050	-1000	-1000	
	C	0	0	+0	-100	-100	-150	
	D	0	0	+0	-150	-200	-200	
1800 psi	A	-1050	-1400	-1000	-1150	-1150	-1100	
	В	-1050	-1400	-850	-1050	-1050	-1000	
	C	+50	+300*	+200*	+100*	+50*	+500*	
	D	-50	+2350*	+1450*	+1350*	+1350*	+1250*	

A – Immediately after pressurization

B - After 24 hours of sustained pressurization

C – Immediately after pressure release

D - After 24 hours of relaxation

^{* -} Questionable values, probably generated by malfunctioning bulkhead penetrators for instrumentation in Model 2000 Nemo Hull, or pressure vessel end closure.

TEST DATA DISCUSSION

Determination of Safe Operational Depth

In order for the chosen operational depth to be safe, many operational, as well as hull performance parameters, must be considered and carefully calculated.

Hull Performance Parameters

The short term critical pressure at which catastrophic implosion of the hull occurs in an uncontrolled dive is the best known and easiest to obtain performance parameter of an acrylic hull. The short term critical pressure represents the ultimate depth beyond which a submersible cannot descend at any time. For the Model 2000 Nemo Hull the short term critical pressure has been experimentally established at approximately 10,000 feet. The actual short term implosion test was performed on the 15-inch Model 34, which imploded at 4700 psi external hydrostatic pressure. Since the scale model is about ten percent thicker than required, the extrapolated short term implosion pressure for the full scale Model 2000 Nemo Hull is around 4000 psi if the same pressurization procedure is used as for the scale model. However, since the pressurization schedule for Model 34 did not correspond to the typical 100 psi/minute short term pressurization rate for acrylic hulls² (recording of strain data at 4500 and 4700 psi pressure levels delayed the pressurization by 5 minutes), the extrapolated short term collapse pressure for the Model 2000 Nemo Hull must be increased from 4000 psi to at least 4500. (Reference 6 indicates that the effect of delay in pressurization at pressures above 4500 psi is to reduce the short term implosion pressure of acrylic hull by about 100 psi for every minute of delay.) The 10,000-feet short term implosion depth gives the Model 2000 Nemo Hull the ability to bounce dive once under extreme emergency conditions probably to at least 8000 feet.

The long term critical pressure of acrylic hulls has been previously established as a function of time and temperature. Because 100 hours is considered the maximum length of time that the crew of a submersible could survive under entrapment without new air support supplies, this time span will be used to establish a long term critical pressure. This pressure can be readily determined from a plot of experimental data generated by implosions of 15-inch OD × 13-inch ID Models 22, 23, 24, 25 and 34, as illustrated in Figure 33. From the plot one can see that the implosion pressure of a scale model Nemo Hull under 2700 psi sustained loading at 70-75°F at ambient temperatures occurs after 100 hours. After application the 0.86 correction factor (based on plastic instability, takes into account the ten percent thicker hull of scale models), the projected 100 hour long term critical pressure of the Model 2000 Nemo Hull is 2320 psi in the 70-75°F ambient temperature range. In terms of depth it can then be stated that the Model 2000 Nemo Hull must be trapped for at least 100 hours at a depth of about 5000 feet before catastrophic failure occurs.

The cyclic fatigue life of acrylic hulls has been the subject of several studies since, as a rule, it is the determining factor in setting the safe operational depth of an acrylic hull. Since the cyclic fatigue life is not only a function of maximum pressure in the pressure cycle but also of duration and temperature, they all must be taken into consideration. Study of typical dive profiles for submersibles has established the fact that a submersible does not

stay at maximum operational depth longer than 4 hours. The rest of the typical dive is taken up by launching, descent, ascent, docking and retrieval. The temperature can vary widely during a dive but at operational depths it is usually below 50 degrees. Since pressure cycling at 70-75°F is not only more conservative, but also more economical, it was used to establish the cyclic fatigue life of the Model 2000 Nemo Hull.

The testing of 15-inch Models 35, 36 and 37 has conclusively shown that crazing appears in the acrylic hull at the polar openings without the polycarbonate gasket only after 1000 pressure cycles of 8-hour duration (4 hours loading followed by 4-hours relaxation) to 1500 psi. No crazing was observed in the polar opening of the acrylic hull protected by the polycarbonate gasket. Judging by these results the minimum crack-free fatigue life of the 15-inch OD \times 13-inch ID Models is 1000 cycles to a maximum operational depth of 3350 feet. Based on the scale model data, the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull can perform 1000 dives to 3000 feet without initiation of cracks in the acrylic hull.

Operational Performance Parameters

In view of the fact that preservation of the crew is the major consideration in the design of pressure hulls it is considered mandatory that the short-term and long-term critical pressures be beyond the depth to which the submersible may be accidentally submerged. Furthermore, it is considered reasonable and customary that the implosion depth for a long term (no more than 100 hours) disabled submersible be at least 50 percent greater than the maximum operational depth (safety factor of 1.5). For a short term loss of control, the implosion depth should be at least 100 percent greater than the maximum operational depths (safety factor of 2).

In addition to preserving the crew there are also the economics of the hull life to be considered. If the fatigue life was set at 100 dives it would prove economically unsound since it would allow the submersibles to operate only for a period of time less than two years, although at greater depths. Similarly, if the fatigue life was stipulated as 10,000 cycles it would give the submersible unlimited life but at the cost of very shallow operational depth, which would significantly lower its operational usefulness. It is the author's opinion that a specified crack-free fatigue life of 1000 cycles represents a sound economical compromise between the operational depth and life of the submersible. For the full scale Model 2000 Nemo Hull the crack-free fatigue life has been experimentally established as 1000 dives to a maximum operational depth of 3000 feet. Since the 3000-foot fatigue life depth is based on 4-hour long simulated dives, there is no need to apply any pressure cycle duration discounting factor to the experimentally established fatigue life depth of 3000 feet.

Based on the factors discussed above, the maximum operational depths should not exceed 4000 feet (8000 feet/2) for short term disablement criterion, 3330 feet (5000 feet/1.5) for long term disablement criterion, and 3000 feet (3000 feet/1) for the fatigue life criterion. Since it is the least permissible operational depth, based on any of the above three criteria, that determines the actual depth rating of the hull, fatigue becomes the determining factor for establishing the operation depth rating of the Model 2000 Nemo Hull. As a result 3000 feet is considered as the maximum operational depth rating for the Model 2000 Nemo Hull.

FINDINGS

- 1. The 66-inch OD X 58-inch ID spherical capsule assembly, Model 2000 Nemo Hull, fabricated from commercial grade (Plexiglas G or equivalent) acrylic plastic and equipped with polycarbonate gaskets between aluminum hatches and the acrylic plastic will withstand a minimum of 1000 dives (4 hours at maximum depth, followed by 4 hours at the surface) from 0 to 3000 feet without initiation of cracks in the acrylic hull.
- 2. At the safe maximum operational depth of 3000 feet the maximum compressive stresses in aluminum hatches and acrylic plastic hull are only equal to 49 and 52 percent of, respectively, aluminum and acrylic plastic yield strengths.
- 3. Model 2000 Nemo capsule assembly will withstand accidental disablement at a depth of 5000 feet for at least 100 hours before catastrophic failure occurs. At greater depths the grace period prior to catastrophic failure is significantly shorter, as shown in Figure 33.
- 4. Model 2000 Nemo capsule assembly will withstand a temporary loss of control to a depth of 8000 feet for about 10 minutes before catastrophic failure occurs.
- 5. Model 2000 Nemo capsule assembly is an active acoustic stress wave emitter whose rate of acoustic emissions increases significantly just prior to short term implosion.
- 6. Permanent deformation of aluminum inserts (top hatch and bottom plate) takes place in areas of high stress concentrations when Model 2000 Nemo Hull is subjected to dives of 4000 feet.
- 7. Long term submersion of 24-hour duration, to 4000-foot depth, does not generate any cracks in the acrylic plastic hull or polycarbonate gaskets at the polar openings and the strains in acrylic plastic after a 24 hour relaxation period at atmospheric pressure return essentially to zero.

CONCLUSION

Spherical acrylic plastic hulls of Nemo Hull design with a t/r_0 = 0.123 thickness can be man-rated for a minimum of 1000 operational dives to a maximum operational depth of 3000 feet.

OPERATIONAL RECOMMENDATIONS

1. The Model 2000 Nemo capsule assembly should, during its operational life, never be subjected to depths greater than 3300 feet. The proof test should preferably utilize a test depth of 3300 feet. Under no conditions should the magnitude of proof-test depth exceed 3600 feet unless stronger polar inserts are substituted for the standard Model 2000 Nemo Hull aluminum inserts.

2. The cyclic crack free fatigue life of the Model 2000 Nemo Hull should be conservatively considered to be in excess of 12,000,000-feet-hours (1000 cycles × 3000 foot depth × 4 hours duty). At the conclusion of each dive, the recorded feet-hours should be subtracted from the initial 12,000,000-feet-hour fatigue life. When the sum of feet hour subtotals generated by dives equals 12,000,000-feet-hours, inserts and gaskets should be removed from the capsule and the entire hull subjected to a detailed visual examination.

If no cracks are observed at the penetrations in the hull, the capsule should be strain gaged, reassembled, prooftested to 3300 feet and resulting strains compared to those generated during the first proof test conducted immediately after fabrication of the capsule. Significant differences in strain behavior will be considered important evidence of hull deterioration and should result in significantly reduced depth rating. Cracks in bonded joints originating at inclusions will be repaired if their length exceeds 0.5 inches. Cracked polycarbonate gaskets should be replaced with new gaskets.

If no significant difference in strain behavior is observed, the capsule assembly will be returned to service with a 3000 foot operational depth rating and an additional 12,000,000-foot-hour fatigue life. When the 12,000,000-foot-hour life is used up the capsule assembly will be subjected to the same inspection and prooftesting procedures conducted at conclusion of the first 12,000,000-foot-hour period. If the results of the new inspection and prooftesting are satisfactory, the capsule will again return to service with a 3,000-foot depth rating and additional 12,000,000-foot-hour life.

The recertification process will be repeated until either cracks are observed in the bearing surfaces of acrylic hull during one of the inspections or the strains change significantly. If cracks are observed they will either be repaired by routing and recasting with resin prior to retesting of the hull, or they will be left in place and the hull's depth rating will be reduced to 600 feet.

Subsequently, the hull will be inspected without disassembly for signs of crack propagation every 100 dives. When the depth of any crack exceeds 1 inch, as pictured in Figure 34, the capsule will be taken out of service immediately and the cracks repaired either by enlarging the polar opening or by recasting cracked areas. If not repaired, such a hull can be recertified for service to 120 feet. If, during periodic inspections conducted every 100 dives, the depth of the crack at the penetration is found to exceed 2 inches the acrylic hull will either be repaired or declared unfit for manned operation at any depth.

3. Attempts should be made to ensure that operators be seated inside the Nemo Hull as close as possible to the center of the sphere in order to minimize optical distortion. 15 Camera mounting should be located at the center of the hull if wide angle panning with the camera is to be performed during the mission.

Objects in hydrospace will appear smaller and closer to the hull than they are in reality. Some experience on the part of the crew will be required to judge the distances correctly between the hull and the objects in hydrospace.

4. Many functions of the equipment mounted externally to the submersible can be controlled by modulated light beams projected from the interior of the hull by the crew. ¹⁶ This type of arrangement will eliminate many electrical penetrators in the bottom plate and make the control of externally stored scientific equipment an operationally easy matter.



Figure 1. Acrylic plastic hull with the typical Nemo polar penetrations, metallic hatches, and spherical pentagon modular construction.

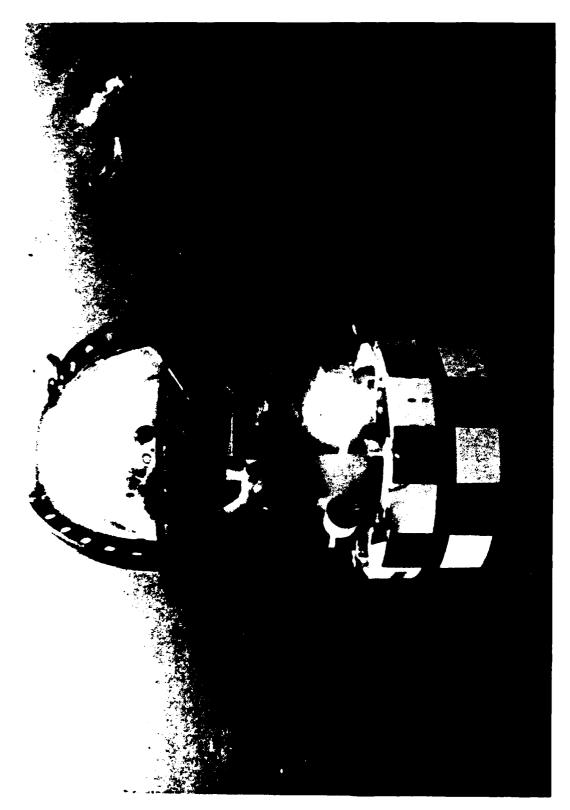


Figure 2. Nemo submersible, approved in 1970 by the U. S. Navy for manned operations to 600 feet.

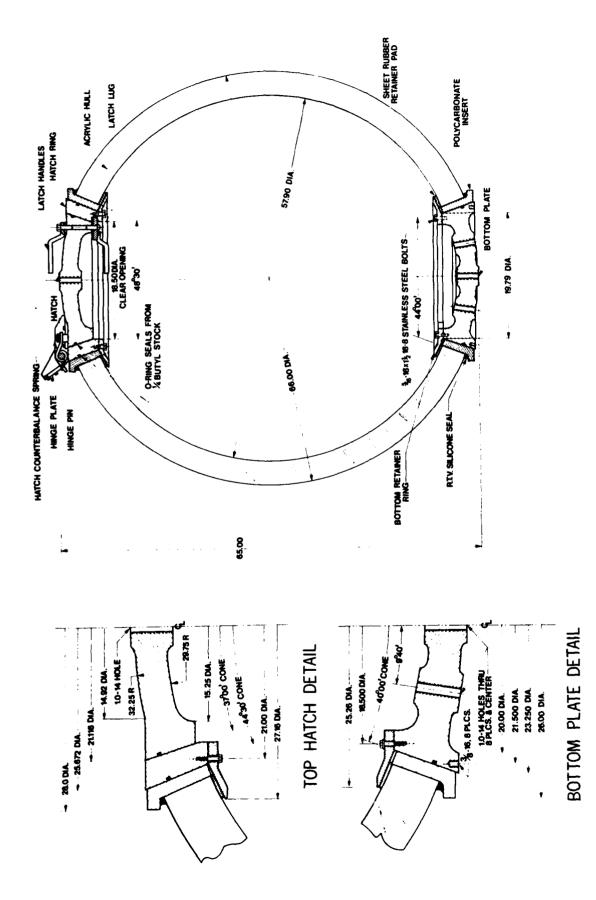


Figure 3. Schematic of the Model 2000 Nemo Hull.

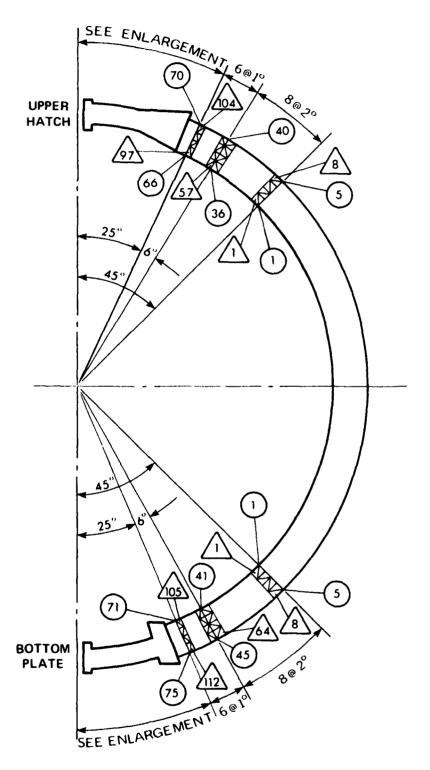


Figure 4. Idealized shape of the Model 2000 Nemo Hull assembly used in the ZP 13 finite element stress analysis.

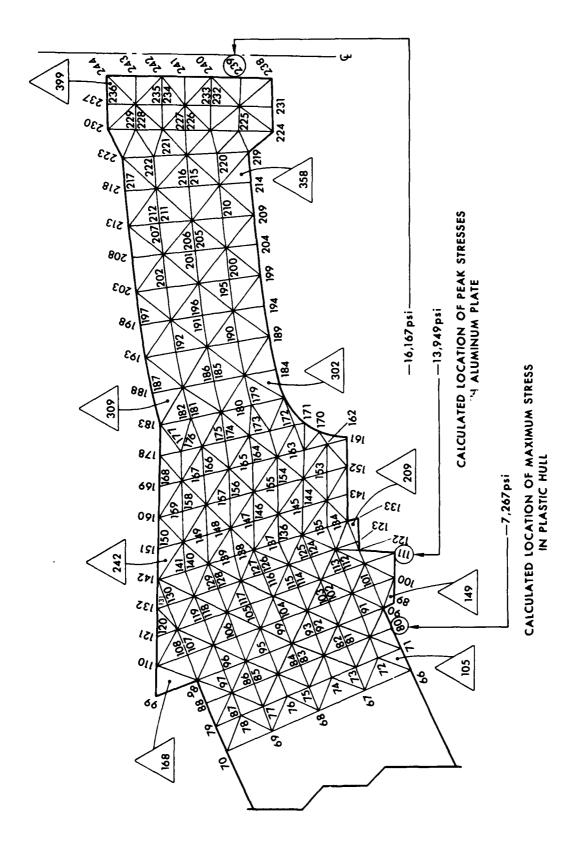


Figure 5. Idealized shape of the top hatch used in the ZP 13 finite element stress analysis of the Model 2000 Nemo Hull under simulated 900 psi external hydrostatic pressure.

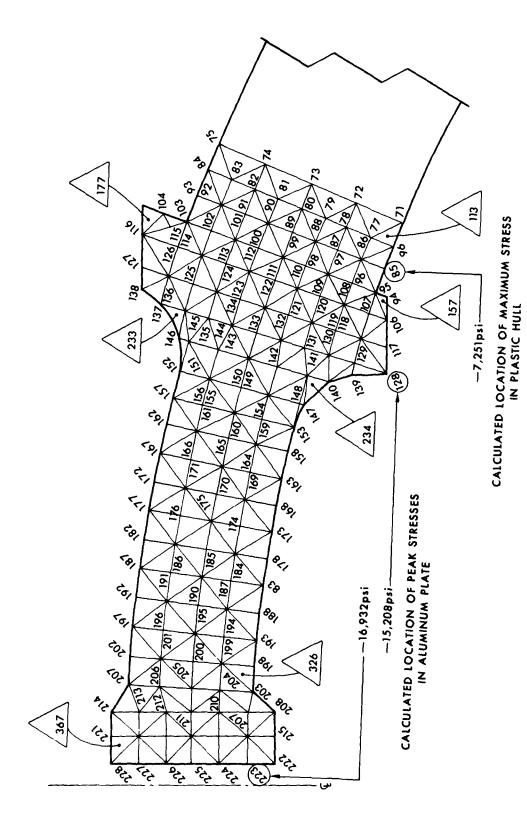


Figure 6. Idealized shape of the bottom plate used in the ZP 13 finite element stress analysis of the Model 2000 Nemo Hull under simulated 900 psi external hydrostatic pressure.

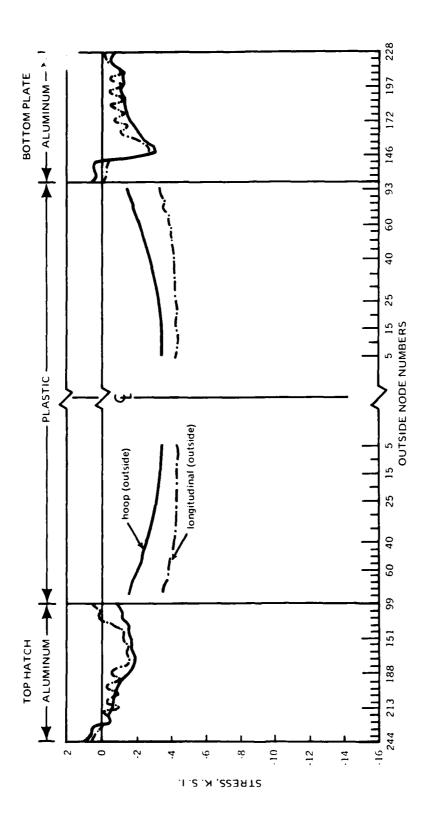


Figure 7. Calculated stress distribution in the Model 2000 Nemo Hull assembly under simulated 900 psi external hydrostatic pressure, outside surface.

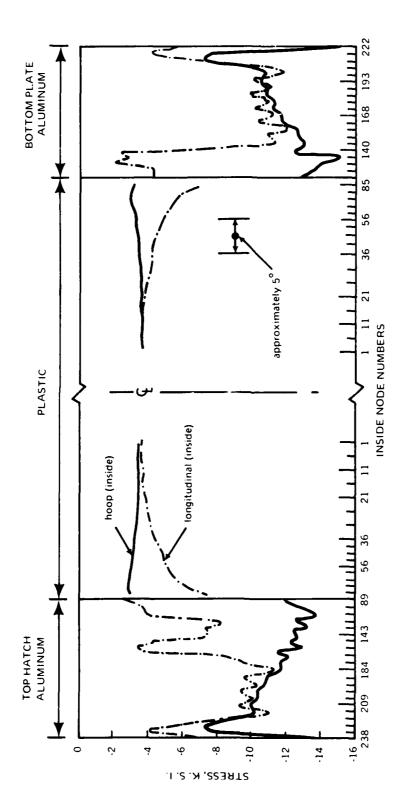


Figure 8. Calculated 'tress distribution in the Model 2000 Nemo Hull assembly under simulated 900 psi external hydrostatic pressure; inside surface.



Figure 9. Assembled Model 2000 Nemo Hull undergoing final polishing.

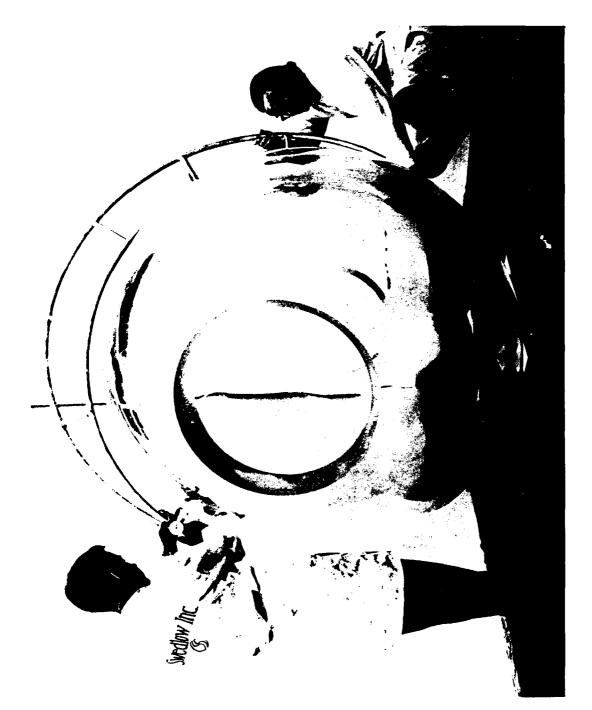
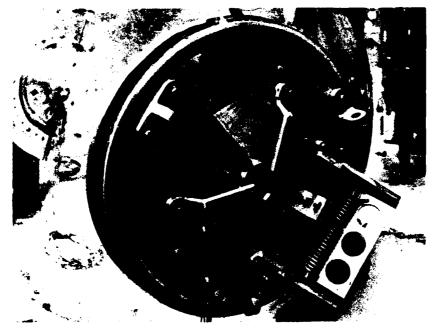


Figure 10. Inspection of Model 2000 Nemo Hull for out-of-roundness at Swedlow Inc.



Figure 11. Typical bonded joint between spherical pentagons.



(a) outside view



(b) inside view

Figure 12. Aluminum hatch for Model 2000 Nemo Hull.



(a) outside view



(b) inside view

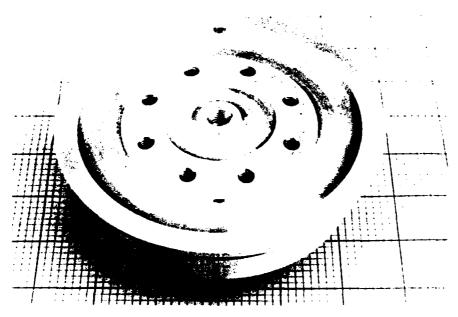
Figure 13. Aluminum bottom plate for Model 2000 Nemo Hull.



Figure 14. Bearing gasket for polar openings in the hull fabricated from polycarbonate plastic plates.



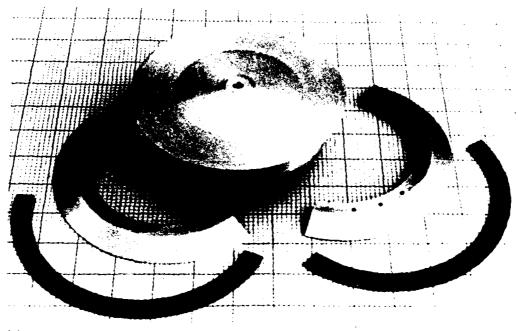
Figure 15. 15 inch OD X 13 inch ID Model 34 serving as scale model of the Model 2000 Nemo Hull assembly.



(a) outside view



Figure 16. Aluminum bottom plate for 15 inch OD \times 13 inch ID Model 34 serving as scale model of Model 2000 Nemo Hull assembly.



(a) outside view of hatch and retaining ring

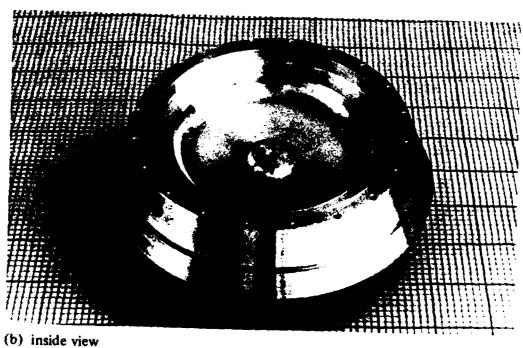
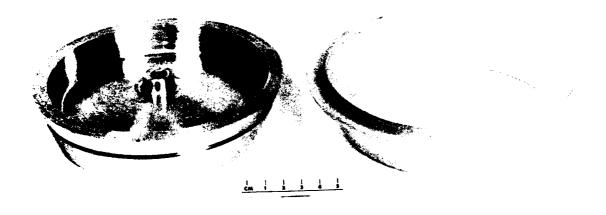
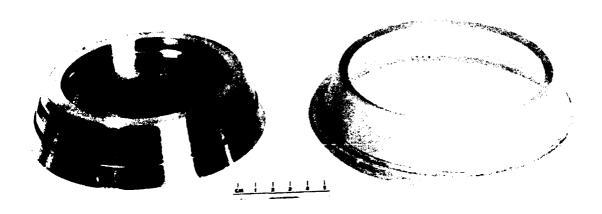


Figure 17. Aluminum hatch for 15 inch OD \times 13 inch ID Model 34 serving as scale model of Model 2000 Nemo Hull assembly.

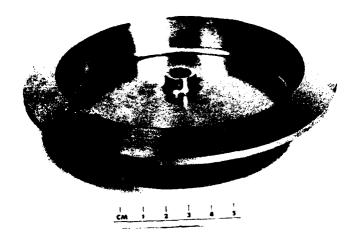


(a) exterior view of hatch designed for service with polycarbonate gasket

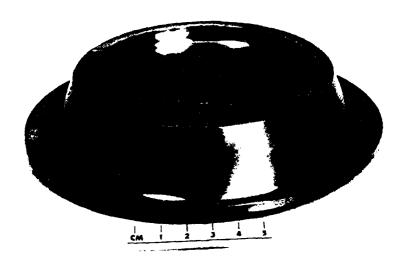


(b) interior view of hatch designed for service with polycarbonate gasket

Figure 18. Titanium hatch for 15 inch OD X 13 inch ID Models 35, 36 and 37.



(c) exterior view of hatch designed for service without a polycarbonate gasket



(d) interior view of hatch designed for service without a polycarbonate gasket

Figure 18. (Continued).

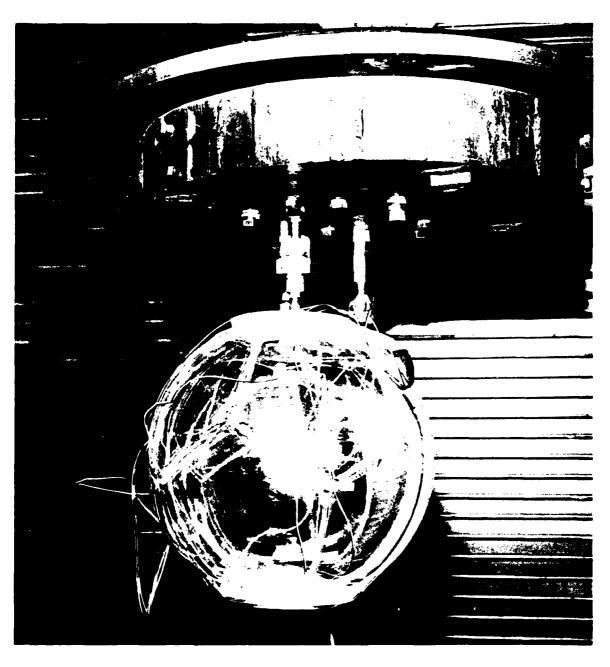
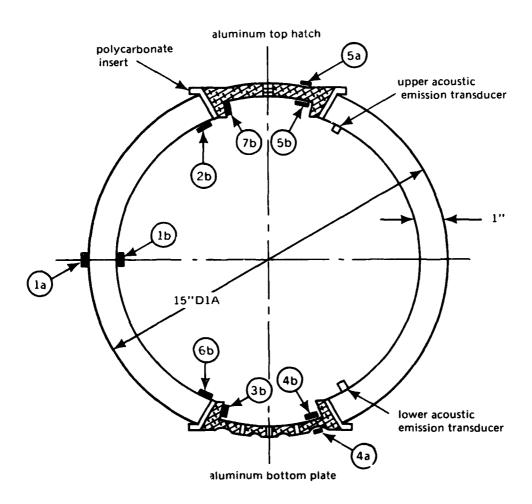


Figure 19. Test arrangement for hydrostatic testing of 15 inch OD \times 13 inch ID Model 34 serving as scale model of Model 2000 Nemo Hull.

2b - 0.500 inches from edge of hatch 6b - 0.700 inches from edge of bottom plate



Note: Each number instrumented with 2 gage 90° rosettes

Figure 20. Location of strain gages on the 15 inch OD \times 13 inch ID Model 34, serving as scale model of Model 2000 Nemo Hull.

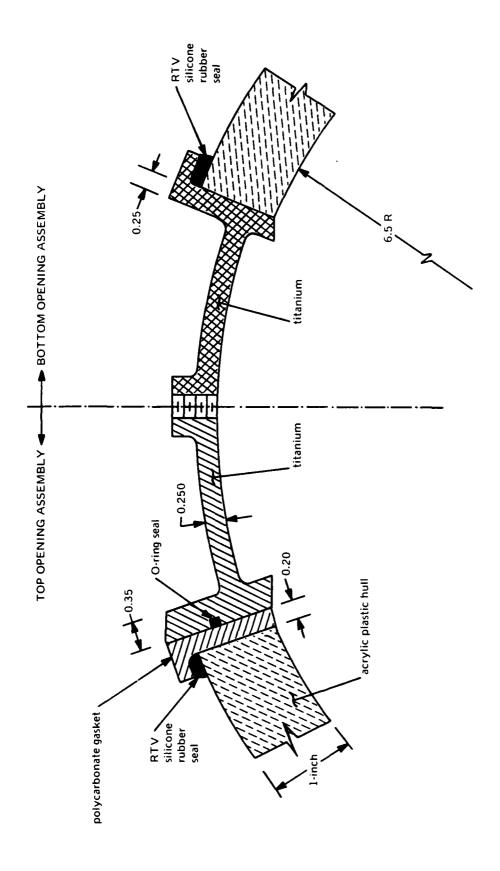
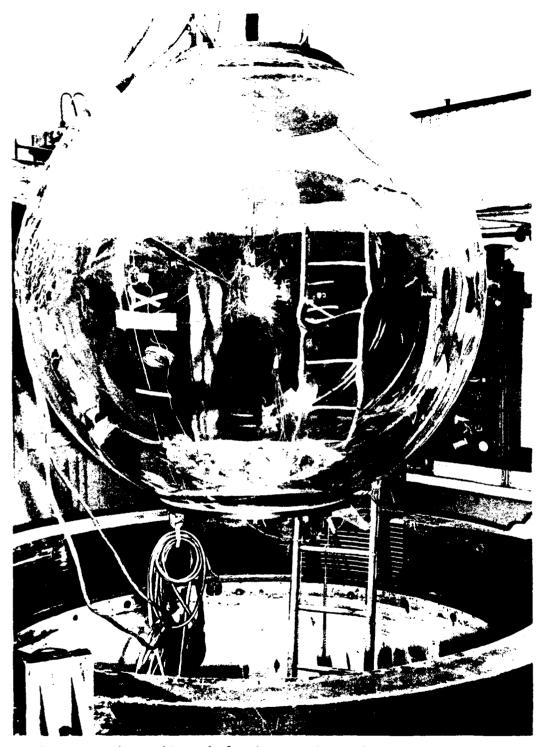
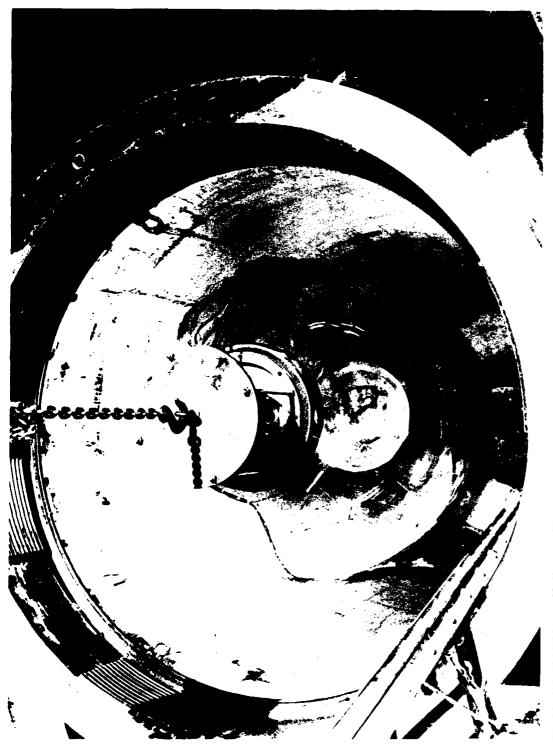


Figure 21. Typical hatches used in the 15 inch OD \times 13 inch ID Models 35, 36 and 37 subjected to pressure cycling.



(a) instrumented assembly ready for placement in vessel

Figure 22. Testing of full scale Model 2000 Nemo Hull assembly in the 90 inch diameter pressure vessel at SWRI.

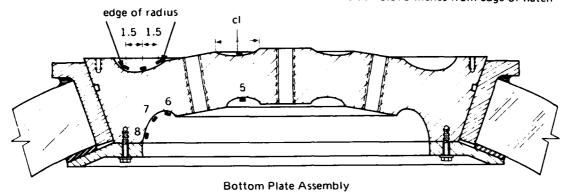


ssemoly in vessel Figure 22. (Continued).

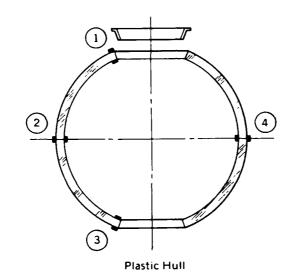
(b) Model 2000 Nemo Hull assembly in vessel

1 outside - 1.500 inches from edge of hatch 1 inside - 1.375 inches from edge of hatch

3 outside - 1.500 inches from edge of hatch 3 inside - 1.375 inches from edge of hatch



,



edge of radius 1" 1" 1" 1" 1" 9, 10 111 12 13 14

Figure 23. Location of strain gages on the 66 inch OD \times 58 inch ID full scale Model 2000 Nemo Hull assembly.



Figure 24. Fragments of the 15 inch OD X 13 inch ID Model 34 after implosion at 4750 psi.

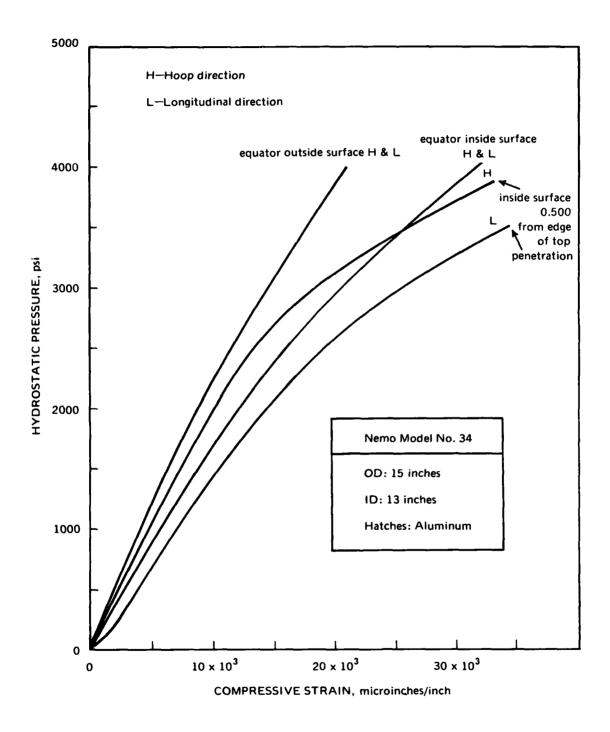


Figure 25. Strains in the 15 inch OD \times 13 inch ID Model 34 serving as scale for Model 2000 Nemo Hull.

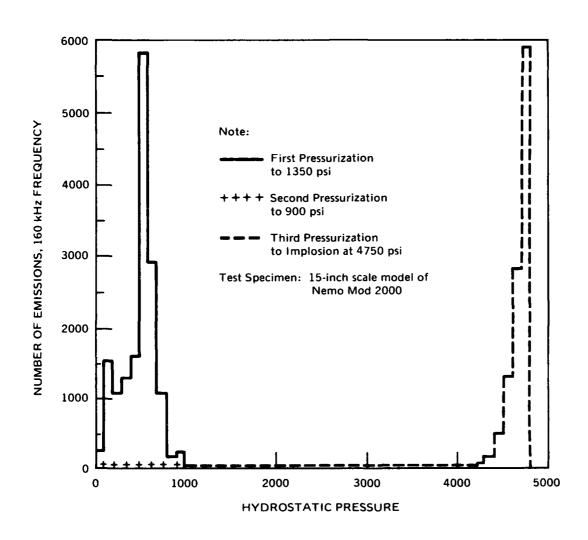
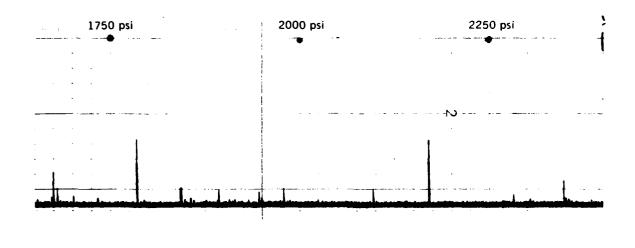


Figure 26. Histogram of stress wave emissions from 15 inch OD \times 13 inch ID Model 34 of undergoing external pressure tests.



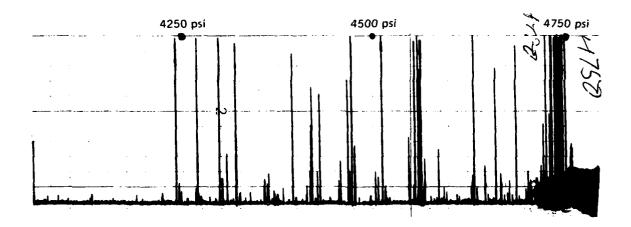


Figure 27. Recording of stress wave emissions preceding the short term implosion of 15 inch OD X 13 inch ID Model 34 assembly at 4750 psi external hydrostatic pressure.



Figure 28. Inspection of bearing surfaces on 15 inch OD X 13 inch ID Models 36 and 37 after 1000 pressure cycles to, respectively, 900 and 1500 psi hydrostatic pressure.

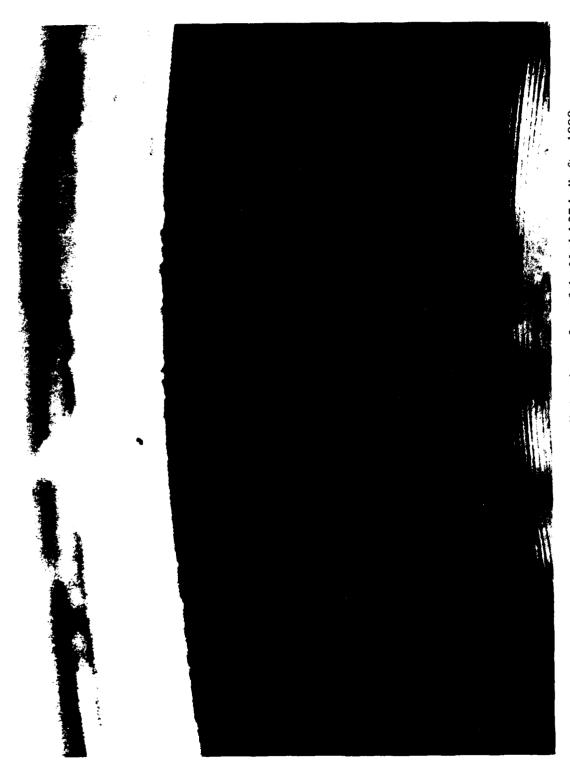


Figure 29. Fatigue crazing of the acrylic bearing surface of the Model 37 hull after 1000 pressurizations of 4 hour duration each to 1500 psi; this acrylic bearing surface was in direct contact with the metallic hatch.

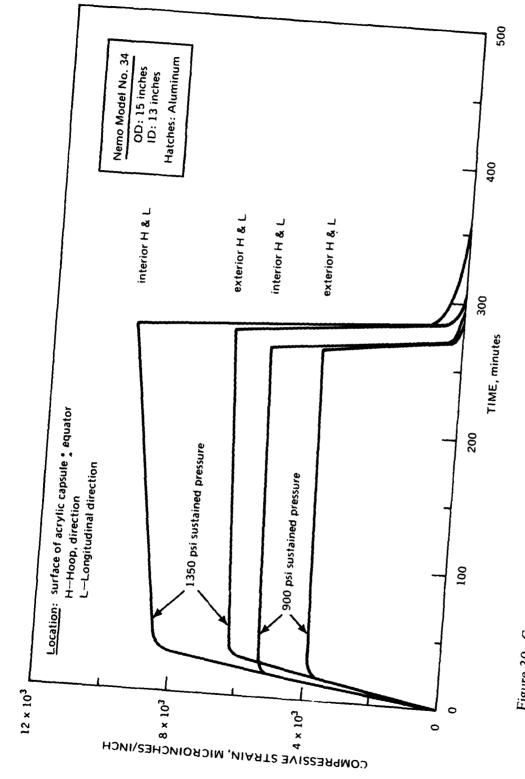


Figure 30. Creep of acrylic hull in 15 inch OD \times 13 inch ID Model 34 under external hydrostatic pressure; measured on the interior and exterior surfaces at the equator.

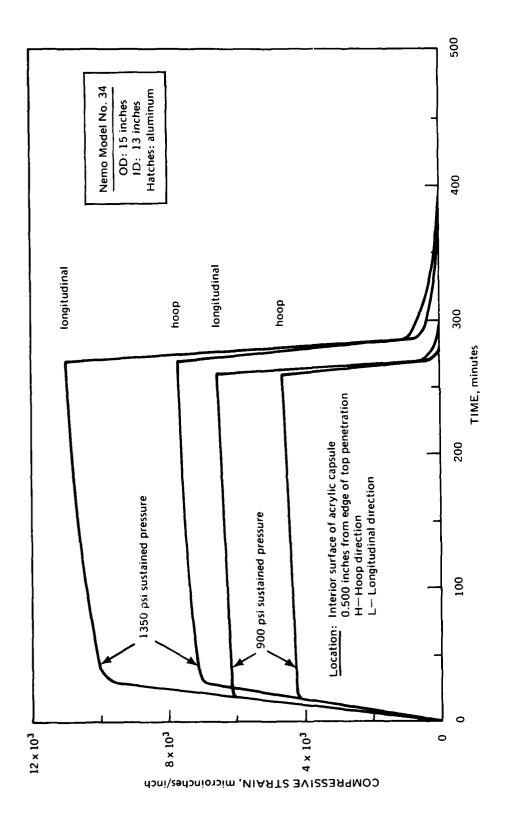
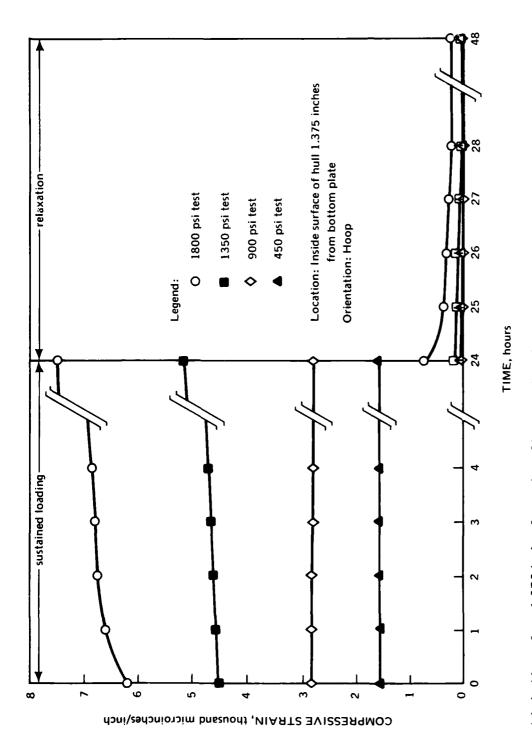
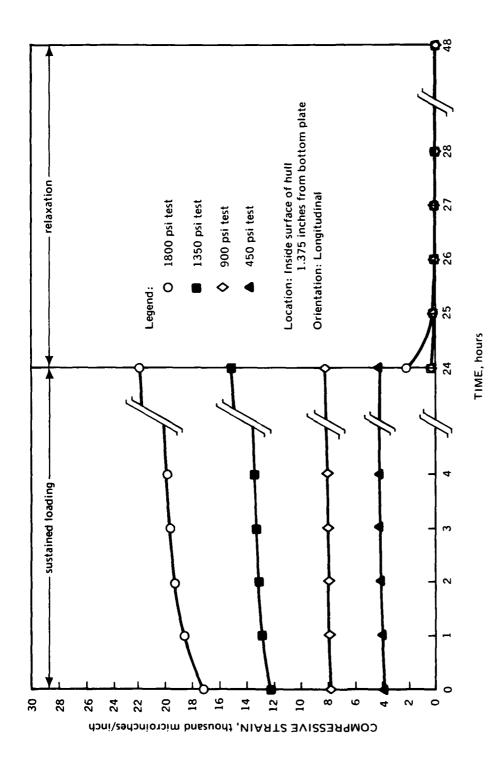


Figure 31. Creep of acrylic hull in 15 inch OD \times 13 inch ID Model 34 under external hydrostatic pressure; measured on the interior surface at the edge of top polar penetration.



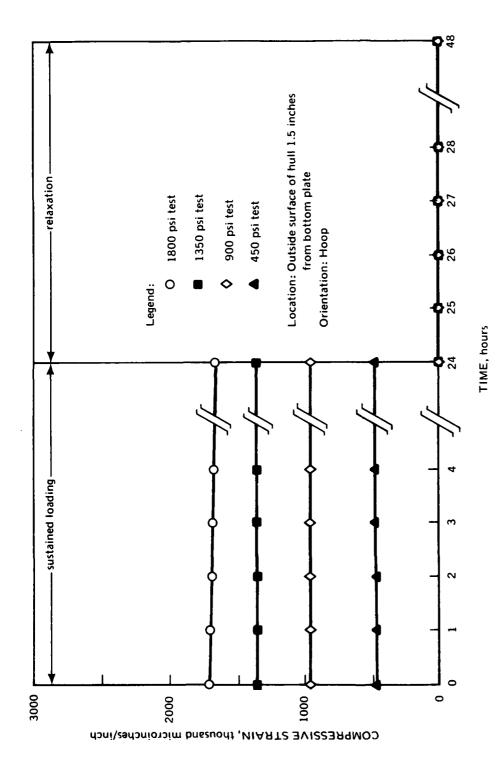
(a) inside surface; 1.375 inches from edge of bottom plate; hoop

Figure 32. Creep measured on the acrylic hull of the 66 inch OD \times 58 inch ID Model 2000 Nemo Hull assembly during 24-hour long sustained loadings under external hydrostatic pressure.



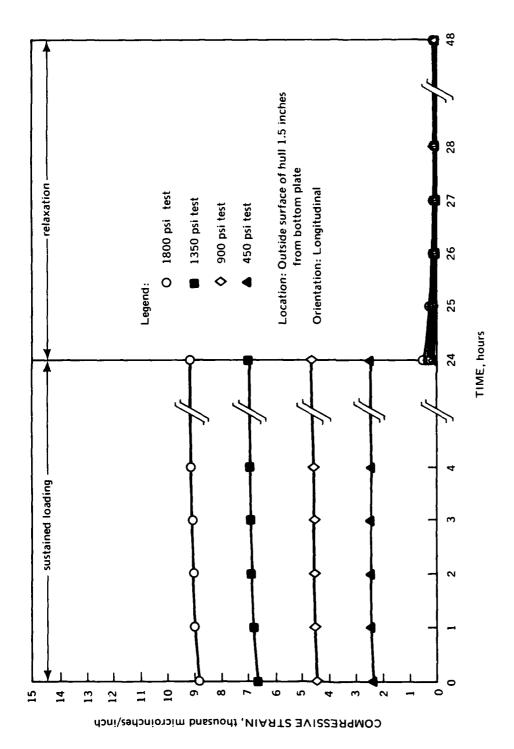
(b) inside surface; 1.375 inches from edge of bottom plate; longitudinal

Figure 32. (Continued).



(c) outside surface; 1.500 inches from edge of bottom plate; hoop

Figure 32. (Continued).



(d) outside surface; 1.500 inches from edge of bottom plate; longitudinal

Figure 32. (Continued).

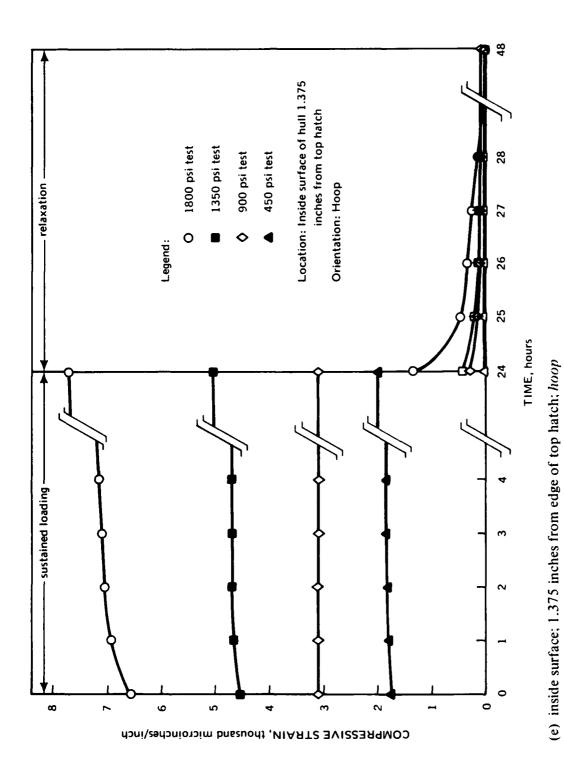
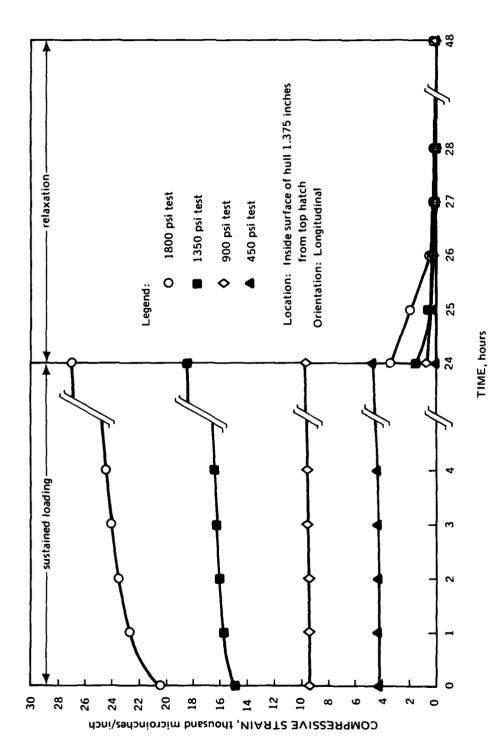
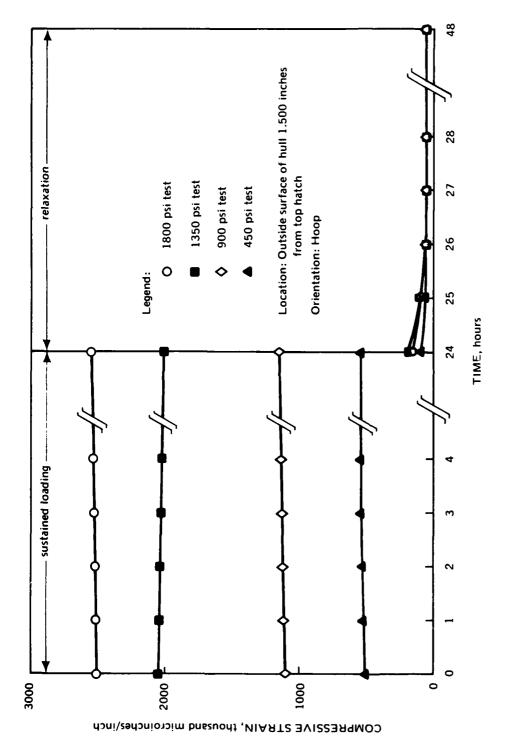


Figure 32. (Continued).



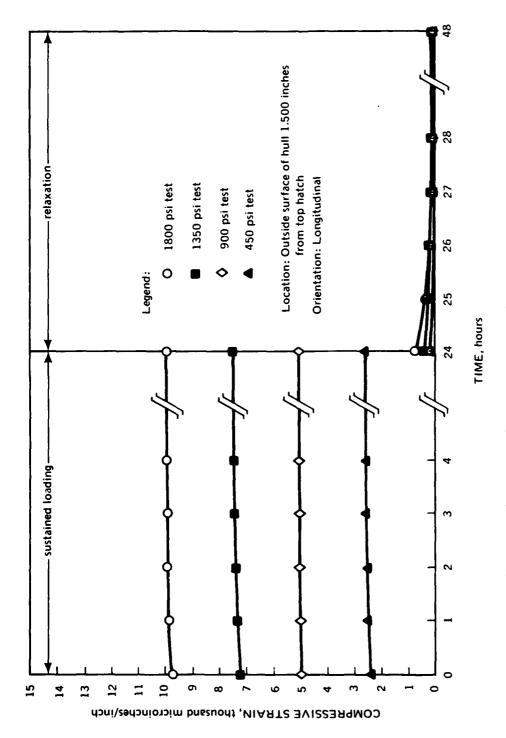
(f) inside surface; 1.375 inches from edge of top hatch; longitudinal

Figure 32. (Continued).



(g) outside surface; 1.500 inches from edge of top hatch; hoop

Figure 32. (Continued).



(h) outside surface; 1.500 inches from edge of top hatch; longitudinal

Figure 32. (Continued).

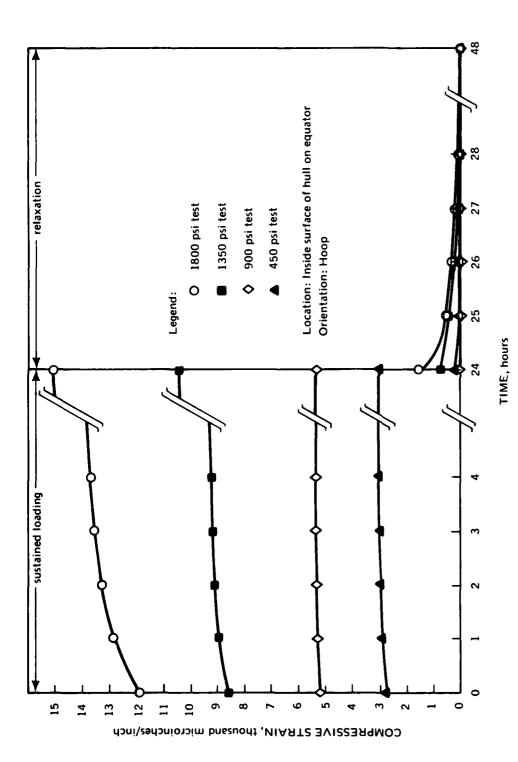
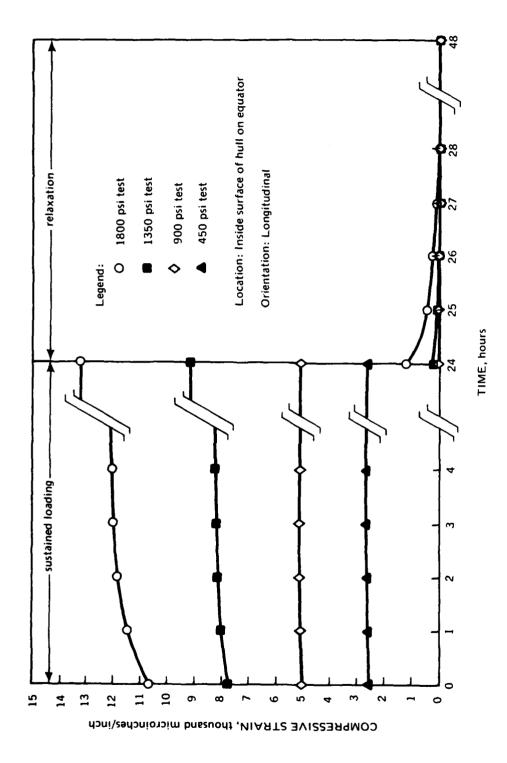


Figure 32. (Continued).

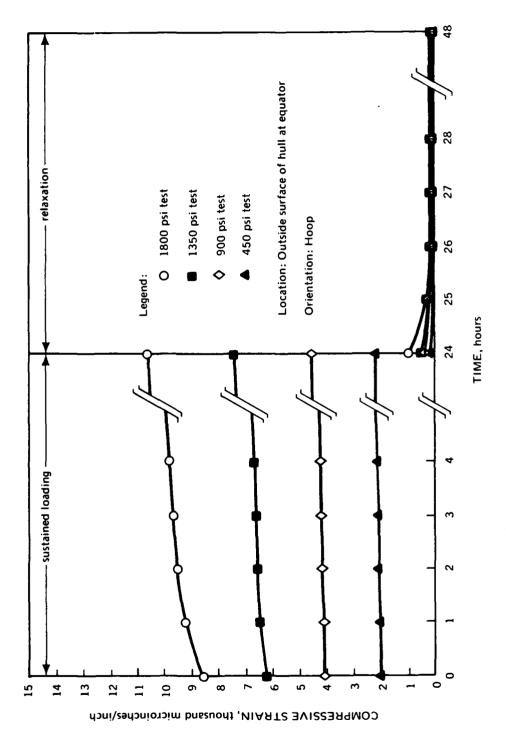
(i) inside surface; equator; hoop



;

(j) inside surface; equator; longitudinal

Figure 32. (Continued).



(k) outside surface; equator; hoop

Figure 32. (Continued).

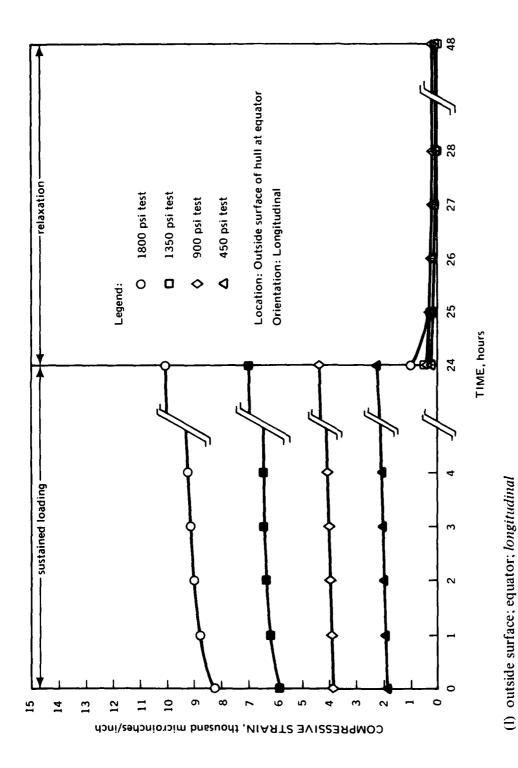


Figure 32. (Continued).

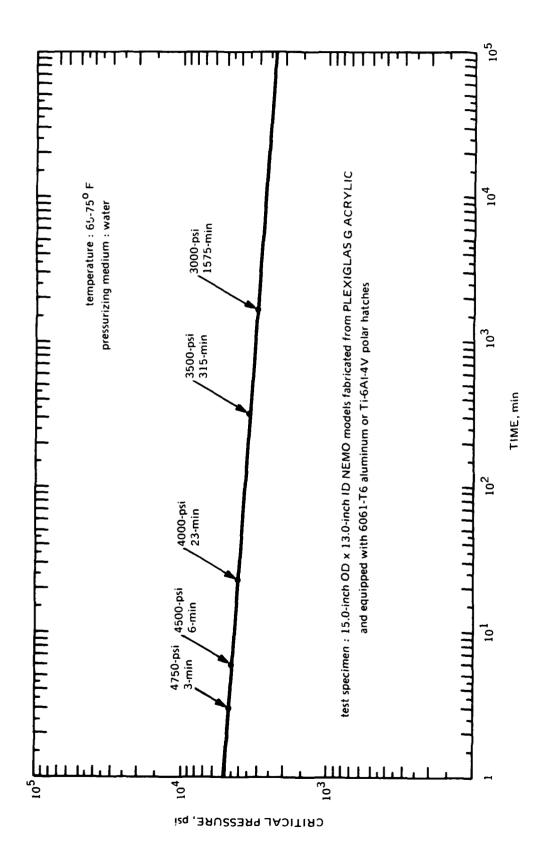


Figure 33. Long term critical pressure as a function of sustained loading duration.



Figure 34. Typical shear cracks in the bearing surface of an acrylic Nemo Hull generated by pressure cycling; when the cracks reach this size the acrylic hull should be removed from service and the bearing surface refinished. (Ref. 5. Nemo Hull Model 600 after pressure cycling to 2000 ft. depth.)

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- 3. Naval Civil Engineering Laboratory, Technical Note N-1113, "The Spherical Acrylic Pressure Hull for Hydrospace Application; Part 2. Experimental Stress Evaluation of Prototype NEMO Capsule," J. D. Stachiw and K. L. Mack, October 1970 (AD 715772)
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- 7. American Society of Mechanical Engineers, Paper No. 71-WA/UnT-6, "Acrylic Pressure Hull for JOHNSON SEA-LINK Submersible," by J. R. Maison and J. D. Stachiw, December 1971
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- 9. American Society of Mechanical Engineers, Paper No. 72-WA/OCT-8, "Transparent Hull Submersible MAKAKAI," by D. W. Murphy and W. F. Mazzone, December 1972
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- 11. Naval Civil Engineering Laboratory, Technical Report R-716, "Structural Analysis of a Full Scale Spherical Acrylic Plastic Pressure Hull," M. R. Snoey and M. G. Katona, March 1971

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APPENDIX A DESIGN DETAILS OF NEMO MODEL 2000

APPENDIX A. DESIGN DETAILS OF NEMO MODEL 2000

15-Inch OD X 13-Inch ID Scale Models

The acrylic hull of the 15-inch OD \times 13-inch ID scale model 34 was designed to be a faithful copy of the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull both in proportions and in method of construction (Figures 1A and 2A). It was to be fabricated in the same manner as the 66-inch OD \times 58-inch ID hull by thermoforming spherical sectors from flat discs, machining pentagons from sectors, and finally assembling the spherical pentagons into a sphere by bonding the joints between adjoining pentagons with PS-30 self-polymerizing adhesive (Figure 3A).

The aluminum plates (Figures 4A, 5A, and 6A) for top and bottom polar openings in the 15-inch OD \times 13-inch ID Model 34 were not faithfully scaled down models of the hatches in the 66-inch OD \times 58-inch ID diameter Model 2000 Nemo Hull. Although structurally the 6061-T6 aluminum plates in the 15-inch OD \times 13-inch ID Model 34 behave identically to the hatches in the 66-inch OD \times 58-inch ID hull, some operational features of the large hatches have been omitted in the scale model plates. Thus, for example, the top aluminum plate 15-inch OD \times 13-inch ID Model 34 has the same rigidity and proportions as the top hatch in the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull but does not disassemble into separate hatch and hatch ring components.

The construction of the 15-inch OD × 13-inch ID Models 35, 36, and 37 was identical to that of Model 34. The only difference between Model 34 and the other models lay in the design of the polar plates. The polar plates for Models 35, 36, 37 were structurally idealized hatches designed in titanium Ti-6Al-4V (Figure 7A). Since these models were to be used in cyclic pressure tests to determine the effect of depth on the performance of the polycarbonate gasket between the hatch and the acrylic bearing surface, each model was equipped with the polycarbonate gasket only for the top plate while the bottom plate in each model was designed to operate without a gasket. In this manner each model was designed to operate both with and without a gasket around the titanium plates. In this manner, each model would provide the data on the performance of acrylic bearing surfaces at a given pressure with and without polycarbonate gaskets (Figure 8A).

66-Inch OD × 58-Inch ID Operational Model

The 66-inch OD X 58-inch ID operational Model 2000 Nemo Hull was designed for economical construction within tight dimensional tolerances to maximize the operational depth of the assembly. The acrylic hull was designed to be constructed from 12 spherical pentagons bonded together with PS 18, PS 30 or any other self-polymerizing adhesive with 5000 psi minimum tensile strength (Figures 9A and 10A).

The polar aluminum assemblies were designed, like the polar insert assemblies in the previous Model 600 and Model 1000 Nemo Hulls, to serve as hatches for personnel entry and feed through plates for electrical and hydraulic control cables. Aluminum was chosen as the

construction material because of its resistance to corrosion and attractive strength to weight ratio. The bottom feed through plate was equipped with 9 holes to accommodate 9 separate electrical or hydraulic feed throughs (Figure 11A). In addition, the feedthrough plate serves also as the foundation for any equipment contained within the capsule. The diameter of the top polar opening was selected to be ample enough even for a heavy set pilot or observer (Figure 12A). Because considerable exertion has been required of the crew in the past Nemo Hull designs to open the heavy hatch, a set of torsion springs was incorporated into the Model 2000 Nemo hatch assembly (Figure 12A). Also latch locks have been incorporated into the hatch handles to lock them securely in the open position when the hatch is open (Figure 13A).

All the parts of the hatch were made from 6061-T6 aluminum, except the Monel K-500 latch shafts, the 17-4 PH stainless hinge pin, steel counter balance springs, and polycarbonate plastic gaskets (Figures 14A, 15A, 16A, 17A, 18A, 19A and 20A). Materials chosen for these applications matched well with the galvanic potential of aluminum, thus preventing unduly severe galvanic corrosion. As a rule all the bevel angle tolerances on polar insert components were specified to be ± 15 minutes, a readily attainable tolerance with standard shop machining practices. During the subsequent assembly of the Model 2000 Nemo Hull structure it was found, however, that the ± 15 -degree angle tolerances result of this finding, it is recommended that the angle tolerance be decreased to ± 7.5 degrees in future Model 2000 Nemo Hull assemblies.

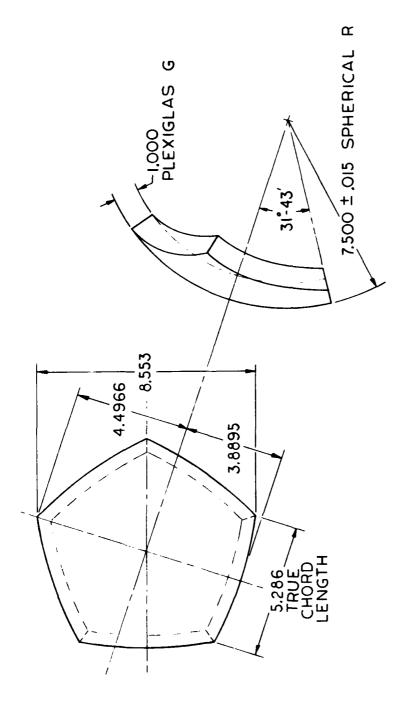


Figure 1A. Spherical pentagon for the 15-inch OD \times 13-inch ID scale model of the Model 2000 Nemo Hull.

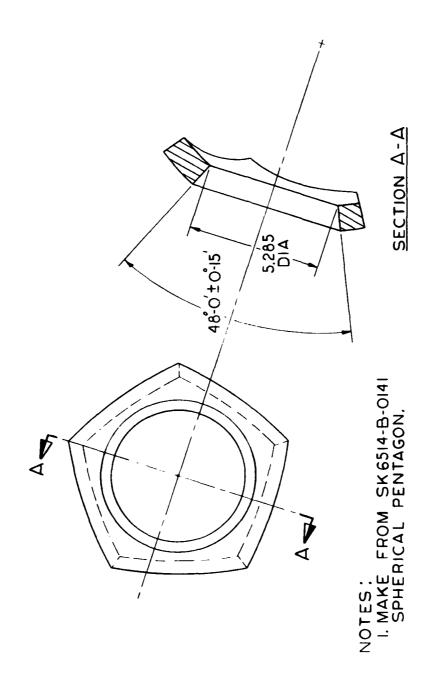
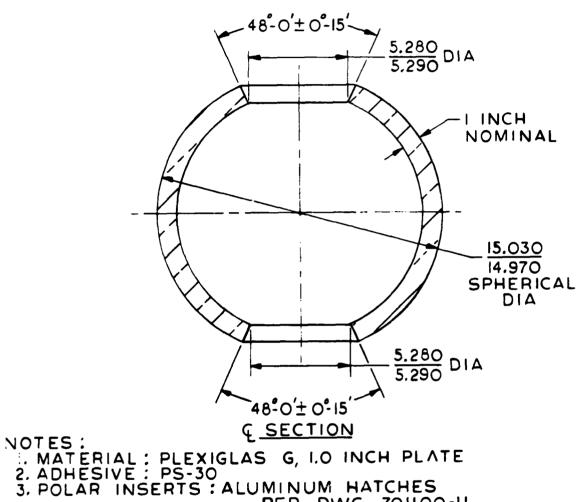


Figure 2A. Polar spherical pentagon for the 15-inch OD \times 13-inch ID scale model of the Model 2000 Nemo Hull.



PER DWG 701100-11
POLYCARBONATE GASKETS
PER DWG 701100-11

Figure 3A. Assembled hull of the 15-inch OD \times 13-inch ID Model 34.

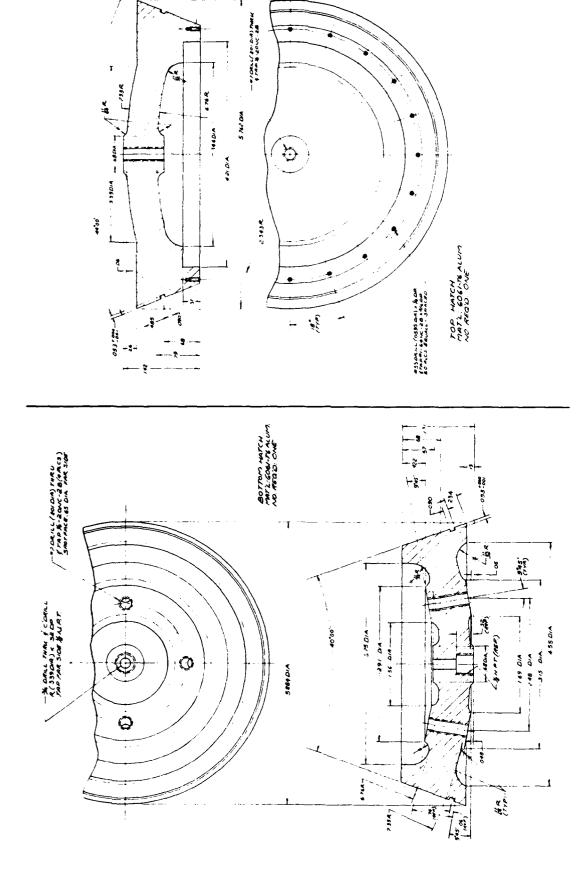


Figure 4A. Polar Aluminum Hatch Assemblies for the 15-inch OD X 13-inch 1D Model 34.

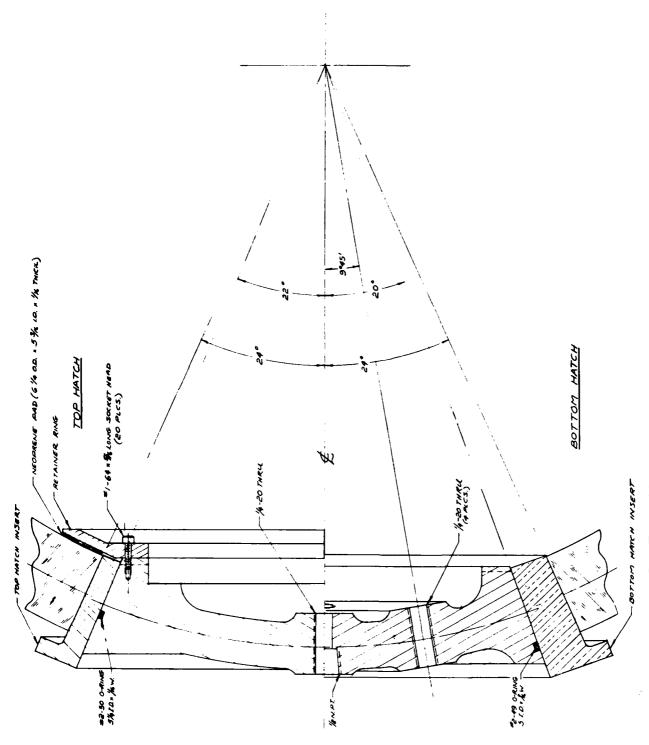


Figure 5A. Top and bottom aluminum hatches for the 15-inch OD X 13-inch ID Model 34.

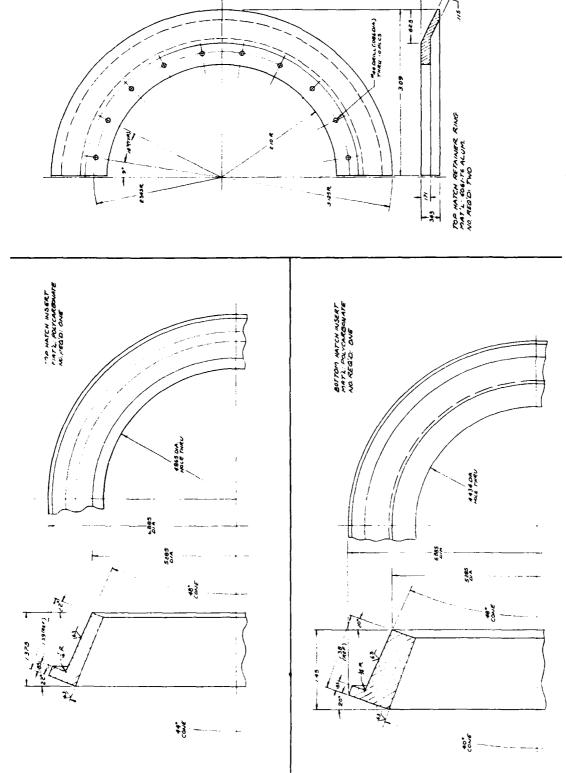


Figure 6A. Polycarbonate polar gaskets and top hatch retaining ring for the 15-inch OD \times 13-inch ID Model 34.

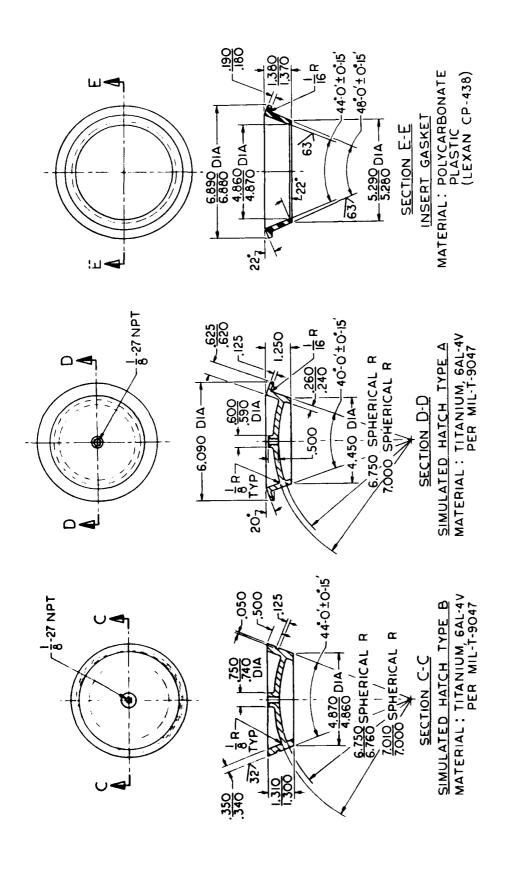
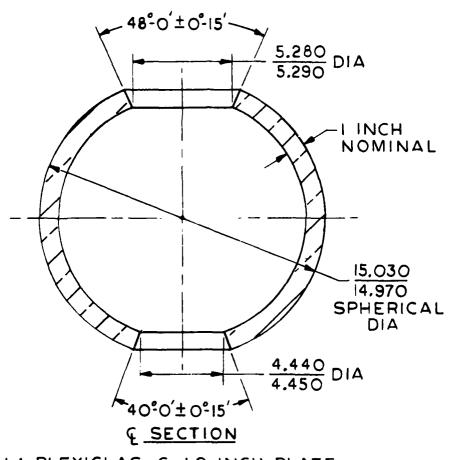


Figure 7A. Polar titanium hatches for service with 15-inch OD × 13-inch ID Models 35, 36 and 37. Note that Type A titanium hatch was designed for service without the polycarbonate gasket while Type B titanium hatch was designed to be used with a polycarbonate gasket.



NOTES:

1. MATERIAL: PLEXIGLAS G, 1.0 INCH PLATE
2. ADHESIVE: PS-30
3. INSERTS: TOP OPENING-TYPE B TITANIUM HATCH
WITH POLYCARBONATE GASKET

BOTTOM OPENING - TYPE A TITANIUM HATCH WITHOUT GASKET

Figure 8A. Typical assembly of 15-inch OD × 13-inch ID Models 35, 36 and 37.

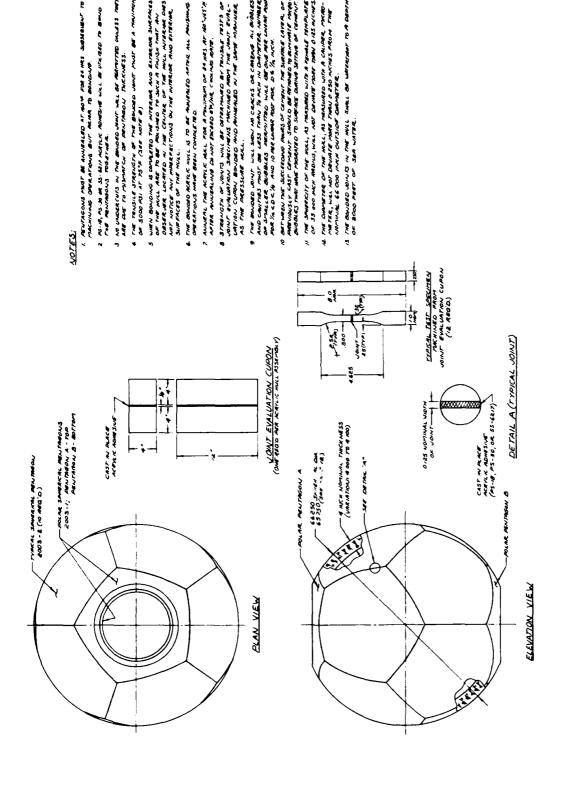


Figure 9A. Model 2000 acrylic Nemo Hull assembly, 66-inch OD X 58-inch ID.

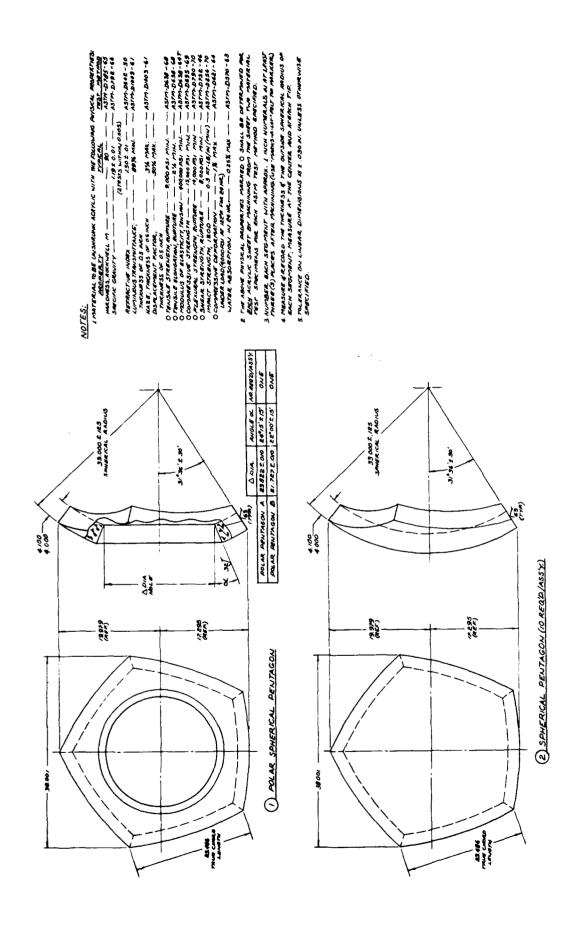


Figure 10A. Spherical pentagons for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.

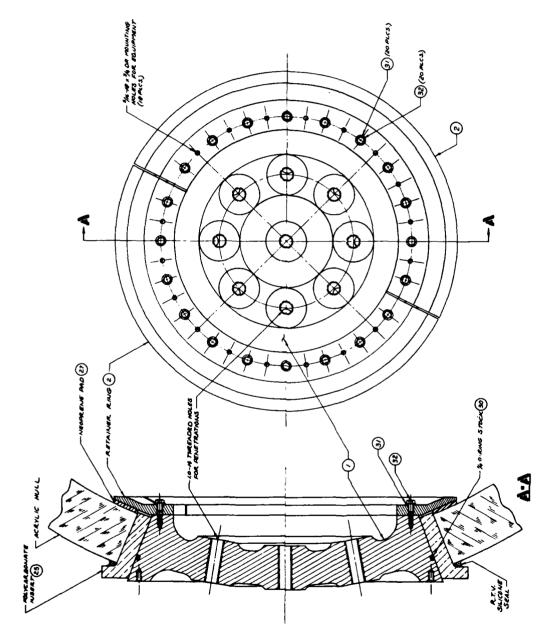


Figure 11A. Bottom plate assembly for the 66-inch OD × 58-inch ID Model 2000 Nemo Hull.

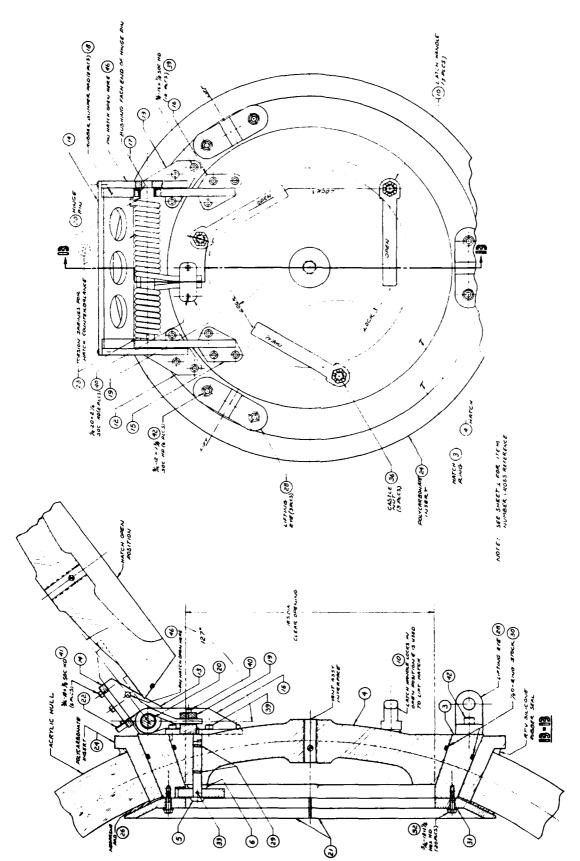


Figure 12A. Hatch assembly for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.

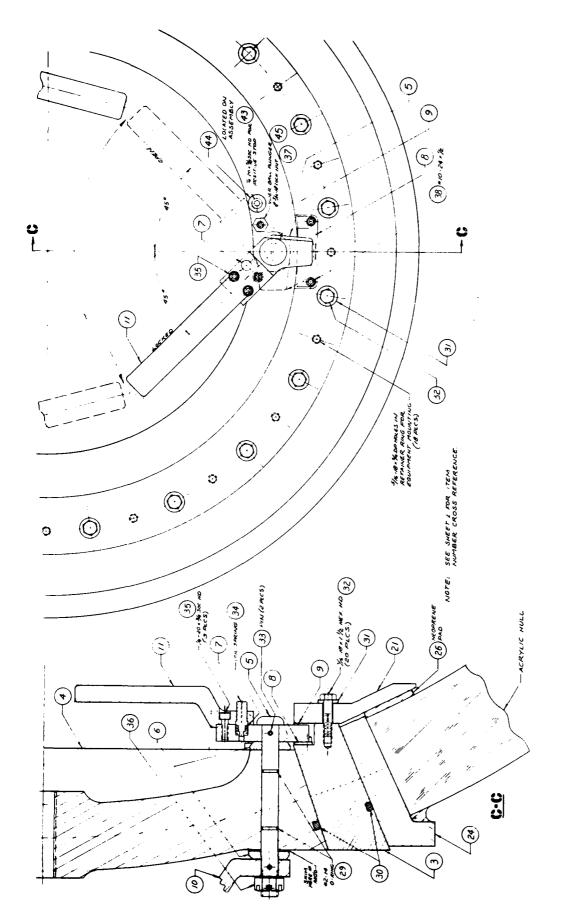


Figure 13A. Hatch lock assembly for the 66-inch OD × 58-inch ID Model 2000 Nemo Hull.

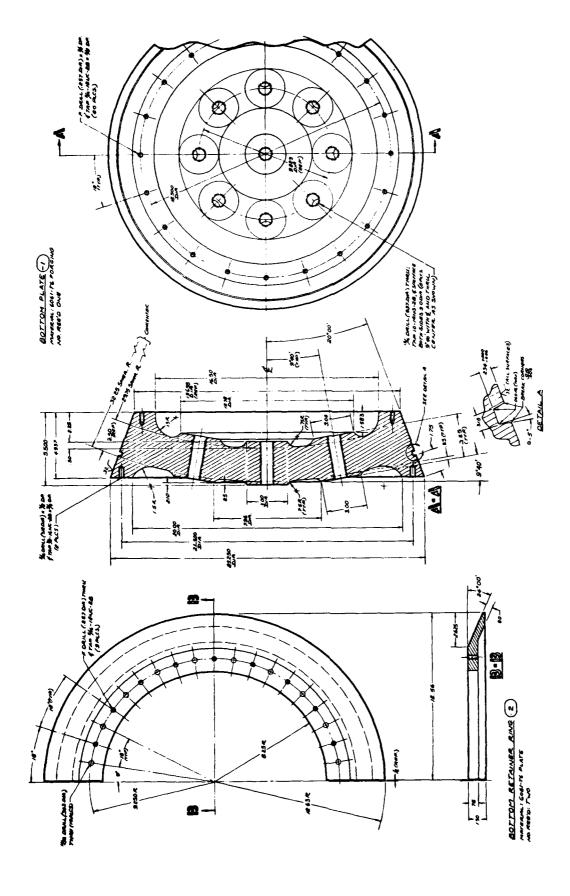


Figure 14A. Bottom plate and bottom retainer ring for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.

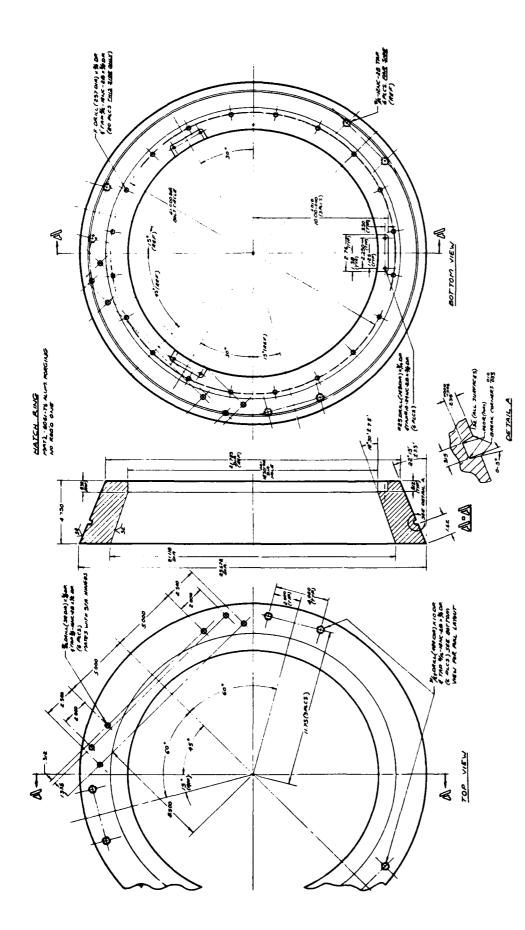


Figure 15A. Hatch ring for the 66-inch OD X 58-inch ID Model 2000 Nemo Hull.

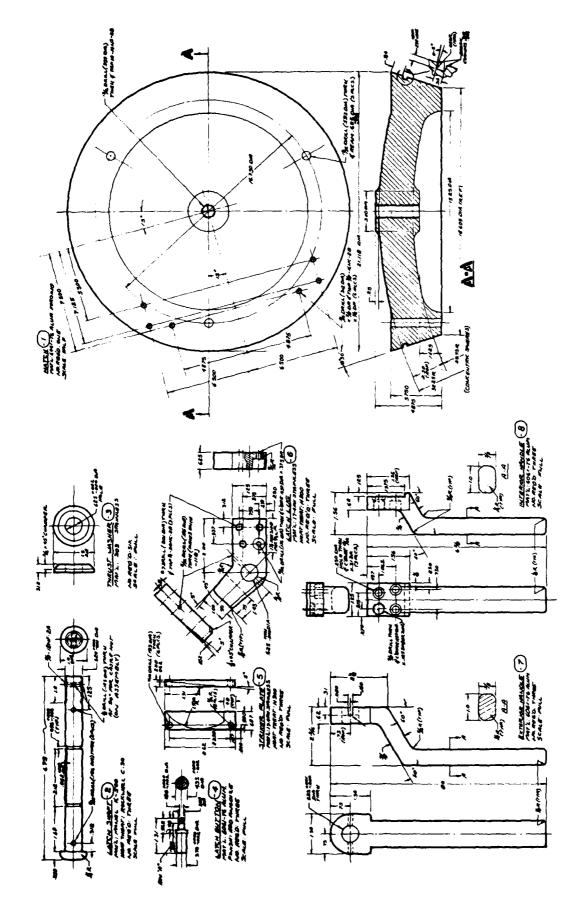


Figure 16A. Hatch and locking mechanism for the 66-inch OD X 58-inch ID Model 2000 Nemo Hull.

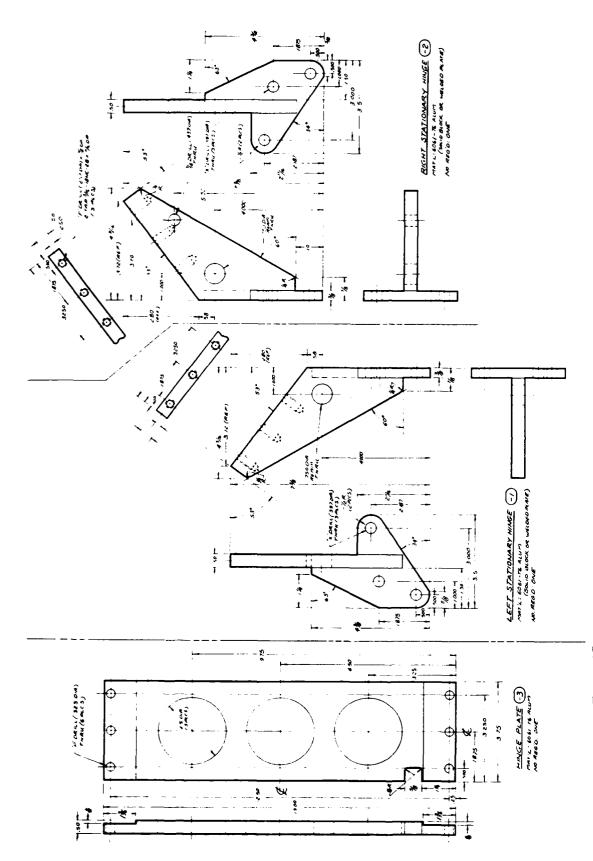


Figure 17A. Stationary hinges and hinge plate for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.

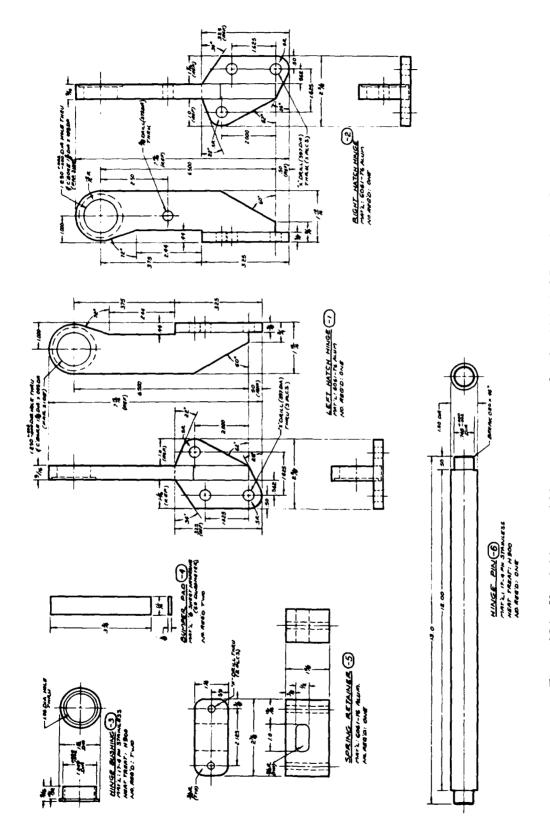


Figure 18A. Hatch hinge and hinge components for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.

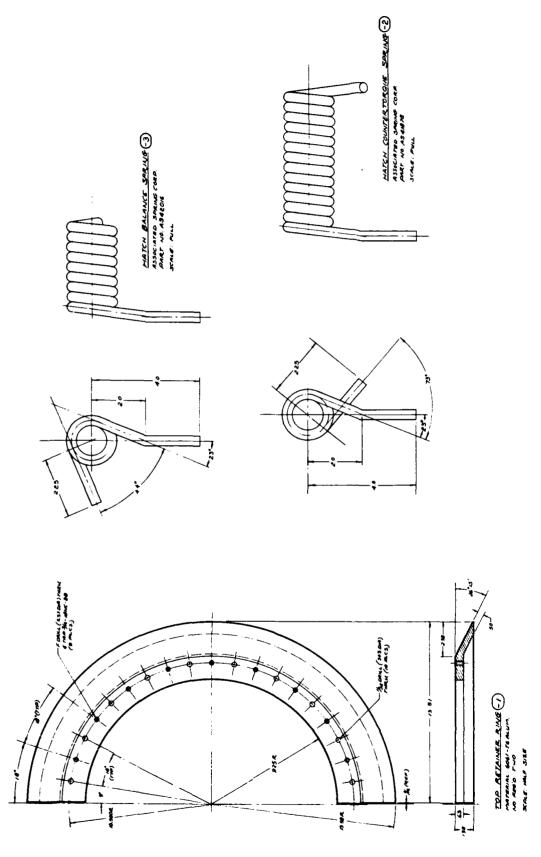


Figure 19A. Top retainer ring and counter balance springs for the 66-inch OD \times 58-inch ID Mode! 2000 Nemo Hull.

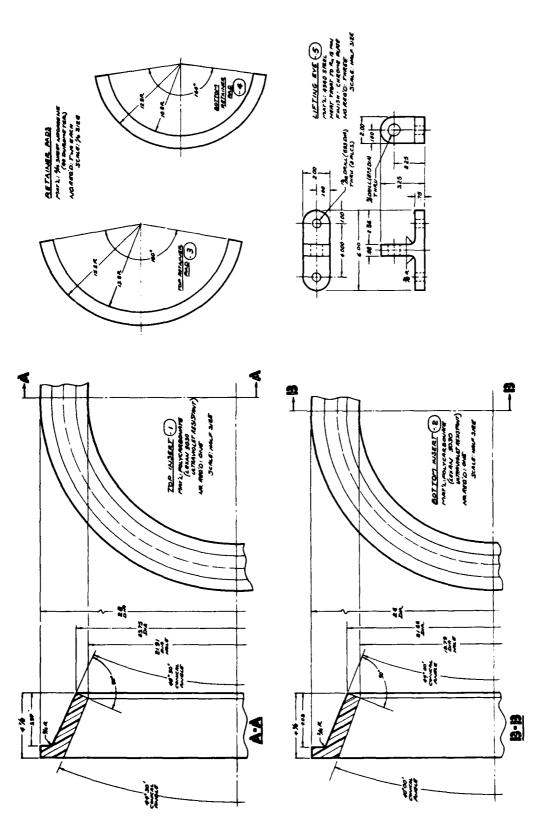
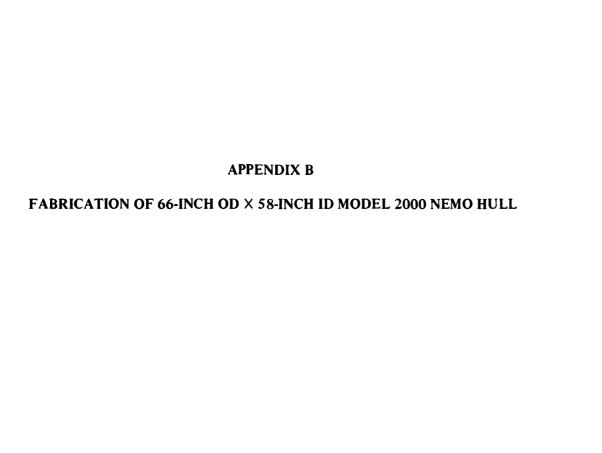


Figure 20A. Inserts, retainer pads, and lifting eye for the 66-inch OD \times 58-inch ID Model 2000 Nemo Hull.



APPENDIX B. FABRICATION OF 66-INCH OD X 58-INCH ID MODEL 2000 NEMO HULL

The 66-inch OD × 58-inch ID Model 2000 Nemo Hull assembly was fabricated basically in the same manner as the first 66-inch OD × 51-inch ID Model 600 Nemo Hull assembly built in 1968 by the Technical Services of Pacific Missile Range, Point Mugu, California. The cardinal features of that fabrication process are (1) cutting of discs from flat acrylic stock (Figure 1B), (2) thermoforming these discs into spherical sectors by means of metallic vacuum mold (Figure 2B), (3) cutting of spherical sectors into spherical pentagons (Figure 3B), (4) bonding of 12 spherical pentagons into a spherical shell (Figure 4B), (5) machining of metallic inserts in the form of top hatch and bottom penetration plate (Figure 5B), and placement of those inserts into polar hull openings (Figure 6B).

One phase of the fabrication process that has given trouble over the years to Nemo fabricators is that of the bonding of assembled 12 spherical pentagons. The problems associated with this phase of fabrication stem from the fact that the thickness of spherical pentagons bordering on a joint is not the same and that the width of joints between pentagons is not uniform. Because of the noncomformity in pentagon thickness and joint width it was difficult to seal the joint effectively so that it would contain the selfpolymerizing adhesive without leakage and yet assure a free flow of adhesive downwards and of displaced air upwards.

Steps were taken during fabrication of the Model 2000 Nemo Hull to eliminate the problems posed by nonuniform pentagon thicknesses and joint widths. These steps consisted of the following operations:

- 1. machining of all formed spherical sectors to uniform thickness in a lathe.
- 2. use of 0.125-inch thick \times 0.25-inch diameter acrylic discs as spacers between individual pentagons during final assembly.
- 3. bonding of acrylic plastic strips to edges of joints for forming of pressure tight joint.
 - 4. placement under pressure of selfpolymerizing adhesive into the joint cavities.

Because of these additional fabrication processing operations the resulting acrylic sphere is more uniform in thickness, sphericity, and diameter. As a result of this uniformity, the finished acrylic huil can be rated to higher operational pressure than was feasibly able prior to this time. Because the improved fabrication process may be of interest to others, a verbatim reproduction of shop fabrication instructions is attached as enclosure 1B at the end of this Appendix.

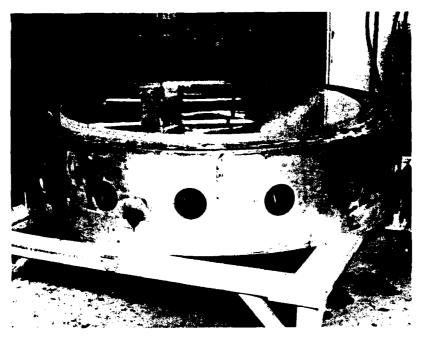
In parallel with the fabrication of the Model 2000 Nemo Hull, stringent quality control measures were a pited to assure a quality product and are attached as enclosures 2B through 9B. The questy control measures consisted of:

1. performing destructive tests on material coupons cut from each sheet of acrylic plastic to ascertain the material properties of plastic (enclosure 2B).

- 2. performing destructive tests on bonded material coupons to ascertain strength of bonded joints (enclosure 3B).
- 3. performing dimensional checks on the spherical hull to ascertain its adherence to specified dimensional tolerances. Samples of dimensional checks are shown for thickness of discs before annealing (enclosure 4B), thickness of disc after annealing (enclosure 5B), thickness of disc after forming (enclosure 6B), sphericity of disc after forming (enclosure 7B), thickness of spherical pentagon after machining (enclosure 8B) and diameter of bonded sphere after annealing (enclosure 9B).



Figure 1B. Sawing of flat plates into circular discs of 46-inch diameter.

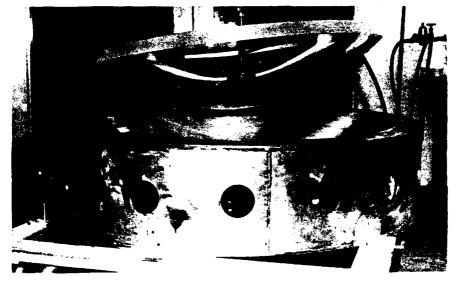


(a) Placement of disc into an aluminum female mold



(b) After forming the spherical sector is ready for removal from the mold

Figure 2B. Thermoforming of flat circular discs into spherical sectors.



(c) The sector is picked up with a vacuum suction disc from the mold



(d) Checking of sphericity on the formed sector

Figure 2B. (Continued).

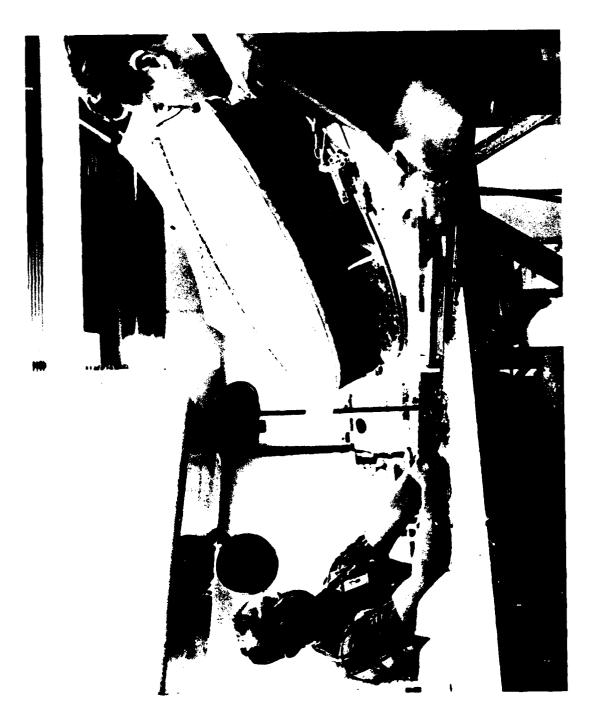
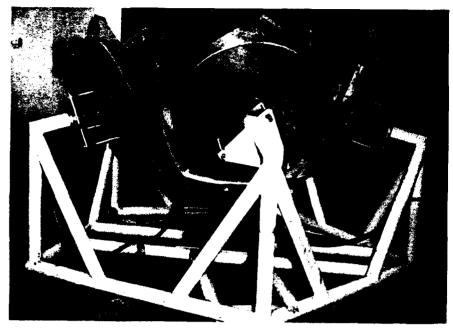
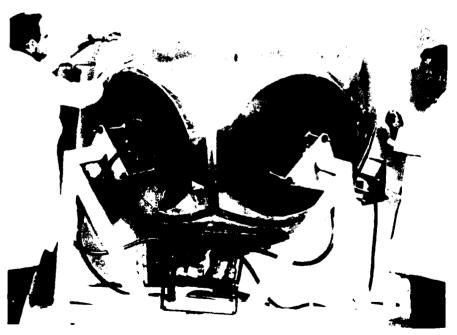


Figure 3B. Sawing the spherical sector into the form of a spherical pentagon.



(a) Bonding of six pentagons to form a hemisphere



(b) Bonding of two hemispheres to form a sphere

Figure 4B. Holding fixture for bonding of spherical pentagons; note the large vacuum suction discs for holding of individual pentagons.



(c) Completed sphere after removal from bonding fixture

Figure 4B. (Continued).

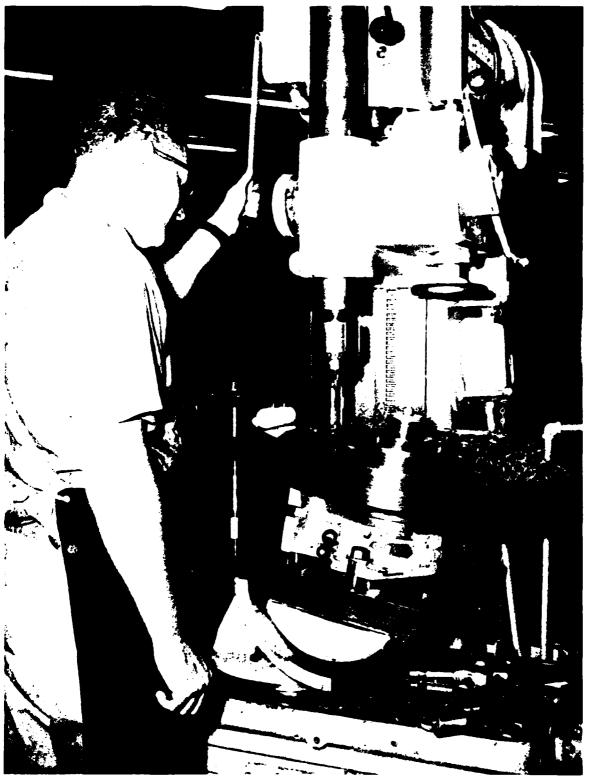
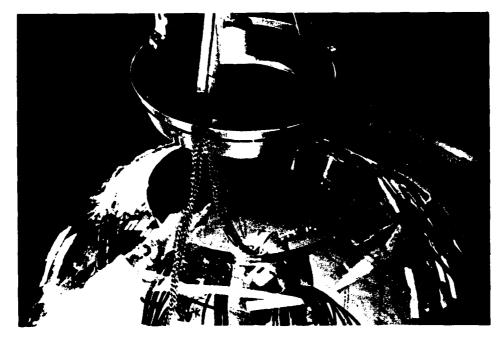


Figure 5B. Machining of metallic inserts for polar openings in the hull.

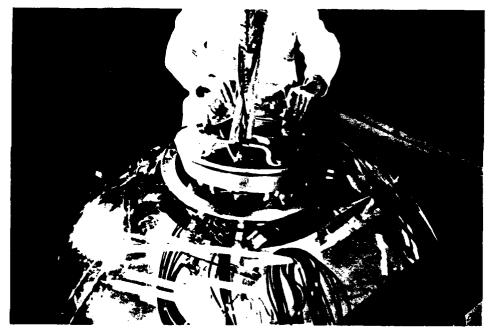


(a) Hatch seat being lowered in place

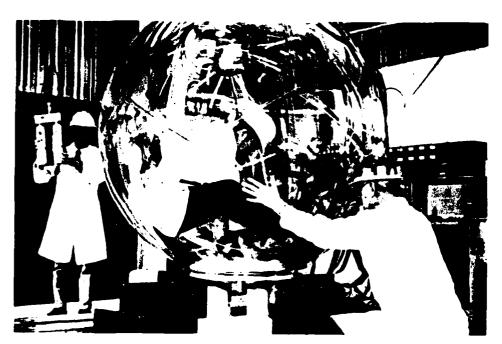


(b) Placement and bolting in place of hatch seat retaining flange

Figure 6B. Placement of metallic inserts into polar opening.



(c) Hatch assembly being attached to the hatch seat



(d) The hull being lowered onto the bottom penetration plate

Figure 6B. (Continued).



(e) Placement in place of retaining flange for bottom penetration plate



(f) Bolting of retaining flange for bottom penetration plate

Figure 6B. (Continued).

SWEDLOW'S INSTRUCTIONS FOR WORKERS FABRICATING NEMO MOD 2000

- Measure thickness of acrylic plate and chart at 12" intervals. Lay out
 a 46" diameter circle at the thickest part of the blank. Lay out nothing
 under 4.050 inches. Scribe SWU # on one cut off.
- Bandsaw trim to line. Save one cut off that is scribed and send to C.Miller in Testing Lab.
- 3. Anneal the part at 325° F for 12 hours on a flat aluminum plate.
- Inspect for thickness and chart at 12" intervals. Do not form plates under 4.170 inches.
- 5. Form per Forming Process Specification
- 6. Inspect for thickness (4.0 4.1) and contour.
- 7. Anneal @ 1750 F for 24 hours.
- 8. Place part in machining fixture and align to marks. Drill three each 1/2" dia. holes and counterbore for 1/2" dia. Allen bolt. Bolt part down in three places.
- 9. Set up part and tracer template in Lucas lathe. Machine inside surface. Remaining thickness to be 4.050 minimum. Check contour readings to be sure part is 4.010 minimum at out of contour places. Part should be machined to 4.100 wherever possible and center section of part should not be cut as it will be under 4.100 thickness. Blend cut and uncut surfaces together with 360 grit wet sandpaper. This part must be polished to an optical finish after machining so the cut must be as smooth as possible.
- 10. Using vacuum lift place part in pentagon machine fixture and even up the edge of the part to the size of the fixture. Vacuum part into place on machining fixture. Make sure that hole in part comes out in the center of the piece to be cut off. Close air toggle clamps. Start saw and set feed at 25 inches per minute.
- 11. While cutting off first piece the saw may have a tendency to slow down or stop. If this happens, immediately press the return button, let the saw return and start the cut over. After cutting off the first piece, open

up the air toggle clamps and press the lever all the way to the left. This will lift the rotating part of the fixture slightly so that the index pin can be pulled and the part rotated to the next position. Continue this procedure until all five cuts have been made. Attach polar hole machining fixture to rotary table and place table on the milling machine. Two pentagons only. Place pentagon into machining fixture and bolt into place. Using a 7/8" dia. x 5" long end mill, cut a hole thru the part per the dimensions on NEMO Model 2000 Drawing #2003

- 12. Inspect for blueprint dimensions.
- 13. Anneal at 175°F for 24 hours.
- 14. Clean up.
- 15. Take to assembly room.
- 16. Place one (1) Polar Pentagon in center of assembly fixture and using the large Starrett No. 656-441 indicator attached to extension, set against 1/2" ball situated in center of assembly pad.
- 17. Making sure indicator is laying flat against holding pad and measure distance from 1/2" ball to the inside of cut out in pentagon.
- 18. When pentagon is centered, place "L" shaped clamps to hold pentagon in place and bolt tightly to keep pentagon from moving during assembly.
- Obtain 1/8" thick x 1/4" diameter acrylic spacers and bond approximately
 4" from each corner centered.

NOTE: Two (2) spacers are needed per side only. If one pentagon has these blocks, then omit blocks to mating side.

Use PS-30 cement - for bonding spacers.

- 20. Using the vacuum hand lift, place five (5) pentagons into assembly fixture and align with polar pentagon.
- After alignment obtain large micrometer with Brown and Sharpe No. 8241-941 dial indicator from inspection box. Obtain the three rods from box (1) 22"
 (2) 23" and (3) 21". Assemble together and set dial indicator to read zero.

- NOTE: Dial indicator will not set at zero-zero dial, but will set at 600, so turn dial indicator to read zero to 600.
- 22. Using dial indicator measure the diameter of the hemisphere and shim where necessary to read $66" \pm .25"$. Measure hemisphere.
 - NOTE: Each pentagon must be measured to one situated directly across from from one to the other.
- 23. When hemisphere is within spherical tolerance of $66" \pm .250"$, obtain Plex G acrylic strips 3/4" thick or S310 material 0.750" wide by 36" long. Take to Machining and rout a groove down the center length 1/8" wide x 1/2" deep.
- 24. Form these strips to fit both outside and inside surfaces of set pentagons.

 Using methylene cloride, bond the strips to the pentagons.
- 25. After all strips have been bonded to hemisphere, place a bead of PS-18 resin around all strips to prevent leaks after hemisphere has been filled with S-49 casting cement.
- 26. Drill an "F" size hole at lowest point of hemisphere. Place a 1/4" OD aluminum tube 3" long into hole and cement into place between pentagons.
- 27. Obtain 2000 grams of basic S-49 resin from resin mixing room-in-a new, clean gallon can: Mix 4 grams of benzoin (.2%) and 10 grams of larurel peroxide (.5%). Place lid on container. Take resin to the NEMO room which is a temperature controlled room @ a constant 72°F.
- 28. Place gallon can on converted edge attachment sander, and set atop the two (2) rollers. Turn switch on and let roll over night.
- 29. Place mixed S-49 as resin into pressure pot and attach nitrogen bottle to pot. Attach fill tube from pot to 1/4" OD Aluminum tube on sphere.
- 30. Apply five (5) pounds pressure.
- 31. Fill joints all around Polar Pentagon and allow resin to rise approximately one (1) inch above the upright joints. (This allows for shrinkage).
- 32. Clamp off tube.
- 33. Allow to cure in NEMO room until hard (approximately 24 hours). Room is

to be kept between $70^{\circ}F$ and $75^{\circ}F$ temperature.

- 34. Remove first hemisphere from cement fixture.
- 35. Rout joints on both outside and inside flush with pentagons.
- 36. Polish up all seams on both inner and outer surfaces.
- 37. Place six (6) new pentagons into assembly fixture and assemble same as first pentagon. Follow steps one (16) through twenty-two (36) above.
- 38. Place hemisphere Number one (1) on top of Number two (2) and check spherical diameter.
- 39. If hemispheres measure within tolerance of $66" \pm .250"$, cut, fit and cement joint strips around equator of sphere leaving a funnel at the top of each bottom pentagon.
- 40. In obtaining and preparing S-49 resin, repeat Steps 27 and 28
- 41. To feed S-49 resin into the equator section of sphere, follow Steps 29 and 30 of Operation 5.
- 42. Allow resin to cure in NEMO room until hard (approximately 24 hours). Room is to be kept between 70° F and 75° F temperature.
- 43. Remove sphere from assembly fixture. Using the cell casting hoist, hook up lifting plate to hoist, fold lifting plate and insert into sphere. Making sure lifting plate will not damage sphere, lift sphere from fixture.
- 44. Machine, sand and polish all bonded areas.
- 45. Anneal sphere in a 1750F oven for 24 hours.
- 46. Inspect all cemented joints for voids larger than 1/4" in diameter.
- 47. Final clean up.
- 48. Final Inspection.
- 49. Wrap with Protec 10V.

SWEDLOW, INC.

Reference: 45-74-121

Date:

March 18, 1974

TEST REPORT

CUSTOMER:

Disbursing Officer, DCASR, Los Angeles

11099 So. La Cienega Boulevard Los Angeles, California 90045

PURCHASE ORDER NO.:

N00123-73-C-1671

MATERIAL TESTED:

Remnants from each sheet used in fabrication

of NEMO Model 2000 Hull

Test specimens were cut to rough dimensions using a bandsaw and to final dimensions (with the exception of tensile specimens) by means of a vertical mill, with a six-flute, carbide-tipped shell end-mill. Tensile specimens were routed to the configuration of a template which complies with dimensions set forth in A.S.T.M. D-638 for Type I specimens. Sharp edges were broken to about .005 inch. Machined surfaces were sanded, first with 280 grit paper, and finally with 600 grit Wet-or-Dry paper to remove tool marks. All specimens were annealed to remove any residual stresses introduced during machining.

Test specimens were conditioned, at a temperature of $73.5 \pm 2^{\circ}F$ and relative humidity of $50 \pm 5\%$, for a period of 40 hours prior to testing.

Tensile, compressive and flexural values were obtained by means of a Tinius-Olsen Elect-O matic Testing Machine. Deformation under load values were obtained on a Tinius-Olsen tester designed for that particular test.

Respectfully submitted,

SWEDLOW, INC.

C. A. Miller, Supervisor Physical Testing Laboratory



TEST REPORT

Date: March 18, 1974

Purchase Order No.: N00123-73-C-1671

FOR: NEMO Model 2000 Hull

Sales Order No.: 3-5940

Page 1

TENSILE: Conditioned 40 hours at 73°F and 50 Percent R. H.

ASTM D-638 tested at 0.05 In/Min

SHEET NO.	SPECIMEN SIZE (INCH)	LOAD (LBS)	ULTIMATE (PSI)	ELONGATION (PERCENT)	MODULUS (PSI)
021	1236 x .483	1330	11,668	7.0	447,000
	2242 x .478	1365	11,800	6.0	465,000
023	1247 x .481	1465	12,331	6.0	467,000
	2251 x .476	1360	11,382	6.0	437,000
024	1255 x .479	1400	11,461	5.5	437,000
	2254 x .482	1420	11,600	5.5	485,000
025	1253 x .481	1425	11,709	5.5	473,000
	2252 x .479	1420	11,764	5.0	490,000
026	1251 x .483	1415	11,671	5.5	474,000
	2252 x .481	1355	11,179	5.5	456,000
027	1252 x .479	1315	10,893	5.5	446,000
	2253 x .477	1275	10,565	5.5	495,000
028	1248 x .472	1235	10,551	5.0	498,000
	2252 x .474	1270	10,632	5.5	492,000
029	1253 x .471	1200	10,070	4.0	446,000
	2247 x .474	1185	10,121	4.0	449,000
034	1254 x .475	1290	10,692	5.5	475,000
	2247 x .475	1265	10,781	5.0	478,000
035	1250 x .479	1250	10,440	5.0	505,000
	2246 x .477	1120	9,545	3.0	428,000
036	1250 x .476	1280	10,756	5.5	454,000
	2251 x .477	1275	10,649	5.0	465,000
037	1249 x .475	1250	10,570	5.5	465,000
	2248 x .474	1235	10,504	5.5	447,000



NEMO Model 2000 Hull S. O. No. 3-5940 Test Report, Continued Page 2

FLEXURAL - Conditioned 40 Hours at 73°F and 50 Percent R. H.

ASTM D-790 - 4 inch span, test speed 0.11 In/Min

SHEET NO.	SPECIMEN SIZE (INCH)	LOAD (LBS)	ULTIMATE (PSI)	MODULUS (PSI)
021	1500 x .258	95.8	17,261	415,000
	2500 x .258	95.5	17,207	443,000
023	1498 x .258	101.3	18,326	444,000
	2491 x .257	82.3	15,238	442,000
024	1500 x .258	103.1	18,577	452,000
	2494 x .258	101.3	18,474	457,000
025	1501 x .257	102.0	18,508	433,000
	2502 x .258	103.1	18,503	445,000
026	1501 x .258	101.1	18,180	460,000
	2498 x .257	101.3	18,492	463,000
027	1493 x .256	96.3	17,883	459,000
	2495 x .256	98.0	18,127	477,000
028	1493 x .249	93.1	18,258	478,000
	2493 x .247	93.7	18,686	474,000
029	1494 x .246	92.4	18,048	479,000
	2495 x .249	93.6	18,306	475,000
034	1492 x .247	88.1	17,588	475,000
	2493 x .246	82.5	16,624	480,000
035	1495 x .248	78.7	15,511	487,000
	2494 x .247	90.6	18,047	484,000
036	1494 x .247	77.0	15,326	473,000
	2493 x .248	92.1	18,210	484,000
037	1492 x .248	91.4	18,113	469,000
	2495 x .248	92.2	18,194	467,000



NEMO Model 2000 Hull S. O. No. 3-5940 Test Report, Continued Page 3

DEFORMATION UNDER LOAD

ASTM D 621 - Tested as received at 122°F for 24 hours under 4000 psi load Test Specimens: 1/2 Inch Cube

SHEET		DEFORMATION (INCH)		DIFFER	MICROMETER READING	CALC. ORIG. THICKNESS	DEFORM.
NO		10 SEC.	24 HOURS	<u>(IN)</u>	<u>(IN)</u>	<u>(IN)</u>	(%)
021	1-	0.0863	0.0889	0.0026	0.4998	0.5024	0.52
	2-	0.0822	0.0848	0.0026	0.5025	0.5051	0.51
023	1-	0.0857	0.0886	0.0029	0.4985	0.5014	0.58
	2-	0.0813	0.0842	0.0029	0.5020	0.5049	0.57
024	1-	0.0819	0.0843	0.002 4	0.5013	0.5037	0.48
	2-	0.0764	0.0788	0.0022	0.5025	0.5047	0.44
025	1-	0.0755	0.0778	0.0023	0.5036	0.5059	0.45
	2-	0.0763	0.0787	0.0024	0.5023	0.5047	0.48
026	1-	0.0765	0.0795	0.0030	0.5014	0.5044	0.60
	2-	0.0752	0.0774	0.0022	0.5007	0.5029	0.44
027	1-	0.0684	0.0708	0.0024	0.5029	0.5053	0.47
	2-	0.0676	0.0699	0.0023	0.5036	0.5039	0.46
028	1-	0.0738	0.0761	0.0023	0.4980	0.5003	0.46
	2-	0.0708	0.0735	0.0027	0.5003	0.5030	0.54
029	1-	0.0691	0.0712	0.0021	0.5014	0.5035	0.42
	2-	0.0707	0.0733	0.0026	0.4997	0.5023	0.52
034	1-	0.0709	0.0743	0.0034	0.4986	0.5020	0. <i>67</i>
	2-	0.0720	0.0751	0.0031	0.4983	0.5014	0.61
035	1-	0.0714	0.0745	0.0031	0.4990	0.5021	0.61
	2-	0.0711	0.0744	0.0033	0.4980	0.5013	0.65
036	1-	0.0715	0.0747	0.0032	0.4975	0.5007	0.63
	2-	0.0698	0.0734	0.0036	0.4996	0.5032	0.72
037	1-	0.0708	0.0740	0.0032	0.4991	0.5023	0.64
	2-	0.0707	0.0740	0.0033	0.4982	0.5015	0.66



NEMO Model 2000 Hull S. O. No. 3-5940 Test Report, Continued Page 4

SHEAR STRENGTH

ASTM D-732

Rate of Test: 0.05 In/Min Punch Diameter: 0.999 In. (1.000 In. Dia. disc punched out)

SHEET NO.		THICKNESS (INCH)	MAXIMUM LOAD (LBS)	SHEAR STRENGTH (PSI)
021	1-	0.259	8,240	10,100
	2-	0.253	8,250	10,400
023	1-	0.255	8,190	10,200
	2-	0.254	8,220	10,300
024	1-	0.254	8,190	10,300
	2-	0.256	8,260	10,300
025	1-	0.257	8,220	10,200
	2-	0.258	8,340	10,300
026	1-	0.254	8,830	11,100
	2-	0.255	8,210	10,300
027	1-	0.250	7,800	9,930
	2-	0.245	7,770	10,100
028	1-	0.245	8,220	10,700
	2-	0.254	8,440	10,600
029	1-	0.250	8,200	10,400
	2-	0.253	8,690	10,900
034	1-	0.253	8,870	11,200
	2-	0.254	8,770	11,000
035	1-	0.252	9,130	11,500
	2-	0.253	8,450	10,600
036	1-	0.255	9,100	11,400
	2-	0.253	8,670	10,900
037	1-	0.254	8,340	10,500
	2-	0.256	7,950	9,880



NEMO Model 2000 Hull S. O. No. 3-5940 Test Report, Continued Page 5

$\underline{\textbf{COMPRESSIVE PROPERTIES}} \;: \; \; \textbf{Tested at Room Temperature}$

ASTM D-695 Rate of Test: 0.05 In/Min

SHEET NO. SPECIMEN SIZE (INCH) YIELD STRENGTH (LBS) YIELD STRENGTH (PSI) 021 1- 0.503 x 0.502 x 1.504 4,880 19,300 2- 0.503 x 0.502 x 1.505 4,950 19,600 19,300 19,600 023 1- 0.500 x 0.503 x 1.502 4,630 18,400 2- 0.501 x 0.503 x 1.502 4,590 18,200 18,400 18,200 18,200 024 1- 0.503 x 0.503 x 1.499 4,740 18,700 18,700 2- 0.504 x 0.503 x 1.499 4,740 18,700 18,200 18,300 18,800 025 1- 0.505 x 0.505 x 1.502 4,660 18,300 18,800 18,800 18,800 026 1- 0.503 x 0.503 x 1.501 4,770 18,100 18,500 027 1- 0.504 x 0.506 x 1.500 4,650 18,500 028 1- 0.501 x 0.501 x 1.500 4,650 18,400 029 1- 0.502 x 0.502 x 1.507 4,710 18,700 029 1- 0.501 x 0.501 x 1.506 4,610 18,300 034 1- 0.501 x 0.502 x 1.505 4,600 18,400 035 1- 0.502 x 0.502 x 1.505 4,600 18,400 036 1- 0.503 x 0.501 x 1.506 4,480 17,700 036 1- 0.503 x 0.500 x 1.505 4,506 4,480 17,700 036 1- 0.503 x 0.500 x 1.506 4,480 17,700 036 1- 0.503 x 0.500 x 1.506 4,480 17,900		
2- 0.503 x 0.502 x 1.505		MODULUS (PSI)
2- 0.501 x 0.503 x 1.502		570,000 530,000
2- 0.504 x 0.503 x 1.499		520,000 500,000
2- 0.504 x 0.504 x 1.500 4,780 18,800 026 1- 0.503 x 0.503 x 1.501 4,570 18,100 2- 0.503 x 0.501 x 1.500 4,650 18,500 027 1- 0.504 x 0.506 x 1.502 4,650 18,200 2- 0.506 x 0.504 x 1.504 4,690 18,400 028 1- 0.501 x 0.501 x 1.500 4,650 18,500 2- 0.502 x 0.502 x 1.507 4,710 18,700 029 1- 0.502 x 0.502 x 1.506 4,810 19,100 2- 0.502 x 0.502 x 1.506 4,610 18,300 034 1- 0.501 x 0.502 x 1.505 4,600 18,400 035 1- 0.502 x 0.501 x 1.506 4,650 18,500 2- 0.503 x 0.503 x 1.506 4,600 18,400 036 1- 0.503 x 0.503 x 1.506 4,480 17,700 036 1- 0.503 x 0.502 x 1.505 4,650 18,500		510,000 500,000
2- 0.503 x 0.501 x 1.500	025	510,000 510,000
2- 0.506 x 0.504 x 1.504	026	510,000 520,000
2- 0.502 x 0.502 x 1.507 4,710 18,700 029 1- 0.502 x 0.502 x 1.506 4,810 19,100 2- 0.502 x 0.502 x 1.506 4,610 18,300 034 1- 0.501 x 0.502 x 1.505 4,600 18,400 035 1- 0.502 x 0.501 x 1.506 4,650 18,500 2- 0.503 x 0.503 x 1.506 4,480 17,700 036 1- 0.503 x 0.502 x 1.505 4,510 17,900	027	510,000 520,000
2- 0.502 x 0.502 x 1.506		540,000 540,000
2- 0.501 x 0.500 x 1.505 4,600 18,400 035 1- 0.502 x 0.501 x 1.506 4,650 18,500 2- 0.503 x 0.503 x 1.506 4,480 17,700 036 1- 0.503 x 0.502 x 1.505 4,510 17,900	029	530,000 500,000
2- 0.503 x 0.503 x 1.506 4,480 17,700 036 1- 0.503 x 0.502 x 1.505 4,510 17,900	034	510,000 530,000
	035	520,000 530,000
	036	510,000 510,000
037 1- 0.501 x 0.502 x 1.505 4,640 18,400 2- 0.503 x 0.502 x 1.503 4,510 17,900	037	530,000 530,000

ENCLOSURE 3B



Date: 21 May 1974

TEST REPORT

For: NEMO Model 2000 Hull

Purchase Order No. N00123-73-C-1671

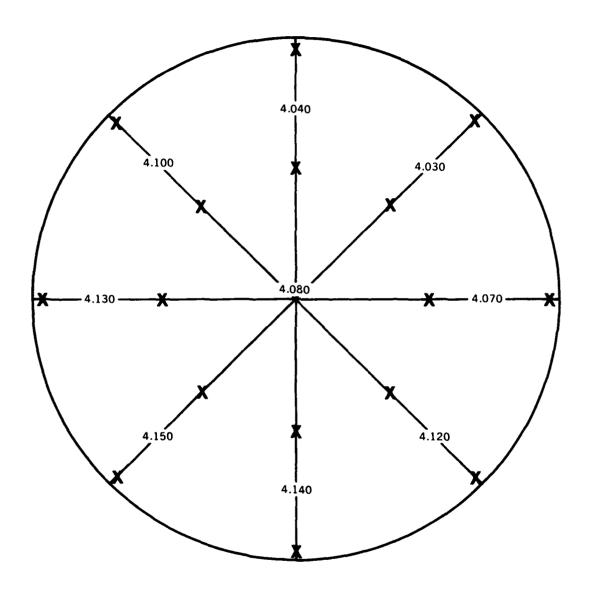
BONDED JOINT TENSILE STRENGTH

(Required: 5000 psi)

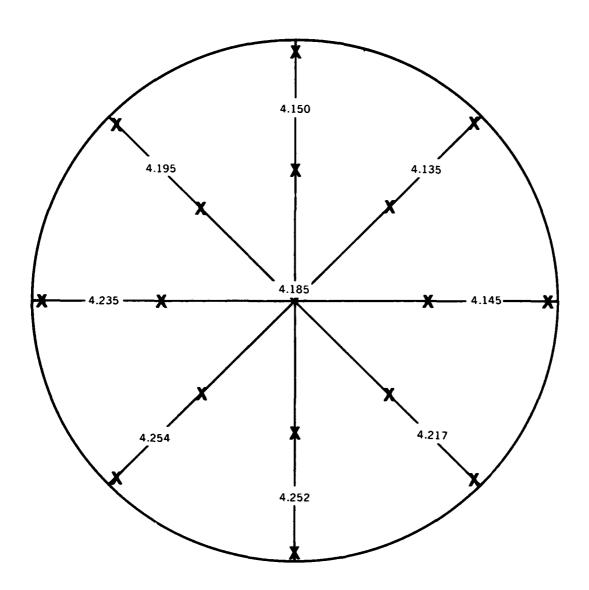
Test specimens were machined from the joint evaluation coupon which had been bonded and annealed in the same manner as the pressure hull. Specimens were of the dimensional configuration set forth in Sketch No. 2002. The testing speed used was 0.05 inch per minute.

<u>Specimen</u>	Width <u>(In)</u>	Thickness (In)	Load (Lbs)	Ultimate Strength (psi)	Mode of Failure
1	0.744	0.529	3015	7661	Cohesive
2	0.747	0.532	3210	8074	Acrylic
3	0.750	0.485	2825	7766	Cohesive
4	0.746	0.547	3720	9116	Cohesive
5	0.750	0.540	2075	5123	Cohesive
6	0.749	0.534	3550	8876	Cohesive
7	0.747	0.531	2920	7362	Cohesive
8	0.748	0.526	3240	8235	Cohesive
9	0.742	0.519	2365	6141	Cohesive
10	0.748	0.536	3320	8281	Cohesive
11	0.745	0.536	3390	8489	Cohesive
12	0.744	0.534	3440	<u>8659</u>	Cohesive
			Average	7815	

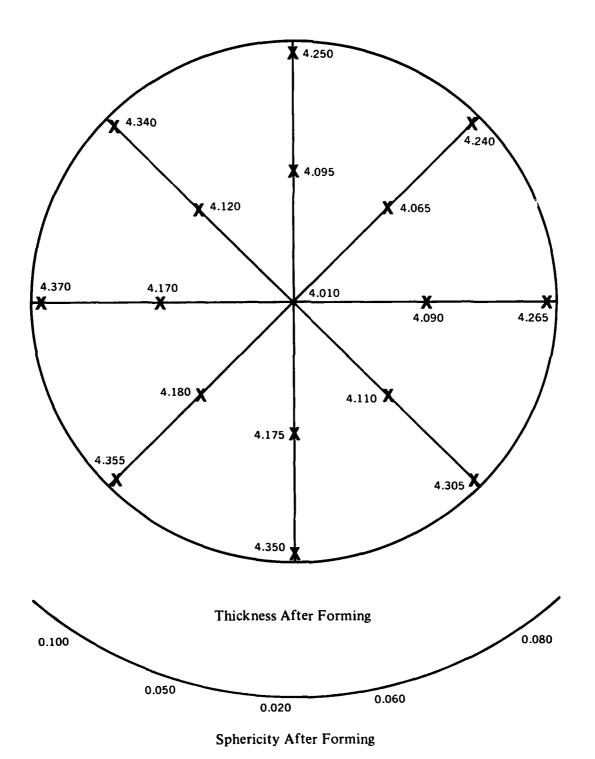
C. A. Miller, Supervisor Testing Laboratory

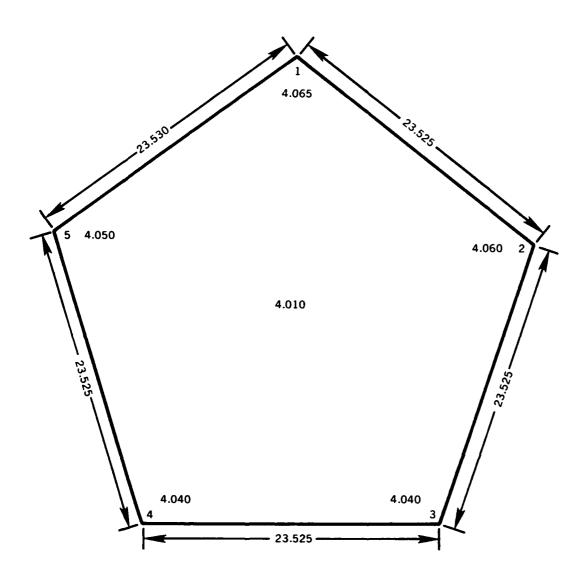


Thickness Before Annealing

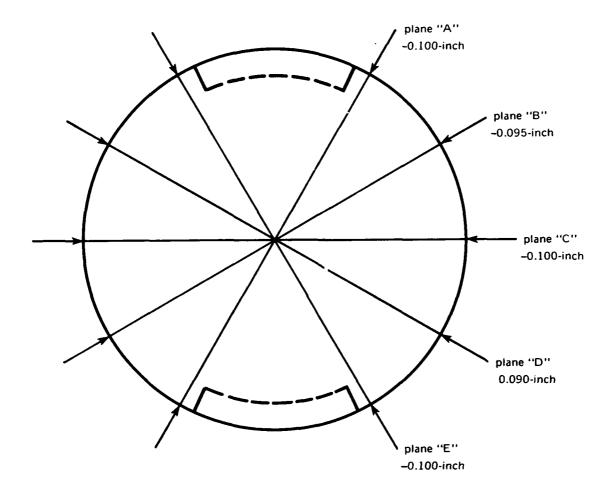


Thickness After Annealing





Thickness After Machining



Diameter of the Finished Sphere

APPENDIX C DATA FROM HYDROSTATIC TESTS

APPENDIX C. DATA FROM HYDROSTATIC TESTS

Both the 15-inch OD × 13-inch ID scale Model 34 and the 66-inch OD × 58-inch ID full scale Model 2000 Nemo Hulls were extensively instrumented with strain-gages (see Figs. 20 and 23 of main text) so that their structural performance under external hydrostatic pressure could be accurately measured and evaluated. Highlights of that data have been summarized (see Tables 4, 5 and 6) and discussed in the main body of the report.

Still, other researchers in the field of acrylic plastic pressure hulls may need to refer to some specific detail of that data not provided in the main body of the report. For this reason the data generated during some of the more severe hydrostatic tests has been selected for presentation here in Appendix C. For the 15-inch OD × 13-inch ID scale Model 34 the most severe as well as the most important test was pressurizing to implosion; this data is presented in Table 1C. The 66-inch OD × 58-inch ID full scale Model 2000 Nemo Hull has not been tested to implosion therefore this data is not available. Instead, the data from the long term pressurization cycles to 1350 and 1800 psi are shown in Tables 2C and 3C.

Data from 15-inch OD X 13-inch ID Model 34

Data Recording

The strain data was generated by subjecting the 15-inch OD X 13-inch ID Model 34 to external hydrostatic pressure rising at 100 psi/minute rate. During recording of strain data (which took about 2 minutes per recording) the pressure was held constant. The temperature of water used in pressurizing of Model 34 was in the 70-75°F range.

Data Interpretation

The most important observation that can be made of the strain data from Model 34 is that all the strains measured on the acrylic plastic as well as on aluminum were linear to well over 1400 psi, indicating that both materials were still in the elastic range when the simulated depth passed the 3000-foot mark. At simulated depth of 4000 feet most of the strains became markedly non-linear, and at 5000 feet the strains became exponential.

The linear behavior of strains to depths in the 3000-3300 foot range substantiates the postulate that the Model 2000 Nemo Hull design (and its scaled down version Nemo Model 34) is operationally safe to 3000 feet as all of the materials in the hull respond elastically in the 0-3000 foot range.

Data from 66-Inch OD X 58-Inch ID Model 2000 Nemo Hull

Data Recording

The strain data was generated by subjecting the 66-inch OD X 58-inch ID Model 2000 Nemo Hull to external hydrostatic pressure rising at 100 psi/minute rate. During recording of strain data (which took about 2 minutes per recording), the pressure was held constant. The temperature of water used in pressurizing of the Model 2000 Nemo Hull was in the 70-75°F range.

Once the maximum pressure was reached further pressurization was stopped and the maximum pressure maintained for 24 hours. Readings were taken at 6 hour intervals, but only the last reading (taken 24 hours after initiation of long term loading) is shown on the data printout.

Readings were taken also during the depressurization which took place at 100 psi/minute rate. Upon reaching 0 pressure the Model 2000 Nemo Hull assembly was allowed to relax for 24 hours while readings were taken every 6 hours (only the last relaxation reading is shown in the data printout).

Data Interpretation

The strains are linear to 3000 feet. The creep in acrylic plastic after 24 hours of sustained loading at 3000 feet was less than 20 percent of the short term strain at that depth. After reduction of pressure to 0 psi and 24 hour relaxation at 0 psi all strains returned to zero. Both the linearity of strain in the 0-3000 foot range and return of strains to zero at conclusion of the pressure cycle indicate that the Model 2000 Nemo Hull can be repeatedly pressurized without permanent deformation to 3000 feet.

When pressurized to 4000 feet there was some nonlinearity at depths beyond 3500 feet. The creep in acrylic plastic after 24 hours of sustained loading at 4000 feet was about 30 percent of the short term strain at that depth. After reduction of pressure to 0 and 24 hours of relaxation at 0 pressure most of the strains in acrylic and aluminum return essentially to zero (are within ± 100 microinches of original zero). Only at some interior locations on aluminum components (location 6, 13, 14) were the remaining stresses positive and significantly high. No explanation has been found for their tensile character, or large magnitude.

The indications of nonlinearity during pressurization in the 3500-4000 foot depth range, excessive creep during long term pressurization at 4000 feet, and residual strains after relaxation at 0 depth indicate that pressurization of the Model 2000 Nemo Hull to 4000 feet subjects the assembly to excessive stresses. It is therefore postulated that the Model 2000 Nemo Hull assembly should not be prooftested in excess of 3600 feet for service at depths to 3300 feet.

Table 1C. Strains Measured on 15-Inch OD × 13-Inch ID Nemo Model 34

Load Psi	Gage No. 1A-C	Gage No. 1A-L	Gage No. Gag	Gage No. 1B-L	Gage No. 2B-C	Gage No. 2B-L	Gage No. 3B-C	Gage No. 3B-L	Gage No. 4,4-C	Gage No. 4A-L
0	0	0	0	0	0	0	0	0	0	0
001	-500	-500	-550	-500	-1.800	006-	-80	0	-50	-50
200	-1,100	-1,000	-1,150	-1,100	-1,250	-2,100	-170	+10	-75	-75
300	-I.400	-1,500	-1,700	-1,800	-1,900	-2,700	-250	+20	-100	-100
400	-1.800	-1,900	-2,250	-2,350	-2,300	-3,300	-330	+30	-115	-120
200	-2,200	-2,250	-2,800	-2,900	-2,800	-3,950	400	+20	-125	-130
009	-2,650	-2,650	-3,400	-3,500	-3,300	4,600	470	+40	-140	-150
700	-3,000	-3,050	-3,950	4,000	-3,750	-5,100	-540	06+	-150	-150
800	-3,400	-3,400	4,500	4,500	4,200	-5,750	009-	+110	-175	-150
006	-3.800	-3,800	-5,000	-5,100	4,600	-6,400	-650	+130	-195	-120
1,000	4,200	4,200	-5,600	-5,700	-5,100	-7,050	-700	+150	-200	-100
1,100	4,600	4,600	-6,200	-6,200	-5,550	-7,650	-780	+150	-225	-75
1,200	-5,000	-5,000	-6,700	-6,800	-6,000	-8,350	-860	+150	-250	-50
1,300	-5,400	-5,400	-7,300	-7,300	-6,400	-8,950	-940	+150	-260	-30
1,400	-5,900	-5,800	-7,900	-7,900	-6,950	-9,700	-1,020	+130	-280	-20
1,500	-6,300	-6,200	-8,600	-8,500	-7,400	-10,400	-1,100	+110		
1,600	-6,700	-6,600	-9,100	-9,200	-7,900	-11,000	-1,190	06+		
1,700	-7,200	-7,000	-9,900	-9,900	-8,500	-11,850	-1,310	+70		
1,800	-7,750	-7,600	-10,600	-10,600	-9,200	-12,650	-1,400	+20		
1,900	-8,200	-8,000	-11,400	-11,500	-9,900	-13,600	-1,490	+30		
2,000	-8,800	-8,600	-12,200	-12,200	-10,500	-14,600	-1,600	+10		
2,200	-9,800	-9,500	-13,700	-13,800	-11,600	-16,500	-1,750	-10		
2,400	-10,000	-10,500	-15,200	-15,200	-12,800	-18,250	-1,900	-30		
2,600	-12,100	-11,700	-17,100	-17,800	-13,200	-21,000	-2,050	-50		
2,800	-13,400	-12,900	-18,600	-18,800	-15,800	-22,850	-2,150	-70		
3,200	-16,400	-15,600	-23,800	-22,700	-19,800	-28,900	-2,200	06-		
3,600	-18,600	-18,000	-27,100		-28,000	>-30,000	-2,250	-110		
4,000	-21,600	-20,200	>-30,000		>-30,000	>-30,000	-2,200	-120		

Location of gages shown on Figure 20, pg. 44.

Table 1C. (Continued).

Gage No. 4B-C	. Gage No. 4B-L	Gage No. 5A-C	Gage No. 5A-L	Gage No. 5B-C	Gage No. 5B-L	Gage No. 6B-C	Gage No. 6B-L	Gage No. Gage No. 7B-L	Gage No. 7B-L
	0	0	0	0	0	0	0	0	0
	-180	-10	0	-80	-10	-800	-1.000	-90	06
\Box	-300	-20	10	-160	-20	-1,500	-1.800	-210	150
0	400	-30	10	-240	-30	-2,000	-2,400	-300	150
\odot		4	10	-330	-50	-2,500	-2,950	400	190
	-590	-50	20	4	09-	-3,000	-3,450	490	190
2	-700	09-	20	490	-80	-3.600	4,000	-590	220
20	062-	-70	20	-570	-100	4,100	4.500	9	230
30	006-	-80	30	-650	-130	4,600	-5,050	-710	230
90	-1,010	06-	30	-710	-170	-5,100	-5.600	006-	240
50		-100	30	-780	-200	-5,700	-6.100	-1.000	240
10		-110	30	-840	-210	-6,150	-6.700	-1,110	240
580	T	-120	40	-920	-220	-6,800	-7,100	-1,220	250
30		-130	40	-1,000	-230	-7.200	-7.700	-1,330	250
96	-1,430	-140	40	-1,090	-240	-7.800	-8,300	-1,440	250
350	-1,520	-150	20	-1,180	-260	-8,250	-8.900	-1,550	270
920	T	-160	20	-1,280	-280	-8,750	009,6-	-1,600	290
-990	-1,710	-170	20	-1,320	-300	-9,300	-10,450	-1,720	320
090,	-1,800	-180	20	-1.370	-320	-9,900	-11,250	-1,830	350
130	ı	-190	09	-1,550	-335	-10,600	-12,200	-1,900	430
1,200	-2,050	-200	09	-1,630	-350	-11,200	-13,050	-2,020	200
80	-2,400	-220	09	-1,800	400	-12,500	-14,700	-2,270	009
,400	-2,600	-280	70	-2,000	450	-13,900	-16,300	-2,470	700
1,500	-2,800	-280	80	-2,200	460	-15,700	-18,100	-2,700	850
50	-2,900	-310	80	-2,340	480	-17,550	-19,800	-2,860	006
8	-3,100	-370	06	-2,500	-550	-23,100	-22,800	-3,250	1,000
550	-3,250	-560	110	-2.500	-640	-27,500		-3,550	1,180
8	-3,400	-530	120	-2,840	-550	>-30,030		-3,750	1,350

Table 2C. Strains Measured on the 66-Inch OD X 58-Inch ID Model 2000 Nemo Hull Under 24-Hour Long Hydrostatic Loading of 1350 Psi

GAGE NO. R 1-0UTSIDE	WIGHA TAL			65.07.5			4754						98¢ 2554 5												
0. 20	MIGMA	•	5	≱ 20€	-538	-764	-1010	61833	1811-	-1714	0502-	おかのかま	-2381	-2381	-2271	E (C) (C)	-1740	-151+	-1319	000	-714	***	967-	•	
POISSONS RATION	~ d.	•	>	056	-1700	-2500	-3300	- 100	0064-	-5750	-6350	- 7200	-7500	-7500	-7300	-6550	-5650	-4700	0086-	0562-	-2000	-1100	-350	0	
	EPL	•	3	•300	0.5 * •	-600	009	055	15	0	-1850	50	9	-≥000	-1850	-1650	-1500	-1300	-1250	T	• 200	05.	-150	0	
0,.	1040	•	5	150	300	05.	00.9	750	006	1050	1200	4 1350	0361	1360	1200	1050	006	750	9 00	05+	300		0 _**	0	ı

^{*} denotes strains at the beginning and conclusion of 24 hour long sustained loading at 1350 psi

EP2 - longitudinal orientation

^{** [} denotes strains at the beginning and conclusion of 24 hour long relaxation at 0 psi

EP1 - hoop orientation

Table 2C. (Continued).

	GAGE NO. R 2-DUTSIDE	D× E× E× E× E× E× E× E× E× E× E× E× E× E×	8	9 (3	~ =	.		~ *	•	+ ₹•	51.	٥	0	6.	F	~•		~-	***	~	0	~-	0
غ ف		MIN	0	K930		9161.	0181.	9622	W + L & .	2126	4846*	48144	e£033	E603=	9484	のナのナー	0.186.1	0 + N O 1	-2676	9861-	-	-867	-276	0
OF A TWO GAGE ROSETTE	RATIOS .40	O I GEA MAX	0	K9 # 8		0661=	+78T+	0622-	45754	******	#14E*	*10**	£603a	-5033	050+1	P. 50 四十二	1362¢	-3257	0592-	+501+	+2+1-	9	062-	0
STRAIN REDUCTION O	POISSONS H	ев 84 М	0	004	00+1+	-2050	0062-	0016	001**	008**	0055=	-6250	0556	-7550	0064-	-6500	-5700	050+-	000+	-2450	-2100	1300	00+-	0
		E P 1	0	- 100	2	8	275	318	415	88	-	635	755	755	35	655	575	6	405	305	25	130	*	0
	£*0	1040	G	150	300	05+	004	750	006	1050	1200	1350	0561	7350	1200	1050	900	750	009	05+	300	150	0	0

Location of gages shown on Figure 23, pg. 48.

Table 2C. (Continued).

=

	GAGE NO. = 3-DUTSIDE	TAU MAX	0 50			- N - M	E S	-9 C 4	000	000) F	104	- A :	에 1 원 주 한 17	+ C C			. D
		₹ Z J Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z	≎ 5	3 9 8 B	20 20 20 20 20 20 20 20 20 20 20 20 20 2	\$0.000 €	•2710	-9 C - 2 C -	C-956-0		-3110	+275+	2002 2003	7/97-			-114	0
STRAIN REDUCTION OF A TWO GAGE ROSETTE	HATIOS . 40	4 × 4 × 4 × 4 × 4 × 4 × 4 × 4 × 4 × 4 ×	0 9 6 5 -	2061		0161.	\$25T	*****	K957.	-1467	-1724	-1510			1576	5-2-	8 + 1	•
STRAIN REDUCTION OF	POISSONS HA	E P &	© 96 € 8	- 1650 - 2350	• 3000	005	0525		0569=	0.000	-6050	0065	14700		1400	-1050	-250	6
		EP1	0 0 •	00.5	0.20	056	011	-1.35¢	1 35	-1350 -1300	720			004	05+	002-	c) 1	o
	0+*	1040	U 20	900 + 20	600 750	9	50	0567	35	1350	7050			450	300	750	0 (>

Table 2C. (Continued).

	STRAIN REDUCTION OF A TWO GAGE	A TWU GAGE ROSETTE	ial	
	Polssons ratios	110* .*0		GAGE NO. # W.OUTSIDE
EP	2	SIGIS	A S I S I S I S I S I S I S I S I S I S	HAK
0	•	0	9	0
• 750	906-	ロケナ・	9640	••
-1400	-1350	サント・	016	••
	0481=	の少れてま	P161.	*I*
-2750	-2600	-1805	-1762	12.
0046.	9300	87448	#155m	⇒ 7•
0501-	0056.	-2671	6294·	120
008 **	004**	-316-	5016.	B ~ ■
0055-	-5250	-361d	8+56.	• 36
3	0465-	980*	*10**	• 36
-2400	3354	60 + 10 + 1	B16 **	* 4
9	0569-	0+0+1	P154-	**
2	0089-	8444	F19+-	***
Š	-6150	のせいナー	-4157	m + 1
595	-5350	49710	43624	M+0
0	005+-	E+16=	-3057	nt.
5	13750	-2545	-2536	F2-
2	0052	-202-	-1971	~
2	0502-	物のナベー	56ET-	-51
35	-1200	-671	-829	
005-	05**	±200 €	016-	~ .
0	0	9	6	0

Table 2C. (Continued).

Table 2C. (Continued).

	-7	STRAIN REDUCTION UF	UF A TWO GAGE HOSETTE	LL.	
£# 10,00		POISSONS RAT	RAT10s , 30		GAGE NO. # 6-DUTSIDE
LOAD	1 d 3	5 d a	O I GEA MAX	O I GHA	TAU
6	9	Đ	0	9	۵
150	95	05=	+T4=	-214	D
300	-100	-150	E 1 2 4 3	8451.	257
450	•150	-150	E+T2+	E+15=	G
009	-200	6 ₹00	-2857	-2857	Đ
750	058	-251	1626-	1456.	ย
007	9300	-250	2	•3736	-145
S	056.	990	-+B35	1544=	-19€
2	001	-350	40	-5165	-145
Ş	05+	007	19.	~	261 ■
0567	00**	-350	B 2 5 4 9		んぎつ
1350		-350	-5544	-5165	5 T T 8
1200	2	086-	-582	-5165	-192
1050	0	006-	-5385	5194-	-385
006	35	006•	-+635	TS++=	+142
750	30	-250	-4121	-3736	~
600	40	-200	トロナの「	-3022	-142
08+	15	-150	E+1a=	E+12=	0
300	5	-100	P C B I	-1264	761
150	10	-50	+14-	+14	6
9	0	0	c	0	0
0	0	0	0	0	0

Table 2C. (Continued).

		STRAIN REDUCTION OF	OF A TWO GAGE ROSETTE	to i	
E= 10.00		POISSONS RATIOS	10. 33		GAGE NO. # 7-0UTSIDE
LOAD	1 43	2	A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A X A Z A Z	WICE MIN	TAU
c	0	0	6	0	5
160	9	05•	-1264	B 2 8 -	251-
000	-150	•100	-1478	£651•	2511
	5	-150	ENTRE	-2143	6
00.	25	-500 -500	*O*E	- 3052	267.
750	8	0.250	1611	-3736	-145
00	E	300	時の位かる	15440	251.
1050	SE		-5000	0005-	9
1200	000	00*	*1450	11650	0
0561	5	057	アルナル		5
1350	Ş	05+	5~+9·	P219.	c
1350	054-	05#*	-648	-6484	o
1200	00+			*029*	251
2050		096	0000	0008-	i 60
00	064	900	5504-	1944	2510
750	006-	092-	-4121	-3736	-142
00,	054-	-200	トロナのコ	-3022	-142
05 +	-150	-150	-2143	£+12-	8
3 00	-100	•100	-1484	P541-	0
750		05:	+14	-214	0
0	0	0	0	0	6
0	0	0	9	•	9

Table 2C. (Continued).

		STRAIN REDUCTION OF	A TWO GAGE RUSETTE	4.4	
10.00	,	POISSONS KATIOS	06. 801		GAGE NO. # 8-DUTSIDE
LOAD	F 9	2	AN NAR	SIE	HAK MAX
Đ	o	o	0	0	0
150	0.60	20	5801	386	5 35 M =
300	007=	20	\$0.5°	022	1577
05+	-150	700	P181-	* O 4	-4b-
009	002•	100	• 1868	0**	-115+
750	-250	100	BT +2-	575	-
006	006-	100	-2967 -	110	*1538
1050	-300	750	#0##	ទ ហ .	~
1500	- 350	700	9756*	\$ S +	-1731
1350	058.	700	9756.	55 B	-1731
1350	900	700	6.246.7	110	_
1380	-300	700	-2967	110	
1200	-300	700	-2967	110	9651-
1050	900	700	-2967	110	9651-
900	058	05	-2582	562-	-1154
750	052-	90	-2582	-275	-1154
009	004=	100	-1868	0**	-1154
05	-150	05	****	S S	-769
300	-100	090	**************************************	220	-577
150	08-	0	T + 10 1	-165	-192
0	0	0	0	0	6
0	0	0	0	•	. 6

Table 2C. (Continued).

		STRAIN REDUCTION OF	OF A THU GAGE HUSETTE	ui	
10.00		POISSONS KATIOS	ATIOR , 30		GAGE NO. R 4-DUTSIDE
LOAD	1	F P	BIGHA MAX	A N I N I N I N I N I N I N I N I N I N	TAU MAX
6	6	9	0	0	9
150	05.	-100	-B79	-1264	767
300	07	057=	£651•	8661.	261
- NO	-150	002-	80£2•	2642	192
009	-150	082*	£642*	2 4 2 E =	385
750	002-	006.	-3182	9566*	385
006	20	-350	- 3352	\$05 to	# P. S
1050	2		TONE	044.	385
0021		001	44000	05550	577
1350	3		084+	かのいいき	275
1350	2	051-	084 **	****	500
1350	C	09**	0824-	-643-	523
1500	w	004	14066	-5250	523
1050	2	-350	4988	\$05 +	573
3 00	20	056-	-3325	505+=	573
750	2	-350	-280s	T+S+*	769
600	57	●300	-2637	1646-	575
05+	-150	002-	-2308	-2692-	192
300	2	-100	P 2 + 1 =	P. S. T. L.	0
750	ED.	0.61	+14-	+72-	0
0	0	0	0	0	0
0	0	0	0	•	0

Table 2C. (Continued).

		STRAIN REDUCTION OF	A THO GAGE HOSETTE	æi	
Es 10,00		POISSONS R	RATIOS , 30		GAGE NO. # 10-0UTSIDE
LOAD	1 6	e 6.	S I CI E E E	₹ Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z	TAU
0	•	0	0	c	Đ
150	05=	•100	£ 6 3 4	-1264	201
300	05•	-150	**01•	•1813	500
05+	001-	-150	-1543	8791.	192
909	001-	-150	6657-	8667.	757
750	•≥00	•€00	-2857	-2857	0
00	052•	002-	K0+6-	250E-	261-
0501	-300	0520	121++	a3736	251-
1500	006.	-250	でになる	•373 6	2574
1360	950	-250	-+670	1066.	\$3E =
1350	• 350	1450	0490	7066-	- 385
1350	•350	-≥50	-+670	1016-	-365
1200	900	052-	181+	-3736	-192
1050	006		-+121	3736	-192
006	006-	002-	4556	-3187	*385
750	054-	-150	-32+5	のとから	500+
6 00	-150	001-	**T-1478	E b 51=	-145
+ 2 0	0.1	•100		-1264	261
300	06-	-100	-87-	-126+	192
150	06-	091	+14-	+16-	•
0	•	0	0	0	0
0	0	0	0	0	ē

Table 2C. (Continued).

GAGE NO. # 11-00181DE -192 267 142 -142 TAU THE WASHINGTON TO A COLUMN TO SHOW THE SIGMA STRAIN REDUCTION OF A TWO GAGE ROSETTE SIGNA 90 POISSONS RATIOS EP2 EP1 1040 Es 10,00

Table 2C. (Continued).

	•	STRAIN REDUCTION OF A TWO GAGE	A THO GAGE ROBETTE	ta é	
10.00		POISSONS RA	RATIOS . 30		GAGE NO. # 12-UUTSIDE
1040	1	3 6	O I GEA MAX	A TO I O	A A A A A A
c	c	0	0	c	0
) 2		05.	+14-	-714	\$
300	001•	-100	B2+1=	P5+1.	6
	15	001-	-1978	£ 1 2 d 3	-19E
004	5	057.	のかて心・	E+12.	0
750	002-	-150	-5645-	90£2•	257-
000	20	00≥•	-2857	-S851	9
1050	0520	-200	ROME =	●302€	261.
1200	2	0020	*O+E+	◆305₹	-142
1350	900	-200	-3456	20	\$500T \$
1360	• 300	200	• 3456	-318·	SOM S
0961	008*	-200	-3456	-3187	986
2021		0002	-3456	-3187	SBC
1050			C0#E=	-3052	-142
		002-	-2857	-2857	0
750	000	-150	-2645	8065·	261-
900	0.81	-100	87678	-1543	261-
05	100	- CO	-756	-824	267=
300	001	-50	+12-	+14	0
750	05-	0	****	-165	261-
0	0	0	0	0	•
0	0	0	0	0	o

Table 2C. (Continued).

.

	GAGE NO. # 13-0UTSIDE	MAM	0 n 0	№ D C	ា មា	រហៈ មា - ភេស - ភេស - ឧ ៖		NN & C C C C C C C C C C C C C C C C C C
		WIN WIN	2 + F 2 N 2 + T 1 1	## ## ## ## ## ## ## ## ## ## ## ## ##		318	# 0 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7 7	
STRAIN REDUCTION OF A TWO GAGE ROSETTE	RATIOS , 30	SIGHA MAX	9 F F W	M # # # # # # # # # # # # # # # # # # #	1 10 10 PC 1 2 2 CC 1 10 M M 1 1 1 1 1	35 M 10 M 1		00 + + + 0 0 5
ITRAIN REDUCTION OF	POISSONS RA	ල ස	1000		1	000		
-		EP1	5 0 1		0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	50	0 0 0 0 0 5 m 0 1 1 1	00000 0000 0111
	10.00	LOAD	1 S O 3 O O S	450 450 450	4000 1000 1000 1000	1 380 1 380	1200 1200 1000 1000	00000 M 0 W 0 W 0 K 4 F M M

Table 2C. (Continued).

		STRAIN REDUCTION OF	REDUCTION OF A TWO GAGE ROSETTE	le.i	
* 10.00		POISSONS RATIOS	710s , 30		GAGE NO. # 14-0UTSIDE
1040	£ P 1	& & & & & & & & & & & & & & & & & & &	SIGHA MAX	A N N N N N N N N N N N N N N N N N N N	D × € E
0	o	D	c	0	9
150	05•	001-	F C 40 1	-1254	R D C
300	001•	051.	£65T=	84610	251
05+	051-	051=	£ + 1 2 •	のナイル・	0
6 00	902•	-150	26424	9062-	261-
750	052-	• • • • •	-3407	#20E •	761
900	-250	002-	€ 0≠ 8 •	2206-	6840
1050	• 300	052•	12140	•3736	261.
1200	• 300	052-	12740	• 3736	-192
1350	980	364 B	0494-	1066.	\$ 30 C
1350	055 •	08.4	• 7527	at 15B	38 00 PP -
1350	-550	054-	-7527	-6758	500
1200	• \$00	00**	-6813	**09-	586-
7050	00+-	006-	5865*	5794=	500+
900	900	-200	-345¢	-9187	-365
750	-250	-150	542C.	-2473	-385
9 00	-200	- 150	-5645 -	8008	-142
450	• 100	-100	62+T-	-1454	0
300	-50	001-	5001	-1264	192
150	05•	09-	+74	*1.C.	•
0	0	0	0	0	0
•	0	0	0	0	0

Table 2C. (Continued).

	GAGE NO. B 1-INSIDE	NA MA	0	05	181	386	557	207	125	1121	1264	*9*1	1874	1074	1750	0051	1200	0001	200	364	157	**	R#T	· •
		AIGIS	c	• 100	2547=	25ナル!	*3324	4574	8615	\$00 9	### S = 1	*164.	57951		-4100	-8167	-6733	mmen.	EES+=	8+78-	0502-	-1214	-776	***
OF A TWO GAGE RUSETTE	RATIOE . 40	A X A X A X A X A X A X A X A X A X A X	0	••00	-1081	-1681	•#210	W + C - C -	2P56.	-3752	57**	3 €5 7 1	-5871	1285 w	-5600	-5167	医医医食	E888.	-3193	# T # 20 *	-1776	-1086	05+	•10
STRAIN REDUCTION O	POISSONS R	E P &	0	-1150	0552-		0019-	1650	0550	-11450	095210	DOR + I +	-18200	-16200	-17150	-15850	-12500	-10750	0028:	05+5=	09464	-1450	-1450	DS *
		EP 1		008•	-1250	-1750	-4200	-2700	0016-	0016-	001**	0554	0\$0\$•	9	2	75	07	35	30	-2400	35	8	05+-	0
	0	1040	0	150	300	0 5 +	004	750	900	80	2	35	0567	1350	1200	1050	00	750	9 00	05+	300	750	0	0

Table 2C. (Continued).

Table 2C. (Continued).

	BTRAIN REDUCTION OF A TWO GAGE ROSETTE	IO GAGE ROSETTE		
	acies as a solution	•		GAGE NO. # 3-INGIDE
EP 1	71 G.	O I GEA MAX	₹ ₹₩ ₹₩ ₹₩	TAN VA
0	6	o	0	6
3	0061-	-533	e 2 3 3	700
9	0552	-1010	+2+1-	202
55	0068-	1811	155	336
000	-5550	•1952	1882.	***
\$	-6600	*5 *5*	• 3610	E \$43
35	0066-	0162•	#2F #=	202
9	0566	00♠€=	0015.	850
•	C\$801 =	8+500	5785 .	T 00 T
50	00621*	98++-	• 17.4	1114
5	00151-	-5354	-B171	1641
-5150	-15100	-6324	-0171	1451
ş	-14800	-5176	0666-	1407
30	-13050	EBS+	-1033	1250
2	+11250	-3424	-6071	101
57	-4550	-331 4	9415-	*Tb
22	0066-	-2681	-4152	986
8	-5850	-2067	-3167	550
35	000*-	-1405	-2162	974
75	-2150	-767	-1167	902
-150	006.	-124	-171	122 231
0	0	0	0	0

Table 2C. (Continued).

:

		STRAIN REDUCTION OF	A TWO GAGE ROSETTE	140	
0.		POISSONS EA	RATIOS . 40		GAGE NO. 8 4-INSIDI
LOAD	E P 1	™	A N A M A M A M A M A M A M A M A M A M	BIGIA	TAU
0	0	0	0	0	6
7.00	-1000	006-	87.91	619ª	***
300	-1450	-1750	2921-	9021-	5° 10° 20° 10° 10° 10° 10° 10° 10° 10° 10° 10° 1
05+		0092	25BT B	1861.	46.
009	-3750	0058-	-2426	1862-	•36
150	004**	D55.4	- 306 7	6962	054
206	-5450	-5.250	9640	92826	A.5.8
S	-6550	-6100	• • • 881	25740	* A *
20	-7550	0074-	物ナケナー	61844	*4*
1350	-855 0	0008*	2554S	86 + 38	-24
3.5	-10400	4500	-126	\$059	b21-
1350	-10400	-4500	-6762	• • 505	-124
1200	0566	0564-	-6514	9464	900
0501	0049-	0069-	F 785.	8+45	•
400	-7750	-7250	-5071	PSP 4-	-71
750	-6600	-6100	SOE +-	-4162	-71
6 00	05+51	-5050	-3557	E++E*	S
+ 5 0		0066	-2814	-2666	**
300	-3150	-2700	+102-	-1886	**
150	-2050	-1500	-1262	-1108	56.
0	- 700	006-	056	962-	-67
0	09-	0	+21	07-	•

		STRAIN REDUCTION OF A TWO GAGE ROSETTE	F A TWO GAGE ROSETT	Ē.	
10.00		PUISSONS RI	RATIUS , 30		GAGE NO. S S-INSIDE
040	E P L	₩ 6.	A E D E E E E E E E E E E E E E E E E E	ATOLO MIR	TAM WAX
0	b	0	6	0	c
150	-50	0.6	+16-	374	0
300	-150	002-	B 06 % a	2692-	761
054	2	● 300	-3187	93454	386
P00	006-	•500	いナザナー	+3+9+	269
280	00**	-600	\$CE96	2166.	P 9/
00	06+	-100	- 7253	96160	462
1050	50	-750	- 746 V	0585	29€
7200	ş	008*	3845	サド ひじ T ®	677
1350	2	058-	56+01-	8+917=	522
056	009•	058•	9585	-11314	296
0981	•	099	-4346	-11314	295
1200	9	008-	-4891	-10764	264
0501	S	-750	-8516	-10055	76.9
900	80	-680	-100	1898-	264
750	00+-	085	-6801	-7363	. 673
6 00	0		-+780	+0-51	575
+ 50	S	006-	-2637	1646-	577
300		002-	8547-	-2527	348
150	LD)	091	-714	+14-	0
0	0	D	0	0	0
0	0	0	0	0	0

Table 2C. (Continued).

		STRAIN REDUCTION OF A	REDUCTION OF A TWO GAGE ROSETTE		
E= 10.00		POISSONS RATIOR	06. 30		GAGE NO. # 1-1NBIDE
1040	£P1	2 2 4	SICMA	SIGNA	TAU
c	s	æ	a	5	
) -) c	001•	62310	62 t [*	0
		056 =	1056-	04940	\$86 \$
		200	55 th 5 th	8+99 •	K-5
		059•	# 7 C C C	48489	468
25.0	005-		• B 1 3 2	0 * 10 1 =	1154
) C	0000	000	0456	-11858	1154
	0020	0501-	-11154	9+861=	9***
		1800	44671-	#585T#	1538
	000	0067	-13626	-17088	1541
1350	058	-1400	-13456	. 8181·	2115
LI)	038	-1400	-13956	-18187	2115
N	008-	-1300	-13077	-16423	1423
1050	-100	-1150	-11+8+	5+6+1-	1621
T	-600	-1000	0585-	-12967	1538
750	-500	040-	-8297	-10484	9467
600	00+=	-200	-6703	-4011	1377
06+	- 350	-550	-5654	9674	592
300	2	00+	**066	-5220	573
150	-100	-100	P2+1+	-1459	•
0	0	0	0	0	0
0	0	0	0	0	6

	ø	STRAIN REDUCTION OF A TWO GAGE ROSETTE	TWO GAGE ROSETTE		
E= 10.00		POISSONS RATIOS	08. 80		GAGE NO. R 7-INSIDE
LOAD	1	4 4	SET	BIGHA	TAR
c	c	G	6	6	5
2 2	2	001•	P5410	P3+1=	0
		0024	-2857	-2857	•
		008	9827	982+	Ð
2 6 4		OSE 2		DE 6 5	* 345
2 5) t	0.00	-8077	* 6 4 6 3	~ C (II)
	000	005*	7 + R & .	= 780s	500
			-1060*	-8681	カンチ・
			-12033	-10110	~35.
7 7 7	0001+	000	C + 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2 - 2	2	
1380	0001-	• 600	-12967	0696	
e e		004-	-12467	0494-	9691-
		000	-11866	19860	-1154
			04401-	-0132	-115
			-9176	-7253	2 5.
		00#	27972	+69-	F3C-
		0 to 10° 1	9444	-5+42	CC9-
		0064	-6385	5794-	-382 -385
		002	C0+81	-3022	261-
	-100	-100	- N+1-	P841-	6
		0	0	0	0
9 6	. 0	0	0	0	0

Table 2C. (Continued).

	GAGE NO. # 8-1481DE	TAC	•							8097			\$19\$E	9194-		2619-	和助于士士					-623		
ROSETTE		MIN	0	220	5623	022	114	011-	011	591	1999年	~ □ + ® • • • • • • • • • • • • • • • • • • •	4566	1546-	-2527	***	9 T +		-165	DEE	27.5	220	0	0
OF A TWU ISAGE	VS RATIOS . 30	VEDIO XVF	•	→R 5 *	~95~	330+	+745ª	8804	435C-	1576	P0211-	-13025	-13187	-13187	-11758	Sent	5568-	-6868		7068-	81+2=	****	0	0
STRAIN REDUCTION	POISSONS	8	0	05	0.5	100	007	002	050	006	100	05	0	•	100		006	092	150	150	700	0 10	0	0
		1 P J	•	•100	25	001	5	2	000	5	105	20	120	-1200	-1100	-1000		-700	0880	001	000 a	-100	•	0
	Es 10,00	1040	•	180	300		004	750		1050	1200	1350	1380	1380	1200	1050	006	750	00.	094	300	150	0	0

	GAGE NO. # 9-INSIDE	TAU	.	2 45	585	573	RL SF	9+67		5772	30 C 2	1731	1521	1791	1231	36ST	1124	797	388	142	745	0	0
		AIGE STE	0 :	2592*	9566	1693	-224	04868	-10824	-13187	59151-	~1 2802	-18802	-12802	+CETT-	-10110	-0247	***	0494		-1564	a	o
STRAIN REDUCTION OF A TWO GAGE ROSETTE	RATIOS .30	A X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X A Z X X X X	0 :	# \$50 PM & 6	-3187	0867	**85.		6466	25FB=	D+507=	T+66=	T+64=	Tembe	2767-	-7033	-8-5:	S+8+=	-3401	-2308	F-01	0	0
STRAIN REDUCTION O	POISSONS R	æ d ⊒	0	000	006-	09**	004-	-750	058*	0507-	-1-00	•1000	-1000	•1000	005-	008=	1650	005-	9320	002=	-100	0	0
		E P 1		051-	0	30	1	0	5	0	60	52	Ø	u	'n	9	S	0	S	S	1	6	0
	E# 10,00	COAD	0 ;	0 0 F	09.7	004	750	400	9	2	35	1350	7350	7500	1050	900	750	90 9	+ 60	300	180	0	0

Table 2C. (Continued).

		STRAIN REDUCTION OF	A TWO GAGE ROSETTE		
		POISSONS RATION	110= ,30		GAGE NO. B ID-INSIDE
E 10.00	EP1	€ 643	A X A Z	MIN	4AU HAX
			•	6	5
6	•	0;		6.0	257-
150	-100		ב ה ה	-1758	- 385
	- 200	0074		-3187	3 E P
05+	006.	002		15440	70 T
009	0563	300	16.71	*165	0 1
750	00 * 1		- T - 17 - 17 - 17 - 17 - 17 - 17 - 17 -	6476	0
005	008		C-500 P	- 1857	י פ
0501	-550			9825	0
1200	0995		5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	-10165	N 5 - 1 =
1350	05.0		58051	1546	2611
1350					
1		099-	308F=	1546-	W C
0567		554	-4266) (
1200			-7887	-282	
1050			-7143	E+14=	
005				10001	267
750	00+			0005*	o
009	-350			30CM	267-
	900				-142
000	002-	DST •			9770
150	001-				0
0	0	י פ	.		0
0	0	D	,	•	

Table 2C. (Continued).

		STRAIN REDUCTION OF A	A THO GAGE ROSETTE		
E# 10.00		POISSONS RATIOS	08. •		GAGE NO. # 11-INBIDE
LOAD	EP1	5 G 3	BIGER AAX AAX	A S I S I S I S I S I S I S I S I S I S	HAK
0	0	0	0	0	0
051	08.	05.0	-714	414-	c
006	•100	002	954T=	-2527	520
000	00%=	05 n	■3022	CO+E •	261
004	-300	058.	75++=	SERT	251
750	001		FC 85 8	****	257
005	-500	085	-7308	9769E	257
1080	0.550	004	520B-	*O+8*	257
1200	054-	-200	15+5	5E 96 P	25-7
1350	-200	• 750	-10165	B + 90 T =	245
1350	004	-750	1106	₩10880	6.79
1350	-600	086	1906-	-10220	877
1200	-600	-200	1068-	-9670	
1080	-550	000	-8187	9850	
900	-500	-650	9066-	-769E	利かべ
750	051		-6424	1211	
009	980	098	-5000	-5000	
05+	-250	000	1486-	-3871	
900	-150	-150	E +TR-	-2143	
750	-50	■ 50	+14-	-714	. 6
0	0	0	o	•	•
0	0	0	0	G	•

Table 2C. (Continued).

GAGE NO. B 12-INSIDE	BIGMA TAU Min max	0	0 576	•			257					-11424	0							_	474	
REDUCTION OF A TWO GAGE ROSETTE POISSONS HATIOS , 30	SER MAK	0	774	# 5 B O B	-3571	75448	-CBS:	#08C*	-8735	00001-	-11454	-11429	-22429	-10714	300 NT 2	C3863	5 N ± 3 1	-5000	12869	E-Ta-	+74-	
STRAIN REDUCTION O	593	0	0.5	•≥00	050	036-	0.9 *•	-550	044	• 100	0000	200	004-	054	064	098*	094	056-		-150	05=	1
	. •	0	•\$0	-150	-≥50	- 300	00**	-500	-100	- 100	000	006-		-750	59	-550	5	35	054	57	09:	
10,00	1040	G	150	300	450	004	750	00	1050	1 2 0 0	1350	7 350	1950	1200	1080	400	36 0	600	450	300	150	•

Table 2C. (Continued).

	STRAIN REDUCTION OF A TWO GAGE ROSETTE Poissons hatios .30	OF A TWO GAGE ROSETT HATIOS .30	w	GAGE NO, # 13-INSIDE
	æ ⊕ ₩)	A X X B X B X B X B X B X B X B X B X B	E E E E E E E E E E E E E E E E E E E	FA
	6	0	c	9
	•100	F2+1-	P541-	0
	002-	# 2002 a	-2857	0
	006.	3827	1871	c
000	00*	-145-	11650	0
	005-	8+76=	アナイルの	5
	004-	\$208 ·	L0+0-	745
	• 100	1066-	00.301	382
	008	06601-	6601T•	500
	006=	85677-	-12827	386
		~5EET•	509+1-	N 2 2
	-1050	-13352	-14505	623
		-11756	-12627	900 000
		-10165	F+801-	251
	009	16871	16871	0
	005	6476	E+14=	0
	004	+128+	+145-	0
		302±1	30N+-	•
	-150	-2642	800 N	261-
		+14-	+74-	•
0	6	0	0	•
		0	0	0

Table 2C. (Continued).

AC A		222	\$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$ \$			24 24 24 24 24 24 24 24 24 24 24 24 24 2
HO GAGE ROSETTE	• •	•		*777	••	2000 2000
STRAIN REDUCTION OF A THO GAGE ROSETTE		 	0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0030		0 0 0 0 0 0 0 0 1 1 1 1
	a	0 0 0 0 0 0 0 0	00000 0000 0000 00000 00000	00000		0 0 0 0 0 0 0 0 0 2 6 4 1 1 1 1
10,00		0 C C C C C C C C C C C C C C C C C C C	0000 4050 * 3 ~ F	1000 1300 1300 1300		

0 \. ' •3		POISSONS RATIOS	0.		GAGE NO. # 1-DUTSIDE
1040	1	£ 6.5	OF GERA	MIN	HAR
•	•	5	6	6	o
100	05~•	000	1680	P2+*	**
00 2	• 300	-1250	196"	254	136
300	00**	0527-	*55*	076-	243
001	05+	-250	649	-1157	484
\$00	•550	-2750	*186	****	216
004	• • • • •	-3250	121	-1671	178
700	• 750	0.586.	0601-	761-	m + +
8	005.	00***	-1267	F922-	003
00	-1050		£ * * 7 •	F-0.55.0-2	857
7000	-1100	0545	-1512	\$0 02	129
1100	-1200	-1050	•1724	0116-	853
1200	0567-	0054	1881-	-3325	726
1 300	-1850	0054	-2145	# 3E 3E	721
00+1	0002-	00+4	•₹36₹	\$056 B	122
1500	-2150	-8000	84528	#12+=	92
1600	006.4	-6700	-2752	185+=	*T6
1 200	00+2-	-4250	*240S	ルタロナー	-25
0087L*	0052-	-4750	#30#B	6115	1036
L 1800	0052-	00001-	- 3045	*5538	101
7 900	-2500	-10000	- 3045	-5238	101
1 200	00+10-	-4650	1867-	#\$0\$#	1036
0041	0562-	-4250	1887-	-+852	30,
1500	0062-	0068	9662-	コトラナー	7 **
00 • 1	-2200	-8500	-2667	F3++1	50 5
0061	-2100	0018-	F > 5 ? .	450+E	157
00~	000~	-2200	P1 +2 •	****	***
0011	0554	0924	->300	00 pm	035.0
0001	0051+		29120	606R	169
00	096.43	0000	94514		N 1
	00274		70010		77 G 1- 1- 1- 1-
					7 :
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90				2000	7 F
300	0521-	-2700	01110	32510	100
002	-1100	0002-	506.	-1166	
700	0.05	-1250	054	-226	m =
0 J**	-150	●750	*12·	186	:
_	05-	150	ĸ	6.2	62.

denotes strains at the beginning and conclusion of 24 hour long sustained loading at 1900 psi

denotes strains at the beginning and conclusion of 24 hour long sustained loading at 0 psi

Table 3C. (Continued).

	GAGE NO. R 2-OUTSIDE	TAU MAX	c	0	^	Ð	0	Э	٠.	5	5	⇒ 1		• " •		→ ;		7	÷ :	₽ (•	6 6	36.	■ 36	5 TU .	T & **	170	7 P	720	737	4 m	120	*T=	***	C -	4 5	=	*		~ `	•
		A I D I R	0	-30u	075-	006-	-1167	1500	-1776	0012•	00+2.	P1620	0106.	9325		2	F-10 - 1	9	5 C D S S		n =	•	8.17.	*159	*6405	- P 3 P S	-100°	25.745 			5 C C C C C C C C C C C C C C C C C C C	2404	P146.	- 32H6	0162.	4045	1812-	+141.	1621	50%	01
F A TWO GAGE RUSETTE	RATIOS . "G	PER X PE X PE	0	006.	-576	006.	-1167	1500	-1740	-2100	00+2-	## No. 1	700 T			**************************************	30 to 10 to	0 · · · · · · · · · · · · · · · · · · ·	905	アリティー		į,	P186 =	9863.	-6662	S0+4=	5019				数のカナモ	5014-	おかんがる	+166-	40B0*	*555*	2512*	9891	₽~?~.	7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	**
STRAIN REDUCTION OF	POISSONS R	243	0	0.6 *	006-	0361-	0541-	0522	0542-	0576-	0046-	040+	005.	0005-				0004					-10650	-10300	3585	4500						- 6050	15850	0064*	3564	9450	0086-	. v	567	0011-	>
		EP1	G	05.4	-650	-1350	-1750	-2250	-2700	- 3150	0045	051+	055+	0015-	0055-			0506					-10400	-10550	-10050	05451	0025		D (C C C C C C C C C C C C C C C C C C	007.61	2 2	16200	-5450		00+++	- 3800	•	-2500	0081-		0.5
	f 40	COAD	6	100	€00	DOF	90,	200	600	200	900	00	1000	1100	0021	0051	00 * 4	0051	0091	000			CORT	1700	0041	1500	1400	006				8008	200	009	500	00≠	300	200	007	۰ ۵	E

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	GAGE NO. 8 3-0UTSIDE	TAU	=	7.7	124	981	£~~	.a. :	7 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	m # ##	* Cy	***	F 72.4	202	6 t k	008	653	# · · · · · · · · · · · · · · · · · · ·	105	101	1021	1601	9801	405	m + e	00*		\ 10 f			: T & G		**************************************	4.4	B/W	1961	250	**	1,	77
•		A S I S I S I S I S I S I S I S I S I S	o	\$0f •	295-	-18-	# 20T*	300 m	9 + F 0	30071	1920	\$152°	56670	610f.	• 3275	-3533	0666-	***	50E **	***	1694-		98 + + 0	-428	4000	006.	m, .		0007.	*2767	6 45 20	• 2 3 1 O	-2026	P181.	-1562	\$061•	-467	*~ 9	-205	6 \$ 7
OF A TWO GAGE RUSETTE	A71040	A X A X A X A X A X A X A X A X A X A X	o	-162	\$0 8 -		765-	\$10.1		0771		400 FT +	8 K 5 T 8	8+41+	-1740	EE-T-	-≥02€	P122.	-2362	30 P	P-252-	B4.57.0	****	*162·	0572-	-5100	P. G. C.	8 1 0 1 5 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1			6561-	*251.	0.01.	おナザー	\$08 +	-662	ルゴナー	9820	~ 1	65
BIRAIN REDUCTION O	POISSONS RATIO	243	ø	004•	0011-	-1600	0002-	0052	0006			005	05+5=	00-5-	00+4-	0054	-7400	0064-	00 + 6 -	000	0515	0.81.8-0	0088-	00+6+	0008	0994	2500				0005-	- 4 N O	001**	- 3600	-3100	-2400	0.661-	0521-		00F
-		EP1	•	-100	002-	300	00**	005	004				-1050	-1100	-1200	0061-	-1,00	-1500	-1400	-1700	-1650	0.1650	-1550	-1500	-1400	-1350	0671				0.58	-750	•650	055-	0.67	056*	-200	•100	05	6
	640	1040	Ð	0	\$00	0	0	0	•	9 9) C			0	2	•	20	3	2	0		0087	1 200	1600	1500	7400	006	0021	1000	300	00	200	009	200	00+	900	00€	100	6	•

Table 3C. (Continued).

STRAIN REDUCTION OF A TWO GAGE HOSETTE

			STRAIN REDUCTION OF A TH	TWO GAGE ROSETTE		
:	0 ••		POISSONS RATIOS	0,.		GAGE NO. # **OUTSIDE
10	90	EPI	₩	ON CHA	SIGRA	E T T T T T T T T T T T T T T T T T T T
	0	0	•	•	0	0
→	700	064-	09**	- 300	300	•
ni n	00	050-	-650	-667	-567	•
m	90	-1350	-1250	188-	258=	***
*	00	-1750	-1700	•1157	£ 4 T T =	
uri	00	-2250	-2150	1961	-1.56	+7+
•	00	0542-	• 2 b 0 0	-1757	-1743	••
•	20	91.80	0016-	-2040	-2076	r•
•	90	-3600	03480	-2371	-2324	12.
5 *	00	-4050	0066	-2671	-2629	12-
9	90	05++•	006 %*	1642-	\$ 582-	72.
77	90	-\$000	008**	* 3245	9636•	₹@ 8
~	90	05+50	-5450	* 3545	●3536	ぎたり
~	9	-5450	-5750	P 3424	-3871	5 h •
=	90	-6400	0024-	ケベルナー	16710	F0.5
51	90	00-9-	-6450	2554		• 3b
2	00	-7500	-7200	***	-4857	m + =
2	90	0505-	-1750	-5310	-5554	物ナー
31	00	-8550	0028-	-56 33	-5533	05.
=	90	-10400	10000	25490	-6781	# TO
	3					10
				W678		
.						
	90			7 L F D 1	1654	
n 31					8000	4 5 6
•				1355		
7	904	-8150	0346-	-5338	3615	17.0
11	00	-7650	- 7200	*105*	9000	- A
7	000	-7050	-6700	**633	66540	05•
•	000	-6500	0014-	- +	E + T + 0	-4.7
_	100	•6000	0595	-3433	*3833	05-
_	004	-5+50	-5150	- 15 7h	0646.	m + B
•	000	058+-	-455G	•3126	0.000	F * •
	000	900	000*-	-2810	42620	₩ + •
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-	900	-3100	0582	-4014	8 + F T •	*3b
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-	001	0541-	0097	Bt 110	-1045	
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•\$0	05•	+14-	*14.	0
04•	05-	77.	174.	9
•100	•100	P 3 4 2 4	6247-	0
007-	•100	F2+T-	P 2 4 2 4	•
-150	150	F + 1 2 -	6 4 7 2 -	
002-	051•	2592.	80E 2•	261-
-200	-150	5692·	#0£2•	251-
• 200	002	-2857	K-5-11-11-11-11-11-11-11-11-11-11-11-11-1	9
-200	002=	-2H24	C5820	0
-250	002•	6046.	◆3022	2670
900	050	18140	- 3736	2010
900	052	12140	•3736	761-
• 300	052-	12140	-3736	2510
-350	006.	ST 8 + 1	15440	2511
056-	906 •	56 8 **	15++•	-
00 **	056.4	355	-5165	-
00*	056-	7 * S * *	•5165	_
00*	- 350	B + 55 S =	99750	2610
00	950	• * * * * * * * * * * * * * * * * * * *	5975-	-
5	97.	5 5 1	31	
136.0				
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006.) 12 mg		7247.	3 T T
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300		45000	6016	
052-		6 P	E 2 2 2 3	
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				2 A A A A A A A A A A A A A A A A A A A
051-	DST •	E + 1 2 4	2000	•
051-	001	50.5°	66510	200
100	001-	52410	-1429	
•100	0.5	-1254	P 28 8	5 T 3
05-	0	B # 15 1	-165	2510
0	0	0	0	0
\$ 0	100	874	1204	-192
05.	0	8 3 3 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	9410	200

Table 3C. (Continued).

10.00		POISSONS RATIOS	96.		GAGE NO. B - DUTSIDE
LOAD	£ P.1	# 6 m	SICIA RAX	SIGNA	TAU MAX
•					
	5 5	9 (0 ;	0	c
000			F (1	*921*	10 F
			**************************************	*92T•	
		2674		8051.	•
905		9410	6657	8451	2 F T
004			B + 1 4 5	64143	0
900		907	7.5000 F	- S-	0
			1000		0 :
000			7/8/1	7285	- :
1000	• 300	052		1/66	•
1100	.300	0064	1007		9 T
1200	980				
1 300	950	CORP	5 C S		
00+1	100			16.4	
1500	1450	0.50	; ;	1100	7 :
1600	05+	200	9 6		n :
1,700	CUST				©
1800					.R (1 20 m ·
200					-
				***	267 a
7800	-\$00	0510	92478		257.
1 700	05**	00+	2	-5874	25.0
1600	35.20	001		_	2510
1500	05.**	00**	-6-64	8	2610
1+00	00+-	056•	B + 55 5 =	2	201-
1 300	-350	-350	• \$000	-5000	0
1200	056-	- 350	-5000	8	0
1100	-300	-350	T5++*	5F # # #	261
1000	-250	900	43736	727 **	657
007	052+	-650	-3571	145F-	Đ
008	-450	-250	-1571	-367I	c
200		9550	●302€	60 hF =	257
P00	051-	052-	所た ナジョ	2 1 2 E =	385
\$00	-100	-500	-1758	-2521	345
00+	001-	-150	£651=	8651.	261
300	05-	-150	**01-	-1813	586
200	05.	-100	50E1	-1254	261
100	Þ	-20	-165	サナル・	261
5	007	0.5	***	8 7.0	142
5	5.0	c	B + S	165	267

		STRAIN HEDUCTION OF A	TWO GAGE MUSELL	Į.	
10.00		POISSONS KATIU	0. , 30		GAGE NO. # 7-0UTSIDE
LOAD	1 63	5 P 2	GRANA	SIGHA	FAC
a	o	G	c	•	e
100	0.6	250	*16-	*14.	
900€	-100	05•	-1264	64.80	2510
300	-150	•100	-1478	61843	2510
00,	-150	-100	-1478	1543	2570
200	00≥•	-150	-2642-	#0£5*	~ T ~ =
009	052-	002.	* D * E *	₹20€•	247-
200	052	0520	-3571	1256-	
00	900	052	-4151	9846	2011
2	004	050	-4121	9886	2511
8	000	052	121	-3736	*61*
2	0 II		-5000	0005	0 (
		056	0005+	0005	o :
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) C) 6
9					7 7 T
20		005	07143	67149	
8	0.55	00%	->-	# 7 W C B	2511
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0081		Q () () () () () () () () () (-75-27	-6758	538 ·
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700	002•	051=	26924	80824	2510
009	002	•100	F-28-1	.1758	2500
200	057-	-50	-1813	**07*	
00 +	-100	05-	-1264	-87	2570
300	-100	۵	PP01-	056.	516-
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001	O ()	05	5 3 A .	386	-385
י פ		150	***	1913	
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•	STRAIN RECUCTION OF POISSONS RAI	OF A TWO GAGE ROSETTE	w	3018100 E = 000 E3989
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	0 (9 4	0
	001	7961	3	19C 1
	001	P161.	*04	
	007	F1011	9 9	RI 74
	001			*******
	001	-186	0 *	****
	001	8 T # 2 8	# C C	### 18
	051			
	2002	-2637	F021	6 2 5 1 B
	200	-2637	1204	-1423
	002	6876.	**01	\$112°
		# 506 # 1666		
	902 908	46.66	- T-C	25.50
	200	-+286	214	-2500
	0 %	1695	1424	2500
	550	14560	F 89 4	00\$2•
	200	-3187	1007	-2115
	0 4	C 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	**01	
				75/70
	500	80 SO CO 1	1324	
	500	8802-	133	-1731
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	_	STRAIN REDUCTION OF	A TWO GAGE ROSETT		
E. 10.00		POISSONS RA	RATIO 30		GAGE NO 4-DUTBIDE
1040	12	8 .	SIGNA MAX	BICKA	- AF
6	•	0	•	•	•
001	05.	0.4.	-174	*16.	0
902	0\$-	001-	P. 0 -	*121-	767
000	001-	097	-154	2471	**************************************
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0047	-380	0040		11950	
1 700	-350	0.6	000		500
7 800	001-	084	-5874	***	261
7800	00+-	054-	9685	****	192
				41616	
200					• •
				9413	3 5 6
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7 300	00m •	086-	15++=	\$60 to	
7200	• 300	900	9827	3027	, 1 0
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007	-250	-200	*D#E	-3054	257.
0	00~-	051-	2642	1067-	2570
700	00~-	051-	2592.	9062	2571
000	002-	001-	~~~	1327-	388.
005	001•	-100	P2+1-	P41.	0
00 >	007•	50 m ·	*927*	***	267-
000	001-	09:	-1264	•624	251-
002 2002	0.5	0	***	591-	2511
007	0	0 (165	7 2 2	251.
0	0	051	**01	181	****
5	D		547	**	** T •

		GASE NO. = 10-001810E	DX Z	5	•	261	9	6 (- 1	25 T	•			2510	500.		.165	500-	CC#-	-64V	• • • • • • • • • • • • • • • • • • • •		P.C.5.0	2200							200	60.0	2510	2510	251.	986+		2510	2570	*385
			41514 417	8	•17.	.151.	アルナイ・	67454	1217	65570				22050	C 0 7 0 0	43187	69760	106.	440+	9.90 + T	, so + •	1101	2500		2000	76 9 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8		* C C C C C C C C C C C C C C C C C C C			196			PC 80						
(Continued).	F A TWO GAGE ROSETTE	08. #011	BIGHA	•	*14.		P211-	500TO	b267.								9556-	0.4.	•5₹50	0225	0285•	-5220	\$05 to	1000 to	5051	15 km 6						*****	\$48.48 \$48.48	21011	***	77010		991	-	900
Table 3C.	STRAIN REDUCTION OF	POISSONS RATIOS	243	•	0.50	-100	001•	001-	001.	007		9614				0020	002+	052-	-250	062-	-250	047-	-200	004	D 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	0570	9674	967.								,	9		001	0.5
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Table 3C. (Continued).

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OF A TWO GAGE	RATIOS , 30	A Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z Z	9 1	3461		****	6247.	(A)		C 400.0	1062-	2206.	220E •	220Fe	-3050	-3736	106.	990+1	1066-	106.	48184	4976	\$6.00°	2206.	~ S & & .	9062-	M () () () () () () () () () (# S & C & C & C & C & C & C & C & C & C &	*14	114	\$7 K.	***	9 :		
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(Continued)	A 780	MATIOE . 30	STORES NAM	6	**01-	-1543	E+720	6591-	#064°	1502 a	7 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0			* 15 F F F	1214	1014	0694	-5520	2 m 2 m	28 E S =	• 5 385	-5385	0225	014	00.971			0 f d f		2042	****			かれまべる	52510	P	5410	4	
	STRAIN REDUCTION OF	POISSONS RATIOS	E P 2	6	051-	051-	097.	-160	002-	002-				0024	-250	-250	-250	052-	•300	900	900	006.	052	0520		202						057.	100	-100	100	001•	05-) t
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Table 3C. (Continued).

STRAIN REDUCTION OF A TWO GAGE ROSETTE

Es 10,00		POISSONS RA	RATIOS .30		GAGE NO. # 13-0UTSIDE
LOAD	E 9 1	243	BIGEA	RICE	Z × C
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1700	00**	6€00	9505*	-3816	#3P.
1600	80**	002-	-5056	9196.	-26-
1500	00+-	081-	0000	-2467	2.5
00+1	-350	-150	7+8+-	2082·	-16
7 300	• 300	091-	T548-	• 2637	657
1200	-300	-150	756E-	-6637	-577
1100	-300	-150	1668	-2637	-623
1000	-250	-100	10m-	62510	-213
00	002-	-100		-1758	200 ·
00	002-	001-	~~~	8561-	\$80.
200	002-	0 6	6462	120	-577
004	051-	0.6	E181*	**01*	\$8C -
\$00	-150	05.	-1813	**07*	-385
00+	-150	05-	#TD1.	**07*	# DC 1
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Table 3C. (Continued).

		Table 30	(Continued).		
		STRAIN REDUCTION OF	A TWO GAGE ROSETTE	•••	
E= 10.00		POISSONS RA	RATIOS .30		GAGE NO. # 14-0UTSIDE
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900	041-	-150	-1543	-143	251
00+	-100	051-	E 1 5 4 3	-1478	N
200	051-	002-	80 C C -	26.92	10 (0)
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0057	056	D (1)			
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,	004	010			~ A A A
	000	D (1)	0.00		
	D	0620			
				,	1
	09+-	-250	-5764	1631-	646
•	00*	-250	-5450	## D# :	229
	00+-	0.8	0225	440+	~
				1016	
•		DDN:			7. GD
m (DOR.		9		
0021	00A+		77.70		#6# I
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			6666	256.10	
200				1961	
			8 10 10	**0"*	
005	901+	05-	-1264		2570
00+	001	0.5	-1264	7.	261-
300	0.5	0	F+5-	597.	2571
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001	0	0.	9		
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Table 3C. (Continued).

GAGE NO. = 1-INSIDE	MARC		
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1100 , 40	BIGE		
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Table 3C. (Continued).

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	GAGE NU. # 2-INBIDE	TAC	5	**	***	720	F 1		# S		9.5	760	• 7.4	•	F -	001-	\$ CT	521 s	.s (732,	1230		100 T	-207	002	J	B 1 0			1		*110	001•	001-	-24	***	0.5	⊅ ¶■
<u>ئمة</u>		BIGIE	6	07.0	-125	-2-1-	\$0\$7·	9057				-362	5000	••••	05051	-5500	~ ~ ~ ~			000.0		****	-200-	9610	0110-	-1767	1964	900C-				C 5 6 7 9	45080	V4560	-2733	e 2 3 3 8	1857-	00	₹5.
OF A THO GAGE ROSETTE	RATIOS . 40	S X X X X X X X X X X X X X X X X X X X	9	121-	196.	-1111	2957-	2957				100 B	2000	F78+	-527¢	0045-	1011-	ST	25120		7.65	-46.2	1625	0.00	ナルカロ・	*	25/10	NJM .					19569	63469	MINE OF B	S5-22-	+171+	•1000	130
STRAIN REDUCTION	PO1850NS	243	•	-100	•1100	0691-	0022-	0082-					900	004-	09+4-	0508-	-1700	00+5-	05007	05407=	-13680	2561	00021-	00621-	05411-	-11280	•10200	05101-					00+4-	004.00	0058	-30\$0	-2550	-1450	05•
		143	•	-180	-1200	0001-	00+2-	- 3000	0046					00+	0019-	0949-	005	-19300	-11000	-11700				05467-	•13200	-18650	00021-	00417-					0024			0096-	-2700	-1600	051•
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			1223		
> ~	3450		1 3 4 4 A	65450	7 P
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1800	-7500	00,12•	-2743	-11857	F505
1 200	- 7250	•=1200	06 \$ 2 •	96411-	E & & T
1600	0014:	00602-	-7152	18501-	****
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0021	0925		C955	4111	7650
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200	0546-	00611-	001+	00+9*	1150
600	0516-	-10750	• 3640	*5776	E + 0 T
200	-3050	-4550	-3271	-	6·2.6
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Table 3C. (Continued).

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		41018 41018	•	001	-754	2511-	1833	07670	1820	WW 750		5168.	486	996+	\$075	-5131	7000	0557	9000			6140	215	-878	946 R•	-8043	F + 4 F =	~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~ ~	56240	P285*	-5 386	076.	4.36.7	-3833	+366-	5082-		•		7
OF A TWO GAGE ROBETTE	RATIOS . 40	ATOTA MAN	6	00+	704-	-1181	-1633	40711	300 V				7-167-B	805 t 1	2985-	974.9		9679	01010	~ P = 0		3000		1805-	C5.48.	6-5-10-10-10-10-10-10-10-10-10-10-10-10-10-	NSF. 4	8176		-4071	*5614	*215*	14547	M#D+	0156-	2952°	_	1891		7
STRAIN REDUCTION	8008810A	E &	G	004	-1100	-1700	-6500	0082-	00 # 8 -					9000	0534	0068-		004	~ •		-			00221-	052210	-11750	06477		0025*	0058-	-7860	-7150	-6350	0555	0000	C\$0+-	0016	-6200	0911:	067
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0021	007	<u>5</u> 11	-13681	-12404	N.2 F
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Table 3C. (Continued).

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ل د.		NIN	9 :	**************************************	のかのでき	508+	**************************************	857K.	11070	•1115	814210	99851-	-14560	-15110	* L 6 3 7 4	-17802	-18216	084bT-	**013*	4 + 22 2 7 4	4 8 8 8	-17582	-16314	06841-	-13077	-11813	-10385	1215	75878 17171		1010	9 4		EE02	5 5 5 6 7 F	0.95	3458	65952
F A TWO GAGE ROSETTE	RATIOS .30	SE S	9	7.21	Mr. #1.0	• 3352	## · · · · · · · · · · · · · · · · · ·			3700 E		-11154	-11868	£6021+	21521-	7 AE 47 =	55051	*E#51*	11511	-16154	• 16154	972570	96647*	-12947	-11423	**011-	97351	3675				******		011) # ## # ##	9.99	1530	8616
BIRAIN REDUCTION OF	POISSONS RA	No.	0	0010	2500	-350	005.					-1050	-1100	-1150	-1250	0487•	00+1+	• 1500	-1500	•1•00	• 1 4 0 0	-1300	-1500	-1100	0.65	058	054	D ()	2 C C C C C C C C C C C C C C C C C C C				pot	000		000	DOM	0562
		4 d d	0.4			002	900	057 =				- 100	054	-750	008•	004		~	-1050	-	1050	0001-	045	0.881	000-	-750	054								•) 3	. S	0.5
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STRAIN REDUCTION OF A TWO GAGE ROSETTE

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Table 3C. Continued).

GAGE NO. 8 8-INSIDE	SIGNA TAU Min		SANA CANA CANA CANA CANA CANA CANA CANA		######################################
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1.4		••									•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•	•											
O GAGE HOBETTE	90	SIGHA MAX	c	•116	50310	176	2643	197		45-40	*2033	-7418	-845	5051	-10.38S	-11424	-12143	• 1 3022	*1335 2	-14231	-13736	-13736	-13082	-12527	-1164	-10764	-10055	7+66.	*854)	* 28 S	*6538	+285+	\$6.45°	400	-3352	1062-	\$12	624T-	***	7 5 -	# * * * * * * * * * * * * * * * * * * *
OF A TWO	HAT10																																								
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•		E .	G) C						096	000		005	-550	-600	-650	-700	-750	- 750	008-	000-	80	034	-750	• >00	0.6.4	004-	-550	-500	05+	00+	098	- 350	052	002-	051-	051-	001•	05.	0 10	091•
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Table 3C. (Continued).

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500	-100	001•	F2+1+	- 7 + S	5
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				1000	2 7 7
0021			1400		•
0061	002-	-700	00001=		: a
00 > 1	056=	056.9	_	*1001	•
1500	008-	-800	P5+11-	P5+11-	• •
1600	044-	058-	-12143	-12143	0
1700	005-	065	-13022	40467	251
1800	056	0001-	-13736	12141-	192
1800	006-	9	-12857	-14851	0
0081	006-	006-	-12857	-12857	0
7200	058-	008-	-11478	£ 6770	~
7600	0081	.750	-11564	10801	~
1500	054-	-700	5 ± 50 T =	-10165	-
00+1	002-	09.9	10 mm	1546	→
006.1		0031	1216	95.430	-
0021			10 m	2008	~
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200		000		\$144a	
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D	כ	0,510			225

Table 3C. (Continued).

BIRAIN REDUCTION OF A TWO GAGE ROSETTE

£= 10.00		POLICEONS RATIOS	06.		GAGE NO. # 11-1NBIDE
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300	-100	001-	62 h 7 e	b2 h 7 •	5
00+	002-	051-	2692	-2308	251-
800	586	002-	-3407	220E •	***
9 9	006.	000	12144	-3736	
007	056	006.			70 T
			>1050		D 6
			45 JP (7 6 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7 6 7	9
			1618	72.40	
,			10 mm mm m m m m m m m m m m m m m m m m	-5450	
•		002	**501*	-1016	25.0
g,	054-	000	-10874	-11364	1 en
•			-12143	-12143	
•	05.00	000	-1-308	-12692	250
•	00-	056-	-13022	10187	451
•	006-	000-	-12857	-15821	0
0001	006-	006.	-12057	-12859	
7 200	004•	0.00	-11543	-11478	25-7
0017	094		PC 801	19211-	257
0097	900	-760	-10165	B+50T=	2 7 7
00+1	094	004	1500		2 F.T
0061	D 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0		-8736	1216	
	046.		22080	40 × 8 •	
D071			3067-	25960	0: 1 -1
					W # G
50 6			1000	18131	\$ 6 m
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0	05	0	\$1°	174	0
5	20	0		165	201

		Table 3C.	Table 3C. (Continued).		
	•	STRAIN REDUCTION OF A TWO GAGE	F A TWO GAGE ROSETTE	ķi	
10.00		POISSONS RATIOS	ATTO8 . 30		GAGE NO. & 12-INSIDE
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1100	- 100	004	02.4	1060	
1 200	000	0.51	5000	1516	2514
1 400	05	• 700	D+507-	•10165	251-
00*7	000	084	-11564	-10374	251.
0057			-12143	-12143	5
009			-12857 -	-12887	6
9921			10861	11321	0
	-	0001+	30 to 10 to	982 × 7 •	S ,
0087			12141-	-11736	
000	0001-	045	72777	-13736	~~.
007			NO+M1=	22061-	20 T
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			# - R		> 6
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0011	• • • • • • • • • • • • • • • • • • • •	055	-8407	220g-	~~~
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		05+1	96739	FF91	~ 1.
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ı	1)			•

Table 3C. (Continued).
STRAIN REDUCTION OF A TWO GAGE ROSETTE

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GAGE NO.# 13+1N8IDE	GEGER TAU	0 0			1					***		440	156	22	**************************************	7 *				P-01	0220													
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Table 3C. (Continued).

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APPENDIX D

DATA FROM FINITE ELEMENT STRESS ANALYSIS
OF 66-INCH OD X 58-INCH ID MODEL 2000 NEMO HULL ASSEMBLY

APPENDIX D. DATA FROM FINITE ELEMENT STRESS ANALYSIS OF 66-INCH OD X 58-INCH ID MODEL 2000 NEMO HULL ASSEMBLY

Although the results of the finite element stress analysis for 66-inch OD × 58-inch ID Model 2000 Nemo Hull under 900 psi hydrostatic loading have been summarized in the text of the report (Figures 4 to 8) it was considered desirable to publish this data for the benefit of other plastic hull investigators. The data details the predicted stress and strains for both the top hatch (Figures 1D) and bottom penetration plate (Figure 2D). To correlate the stresses and strains shown on Figures 1D and 2D with physical locations on the Model 2000 Nemo Hull one has only to locate the corresponding node numbers on finite element meshes for top hatch (Figure 5) and bottom penetration plate (Figure 6).

Since the finite element stress analysis was based on the assumption that the stress-strain relationship of acrylic plastic is linear under short term loading in the 0-10,000 psi stress range, the calculated values for 900 psi hydrostatic loading (Figure 1D and 2D) can be extrapolated with reasonable confidence to 1350 psi hydrostatic loading representing the 3000 foot operational depth of the Model 2000 Nemo Hull. For hydrostatic loadings in excess of 1500 psi the extrapolation of values calculated for 900 psi is not recommended as the stress-strain relationship for acrylic plastic becomes non-linear at the stress values encountered in this loading range.

Strain							
20.50b -5.4876c-03			,	Longitudinal	Longitudinal	Ноор	Moop
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20, 10, 10, 10, 10, 10, 10, 10, 10, 10, 1		21.7132	21.714	-b.2274E-03	-3771.071	-3420.7828	-5.53927E-03
22, 1375 22, 1375 24, 1375 25, 1375 26, 1375 27, 13		21.4203	51.42n3	-6.5743E-03	100 200	-3440.6879	-5.23011E-0
21, 4975 21, 4976 22, 1747 24,		22.1774	27.1676	E0-34/10-4-	-4237.637	-3462.9044	-4 60806F-0
22.6727		14.2280	7500,10	77676-0	-3573,897	-3470.0490	-5.873116-03
22,1720		20.4600	40 4 5 1 2	-6.24676-03	-1774,153	-3404.9458	-5.49853E-(
7-3-4-0337-0818E-034-44-54342.0766 7-3-4-0337-0818E-034-44-064332.0766 7-3-4-0337-0818E-033-7-8-643-7-8-8-643-7-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8-8-		21.1419	22.4720	-h. 5653E-03	P12.E1:14.	-3411.7487	-5.15843E-(
71,887 72,887 71,887 72		21.4234	6600.65	-7.1681E-03	*E4.***	-3427.5842	-4.85680E-
71 REST		55.5054	20.1347	20 - 20 TEC - 00 TEC	rec. russi	-3392.0766	-4.5/65/2
75,4814		19,0252	71.005	E0-16841-9-	190 C 190 (-3505.584	-5.82443E-
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22. 16.00 - 1.00			24. RhB2	-7.2487E-03	C+5.8/2+-	-3301.9317	-4.52102E
74,1667		20.26.26	4544.24	-7.62146-03	-61.86.11	-3235.4173	-4.18648E
74.4541 -6.7844E-73 -348.48 -35. 1203.0848 74.5547 -7.7701E-0.3 -4249.554 -1200.5057 75.7554 -7.7701E-0.3 -4249.554 -1200.5057 75.7554 -7.7701E-0.3 -4247.415 -1300.5057 75.7554 -7.7501E-0.3 -4474.415 -1300.3057 75.7032 -7.701E-0.3 -4474.415 -1200.3057 75.7032 -7.701E-0.3 -4474.415 -1200.3057 75.7032 -7.701E-0.3 -4474.415 -1200.3057 75.7032 -7.701E-0.3 -4474.417 75.7032 -7.701E-0.3 -4477.417 75.7032 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7032 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7032 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.7012 -7.7012 -7.7012 -7.7012 -7.7017.7017 75.7012 -7.		17.45.6	\$19T.16	-4.6143E-03	-3411.604	-3460.3127	-5.57817E
74,7577 -7,7344E-03 -4,849.154 1200.5057 74,7571 -7,1344E-03 -4,949.154 1200.5057 74,7571 -7,1344E-03 -4,933.244 -2092.717 24,7451 -7,14612E-03 -4,933.244 -2092.717 24,7497 -7,1614E-03 -4,935.241 -2206.57 25,7497 -7,1614E-03 -4,935.241 -2206.57 25,7497 -7,1614E-03 -4,935.241 -2206.57 25,7497 -7,1614E-03 -4,935.241 -2206.57 25,7498 -7,5484E-03 -4,934.46 25,7484 -7,5484E-03 -4,934.46 25,7484 -7,5484E-03 -4,934.46 25,7484 -7,5484E-03 -4,934.46 25,7487 -8,1372E-03 -4,934.46 25,7487 -9,1487 -9,1484.46 25,7487 -9,1487 -9,1484.46 25,7487 -9,1487 -9,1484.46 25,7487 -9,1487 -9,1484.46 25,7487 -9,1487 -9,1484.46 25,7487 -9,1487 -9,1486.03 -4,134.46 25,7487 -9,1487 -9,1486.03 -4,134.46 25,7487 -9,1487 -9,1486.03 -4,134.46 25,7487 -9,1487 -9,1486.03 -4,134.46 25,7487 -9,1487 -9,1486.03 -4,134.46 25,7487 -9,1487 -9,1486.03 -4,134.46 27,1139 -9,1486.03 -4,134.46 27,1139 -9,1486.03 -4,134.46 27,1137 -9,1486.03 -4,134.46 27,1137 -9,1486.03 -4,134.46 27,1137 -9,1486.03 -4,148.14 27,1137 -9,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 -4,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.03 28,1486.		5450.81	1930.64	-6.7894E-13	-3448.435	-3293.8848	-5.08613E
74,4553		18.656 3	24.7527	-7.0700F-03	-4111.619	-3242.9213	-4.65891E
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25, 481		18.9260	27,0320	-7.5RB1F-03	-+356.546	-2966.5721	-3.57166E
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		14.5434	\$7.3714	-A. 55.74F-03	4E+ BUS#	-2615.2823	אווסל כ-

UPPER HATCH NODES 1-54

Figure 1D. Predicted stress and strains for the top hatch.

																												!	COPER INTILI		27-10																							
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		2004.3193	- 2736 9161	-2525, 6225	-2300.5618	-2017.1828	-2971.6538	-2680. 1239	-2177 7783	1929 2457	-2947, 3187	-2599.5985	-2385.1660	-5060.5552	-1725.8002	2575 6930	-2650 1013	-2425.6255	-2345.3813	-2106.8106	-2005. 1022	-1769.5195	-1627.6608	-306/.3956	-2563.9970	-2414.2018	-2240.6718	-2071.4554	-1897.4569	-1720.7316	11203 2401	-8859 9858	-6888.6165	-5979.6645	-5366.5999	4298 7406	-3748.6515	-3289.7120	-2762.7734	12066 3613	-11782 8561	-10594.6768	-9373.6264	-8242.5875	-/195.7768	-6239.9361	-4186.1647	-2839.3518	-1111.2789	-13949.7961	-11724.3149	-10694.9814	1973.814/	-8109.7783
Low CITUDINAL	144 - 5 × 15 × 16	-5157.hah	* T * T B B * -	-41,14.544	-4229. AR2	-3747.HRV	-5243.186	*******	136.1.540	C#C*116E=	-5746.H39	5411.072	-4823.434	-4145.U47	PC4.P45F=		-5430.744	-5186.463	-4674.155	-++47.508	-4196.425	-3787.640	225 -110 11	2007	*10,6345	694.01145-	-4Kil.126	*14.55***	-4150.475	O# 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	461.1156	*5555*	+10.9Pe.	-5554° #35	127.661	850 6195	-405.544	-3654.703	-2548.735	574.681	-5018.653	-5+48.123	-5315.120	**C . 885 **	707 2071	-3727,474	501.466-	-1788.544	76.116	142.8PP.	-45.80.733	*5806.56*	C44 #50##	*******
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H	24.1373	25. H 34.2	24.73n2	57. K. 21. 2	58.5125	24.4032		22.05.25	28.7529	29.8402	24.2824	27.1842	5560°86	8100.62	64°408°4	24.4322	27.1887	27.8451	28.3016	24.7581	24.2146	014.6710	31. 1675	27.1214	77.5811	BUTU Be	28.5mm	1098.85	# P 1 9 2 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	30 3309	24.6043	25.84:7	27.3034	27,7642	8 - 8 4 · 8 6	24.1545	24.6173	30°UL	30.5428	24.5425	27.0147	27.4826	*B**	2	20 345 B	24.8116	30.224	10.7432	41.3487	26.6231	27,2325	20.77	48.4.46.	24.1073
ď	15.4424	13.1657	13.6147	14.0737	14.5277	14.4817	12.7161	13.5845	14.0279	10,0662	12.2559	12.6785	13.1012	13.5638	11.464	12.0378	12.2418	12.458	12.6498	12.8539	13.0574	13,4614		11.6052	11.6019	11.9985	12,1951	12,3920	2000	15.9851	10.0841	ID. SRCA	11,1701	5556.11	1862-11	11.9274	12.1168	12,3061	16,1454	10.3766	10,5402	P1-6.01	10,4034	11.0854	EE **	11.6305	11.8123	U+66.11	12.2302	9.88.6	10,1141	2000	10.6364	10.8105
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Figure 1D. (Continued).

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4001																														-1620.1791	-17188.6/16	-9750.3437	-8357.3207	-6949.5901	-2000-952/	-3013.8203	-1680.1903	-13080.6782	-10026.8617	-8825.0017	-7346.5747	-5906.9394	-4484.2198									-1918.9998
SENDI PAPE	-40115, 157	-3339.676	ゆきに、よのふやし	-1475,945	**************************************	104 (1) (8X)	2011001	-5277.624	148:14.601	-4254.HD3	COM. PULCE		-1205.254	+63,834	- 7355. P31	050.020	144.244	+50° 555+	-4025.336	3447.446	B C C C C C C C C C C C C C C C C C C C	170000	*500.000	-6270.792	5 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0 0	-4-00° 05'	-3558,133	-2813.6RZ	-20% 158	11034.703	8 M 8 4 5 C 5 -	-5614.473	15 4 A B . O B B	-4660.856	-3004,121	-2206.409	-1440,7k3	-3450-401	******	-5754.164	15164.75		-2341,003	-1347.474	ř	-9659.361	19565.857	+15.96.7.1 •646 943	107.000	-3528.736	224.1862-	-1550.659
TANGOTONO NICOTO	-2.0223E-D*	-1.5A105-0*	-1.33276-04	-5.4326F-05	8.4780F-05	*0-3558***		-2.0754E-04	-2.0725E-0*	-1.942hE-04	-1.7757E-04	-0.35.45.04 -0.35.46.04	-3.72176-05	3.8*** 5-05	-2.83356-04	-3.05436-04	-2-1540F=04	-1.99186-09	-1.84126-04	-1.61726-04	*0=3*8C2*T=	-621016-0-	-3.1.35E-0*	-2.4425E-04	-2. 10046-04	*D-31*51.2*	-1.5150E-09	*1.2364E=04	-8.6406E-05	50-11/28-00	*0-30585.1-	*2.34086-0*	-5.6088E-0*	*2.1426E=04	*1.3514E=04	-4.4019E-05	-7.6581E-05	1.60162-04	-2.32446-04	-2.7854E-04	-2.6196E-04	-6.01200-04	-1.03325-04	-6.17335-05	-3.5024F-05	-4.5154E-04	-4.5778E-04	# 1 4 1 4 2 5 5 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	-2. (80.25.04	-1-63236+04	-4.88165-05	-7.24216-05
H	24.5747	BASC CE	30.5135	30.4622	31.4040	27.3640	27.8658	28.3373	28.8088	24.2803	24,7518	8 + B 4 - U.E	31,1663	41.3549	27.3905	27.6411	78.1157	24.0780	24,5403	34,0026	100000	31.1845	27.6844	24.1513	28,6163	2180.45	30,0111	30,4760	30.4410	31.4034	28.1340	28.6064	24.738	34.5416	30.4761	36 44 06	31,4104	28.1250	28,5457	29.0655	24.5353	30,000	30.946	31. * LOW	28.2116	28.5248	28.6701	5117.50	30.05	30.4588	10.401	31.3532
œ	5 + 8 + 01	11.1584	11.3327	11.5069	11.6635		4.8336	10.000	10.1664	10,3328	10.4992	10. N3.9	10,4983	11.06*4	9.2174	F. 44.5	6.579	4.724	C+88.F	10.0387	10.1434	10.5078	6.7245	A.8761	C22U*		5.44.6	4.6041	4,7556	200	8,3351	8.473b	8.6121		4.057	4.1654	4,3043	2,802,7	7.9303	9.000	6067°	3136.8	8.5817	8.7120		7,3239	7,3612	2000	2.205.2	7,8205	7,9353	1050.8
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Figure 1D. (Continued).

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សសម <i>ា</i> សសសសសស <i>និនិនិនិ</i> ភព៣៩៩៣៣៣៣៧៧៧សំស៊ីជីជីជីជីជីជីជីជីជីជីជីជីជីជីជីជីជីជីជ	24, 8627 31, 14827 24, 36 8 8 2 24, 36 8 8 2 31, 14 8 3 3 31, 14 8 3 3 31, 14 8 3 3 31, 14 8 3 3 31, 14 8 8 8 8 8 31, 14 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8 8	1. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2. 2.	10 9 4 5 5 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	-1083.18/0 -5775, 5263 -4878, 5263 -148.0, 5924 -1081, 13092 -1966, 8309 -5539, 6412 -1036, 6344 -1036, 7770, 4775 -537, 7705 -537, 7705 -537, 7705 -537, 7705 -539, 1835 -589, 1835 -589, 1835 -589, 8776	-7.3880E-04 -3.8820E-04 -3.83251E-04 -7.3948E-04 -7.3948E-05 -7.3948E-04 -5.3560E-04 -5.3560E-04 -5.8260E-04 -5.8260E-04 -5.8260E-04 -5.8260E-04 -1.7060E-04 -1.2624E-05 -1.3660E-04 -1.2624E-05 -1.3660E-04 -1.2624E-05 -1.3660E-04
<i>๛๛๛</i> ๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛๛	31. 16 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6 6	1	1 1 1 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2 2	-27.5 .263 -37.5 .263 -3.40 .524 -1.10 .4.64 -1.10 .4.64 -1.20 .539 .841 -2539 .841 -2539 .841 -2539 .841 -2539 .841 -2537 .276 -2337 .276 -337 .276 -336 .475 -586 .833 -586 .633 -586 .633 -586 .633 -589 .835 -589 .835	-5.5607F.C.4 -3.8325IE-04 -2.24728E-04 -7.3943E-05 -7.1943E-05 -7.1943E-04 -3.5608E-04 -5.8228E-04 -5.9228E-04 -5.9228E-04 -1.91794E-04 -1.91794E-04 -1.20241IE-04 -3.36440E-04 -3.36440E-04
๚๛๚๚๚๚๛๛๛๛๛๛๛๛๛๛๛๛๚๚๚๚๚๚๚๚๚๚๚๚๚๚๚๚๚๚๚	33. 56 56 56 56 56 56 56 56 56 56 56 56 56	1	100 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4 4	-5/7.283 -3430,5924 -1181,4646,- -10811,3092 -7968,8309 -5539,8412 -5230,6344 -688,056 -1770,4775 -537,2792 -337,2792 -336,4775 -580,633 -5191,3118 -589,1535 -589,1635	-3.84251E-04 -2.287251E-04 -7.39433E-05 -7.1948E-04 -5.3550E-04 -5.3550E-04 -2.05800E-04 -2.05800E-04 -2.9286E-04 -5.9260E-04 -1.19760E-04 -1.19760E-04 -3.64411E-04 -3.64411E-04 -3.36411E-04 -3.36411E-04 -3.36411E-04
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	10.4334	-2.5155E=0+	0 p p . 9 p c p =	~4888.8967	-3.30241E-04
	10.1828	\$15371717F		~2598.4038	-1.70705E-04
444444444	24,3500	#0161-04:41	501°10'10	-396.4596	-1.98068E-05
	29.4910	*********	-7340,752	-/146./916	-5.29306E-04
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*	30.9730	•	E88.649F-	4894 3954	-4.89082E-04
	041.0		*EB. 50+0.	-2938, 7720	-2.06812F_04
iddddd	35.5850		-410.040	-909.3383	-5.83982E-05
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•	31.5140	-7.7510E-05	-1845.844	-3248.5306	
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•	36.348	•	415.417	566.6736	
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5*0		13 00306105	110 7013	-16167.6661	-1.33739£-03
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UPPER HATCH NODES 178-240

Figure 1D. (Continued).

Figure 1D. (Continued).

			land burdens.		
	2	Longicudinel Strain	Stress	Hoop	Hoop Strein
	20.5061	-5.8484E-03	-3671,763	-3676.4302	-6.31204E-03
	21.2132	-6.0453E+03	646.1266-	-3577.2228	-5.94492E-03
	21.4203	-6.4328E-03	166.166.	-3584.4522	-5.61919E-03
3966	23.23.67	E0-3015-4-	*67.00.5**	3596.0460	-5.32294E-U3
	21.2043	-5.6478E-03	-3576.233	-3648.8519	-6.31812E-03
	9046.15	0-352	-3752,318	-3570.0445	-5.92599E-03
	22.6720	*6.4572E-03	-3474.803	-3565.4069	-5.57055E-03
21.1234	23.4033	-6.8886.403	665°061#=	-3569.6790	-5.254586-03
	1134	**************************************	010.0000	-3564.0034	-4.3024/E-03
	22.44.3	-5-1480E-03	*1782.072	-3000.1243	-6.29003E-U3
	24.446	-4-5303E-03	228.898-	1517 5025	-5.0000E-03
	24,1507	-6.8841E-03	169.8514-	-3525, 9736	-5.13963E-03
	24.4054	-7.0823E-03	-4273,136	-3444.2123	-4.82618E-03
	22.5372	-5.87116-03	-3661.114	-3641.6826	-6.23128E-03
	23.3144	-6.2710E-03	-3612.450	-3525.9564	-5.76900E-03
	24.0415	-6.62876-03	+11.4104-	-3491.5672	-5.35463E-03
	24.8587	-6.4738E-03	******	-3467.7744	-4.98334E-03
20.7676	25.6458	-7.3331E-03	-4358.201	-3394.6239	-4.63768E-03
	23.1604		-3837.068	-3658.9880	-6.21862E-03
. 0343	163.434	-6.46135.03		-3486.03/6	-5.62220E-U3
287	25.5583	-2.04716-03	164°C024-	-3383.4606	-4.77425E-03
14.8544	26.3550	-7.21516-03	-+260.324	-3269.6393	-4,40229E-03
337	4554.62	-6.414E-03	-3854.667	-3599,1075	-5,97170E-03
673	24.5746	-6.6922E-03	E # 30 * C * :	-3428.4066	-5.42715E-03
2000	26.21.20	ED-16967-03	109.6/044	-3348.4152	-4.34.201-03
4280	27.0320	-7.4224E-03	- 430B-1-47	1168 6544	4 109345-03
15.7445	24,3214	-6.4346E-03	*U8. P40**	-3585.3381	-5.76046E-03
75	25.1601	-7.02466-03	-+0+5.438	-3358.1778	-5,17031E-03
838	25.4488	-7.1482E-03	-4154.169	-3247.8314	-4.65846E-03
	26.8375	50-30842°C	648-7-14-	-3150.5676	-4.19596E-03
1110	19/90/20	13 total 03	100 COR 100 CO	2506.4109	-3,76068E-03
115	25.2150	-7.4343E-03	P. 2. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4. 4.	3266 0000	-5.47.301E-03
200	26.5722	-7.4189E-U3	-4183.568	-3123, 1644	-4.30396E-03
16.4812	27.4244	-7.4048E-03	-+202. +41	-2998.9216	-3.81403E-03
	59.28e5	-7.4425E-03	-4184.50B	-2830.8200	-3.35056E-03
5 5	25.3640	-7.45565-03	T#E 0144	-3429.6796	-5.10823E-03
1960	00.000	E0131611.11	COT OTHER	-3143.3912	-4.43923E-U3
5139	27.9878	e7.5155F=03		2005 2000	-3.5/U92E-U3
182	29.86.55	-2.457bE-03		2500.0035	23.30350E-U3
1.6147	25.6055	-8.5116E-03	-46-8-835	-3402 70K5	-4 88617F-03
T+80°	26.4884	-8.28+8E-03	808.2+++-	-3085.9731	-4.19696E-03
.5536	27.3714	-8.0014E-03	042.6564-	-2891.3813	-3.62342E-03
.0231	28,2543	-7.6473t-n3	-+187.644	-2710.6170	-3.10717E-03
-S-6	24.1373	-7.36*4E-03	-+0-520+-	-2483.5073	-2.61463E-03
3.1657	25,8392	825bE-0	-4733.421	-3337.6310	-4.63183£-03
-6147	26.7302	64435-0	8	-3022.0703	-3.92902E-03
16/00	67.6616			-2798.1215	-3.34580E-03

LOWER PLATE NODES 1-54

Figure 2D. Predicted stress and strains for bottom penetration plate.

ď	Ļ	774625			•
14.9817	54,4032	-7.4815E-n3 -402	-4054.424	-2368.1943	2.33432E-03
12.7128	26.1651	-4.4785E-03	-4477.742	-3291.9706	-4.344B6E-03
151	26.4638	536	-4700.856	-2935.1585	-3.62443E-03
.58	27.9626	15E-n	16+.64++	-2705.6408	-3.04676E-03
5	28.7614	0-3+B		-2480.4011	-2.53534E-03
7 . 4 F F S	<u>.</u>	45E=0	5	-2201.8182	-2.04616E-03
2	50° 50°	-4. B165E-03	•	-3198.4056	-4.01608E-03
•	200.00	0-2562	•	2867.0120	-3.28631E-03
	0.00.00	F0-2-10-6	· .	-2352 3312	2 222315 03
				2100.0426	1 426631 03
11.795		-1 05456	:	3139 0050	2 543301 03
12.2021	27.4054	•	::	2750 0626	2.04330E-U3
		346.01	900 95 181	2505.5050	2 347345 03
•		30000	******	2227 6470	1 034505 00
2000		7 7 7 7 7			-1.6/4062-03
11 2310		ED#366394	BOT - 0095	-1692. 3343	-1.423401-03
3766 - 11			5548	-3035. 154	-3.21224E-03
11.7614	27.6151	-9*	-5+04.871	-2751.4027	-2.45929E-03
16.1127	ä	÷	-4764.554	-2416.1246	
12,5034	2454.62	-7.4850£-03	-4159.980		-1.52273E-03
12.841	30.3767	-7.0300E-03	.48	-1771.5039	-1.10184E-03
10,8636	26.98B3	-1.2620E-02	;	-2994, 2539	-2.71028E-03
11,0509	27.3519	*1.2135t*n2	-5882-632	-2775.6018	-2.28787E-03
11,2382	27.8155	-1.1279E-02	-5648.477	-2735. 1007	-1.96314E-03
11,4255	28.2791	-1.03+0E-02	-5163.021	-2461.6637	-1.71860E-03
11,6128	28.7427	-9.428BE-03		-2392, 5822	_
11,8001	24.2063	-8.75.46-03	7	-2155.8415	-1 32306F-03
11,9874	29.6649	-7.44065-03	-4162.041	-2029.1135	-1.14601F-03
12,1747	30,1345	-7.14816-03		-1762.6422	-0 64454F-04
12,3620	30.5471	-6.4784E-03	1001101110	-1624.4482	-7 04959F-04
10.3927	27.0738	-1.5248E-02	0.541.152	-3136, 2059	-2.06682E-03
10.5719	27.540h	-1.30336-02	-6466-421	-2921.6216	
10,7510	28,007*	-1.1165E-02	-5647,304	-2614.2474	-1.42822E-03
10.4302	5 + C + C + E	-1.0121E-02	-5222.125	-2447.7751	
11.1044	28.9410	-4.216BE-03	*606.408	-2264.5843	-1.06828E-03
11.2886	29.4078	-8.4518E-03	-++55,312	-2091.8193	-8.99796E-04
11.4678	24.8746	-7.73666-03	-+085.230	-1896,9618	
11.6470	*1 *E * OE	-6.4733E-03	+22.807E-	-1702.0973	-6.17725E-04
11.8261	30.8085		-3344.163	-1519.2830	-5.02206E-04
•	27.0162	ø	-4334.719	-12904.3074	-1.14847E-03
9816	27.2511		-5418.467	-9424.8996	-1.09603E-03
10.0846	27,7204	-6.56716-03	-6384.380	-7186.4579	-9.69679E-04
10.6504	80 T 80 E		-5564.630	-6122.3294	-8.36668E-04
916	6H-66U6		-5218.320	-5405.0885	-7.16721E-04
90000	5061.50		-4747.523	-4713.2604	-6.01837E-04
9840 04	6000	E0#10#10##	290.000	-4191.2160	
٠	2010.00	FO 1 10 C 1 T C C	F.S.T. **********************************	-3601.1412	-3.976365-04
11.2863	6600	COLUNCTION OF THE	FTD-acce	-3102.0460	-3.003135-04
11.4577	31.4747	9.74545-05	100.001.3	1510 7170	-1.904995-04
11.6287	31.9495		9 1 2 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1 1	100 3400	-7.30021E-UZ
9.3178	27.9807	2,70936-05	065"2264-	13601 2316	1 20620F-03
9.4415	27.4200	-8.1995E-05	28E-E02+-	12378 8248	1 102715-03
9,6043	27.8928	43606	-52hb, 258	11052 4048	0 62755F
~	8.365	310+S	-5374.510	-9675, 8467	-8.25726F-04
4.9298	.838	. 6625E	-5071.719	-8319, 2223	-6.97888E-04
10.0426	•	.5806E	4724.427	-7158.6530	-5.81623E-04
	783	. * 0? 2E	4304-016	-6057,7859	-4.72541E-04
10.41RZ	30.2566	.1234E	-3773.673	-4958,3352	-3.69785E-04
10,5810	. 28	-1.3+4+E-0+	-2820.445	-3830.5302	-2.69106E-04
10,7437	31.2021	06.3434E=05			
				2441 170	1 68171F_04

LOWER PLATE NODES 55-116

Figure 2D. (Continued).

			A MEN	256
	-	133.	-14022.0108	-1.30900£-03
200,000 200,000 200,000 200,000 200,000		; ;	-12176.6055	_
C00, 645 468 468 468 468 468 468 468 468 468 468		100	-11110.2074	-9.63350E-04
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30,433 30,433 30,433		-4744,435	7074 0337	-6.81830E-04
30.409		-4266.	-5837, 0761	-4. 474315-04
*0**O*	-1.4600		-4602.4514	-3.39962E-04
20 304	-1.42676	45658.266	-3348, 1486	-2.38362E-04
100.10	-3°.7¢45E-0		-1932,1793	-1.36846E-04
22.063		757 - 97 - 1 7 - 9 - 9 - 1	-273.9907	-2.48937E-05
22.732	36196.3	100.000	-15208.8778	-1.39295E-03
110.00		0.000	-12547.6457	-1.14761E-03
	DADA TI	400000	-11271.6440	-9.680435-04
20 162	1	VER - 11 COUNTY	- 10089.5583	-8.061636-04
944	0-16/0/25		-8461.9845	-6.63952E-04
20.00	78.586.54		-7066.4837	-5.38178E-04
	22.20		-5626.9861	-4.20749E-04
700.00	3,296	-3340.145	-4361.5820	-3.12078E-04
<b>:</b> ,	-1.19676-0	PP.84	-2934.0408	-2.06787E-04
31.5541	•	-1094.846	-1528.1493	-1.065645-04
	4.3407E-U	163.149	-363.8600	-2.89866F-05
•	5.24305-0	Ξ	-12406 6078	1 167695-03
28.2130	ï	-6561.645	12722 4868	73026
28.547	•	5	11515 2920	2002
		ě	8000 Jay	-0./9033E-U
24.7491	C-145684 6-	-502 - 503-	2007.707	-0.823916-04
30.344	37565	38.0	201 1505	-5.153518-04
31,0007	3.645.10	256. (886-	2420 0180	-3.683635-04
31.1929		-2502-847	2016 0206	3 696075 04
28.634	-6.765¥E	-11677,611	-13077 4914	1.0000.1-
28.736	-6.1301E	-10618,975	12575 2348	9 23000E 04
24.340	-3,72186	-714,243	-9409,1195	-6. 66415E-04
*		-5170.548	-7089 9932	-4 9685F-04
30.5424		-3840,264	-5012, 9351	-3.432325-04
11.1511		-2730.647	-3134, 1199	-2 no1725-04
670	1 -6.18	*10980.454	-12731.9406	-8.50584F-04
•	T	-7659.6h2	-9386, 6405	
1680.06	-5.7465E-04	-5+81,304	-6965, 2212	
	7	-3558.511	-4725.6134	-3.20516E-04
- 1	7	9++* B+ [2-	-2554, 4933	
	î	=	-12550.7194	
0000			-9344.1831	-6.31884E-04
20.00		*C******	-6763.2408	
		20+ * LES	-4425.9522	-2.98043E-04
	9900.6	*55° +877-	-2070.2821	
•	35841.01	980.	-11747.2758	
٠,	59.0.5	5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5 5	-9021.3042	~6.13920E-04
•	39.64066	CC - 4845	-6556.8679	
31 6669	2/ FOS. I	770.000	-4189.4720	-2.79689E-04
: .	2004.84	366.1561-	-1908.6411	-1.26952E-04
	36081.7-	505 - 8021I -	-11738.6278	-7.80317E-04
2000	-5-18616	C+0" HUUX-	-8800.4791	-5.94060E-04
	3,300.6	100 min 100 min	-6328.9360	-4.23064E-04
:.		1000.400	-3956.6594	-2.61206E-04
	•	160.367	-1554.2827	-1.11804E-04
74.4222	13356	366 96 197	-11243.0807	-7.59624E-04
	46.64		-8596.4122	-5.78009E-04
3	0049-1-	-3163.893	2700 2423	-4.05558E-04
1:7	24945		1562.340/	*2.4/)32E-US
•	9	1	- 1997.8149	-9.65694E-05

LOWER PLATE NODES 116-178

Figure 2D. (Continued).

LOWER PLATE

Figure 2D. (Continued).